# 7th IAA Planetary Defense Conference 26-30 April 2021, Online Event

Hosted by UNOOSA in collaboration with ESA





# Session 3: DART Chairs: Andy Rivkin | Dawn Graninger

Presenters: T. Statler, NASA | E. Adams, JHU/APL | E. Dotto, INAF | A. Rivkin, JHU/APL | A. Cheng, JHU/APL









# Overview of the DART Mission Seven Months to Launch

### **Dr. Thomas S. Statler**

DART Program Scientist Planetary Defense Coordination Office NASA Headquarters Washington, DC





# NASA's Planetary Defense Coordination Office (PDCO)



- Established January 2016 at NASA HQ
- Manages planetary defense related activities across NASA
- Coordinates with U.S. interagency and international efforts in planetary defense

## **Mission Statement**

- Lead national and international efforts to:
- Detect any potential for significant impact of planet Earth by natural objects
- Appraise the range of potential effects by any possible impact
- Develop strategies to mitigate impact effects on human welfare





[CENTER FOR NEAR EARTH OBJECT STUDIES]

### **SEARCH, DETECT & TRACK**

[SPACE-BASED & GROUND-BASED OBSERVATIONS, IAWN]

PLANETARY DEFENSE

IAU

Planet

# MITIGATE

CHARACTERIZE

[NEOWISE, GOLDSTONE, IRTF]

PLAN & COORDINATE

[SMPAG, PIERWG, NITEP IWG]





[CENTER FOR NEAR EARTH OBJECT STUDIES]

### **SEARCH, DETECT & TRACK**

[SPACE-BASED & GROUND-BASED OBSERVATIONS, IAWN]



IAU

MITIGATE [dart, fema exercises]

DART is the first full-scale flight demonstration of an asteroid deflection technology: kinetic impact

DART – Double Asteroid Redirection Test

PLAN & COORDINATE

[SMPAG, PIERWG, NITEP IWG]

CHARACTERIZE

[NEOWISE, GOLDSTONE, IRTF]

## Part of a Larger Strategy

### National Near-Earth Object Preparedness Strategy and Action Plan

A Report by the Interagency Working Group for Detecting and Mitigating the Impact of Earth-Bound Near-Earth Objects of the National Science & Technology Council, June 2018



**DAMIEN (Detecting and Mitigating the Impact of Earth-Bound Near-Earth Objects) Membership:** Department of Commerce, Department of Defense, Department of Energy, Department of Homeland Security, Department of the Interior, Department of State, NASA, National Science Foundation, Office of the Director of National Intelligence, National Security Council, Office of Management and Budget, Office of Science and Technology Policy





# AIDA international collaboration







- There is no known asteroid that poses an actual impact risk to Earth.
- The current impact hazard comes from asteroids not yet discovered.
- The test is being conducted to develop a deflection capability, in case one is needed in the future.



## Launch Window

Nov. 18, 2021 – Feb. 15, 2022

- Target the binary asteroid Didymos system
- Impact Dimorphos and change its orbital period
- Measure the period change from Earth





### **The Ideal Target: A Natural Laboratory**



A binary asteroid allows a detectable deflection of an asteroid of relevant size.

- DART's kinetic impact will change the orbital period about the larger asteroid by ~1%.
  - Detectable in weeks to months.
- Same kinetic impact on a non-binary asteroid would change the orbital period about the Sun by ~0.00006%.
  - Would take many years to detect.

### **DART's Level 1 Requirements**

Defining the Mission's Planetary Defense Investigation









### **Impact Dimorphos**

During its Sept/Oct 2022 close approach to Earth

# Change the binary orbital period

Cause a ≥73-second change in the orbital period of Dimorphos

# Measure the period change

To within 7.3 seconds, from ground-based observations before and after impact

### Measure "Beta" and characterize the impact site and dynamics

**Beta** = the momentum enhancement factor



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No DART spacecraft ops

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**Beta** = the momentum enhancement factor

DART spacecraft ops



### DART – Double Asteroid Redirection Test





- 1. Test the ability to achieve a kinetic impact on a real asteroid
- 2. Test how a real asteroid responds to a kinetic impact

Test 1 ends, and Test 2 begins, at the moment of impact.



## **DART Spacecraft**





### **DART Operations**

### How DART targets Dimorphos





## **DART Operations**

### How DART targets Dimorphos





## Light Italian CubeSat for Imaging of Asteroids





- LICIACube Goals
- Obtain multiple (at least three) images of the DART impact ejecta plume over a span of times and phase angles, to allow estimation of plume density structure
- 2. Obtain multiple (more than three) images of the DART impact site with sufficient resolution to allow measurements of impact crater size and morphology
- 3. Obtain multiple (at least three) images of the non-impact hemisphere of Dimorphos
- 4. Obtain images of the ejecta plume and of the asteroid target to characterize color and spectral variations



Capable 6U CubeSat provided by Agenzia Spaziale Italiana (ASI)

Based on Argomoon CubeSat that will be flying on EM-1 mission (first flight of SLS in 2020)

Two cameras (goal of 2 m/pixel resolution imagery)

*Current concept of operations includes flyby of Didymos* ~3 *minutes after DART impact and downlinking data after event* 



## From Kinetic Impact to Beta – Looks Easy, but Isn't!

$$\beta = \frac{\frac{M}{m_{sc}} \Delta V_T - \vec{V}_{\infty_{\perp n}} \cdot \hat{e}_T + V_{\infty_n} \vec{\epsilon} \cdot \hat{e}_T}{V_{\infty_n} (\hat{n} + \vec{\epsilon}) \cdot \hat{e}_T}$$



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Determined directly from ground-based measurement of period change

Known spacecraft & binary asteroid parameters

Constrained by DART & LICIA imaging of Dimorphos and impact site

Constrained by impact simulations and LICIA imaging of ejecta plume



- DART is 7 months from launch, and 17 months from its kinetic impact on Dimorphos.
- DART will be a historic first test, both of humanity's ability to deflect a real asteroid and of a real asteroid's response to a deflection.
- DART will enlarge our understanding of NEOs for planetary defense and planetary science, both on its own (along with LICIACube) and synergistically with Hera.







## Double Asteroid Redirect Test (DART) Mission Status

Dr. Elena Adams Elena.Adams@jhuapl.edu 26 April, 2021



### Launch Target the binary asteroid Didymos system Impact Dimorphos and change its orbital period November 18, 2021 – February 15, 2022 Measure the period change from Earth September 25 -October 1, 2022 LICIACube (Light Italian Cubesat for Imaging of Asteroids) **DART Spacecraft** ASI contribution 676 kilograms wet mass 15,000 miles per hour (6.6 kilometers per second) 160 meters 11.92-hour orbital period 1,180-meter separation Earth-Based Observations between centers 780 meters 7.1 million miles (~0.07 AU) from 2.26-hour rotation period Earth at DART impact

**1** 

### **DART** moves to the secondary launch opportunity

- DART was directed by NASA to move to secondary launch opportunity with baseline launch readiness date: 18 November, 2021
- This secondary opportunity allows DART to meet its launch and objectives
  - Launch ground coverage becomes simpler, as DART will have a good contact with ESTRACK's New Norcia ground station
  - Cruise portion of the mission is a few months shorter
  - Impact geometry and velocity is comparable
  - Spacecraft still arrives at Dimorphos near the asteroid's closest approach to Earth, and guarantees good ground telescope and radar coverage
    - New impact dates: 25 Sept, 2022- 01 Oct, 2022



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## **DART: Key and New Technologies**

DART will mature key new technologies for future planetary missions



Autonomous Navigation using SMART Nav



**DRACO** Telescope



LICIACube Cubesat



Ion Propulsion Engine (NEXT-C)



Radio Line Slot Array



Roll Out Solar Arrays (ROSA)



Transformational Solar Array Concentrators



**Coresat Avionics** 



DART Mission Status, PDC 2021

## **DART** mission status overview

Spacecraft integration and test is proceeding, even in COVID-19 environment



DART in thermal chamber



### **Timeline to Launch** $\leq$ Integration Spacecraft **NEXT-C Panels/Harness Coresat Avionics Delivered and** Delivered, Readiness Delivered Core I&T began! Integrated **Review** Delivered April 2020 March 2020 April 2020 May 2020 June 2020

1

Solar Arrays Engineering Uni Deployed (@ DSS) Jul 2020Comprehensive Parformance Text (CPT) 1 Aug 2020Mission Simulation Aug 2020Closed Solar Arrays Closed Solar Arrays Closed Solar Arrays Closed PanelsDSN Closed Solar Arrays Closed Solar Arrays Closed PanelsDSN Closed Solar Arrays Closed PanelsNEXT-C Chegration Closed 	Timeline to Launch								
Solar Arrays Comprehensive Mission Closed DSN NEXT-C CPT 2 PER   Init Deployed Performance Image: Comprehensive Mission S/C DSN NEXT-C CPT 2 PER   Image: Solar Arrays Image: Comprehensive Mission S/C DSN NEXT-C CPT 2 PER   Image: Solar Arrays Image: Comprehensive Image: Compation Image: Compatio									
	Solar Arrays Engineering Unit Deployed (@ DSS) Jul 2020	Comprehensive Performance Test (CPT) 1 Aug 2020	Mission Simulation (MSIM) 1 Aug 2020	Closed S/C Panels Sept 2020	DSN Compat Testing Oct 2020	NEXT-C Integration Oct 2020	CPT 2 MSIM 2	S PER Jan 2021	

## **DART** mission status overview

- Spacecraft integration and test is proceeding, even in COVID-19 environment
- Move to the secondary launch opportunity is caused by waiting on two hardware articles to be delivered to the spacecraft: the DRACO telescope and the Solar Arrays
  - Spacecraft structural testing cannot commence without them
- DRACO telescope had a mirror failure during vibration of a spare flight telescope, and the failure investigation board concluded that the flight telescope mirror mounts needed redesign to ensure that the flight telescope survives spacecraft testing and launch. Redesign was completed, and new mirror mounts manufactured. Telescope is expected to arrive for integration in June 2021
- Rollout Solar Arrays (ROSA) have experienced delays in the build due to COVID and as part of the new technology development, but now have been received at APL for integration on the spacecraft in May 2021



DART in thermal chamber



# **Timeline to Launch**




**Double Asteroid Redirection Test** 

6



# LICIACube

## the Light Italian Cubesat for Imaging of Asteroids

Elisabetta Dotto (INAF-OAR) on behalf of the LICIACube Team



PDC- April 2021



## The LICIACube team:

INAF:	E. Dotto (Science Team Lead)
	V. Della Corte (Instrument Team Lead)
	E. Mazzotta Epifani (WP Observations Lead), S. Ieva, D. Perna
	J.R. Brucato (WP Laboratory experiments Lead), A. Meneghin, G. Poggiali
	S. Ivanovski (WP Ejecta Lead)
	A. Lucchetti (WP Impact Simulation Lead), G. Cremonese, E. Simioni
	M. Pajola (WP Proximity Lead)
<b>IFAC-CNR:</b>	A. Rossi (WP Dynamics Lead)
<b>Politecnico Milano:</b>	M. Lavagna (WP Mission Analysis Lead), A. Capannolo, G. Zanotti
Univ. Bologna:	M. Zannoni (WP Orbit determination Lead), P. Tortora, D. Modenini, I. Gai
Univ. Parthenope:	P. Palumbo, I. Bertini,
Argotec:	V. Di Tana (Argotec Program Manager), S. Simonetti (System Engineer),
	B. Cotugno, F. Miglioretti
ASI:	S. Pirrotta (Program Manager), M. Amoroso, G. Impresario,
ASI SSDC:	A. Zinzi (SOC Lead)

Agenzia Spaziale Italiana

argotec

1Sİ

POLITECNICO DI MILANO

Aerospace Science and Technology Department

ALMA MATER STUDIORUM Università di Bologna + INAF BITUTO MAZIONALE MATCHANGUNALE โเรกะ



# LICIACube



**Orbit:** Heliocentric (~10M km from the Earth)

Mass: 14 kg

Volume: 6U+

366 mm x 239 mm x 116.2 mm (stowed) 911.5 mm x 366 mm x 239 mm (deployed)

	Focal length (mm)	FoV (°)	IFoV (μrad/px)	Spat. scale at 55.2km (m/px)
LEIA	220	± 2.06	25	1.38
LUKE	70.55	±5	78	4.31



#### LEIA: a catadioptric camera spatial scale at C/A (~55km) 1.38 m/px

**LUKE**: a camera with a RGB Bayer pattern filter



C LICIA

E. Dotto and the LICIACube team

# LICIACube



LICIACube is carried by DART until close to Didymos and then released to perform a fly-by of Dimorphos after DART impact.

LICIACube downlinks images directly to Earth after the target fly-by.

E. Dotto and the LICIACube team





LICIACube performs maneuvers and acquires pictures of Dimorphos and plume generated by the DART spacecraft impact







CIACub

# LICIACube acquisition strategy

Phase	Start	End	LEIA	LUKE
1 – DART Impact	-45 s to T0	T0+136.11 s	yes	Not operative
2 - Ejecta Observation	-25 s to T0	T0 + 169.1 s	yes	yes
<b>3 - High resolution (surface properties/crater) observation</b>	T0 + 157.5 s	T0 + 169.1 s	yes	yes
4 – Non-impact hemisphere observation	T0 + 165.41s	T0 + 179.1 s	yes	yes
5 – Plume evolution in forward scattering	T0 + 179.1 s	T0 + 600 s	yes	yes





E. Dotto and the LICIACube team

# Scientific Objectives





E. Dotto and the LICIACube team

# **Ground Segment**



The mission Ground Segment architecture includes DSN antennas and the two main elements located in Italy:

- Mission Control Center (MCC): @ Argotec (Turin)
- Science Control Center (SOC): @ ASI SSDC (Rome) <u>https://www.ssdc.asi.it/liciacube/</u>
- The raw data, coming from MCC, will be calibrated by SOC, using the calibration procedure provided by INAF.
- A PDS4 archive will be designed, populated and released to the public after the end of the mission.
- Data will also be accessible (first of all to the team and in a second time to public) by means of the SSDC MATISSE webtool (<u>https://tools.ssdc.asi.it/Matisse</u>), with advanced 2D and 3D visualization capabilities.





# Status of the Project



- Qualification test campaign at Cubesat level in progress:
  - Full functional test succesfully completed;
  - TVAC test succesfully completed;
- Integration into the dispenser, expected on 12th May
- Qualification and Acceptance test campaign at System level (cubesat + dispenser + EBC)
  - Vibrations test
  - Deployment test
  - Thermal balance test
- Readiness of the System, expected on mid June
- Reviews and logistics
- Delivery to APL in Baltimore, expected on 23rd July



#### LICIACube spacecraft in TVAC chamber, after test



E. Dotto and the LICIACube team





# LICIACube

## the Light Italian Cubesat for Imaging of Asteroids

Elisabetta Dotto (INAF-OAR) on behalf of the LICIACube Team



PDC- April 2021







#### DART: How will we know what we've done?

Observations and Dynamics

DART

Andy Rivkin DART Investigation Lead



#### **DART's Level 1 Requirements**

Defining the Mission's Planetary Defense Investigation









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I'll be talking mostly about #3, just touch on #4



2

### **Current Knowledge about Didymos**

#### Lightcurves



Didymos, moving through star field Taken from Keck Observatory, January 2021

#### **Radar Shape Model**



Shape model of the Didymos primary asteroid from combined radar and light curve data (Naidu et al. 2020)

#### Composition



Spectral parameters from observations by de Leon et al. (2009) (yellow diamond) found by Dunn et al. (2014) to be most consistent with L/LL meteorites.



3

#### Measuring the Binary Asteroid System from Earth Dimorphos Orbiting about Didymos

Time



**€**∂



#### **DART at Scale**

Burj Khalifa 830 meters





5



- Focused on improving precision of Dimorphos' orbit to allow extrapolation and targeting of a particular orbit phase at time of DART arrival
- Observing season December 2020—March 2021
  - Early observations only had short observing window per night
- Deadline-driven data analysis included observations from LDT and Keck
  - Only data reduced through mid-January included in current fits
  - Additional observations still under analysis



- New data reduces the uncertainty on Dimorphos' orbit period to 0.01 seconds
- 3-σ uncertainty on Dimorphos orbit phase < 7° when extrapolated to time of DART arrival
- GM value corresponds to mass of roughly 5.5 x 10<sup>11</sup> kg
  - 95%+ of mass in Didymos, remainder in Dimorphos
  - Density 2170 ± 350 kg/m<sup>3</sup>
- Best estimate for binary YORP ("BYORP") is small but non-zero.



Lowell Discovery Telescope, December 2020 & January 2021 Nick Moskovitz, Matthew Knight, Tony Farnham

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Parameter	Value	$1\sigma$ uncertainty
<i>M</i> <sub>0</sub> (°)	78.9	1.9
Period (h)	11.9216287	0.0000031
$n_0 ({\rm rad}\;{\rm s}^{-1})$	1.46400235e-04	0.0000038e-4
$\dot{n}$ (rad s <sup>-2</sup> )	4.9e-18	1.1e-18
Epoch (TDB)	2011-08-21.5	
$\chi^2$	20.0	
$\chi^2_{\nu}$	0.42	
$(\boldsymbol{\lambda},\boldsymbol{\beta})^{\circ}$	$(320, -79)^{\circ}$	
$GM_{sys} (m^3 s^{-2})$	37.0362739237411501789	



#### Also see e-Lightning Talks:

"Constraining the Orbital Parameters of the Didymos-Dimorphos System: Lightcurve Observations in Preparation for AIDA/DART" by Thomas et al. "Estimation of orbital parameters of Dimorphos from lightcurve mutual events" by Naidu et al.

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## The 2022-2023 apparition

- Didymos spends very long period at >90° solar elongation, roughly 6 months at V < 17.5</li>
  - At V < 17.5 should be able to get good lightcurve data with m-class telescopes
- DART supporting observations from 4 observatories
  - Lowell Observatory
  - Magdalena Ridge Observatory
  - Las Cumbres Observatory (network)
  - Las Campanas Observatory
- Access to additional observatories through team membership around the world
- At time of DART arrival, Didymos is in southern skies. It moves north rapidly over following weeks
  - Distribution of supported telescopes allows good coverage during entire period
- Anticipate reaching required precision on period change by end of calendar year 2022





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3

### **Additional measurements**

- Radar: High SNR measurements available from Goldstone 2-16 October 2022, 150 m/pixel monostatic, 75 m/pixel bistatic with Green Bank possible.
  - Can potentially measure offset between Dimorphos position and unperturbed position within a few days of DART impact
- Ejecta:
  - Visualization models still being run for LICIACube and other observing platforms
  - Expect brightening of system while ejecta is still unresolved within same pixel as Didymos (duration dependent upon imaging system used as well as ejecta model)
  - Hope to study the ejecta evolution from Earth-based systems, which will help constrain particle sizes and ejecta mass

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#### Also see e-Posters:

"Dynamics of ejecta plume after the DART impact on Dimorphos" by Ferrari et al.;

"Non-spherical dust dynamics of the ejecta plume in support of DART/LICIACube mission" by Ivanovski et al.

"Incorporating a gravity field model based on radar observations into the rebound ejecta dynamics package" by Larson et al.

e-Lightning talks: "Simulating planned LICIACube imagery of DART impact ejecta based on ejecta dynamics simulation output" by Fahnestock et al.;

"Influence of the body composition on the evolution of ejecta in the Didymos-Dimorphos binary system" by Rossi et al.

### **Dynamical Possibilities**

- The impact into Dimorphos will inevitably cause librations
  - Even if perfectly circular/perfectly tidally locked pre-arrival, DART will cause some amount of eccentricity and thus libration
- The amount of libration is dependent upon Dimorphos' shape
  - Not clear if likely amounts of libration will be observable from Earth
  - May be observable with Hera's arrival
  - If observable, will provide information about Dimorphos' interior
- Dynamics of system very sensitive to initial conditions, shape of Didymos and Dimorphos
  - Close enough to one another that they do not appear as point sources for gravitational purposes
- · Hera's visit to Didymos will enable dynamical models to be tested
  - Another case were the combined data sets will be "more than the sum of parts"

## **Dynamical Possibilities**

The increase tists Direction will incruitably course librations

#### Also see e-Lightning talks:

"Consequences of the DART impact on Dimorphos' spin state and surface mass", Benavidez et al.;

"On the post-impact spin state of the secondary component of the Didymos-Dimorphos binary asteroid system" by Agrusa et al.;

"Changing the heliocentric orbit of the Didymos system with DART" by Makadia et al.

- · Hera's visit to Didymos will enable dynamical models to be tested
  - Another case were the combined data sets will be "more than the sum of parts"

## Summary

- Telescopic observations of the Didymos system are integral to the DART project
- Dynamical studies have been showing what can/can't be extrapolated or determined from 1<sup>st</sup> principles, and studying the lifetime of DART effects
- Observations prior to DART's arrival are necessary to characterize its undisturbed nature and to enable targeting at a desired Dimorphos orbit phase
- Observations subsequent to DART's arrival are how the experiment will be evaluated
- Preparations for the 2022-2023 observations are beginning, with opportunity to participate



Lowell Discovery Telescope (credit: Lowell Observatory)



# DART

**Double Asteroid Redirection Test** 





## DART Legacy: Determination of beta, data archive

DART

Kinetic Impactor Demonstration

Andrew Cheng andy.cheng@jhuapl.edu



## DART

Kinetic impactor demonstration at Didymos

- Demonstrate orbital deflection of a representative threat asteroid
- The DART target is a realistic-sized asteroid (160 m) of the most common NEO composition (S-type)
- A controlled impact experiment to increase confidence of kinetic impact predictions and improve understanding of asteroid physical properties and high-speed collisions

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# Measure "Beta" and characterize impact site and dynamics

**Beta** = the momentum enhancement factor



# **Beta: Momentum Transfer Efficiency**



- β is defined as momentum transfer divided by momentum input
  - If no ejecta, then  $\beta = 1$
  - Ejecta release *enhances* momentum transfer,  $\beta > 1$
- Momentum transfer from kinetic impact is enhanced by impact ejecta
- Estimation of β important for asteroid deflection. How large a kinetic impact is needed, or how many kinetic impacts are needed, depend on β

## For DART impact in 3D:

$$\Delta \vec{v} = \frac{m}{M} \left( \vec{u} + (\beta - 1)(\hat{n} \cdot \vec{u})\hat{n} \right)$$

$$\beta = \frac{M(\hat{n} \cdot \Delta \vec{v})}{m(\hat{n} \cdot \vec{u})}$$

 $\beta$  is the transferred momentum component along  $\hat{n}$  divided by the incident momentum component along  $\hat{n}$ .

With some assumptions, the vector  $\hat{n}$  is the target surface normal



# **DART Beta determination**

- Measure transverse component of velocity change (from period change)
- Use full 2-body numerical modeling of Dimorphos binary dynamics to relate period change to vector velocity change and beta (using mass)
- Determine size, shape and volume of Dimorphos to infer mass (with assumed density given by bulk density of Didymos)
  - DART approach imaging of approach hemisphere
  - LICIACube imaging of approach and departure hemispheres
- Determine DART impact site location, and geology and topography of impact site, from DART approach imaging
- LICIACube plume observations to constrain plume direction  $\hat{n}$
- Use impact simulations for
  - Effects of spacecraft geometry
  - Effects of impact site topography, slopes, rubble structures
  - Estimation of uncertainty



# **DART Data Archive**

#### DART, LICIACube and Ground-based telescopic data

Data Source	Dataset	Derived data	Archive
DART DRACO	Monochrome imaging	Mosaics, light curves, image backplanes, shape models, facet data	PDS SBN
LICIACube LEIA	Monochrome imaging	Image backplanes	PDS SBN
LICIACube LUKE	3-color imaging	Image backplanes	PDS SBN
Las Cumbres	Ground-based 2m/spectral images	Light curves	PDS SBN
Las Campanas	Magellan/IMACS images	Light curves	PDS SBN
Lowell	DCT/LMI images	Light curves	PDS SBN
Magdalena Ridge	2.4m images	Light curves	PDS SBN
Radio Science	Doppler and ranging		PDS SBN
SPICE data	Ancillary data		NAIF



5

# **Anticipating Hera science at Didymos**

#### AIDA: DART and Hera joint studies of kinetic impact demonstration results

- Hera ESA mission to Didymos, arriving late 2026, will
  - Measure mass of Dimorphos
  - Measure DART impact crater
  - Study spins, search for shape changes, and measure librations excited by DART impact
- Didymos telescopic observations
  - Light curve observations of mutual events, spins and orbital period
  - Ejecta imaging from Earth to study activation of comet-like dust tails by DART impact
- Impact simulations
  - DART crater predictions
  - Modeling of spacecraft structure and target structures (rubble, boulders, topography)
  - Inference of Dimorphos target properties (like strength, porosity, friction, crush properties), with Hera data
  - Beta determination and uncertainties, with Hera data
- Didymos binary system dynamics
  - Inference of interior structures
  - Tidal interactions and BYORP



# **DART Legacy**

- DART demonstrates planetary defense capability
  - Showing that a kinetic impactor accomplishes asteroid deflection
  - Demonstrating kinetic impactor technology and concept of operations
- DART determines momentum transfer efficiency for a target asteroid of representative size and most common spectral type
- DART improves and validates models of kinetic impact deflection
  - More reliable predictions of kinetic impactor effects on other asteroids
- DART informs planetary defense decision and policy definition processes and reduces risks


## DART

**Double Asteroid Redirection Test** 

## 7th IAA Planetary Defense Conference 26-30 April 2021, Online Event

Hosted by UNOOSA in collaboration with ESA





**Q&A** Session 3 - DART



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## Break

Up Next: Session 4 - OSIRIS-REx

