TECHNICAL MANUAL

Operator's and Crewmember's Checklist

ARMY C-12C AIRCRAFT NSN 1510-01-070-3661

ARMY C-12D AIRCRAFT NSN 1510-01-087-9129

ARMY C-12T AIRCRAFT NSN 1510-01-470-0220

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited.

*This manual supersedes TM 55-1510-218-CL dated 22 April 1985.

HEADQUARTERS DEPARTMENT OF THE ARMY 04 September 2001

GENERAL INFORMATION AND SCOPE

SCOPE

This checklist contains the operator's and crewmember's checks to be accomplished during normal and emergency operations.

GENERAL INFORMATION

This checklist consists of two parts, Part I for the C-12C and the C-12D model aircraft and Part II for the C-12T1 and C-12T2 model aircraft. Both Parts I and II consist of three parts: normal procedures, emergency procedures, and performance data. Normal procedures consist of the procedures required for normal flight. Emergency procedures are subdivided into seven classifications as follows: engine, propeller, fire, fuel, electrical, landing and ditching, and flight controls. Performance data consists of performance checks.

This checklist, printed from CD, must be printed on 4 1/2" x 8" paper and assembled in a checklist binder. This manual must be carried with the aircraft at all times. Users are authorized to remove those parts that are not applicable to their aircraft model and are not required to carry them on the aircraft.

NOTE

This checklist does not replace the amplified version of the procedures in the operator's manual, TM 1-1510-218-10, but is a condensed version of each procedure.

NORMAL PROCEDURES PAGES

The normal procedures checklist is a condensed version of the amplified checklist appearing in the normal procedures or crew duties portion of the applicable operator's manual.

EMERGENCY PROCEDURES PAGES

The requirements for this section of the condensed checklist manual (CL) are identical to those for the normal procedures, except that the information is drawn from the amplified checks in the emergency procedures portion of the operator's manual. The emergency requirements are subdivided into the seven classifications listed in the General Information paragraph. Immediate actions are <u>underlined</u> and shall be memorized.

PERFORMANCE PAGES

The contents of the performance checks procedures of this manual are a detailed version of the procedure from the Normal Procedures pages designated by a \bigstar . The detailed procedures in the performance checklist are the same as those annotated with a \bigstar in the amplified normal procedures checklist in the operator's manual. The condensed normal procedures checklist has only the title of the procedure annotated with a \bigstar , which indicates that the detailed procedure is included in the performance checklist.

Symbols Preceding Numbered Steps:

- N Indicates performance of step is mandatory for night flights.
 - Indicates a mandatory check for instrument flights.
- O Indicates if installed.
- ★ Indicates a detailed procedure for this step is included in the performance checks section, located at the back of the checklist.
- * M Indicates performance of step is mandatory for all through flights. During the through flight walk-around inspection, refer to TM 1-1510-218-10 for the specific items to be inspected.

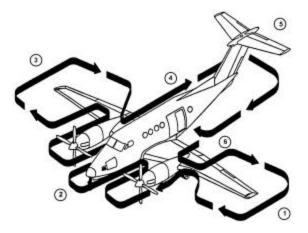
Immediate action emergency items are underlined.

REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS

You can help improve this checklist. If you find any mistakes or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms) or DA Form 2028-2 directly to: Commander, US Army Aviation and Missile Command, ATTN: AMSAM-MMC-ma-np, Redstone Arsenal, AL 35898-5230. A reply will be furnished to you. You may also send your comments electronically to our e-mail address, <u>2028@redstone.army.m</u>il or by fax 256-842-6546/ DSN 788-6546.

OZONE DEPLETING CHEMICAL INFORMATION

This document has been reviewed for the presence of Class I ozone depleting chemicals. In the base document dated 25 April 1985, all references to Class I ozone depleting chemicals have been removed from this document by substitution with chemicals that do not cause atmospheric ozone depletion.



Exterior Walkaround Diagram

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OPERATOR'S AND CREWMEMBER'S CHECKLIST

PART I

ARMY C-12C AIRCRAFT NSN 1501-01-070-3661

ARMY C-12D AIRCRAFT NSN 1510-01-087-9129

NORMAL PROCEDURES

BEFORE EXTERIOR CHECK

- *1. Forms/publications Check.
- *2. Oxygen system Check.
- *3. Flight controls Unlocked and checked.
- *4. Parking brake Set.
- *5. Manual trim Zero.
- *6. GEAR DN.
 - 7. ICE VANE handles As desired.
 - 8. Key lock switch ON.
 - 9. Battery switch ON.
- 10. Lighting and Heats Check.
- 11. Fuel gauges Check fuel quantity and gauge operation.
- 12. Battery switch OFF.
- O 13. Galley power switches OFF.
 - 14. Electric toilet Check.
 - 15. Emergency equipment Check.

FUEL SAMPLE

1. Fuel sample - Check.

LEFT WING, AREA 1

- 1. Left wing area Check.
- 2. Left main landing gear Check.

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- 3. Left engine and propeller Check.
- 4. Left wing center section Check.
- 5. Fuselage underside Check.

NOSE SECTION, AREA 2

1. Nose section – Check.

RIGHT WING, AREA 3

- 1. Right wing center section Check.
- 2. Right engine and propeller Check.
- 3. Right main landing gear Check.
- 4. Right wing Check.

FUSELAGE RIGHT SIDE, AREA 4

1. Fuselage right side – Check.

EMPENNAGE, AREA 5

1. Empennage – Check.

FUSELAGE LEFT SIDE, AREA 6

- 1. Fuselage left side Check.
- *2. Chocks and tiedowns Check removed.

INTERIOR CHECK

- 1. Cargo/loose equipment Check secure.
- 2. Cabin door Locked and checked.
- 3. Cargo door Locked and checked.

- 4. Emergency exit Check.
- ★ 5. Crew/passenger briefing As required.

BEFORE STARTING ENGINES

- *1. Oxygen system Set.
 - 2. Circuit breakers Check.
- *3. Overhead panel Check.
- *4. Fuel panel switches Check.
 - 5. Magnetic compass Check.
 - 6. CLOCK and MAP lights OFF.
- *7. Pedestal controls Set.
 - 8. Lower console switches Set.
 - 9. Gear alternate engage and ratchet handles Stowed.
- 10. Free air temperature gauge Check.
- 11. Pilot's instrument panel Check and set.
- 12. Copilot's instrument panel Check and set.
- 13. Subpanel Check and set.
- ★ 14. Fuel pumps/crossfeed operation Check.
 - *15. GPU As required.
 - *16. EXTERNAL POWER advisory light As required.
 - *17. Battery switch ON.
 - 18. Annunciator panels Test.
- ★ 19. Stall and gear warning system Check.
 - 20. Engine fire protection system Check.

FIRST ENGINE START (BATTERY START)

- 1. **EXTERIOR LIGHTS** switches As required.
- 2. Propeller Clear.
- 3. Engine Start.
- 4. Engine and systems instruments Check.
- CONDITION lever HIGH IDLE. Monitor TGT as the condition lever is advanced.
- 6. GEN switch RESET, then ON.

SECOND ENGINE START (BATTERY START)

- 1. First engine generator load 50% or less **GEN** switch **OFF**.
- 2. Propeller Clear.
- 3. Engine Start.
- 4. Engine and systems instruments Check.
- 5. BATTERY CHARGE light on Check.
- 6. **INVERTER** switches **ON**, check, **INVERTER** lights **OFF**.
- 7. Second engine generator switch **RESET**, then **ON**.
- 8. **CONDITION** levers As required.
- 9. Red anticollision light Reset.

ABORT START

- 1. CONDITION lever FUEL CUTOFF.
- 2. IGNITION AND ENGINE STARTER switch STARTER ONLY.
- 3. **TGT** Monitor for drop in temperature.
- Ignition and engine starter switch OFF after TGT is below 750 °C.

ENGINE CLEARING

- 1. CONDITION lever FUEL CUTOFF.
- IGNITION AND ENGINE STARTER switch OFF (1-minute minimum).
- IGNITION AND ENGINE STARTER switch STARTER ONLY (15 seconds minimum, 40 seconds maximum).
- 4. IGNITION AND ENGINE STARTER switch OFF.

FIRST ENGINE START (GPU START)

- 1. EXTERIOR LIGHT switches As required.
- 2. Propeller area Clear.
- 3. Engine Start.
- 4. Engine and systems instruments Check.
- CONDITION lever HIGH IDLE. Monitor TGT as the condition lever is advanced.
- 6. GPU disconnect As required.
- 7. GEN switch RESET, then ON.

*SECOND ENGINE START (GPU START)

- 1. Propeller area Clear.
- 2. Engine Start.
- 3. Engine and systems instruments Check.
- 4. Right **PROP** lever **FEATHER**.
- 5. GPU Disconnect.
- 6. Right **PROP** lever **HIGH RPM**.
- INVERTER switches ON, check INVERTER lights OFF.
- 8. GEN switches RESET, then ON.
- 9. CONDITION levers As required.
- 10. Red anticollision light Reset.

BEFORE TAXIING

- *1. BLEED AIR VALVES As required.
- *2. BRAKE DEICE As required.
- *3. CABIN TEMP MODE and temperature switches Set as desired.
- ★ 4. AC/DC power Check.
 - *5. AVIONICS MASTER POWER switch ON.
 - *6. Avionics controls As required.
- ★ 7. Electric elevator trim, autopilot/flight director operation
 C D1 Check as required.
- ★ 8. Autopilot trim fail system C D1 Check.
- ★ 9. Automatic flight control system D2 Check as required.

- 10. Electric elevator trim D2 Check.
- 11. Ground Proximity Altitude Advisory System (GPAAS) Check.
- 12. Avionics Check and set as required.
- 13. Flaps Check.
- 14. Altimeters Set and checked.

***TAXIING**

- 1. Brakes Check.
- 2. Flight instruments Check for normal operation.

ENGINE RUNUP

- 1. Parking break As required.
- 2. Propeller feathering Check.
- ★ 3. AUTOFEATHER/AUTO IGNITION Check.
- ★ 4. Overspeed governors and rudder boost Check.
- ★ 5. Primary governors Check.
- ★ 6. Ice vanes Check.
 - 7. CONDITION levers HIGH IDLE.
 - 8. POWER levers IDLE.
- ★ 9. Anti-ice and deice systems Check.
- ★ 10. Pneumatic system Check.
- ★* 11. Pressurization system Check.
 - 12. CONDITION levers As required.

*BEFORE TAKEOFF

1. AUTOFEATHER switch – ARM.

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- 2. BLEED AIR VALVES As required.
- FUEL panel Check fuel quantity and switches positions.
- 4. Flight and engine instruments Check for normal indications.
- 5. CABIN CONTROLLER Set.
- 6. Annunciator panels Check and note indications.
- 7. PROP levers HIGH RPM.
- 8. FLAPS As required.
- 9. Trim Set.
- 10. Avionics Set.
- 11. Flight Controls Check.
- ★ 12. Departure briefing Complete.
 - 13. CABIN SIGNS switch As required.

*LINE UP

- 1. ICE & RAIN switches As required.
- 2. Altitude alerter **D2** Check.
- O 3. Transponder/TCAS/Weather Radar As required.
 - 4. ENG AUTO IGN switches ARM.
 - 5. Lights As required.
 - 6. CONDITION levers HIGH IDLE.
 - 7. Power stabilized Check 27% minimum.

AFTER TAKEOFF

1. GEAR – UP.

- 2. FLAPS (105 KIAS) UP.
- 3. LANDING LIGHTS OFF.
- 4. Climb power Set.
- 5. **PROP SYNC** switch As required.

CLIMB

- 1. **YD** switch As required.
- 2. Cabin pressurization Check.
- 3. AUTOFEATHER switch As required.
- 4. BRAKE DEICE As required.
- 5. WSHLD ANTI-ICE As required.
- 6. Wings and nacelles Check.
- O 7. TCAS Set range.

CRUISE

- 1. Power Set.
- 2. ICE & RAIN switches As required.
- 3. CABIN SIGNS switch As required.
- 4. AUXILIARY fuel gauges Monitor.
- 5. Altimeters Check.
- 6. Engine instrument indications Noted.
- 7. **RECOG** lights As required.
- O 8. TCAS Set for en route.

DESCENT – MAX RATE (CLEAN)

- 1. Cabin pressurization Set.
- 2. CABIN SIGNS switch As required.
- 3. POWER levers IDLE.
- 4. **PROP** levers **HIGH RPM**.
- 5. **GEAR UP**.
- 6. FLAPS UP.
- 7. Airspeed V_{mo} maximum.
- 8. ICE & RAIN switches As required.
- 9. **RECOG** lights As required.

DESCENT – MAX RATE (LANDING CONFIGURATION)

- 1. Cabin pressurization Set.
- 2. CABIN SIGNS switch As required.
- 3. POWER levers IDLE.
- 4. PROP levers HIGH RPM.
- 5. FLAPS APPROACH.
- 6. GEAR DN.
- 7. Airspeed 181 KIAS maximum.
- 8. ICE & RAIN switches As required.
- 9. **RECOG** lights As required.

DESCENT – ARRIVAL

- 1. Cabin pressurization Set.
- 2. CABIN SIGNS switch As required.
- 3. ICE & RAIN switches As required.
- 4. WSHLD ANTI-ICE As required.
- 5. **RECOG** lights **ON**.
- 6. Radio altimeter As required.
- 7. Altimeters Set to current altimeter setting.
- O 8. TCAS Set as required.
- ★ 9. Arrival briefing Complete.

APPROACH CHECK

- 1. HSI NAV SOURCE As required.
- O 2. TCAS As required.

BEFORE LANDING

- 1. CABIN SIGNS switch BOTH.
- 2. PROP SYNC switch OFF.
- 3. AUTOFEATHER switch ARM.
- 4. BRAKE DEICE As required.
- 5. **PROP** levers As required.
- 6. FLAPS (below 199 KIAS) APPROACH.
- 7. GEAR (below 181 KIAS) DN.
- 8. LANDING/TAXI lights As required.

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9. CONDITION levers - HIGH IDLE.

O 10. TCAS - Set as required.

LANDING

- 1. AP & YD Disengaged.
- 2. GEAR DOWN lights Check.
- 3. **PROP** levers **HIGH RPM**.

TOUCH AND GO LANDING

- 1. PROP levers HIGH RPM.
- 2. FLAPS As required.
- 3. Trim Set.
- 4. Power stabilized Check 27% minimum.
- 5. Takeoff power Set.

GO AROUND

- 1. Power As required.
- 2. GEAR UP.
- 3. FLAPS APPROACH.
- 4. FLAPS (105 KIAS) UP.
- 5. LANDING LIGHTS OFF.
- 6. Climb power Set.
- 7. YD As required.
- 8. BRAKE DEICE OFF.

AFTER LANDING

- 1. CONDITION levers As required.
- 2. ENG AUTO IGN OFF.
- 3. ICE & RAIN switches As required.
- 4. FLAPS As required.
- 5. **XPNDR** Standby.
- 6. RADAR Standby.
- 7. LIGHTS As required.

ENGINE SHUTDOWN

- 1. BRAKE DEICE OFF.
- 2. Parking brake Set.
- 3. LANDING/TAXI light OFF.
- 4. CABIN TEMP MODE switch OFF.
- 5. AUTOFEATHER switch OFF.
- 6. VENT and AFT VENT BLOWER switches AUTO.
- 7. **INVERTER** switches **OFF**.
- 8. Battery condition Check.
- 9. TGT/ITT Check.
- 10. CONDITION levers FUEL CUTOFF.
- 11. **PROP** levers **FEATHER**.
- 12. EXTERIOR LIGHTS Off.
- 13. MASTER PANEL LIGHTS Off.

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- 14. AVIONICS MASTER switch Off.
- 15. MASTER SWITCH Off.
- 16. Key lock switch OFF.
- 17. Oxygen system OFF.
- 18. Chocks As required.
- 19. Parking break As required.
- 20. Control locks As required.

BEFORE LEAVING AIRCRAFT

- 1. Wheels Chocked.
- 2. Parking brake As required.
- 3. Flight controls As required.
- N 4. OVERHEAD FLOOD LIGHTS OFF.
 - 5. STANDBY PUMPS OFF.
 - 6. Windows **C** As required.
 - 7. Emergency exit lock As required.
 - 8. Galley power switches OFF.
 - 9. Aft cabin light OFF.
 - 10. Door light OFF.
 - 11. Walk-around inspection Complete.
 - 12. Aircraft forms Complete.
 - 13. Aircraft secured Check.



EMERGENCY PROCEDURES

ENGINE MALFUNCTION

ENGINE MALFUNCTION BEFORE V (ABORT)

- 1. POWER levers IDLE.
- 2. Braking As required.

ENGINE MALFUNCTION AFTER V1

- 1. GEAR (positive climb) UP.
- 2. POWER As required.
- 3. FLAPS (105 KIAS) UP.

IF THE PROP DID NOT FEATHER, PERFORM STEP 4.

4. PROP (dead engine) - FEATHER.

ONCE THE PROP IS FEATHERED, PERFORM STEPS 5 THROUGH 8.

- O 5. <u>TCAS Set TA</u>.
 - 6. LANDING/TAXI LIGHTS OFF.
 - 7. BRAKE DEICE OFF.
 - 8. Engine cleanup Perform.

ENGINE MALFUNCTION DURING FLIGHT

- 1. Autopilot/yaw damp Disengage.
- 2. **POWER** As required.
- 3. Dead engine Identify.

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- 4. PROP lever (dead engine) FEATHER.
- 5. <u>GEAR As required</u>.
- 6. FLAPS As required.
- 7. TCAS Set TA.
 - 8. Power Set for single-engine cruise.
 - 9. Engine cleanup Perform.

ENGINE MALFUNCTION DURING FINAL APPROACH

- 1. POWER As required.
- 2. <u>GEAR DN</u>.

ENGINE MALFUNCTION (SECOND ENGINE)

- 1. Airspeed As required.
- 2. PROP lever As required.

ENGINE SHUTDOWN IN FLIGHT

- 1. **POWER** lever **IDLE**.
- 2. **PROP** lever **FEATHER**.
- 3. CONDITION lever FUEL CUTOFF.
- 4. Engine cleanup Perform.

ENGINE CLEANUP

- 1. CONDITION lever FUEL CUTOFF.
- 2. ENG AUTO IGN OFF.
- 3. AUTOFEATHER switch OFF.
- 4. GEN switch OFF.



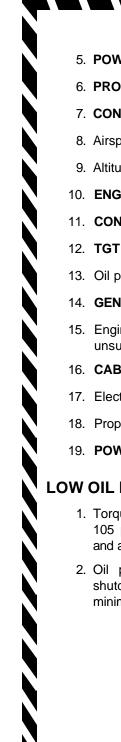
5. PROP SYNC switch - OFF.

ENGINE RESTART DURING FLIGHT (USING STARTER)

- 1. CABIN AIR/TEMP MODE switch OFF.
- 2. Electrical load Reduce to minimum.
- 3. FIRE PULL handle In.
- 4. POWER lever IDLE.
- 5. **PROP** lever **FEATHER**.
- 6. CONDITION lever FUEL CUTOFF.
- 7. TGT (operating engine) 700 °C or less.
- 8. Engine Start.
- 9. GEN switch RESET, then ON.
- 10. Engine cleanup Perform if engine restart is unsuccessful.
- 11. CABIN AIR/TEMP MODE switch As required.
- 12. Electrical equipment As required.
- 13. ENG AUTO IGN switch ARM.
- 14. **PROP SYNC** switch As required.
- 15. **POWER** As required.

ENGINE RESTART DURING FLIGHT (NOT USING STARTER)

- 1. CABIN AIR/TEMP MODE switch OFF.
- 2. Electrical load Reduce to minimum.
- 3. GEN switch (affected engine) OFF.
- 4. FIRE PULL handle In.



- 5. POWER lever IDLE.
- 6. PROP lever HIGH RPM.
- 7. CONDITION lever FUEL CUTOFF.
- 8. Airspeed 140 KIAS minimum.
- 9. Altitude Below 20,000 feet.
- 10. ENG AUTO IGN switch ARM.
- 11. CONDITION lever LOW IDLE.
- 12. TGT 1000° 5 seconds maximum.
- 13. Oil pressure Check.
- 14. GEN switch RESET, then ON.
- Engine cleanup Perform if engine restart is unsuccessful.
- 16. CABIN AIR/TEMP MODE switch As required.
- 17. Electrical equipment As required.
- 18. Propellers Synchronized.
- 19. **POWER** As required.

LOW OIL PRESSURE

- 1. Torque 49% maximum. Oil pressure less than 105 psi below 21,000 feet or 85 psi 21,000 feet and above.
- 2. Oil pressure below 60 psi. Perform engine shutdown or land as soon as practicable using minimum power to ensure safe arrival.



CHIP DETECTOR WARNING LIGHT

If a **CHIP DET** warning light illuminates and safe single-engine flight can be maintained, perform engine shutdown.

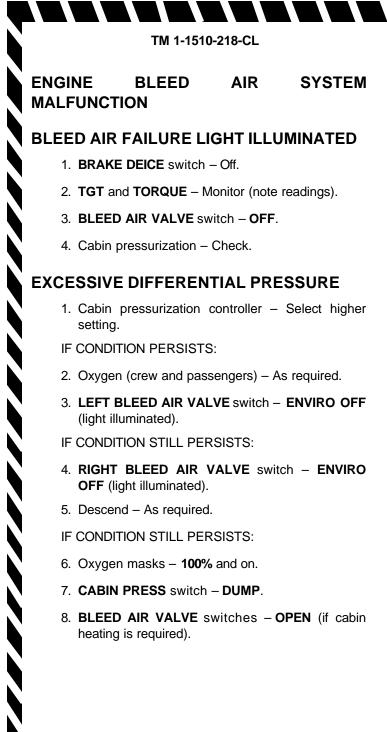
DUCT OVERTEMP CAUTION LIGHT

- 1. CABIN AIR control In.
- 2. CABIN AIR/TEMP MODE switch AUTO.
- 3. CABIN AIR/TEMP switch Decrease.
- 4. VENT BLOWER switch HI.
- 5. CABIN AIR/TEMP MODE switch MAN COOL.
- 6. MANUAL TEMP switch DECREASE (hold).
- LEFT BLEED AIR VALVE switch PNEU & ENVIRO OFF.
- Light still illuminated (30 seconds) LEFT BLEED AIR VALVE switch – OPEN.
- 9. RIGHT BLEED AIR VALVE switch PNEU & ENVIRO OFF.
- Light still illuminated (30 seconds) RIGHT BLEED AIR VALVE switch – OPEN.

ICE VANE FAILURE

- 1. Airspeed 160 KIAS or below.
- 2. ICE VANE CONTR circuit breaker Pull.

- 3. ICE VANE Operate manually.
- 4. Airspeed Resume normal airspeed.



BLEED AIR FAILURE LIGHT ILLUMINATED

- 1. BRAKE DEICE switch Off.
- TGT and TORQUE Monitor (note readings).
- BLEED AIR VALVE switch OFF.
- Cabin pressurization Check.

EXCESSIVE DIFFERENTIAL PRESSURE

- 1. Cabin pressurization controller Select higher setting.
- IF CONDITION PERSISTS:
- Oxygen (crew and passengers) As required.
- LEFT BLEED AIR VALVE switch ENVIRO OFF (light illuminated).
- IF CONDITION STILL PERSISTS:
- 4. RIGHT BLEED AIR VALVE switch ENVIRO OFF (light illuminated).
- 5. Descend As required.
- IF CONDITION STILL PERSISTS:
- 6. Oxygen masks 100% and on.
- CABIN PRESS switch DUMP.
- 8. BLEED AIR VALVE switches OPEN (if cabin heating is required).



LOSS OF PRESSURIZATION (ABOVE 10,000 FEET)

- 1. Crew oxygen masks 100% and on.
- Passenger oxygen ON and checked to ensure all passengers have oxygen masks on and are receiving supplemental oxygen if required.

CABIN DOOR CAUTION LIGHT

- 1. CABIN SIGNS switch BOTH.
- 2. BLEED AIR VALVE switches ENVIRO OFF.
- 3. Descend below 14,000 feet as soon as practicable.
- 4. Oxygen As required.

SINGLE-ENGINE DESCENT/ARRIVAL

- 1. CABIN CONTROLLER Set.
- 2. CABIN SIGNS switch As required.
- 3. ICE & RAIN switches As required.
- 4. Altimeters Set.
- 5. RECOG/BEACON/NAV lights ON.
- ★ 6. Arrival briefing Complete.

SINGLE-ENGINE BEFORE LANDING

- 1. CABIN SIGNS switch BOTH.
- 2. BRAKE DEICE Off.
- 3. **PROP** lever As required.
- 4. FLAPS (Below 199 KIAS) APPROACH.



7. CONDITION lever (operating engine) – HIGH IDLE.

SINGLE-ENGINE LANDING CHECK

- 1. AP/YD Disengaged.
- 2. GEAR DOWN lights Check.
- 3. PROP lever (live engine) HIGH RPM.

SINGLE-ENGINE GO-AROUND

- 1. POWER As required.
- 2. <u>GEAR UP</u>.
- 3. FLAPS APPROACH.
- 4. FLAPS (105 KIAS) UP.
- 5. LANDING/TAXI LIGHTS OFF.
- 6. Climb power Set.
- 7. YD As required.

PROPELLER FAILURE (OVER 2120 RPM)

- 1. **POWER** lever (affected engine) IDLE.
- 2. PROP lever FEATHER.
- 3. CONDITION lever As required.
- 4. Engine cleanup As required.





FIRE

ENGINE/NACELLE FIRE DURING START OR GROUND OPERATIONS

- 1. <u>PROP levers FEATHER</u>.
- 2. CONDITION levers FUEL CUTOFF.
- 3. FIRE PULL handle Pull.
- 4. PUSH TO EXTINGUISH switch Push.
- 5. MASTER SWITCH OFF.

ENGINE FIRE IN FLIGHT (IDENTIFIED)

1

- 1. POWER lever IDLE.
- 2. <u>PROP lever FEATHER</u>.
- 3. CONDITION lever FUEL CUTOFF.
- 4. FIRE PULL handle Pull.
- 5. Fire extinguisher Actuate as required.
- 6. Engine cleanup Perform.

FUSELAGE FIRE

- 1. Fight the fire.
- 2. Land as soon as possible if fire continues.

WING FIRE

- 1. Perform engine shutdown on affected side.
- 2. Land as soon as possible if fire continues.



ELECTRICAL FIRE

- 1. Crew oxygen masks As required.
- 2. Passenger oxygen As required.
- MASTER SWITCH OFF. (Visual conditions only).
- 4. All nonessential electrical equipment OFF.
- 5. **BATT** switch **ON**.
- Generator switches (individually) RESET, then ON.
- Circuit breakers Check for indication of defective circuit.
- Essential electrical equipment **ON** (individually until fire source is isolated).
- 9. Land as soon as practicable.

SMOKE AND FUME ELIMINATION

- 1. Crew oxygen masks 100% and on.
- 2. Passenger oxygen ON.
- 3. BLEED AIR VALVE switches ENVIRO OFF.
- 4. VENT BLOWER switch AUTO.
- 5. AFT VENT BLOWER switch OFF.
- 6. CABIN AIR/TEMP MODE switch OFF.
- If smoke and fumes are not eliminated, CABIN PRESS switch – DUMP.
- Passenger oxygen masks Check. Confirm that all passengers are receiving supplemental oxygen.

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9. Engine oil pressure – Monitor.



FUEL SYSTEM

FUEL PRESS WARNING LIGHT

- 1. STANDBY PUMP switch ON.
- 2. FUEL PRESS light out Check.
- 3. FUEL PRESS light still on Record unboosted time.

NO FUEL TRANSFER CAUTION LIGHT ILLUMINATED

- AUX TRANSFER switch (affected side) -OVERRIDE.
- 2. Auxiliary fuel quantity Monitor.
- 3. **AUX TRANSFER** switch (after respective auxiliary fuel has completely transferred) **AUTO**.

NACELLE FUEL LEAK

- 1. Perform engine shutdown.
- 2. FIRE PULL handle Pull.
- 3. Land as soon as practicable.

FUEL CROSSFEED

- 1. AUX TRANSFER switches AUTO.
- 2. STANDBY PUMPS OFF.
- 3. **CROSSFEED** switch As required.
- 4. **FUEL CROSSFEED** annunciator illuminated Check.

E-11

5. FUEL PRESS light extinguished – Check.

6. Fuel quantity - Monitor.

NAC LOW LIGHT ILLUMINATED

- 1. Usable fuel remaining Confirm.
- 2. Land as soon as possible.

ELECTRICAL SYSTEM EMERGENCIES

DC GEN LIGHT ILLUMINATED

- 1. GEN switch OFF, RESET, then ON.
- IF THE GENERATOR DOES NOT RESET:
- 2. GEN switch (no reset) OFF.
- 3. Operating loadmeter 100% maximum.

BOTH DC GEN LIGHTS ILLUMINATED

- 1. All nonessential equipment Off.
- 2. Land as soon as practicable.

EXCESSIVE LOADMETER INDICATION (OVER 100%)

- 1. Loadmeter Monitor.
- 2. BATT switch OFF (monitor loadmeter).
- IF LOADMETER STILL INDICATES ABOVE 100%:
- 3. Nonessential electrical equipment off.
- IF LOADMETER INDICATES 100% OR BELOW:
- 4. BATT switch ON.





INVERTER LIGHT ILLUMINATED

1. Affected **INVERTER** switch – **OFF**.

INST AC LIGHT ILLUMINATED

- 1. N_1 and TGT indications Check.
- 2. Other engine instruments Monitor.

CIRCUIT BREAKER TRIPPED

1. **BUS FEEDER** breaker tripped – Do not reset.

- 2. Nonessential circuit Do not reset.
- 3. Essential circuit Reset once.

BATTERY CHARGE LIGHT ILLUMINATED

- 1. Loadmeter Check and note indication.
- 2. BATT switch OFF.
- Loadmeter Check. If loadmeter indicates less than 2.5% change (one needle width), turn BATT switch ON and monitor for increasing load. If load continues to increase, turn BATT switch OFF.
- 4. **BATT** switch (landing gear/flap extension only) **ON**.

AVIONICS MASTER POWER SWITCH FAILURE

1. AVIONICS MASTER CONTR circuit breaker – Pull.

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EMERGENCY DESCENT

- 1. POWER levers IDLE.
- 2. <u>PROP levers HIGH RPM</u>.



- 3. FLAPS APPROACH.
- 4. <u>GEAR DN</u>.
- 5. Airspeed 181 KIAS maximum.

LANDING EMERGENCIES

LANDING GEAR UNSAFE INDICATION C

- 1. LDG GEAR CONTR switch DN.
- 2. LANDING GEAR RELAY and LANDING GEAR IND circuit breakers Check in.
- 3. GEAR DOWN lights Check.
- IF INDICATOR REMAINS UNSAFE:
- 4. Landing gear emergency extension Perform.

LANDING GEAR UNSAFE INDICATION D2

- 1. LDG GEAR CONTROL switch Check DN.
- 2. LANDING GEAR CONTROL and LANDING GEAR IND circuit breakers Check in.
- 3. GEAR DOWN lights illuminated Check.
- IF INDICATOR REMAINS UNSAFE:
- 4. Landing gear emergency extension Perform.

LANDING GEAR EMERGENCY EXTENSION

- 1. Airspeed 130 KIAS.
- 2. LANDING GEAR RELAY circuit breaker Out.

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3. LDG GEAR CONTR switch - DN.



 Landing gear alternate engage handle – Lift and turn clockwise to the stop.

- 5. Alternate landing gear extension handle Pump.
- 6. GEAR DOWN lights illuminated Check.

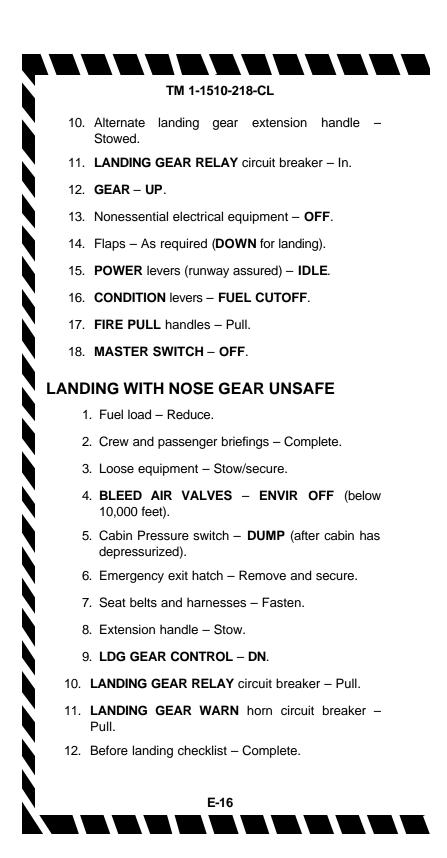
LANDING GEAR EMERGENCY EXTENSION

- 1. Airspeed Below 181 KIAS.
- 2. LANDING GEAR CONTR circuit breaker Pull.
- 3. LDG GEAR CONTROL switch DN.
- 4. Alternate extension lever Unstow.
- 5. Alternate extension lever Pump up and down until the three green **GEAR DOWN** lights illuminate or resistance is felt.
- 6. Alternate extension lever As required.

GEAR-UP LANDING (GEAR UP OR UNLOCKED)

- 1. Fuel load Reduce.
- 2. Personnel emergency briefing -Completed.
- 3. Loose equipment Stowed.
- 4. BLEED AIR VALVES ENVIRO OFF.
- 5. CABIN PRESS switch DUMP.
- 6. CABIN SIGNS switch BOTH.
- 7. Cabin emergency hatch Remove and stow.
- 8. Seat belts and harnesses Secured.
- O 9. Landing gear alternate engage handle -Disengaged.







AFTER TOUCHDOWN:

- 13. POWER levers IDLE.
- 14. **PROP** levers **FEATHER**.
- 15. CONDITION levers FUEL CUTOFF.

AFTER STOPPING:

16. Fuel **FIREWALL SHUTOFF VALVES CLOSED**. 17. MASTER SWITCH – OFF.

LANDING WITH ONE MAIN GEAR UNSAFE

1. Retract the gear and make a gear up landing.

IF THE GEAR WILL NOT RETRACT:

- 2. Fuel load Reduce.
- 3. Personnel emergency briefing Completed.
- 4. Loose equipment Stowed.
- 5. BLEED AIR VALVES ENVIRO OFF.
- 6. CABIN PRESS switch DUMP.
- 7. CABIN SIGNS switch BOTH.
- 8. Cabin emergency hatch Remove and stow.
- 9. Seat belts and harnesses Secured.
- 10. Nonessential electrical equipment OFF.
- 11. Touchdown On safe main gear first.
- 12. POWER levers IDLE.
- 13. CONDITION levers FUEL CUTOFF.

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14. FIRE PULL handle - Pull.

15. MASTER SWITCH - OFF.

CRACKED WINDSHIELD

INTERNAL CRACK

- 1. Descend Below 25,000 feet.
- Cabin Pressure Reset pressure differential to 4 psi or less within 10 minutes.

CRACKED CABIN WINDOW

- 1. Crew oxygen masks **100%** and on, if above 10,000 feet.
- 2. CABIN SIGNS switch BOTH.
- Passenger oxygen ON and checked, if above 10,000 feet.
- 4. Cabin pressure Depressurize.
- 5. Land as soon as practicable.

DITCHING

- 1. Radio calls/transponder As required.
- 2. Personnel emergency briefing As required.
- 3. BLEED AIR VALVES ENVIRO OFF/PNEU ONLY.
- 4. CABIN PRESS switch DUMP.
- 5. CABIN SIGNS switch BOTH.
- 6. Cabin emergency hatch Remove and stow.

- 7. Seat belts and harnesses Secured.
- 8. GEAR UP.
- 9. FLAPS DN.



- 10. Nonessential electrical equipment OFF.
- 11. Approach Normal, power on.
- 12. Emergency lights As required.

FLIGHT CONTROLS MALFUNCTION

UNSCHEDULED RUDDER BOOST ACTIVATION

- 1. RUDDER BOOST OFF.
- IF CONDITION PERSISTS:
- 2. RUDDER BOOST circuit breaker Pull.
- 3. BLEED AIR VALVE OFF (Below 10,000 feet).
- 4. Rudder trim Adjust.

UNSCHEDULED ELECTRIC ELEVATOR TRIM

1. Control wheel disconnect switch - Depress.

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- 2. ELEV TRIM switch OFF.
- 3. ELEC TRIM circuit breaker OUT.

PERFORMANCE CHECKS

FIRE EXTINGUISHER PRESSURE

A gauge, calibrated in psi, is mounted on each supply cylinder for determining the level of charge and should be checked during preflight. Refer to Table P-1.

Table P-1. Engine Fire Extinguisher Gauge Pressure

Temp °C	-40	-29	-18	-06	04	16	20	38	48
	190	220	250	290	340	390	455	525	605
PSI	to								
	240	275	315	365	420	480	550	635	730

CREW/PASSENGER BRIEFING

- 1. Crew Introduction.
- 2. Equipment.
 - a. Personnel to include ID tags.
 - b. Professional (medical equipment, etc.).
 - c. Survival.
- 3. Flight Data.
 - a. Route.
 - b. Altitude.
 - c. Time en route.
 - d. Weather.
- 4. Normal Procedures.
 - a. Entry and exit of aircraft.

- b. Seating and seat position.
- c. Seat belts.
- d. Movement in aircraft.
- e. Internal communications.
- f. Security of equipment.
- g. Smoking.
- h. Oxygen.
- i. Refueling.
- j. Weapons and prohibited items.
- k. Protective masks.
- I. Toilet.
- m. Polarized windows.
- 5. Emergency Procedures.
 - a. Emergency exits.
 - b. Emergency equipment.
 - c. Emergency landing / ditching procedures.

FUEL PUMPS/CROSSFEED OPERATION

- 1. FIRE PULL handles Pull.
- 2. STANDBY PUMP switches ON.
- 3. Battery Switch **ON**.
- 4. **#1** and **#2 FUEL PRESS** warning lights Illuminated.
- 5. FIRE PULL handles In.

- 6. **#1** and **#2 FUEL PRESS** warning lights Extinguished.
- 7. **STANDBY PUMP** switches **STANDBY PUMP**.
- 8. **#1** and **#2 FUEL PRESS** warning lights Illuminated.
- CROSSFEED Check. Check system operation by activating switch momentarily left then right, noting that #1 and #2 FUEL PRESS warning lights extinguish and that the FUEL CROSSFEED advisory light illuminates as switch is energized.
- 10. **BATT** switch **OFF** (GPU start).

STALL AND GEAR WARNING SYSTEM

- STALL WARN TEST switch TEST. Check that warning horn sounds.
- LDG GEAR WARN TEST switch TEST. Check that warning horn sounds and that the two LDG GEAR CONTR handle warning lights illuminate.

AC/DC POWER

- 1. AC frequency 394 406 Hz.
- 2. AC voltage 104 124 Vac.
- 3. DC load 85% maximum per generator.
- 4. DC voltage 28 28.5 Vdc.

ELECTRIC ELEVATOR TRIM, AUTOPILOT

- Pilot and copilot PITCH TRIM switches Press to NOSE UP and NOSE DN positions, singularly and in pairs. Check that trim wheel moves in proper direction and operates only when trim switches are pressed in pairs. Any deviation requires that electric elevator trim be turned off and flight conducted using only manual trim.
- DISC TRIM switch Press to second detent and verify that electric trim disconnects and that ELEC TRIM light extinguishes.
- 3. Flight Director (FD) and Radio Magnetic Indicator (RMI) warning flags masked Check.

NOTE

Since the pressure of airflow that normally opposes movement of control surfaces is absent during preflight check, it is possible to get a hard over control surface deflection if an autopilot command is allowed to remain active for any appreciable length of time. Move turn knob and pitch thumbwheel only enough to check operation, then return them to the center position.

- 4. Select HDG mode Check.
- 5. Horizontal Situation Indicator (**HSI**) heading marker under lubber line and vertical needle centered Set.
- 6. Engage autopilot and check controls stiff Check.
- Move HSI heading marker 10° left and right and verify that FD and control wheels respond in the appropriate direction – Check.

- Press AP/YD disengage switch to first detent and verify that autopilot disengages and flight controls are free – Check.
- 9. Elevator Trim Check on.
- 10. Engage autopilot Check.
- Command 5° trim UP with AP pitch wheel and verify that manual trim wheel moves nose UP and AP trim light indicates UP trim – Check.
- 12. Press **PITCH TRIM** switch **NOSE DN** and verify that autopilot disengages and **AUTO PILOT TRIM FAIL** and **MASTER WARNING** lights illuminate Check.

NOTE

The AP TRIM FAIL annunciator will extinguish by pressing the AP/YD disconnect button on the control wheel to the first detent.

- 13. Engage autopilot Check.
- 14. Move **HSI** heading marker to command a bank on FD Check.
- 15. Press go-around switch and verify that **GA** light illuminates, autopilot disengages, and that FD commands a wings level, 7° nose-up attitude Check.
- Press **TEST** switch (pilot's HSI) and verify that attitude display indicates an additional 10° pitch up and 20° right bank – Check.

AUTOPILOT TRIM FAIL SYSTEM C D1

1. Engage autopilot command **DN** with **AP** pitch wheel and engage **AUTO PILOT TRIM TEST** switch when elevator trim wheel starts to rotate.

 Verify that autopilot disengages and AP TRIM FAIL and MASTER WARNING lights illuminate within 10 seconds.

AUTOMATIC FLIGHT CONTROL SYSTEM D2

1. Altitude alert.

NOTE

Pause a few seconds after each step to allow time for the proper indications.

- a. Set alert controller more than 1000 feet above altitude indicated on pilot's altimeter. The pilot's altimeter alert light should be extinguished.
- b. Decrease the alert controller to within 1000 feet of the pilot's altimeter setting. The alert light should illuminate.
- c. Decrease the controller to less than 250 feet above the pilot's altimeter setting. The alert light should extinguish.
- d. Increase the controller to 300 ± 50 feet above the pilot's altimeter indication and check that the alert light illuminates.
- e. Set the desired altitude.
- 2. Autopilot.
 - a. Autopilot controller **UP TRIM**, **DN TRIM** annunciators Check not illuminated.

CAUTION

A steady illumination of UP TRIM or DN TRIM annunciator indicates that the automatic synchronization is not functioning and the autopilot should not be engaged.

- b. Turn knob Center.
- c. Elevator trim control switch ON.
- d. Control wheel Hold to mid travel.
- e. AP button Press. AP ENGAGE and YD ENGAGE annunciators on autopilot controller will flash. Servo clutches will engage. FD flag on ADI should be in view.
- f. Control movement Check.
 - Rudder pedals Overpower slowly. YD ENGAGE annunciator stops flashing.
 - (2) Control wheel Overpower slowly in both pitch and roll axis. AP ENGAGE annunciator stops flashing. FD flag on ADI retracts.

WARNING

If autopilot or yaw damper disengages during overpower test or if AP ENGAGE or YD ENGAGE annunciator continues to flash, the system is considered non-operative and should not be used. The elevator trim system must not be forced beyond the limits indicated on the elevator trim tab indicator.

- g. Elevator trim follow-up Check.
 - Control wheel Hold aft of mid travel. Trim wheel should run nose down after approximately 3 seconds. Trim down annunciator should illuminate after approximately 8 seconds.

- (2) Control wheel Hold forward of mid travel. Trim wheel should run nose up after approximately 3 seconds, trim up annunciator should illuminate after approximately 8 seconds, and AP TRIM FAIL annunciator and MASTER WARNING lights should illuminate after approximately 15 seconds.
- h. AP/YD & TRIM DISC Button Press through second level. Autopilot and yaw damper should disengage and ELEC TRIM OFF annunciator should illuminate. AP ENG and YD ENG annunciators on instrument panel should flash five times.
- i. Elevator trim control switch **OFF**, then on. **ELEC TRIM OFF** annunciator should extinguish.
- j. **AP** Re-engage and overpower another time.
- k. Turn controller Check that control wheel follows in each applied direction, then center.
- Pitch wheel Check that trim responds to pitch wheel movement. UP TRIM and DN TRIM annunciators may illuminate.
- m. Heading bug Center and engage HDG. Check that control follows a turn in each direction.
- n. Disengage AP by selecting GA. Check that AP disengages and FD commands 7° nose up, wings level attitude.

AUTOFEATHER/AUTO IGNITION

- 1. Condition levers LOW IDLE.
- 2. ENG AUTO IGN Switches On.

- 3. AUTOFEATHER switch Hold to TEST.
- POWER levers Advance until IGN ON lights are extinguished and AUTOFEATHER lights are illuminated, approximately 22% torque.
- 5. **#1 POWER** lever Retard and check.
 - a. At 16 21% torque #2 **AUTOFEATHER** light out, #1 **IGN ON** light illuminated.
 - b. At 9 14% torque Both **AUTOFEATHER** lights out (propeller starts to feather).
- 6. #1 POWER lever Approximately 22% torque.
- 7. Repeat steps 1 through 6 for #2 engine.
- POWER levers IDLE (both AUTOFEATHER lights out, neither propeller feathers and both IGN ON lights illuminated).
- 9. AUTOFEATHER switch ARMED.
- 10. ENG AUTO IGN switches OFF.

OVERSPEED GOVERNORS

- 1. RUDDER BOOST switch On.
- 2. PROP levers HIGH RPM.
- PROP GOV TEST switch Hold in TEST position.
- Left **POWER** lever Increase until propeller is stabilized at 1830 – 1910 RPM. Continue to increase until rudder movement is noted. (Observe **ITT** and torque limits.)
- 5. **POWER** lever Retard to **IDLE**.
- 6. Repeat steps 3, 4, and 5 for the right engine.

PRIMARY GOVERNORS

1. POWER - Set 1800 RPM.

NOTE

Reduce PROP levers gently to the detent to prevent the PROP RPM dropping abruptly below 1600 RPM.

- 2. PROP levers aft gently to detent Set.
- 3. Propeller RPM 1600 to 1640 Check.
- 4. PROP levers HIGH RPM.

ICE VANES

- ICE VANE switches EXTEND. Verify torque drop, TGT increase, illumination of VANE EXT light and visually confirm the bypass doors are open. Maximum allowable time for the complete operation is 15 seconds – Check.
- ICE VANE switches RETRACT. Verify return to original torque and TGT, VANE EXT light extinguished, and visually confirm the bypass doors are closed. Maximum allowable time for the complete operation is 15 seconds – Check.

ANTI-ICE AND DEICE SYSTEMS

- WSHLD ANTI-ICE switches NORMAL. Check PILOT and COPILOT (individually) for loadmeter rise, then OFF.
- PROP deice switches Check when MANUAL is selected. Note rise on DC loadmeter. When AUTO mode is selected, monitor prop ammeter for 2 minutes and ensure the indicator remains in the normal operating range for entire time.

 DEICE switch – SINGLE CYCLE AUTO. Check for a drop in pneumatic pressure and wing-deice boots inflation and after 6 seconds for a second drop in pneumatic pressure.

PNEUMATIC SYSTEM

- 1. LEFT BLEED AIR VALVE switch OFF.
- 2. Pneumatic pressure 12 to 20 psi Check.
- 3. L BL AIR OFF light on Check.
- 4. **RIGHT BLEED AIR VALVE** switch **OFF**.
- 5. L & R BL AIR OFF, L & R BL AIR FAIL lights and MASTER WARNING light ON.
- 6. LEFT BLEED AIR VALVE switch OPEN.
- L BL AIR OFF and L & R BL AIR FAIL lights OFF and pneumatic pressure at 12 to 20 psi – Check.
- 8. RIGHT BLEED AIR VALVE switch OPEN.
- 9. **R BL AIR OFF** light off Check.

PRESSURIZATION SYSTEM

- 1. CABIN DOOR caution light extinguished Check.
- 2. Vent windows closed Check.
- 3. BLEED AIR VALVE switches OPEN.
- 4. Cabin altitude 500 feet lower than field pressure altitude Set.
- 5. CABIN PRESS switch TEST (hold).
- CABIN CLIMB gauge descending indication Check, then release TEST switch.

- 7. ACFT ALT Set to pressure altitude plus 200 feet.
- 8. RATE control Set between 9 and 12 o'clock.

DEPARTURE BRIEFING

- 1. ATC clearance Review.
 - a. Routing.
 - b. Initial altitude.
- 2. Departure procedure Review.
 - a. Named departure procedure.
 - b. Obstacle clearance departure procedure / noise abatement procedure.
 - c. Visual flight rules departure route.
- 3. Copilot duties Review.
 - a. Adjust takeoff power.
 - b. Monitor engine instruments.
 - c. Power check at 65 knots.
 - d. Call out engine malfunctions.
 - e. Tune/identify all nav/comm radios.
 - f. Make all radio calls.
 - g. Adjust transponder and radar as required.
 - h. Complete flight log during flight and note altitudes and headings.
 - i. Note departure time.
- 4. TOLD card Review.

- a. Takeoff power.
- b. V_1/V_r .
- c. V_2 +10 (climb to 1500 feet AGL).
- d. V_2/V_{yse} .

ARRIVAL BRIEFING

- 1. Weather/altimeter setting.
- 2. Airfield/facilities Review.
 - a. Field elevation.
 - b. Runway length.
 - c. Runway condition.
- 3. Approach procedure Review.
 - a. Approach plan/profile.
 - b. Altitude restrictions/VDP.
 - c. Missed approach.
 - (1) Point.
 - (2) Time.
 - (3) Intentions.
 - d. Decision height or minimum descent altitude.
 - e. Lost communications.
- 4. Backup approach/frequencies.
- 5. Copilot's duties Review.
 - a. Nav/comm set-up.
 - b. Monitor altitude and airspeeds.

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- c. Monitor approach.
- d. Call out visual/field in sight.
- 6. Landing performance data Review.
 - a. Approach speed.
 - b. Runway required.
- 7. Passenger briefing As required.

OPERATOR'S AND CREWMEMBER'S CHECKLIST

PART II

ARMY C-12T AIRCRAFT NSN 1501-01-470-0220

NORMAL PROCEDURES

BEFORE EXTERIOR CHECK

- *1. Forms/publications Check.
- *2. Oxygen system Check.
- *3. Flight controls Unlocked and checked.
- *4. Parking brake As required.
 - 5. Manual trim Check and set to zero.
- *6. LDG GEAR CONTR DN.
- *7. EFIS POWER OFF.
- ★ 8. Fuel pumps/crossfeed operation Check.
 - *9. Fuel gauges Check quantity.
- ★* 10. EFIS POWER and INVERTERS ON, check, OFF.
 - 11. Subpanel Check and set.
 - 12. Lighting and Heating Systems Check.
 - 13. **BATT** switch **OFF**.
 - 14. Galley power switches Off.
 - 15. Toilet Check.
 - 16. Emergency equipment Check.

FUEL SAMPLE AND OIL CHECK

1. Fuel sample - Check.

LEFT WING, AREA 1

- 1. Left wing Check.
- 2. Left main landing gear Check.
- 3. Left engine and propeller Check.

- 4. Left wing center section Check.
- 5. Fuselage underside Check.

NOSE SECTION, AREA 2

1. Nose section – Check.

RIGHT WING, AREA 3

- 1. Right wing center section Check.
- 2. Right engine and propeller Check.
- 3. Right main landing gear Check.
- 4. Right wing Check.

FUSELAGE RIGHT SIDE, AREA 4

1. Fuselage right side – Check.

EMPENNAGE, AREA 5

1. Empennage - Check.

FUSELAGE LEFT SIDE, AREA 6

- 1. Fuselage left side Check.
- *2. Chocks and tiedowns Removed.

INTERIOR CHECK

- 1. Cargo/loose equipment Check secure.
- \star 2. Cabin door Locked and checked.
 - 3. Cargo door Locked and checked.
 - 4. Emergency exit Check.
- ★ 5. Crew/passenger briefing Complete.

BEFORE STARTING ENGINES

- *1. Parking brake Set.
- *2. Oxygen system Set.
 - 3. Circuit breakers Check.
- *4. Overhead panel Check.
- *5. Fuel panel switches Check.
 - 6. Magnetic compass Check.
 - 7. CLOCK and MAP lights OFF.
- *8. Pedestal controls Set.
 - 9. Lower console switches Set.
- 10. Gear ratchet handle Stowed.
- 11. Free air temperature gauge Check.
- 12. Pilot's instrument panel Check and set.
- 13. Copilot's instrument panel Check and set.

FIRST ENGINE START (BATTERY START)

- 1. BATT switch ON.
- 2. EXTERIOR LIGHTS As required.
- 3. Propeller Clear.
- 4. Engine Start.
- 5. Engine and systems instruments Check.
- 6. CONDITION lever HIGH IDLE.
- 7. GEN switch RESET, then ON.

SECOND ENGINE START (BATTERY START)

- 1. First engine generator load 50% or less **GEN** switch **OFF**.
- 2. Propeller area Clear.
- 3. Engine Start.
- 4. Engine and systems instruments Check.
- 5. BATTERY CHARGE light on Check.
- 6. **INVERTER** switches **ON**, check, **INVERTER** lights **OFF**.
- 7. Second engine **GEN** switch **RESET**, then **ON**.
- 8. CONDITION levers As required.
- 9. Red anticollision light Reset.

ABORT START

- 1. CONDITION lever FUEL CUTOFF.
- 2. IGNITION AND ENGINE STARTER switch STARTER ONLY.
- 3. ITT/TGT Monitor for drop in temperature.
- 4. IGNITION AND ENGINE START switch OFF.

ENGINE CLEARING

- 1. CONDITION lever FUEL CUTOFF.
- IGNITION AND ENGINE START switch OFF (1-minute minimum).
- 3. IGNITION AND ENGINE START switch STARTER ONLY (15 seconds minimum,

40 seconds maximum).

4. IGNITION AND ENGINE START switch - OFF.

*FIRST ENGINE START (GPU START)

- 1. BATT switch OFF.
- 2. GPU Connect.
- 3. EXTERNAL POWER advisory light On.
- 4. BATT switch ON.
- 5. EXTERIOR LIGHTS switches As required.
- 6. Propeller Clear.
- 7. Engine Start.
- 8. Engine and systems instruments Check.
- 9. CONDITION lever HIGH IDLE.
- 10. GPU disconnect As required.
- 11. GEN switch RESET then ON.
- 12. BATTERY CHARGE light Monitor.

*SECOND ENGINE START (GPU START)

- 1. Propeller area Clear.
- 2. Engine Start.
- 3. Engine and systems instruments Check.
- 4. Right **PROP** lever **FEATHER**.
- 5. GPU Disconnect.
- 6. Right **PROP** lever **HIGH RPM**.
- 7. INVERTER switches ON, check INVERTER

lights OFF.

- 8. GEN switches RESET, then ON.
- 9. CONDITION levers As required.
- 10. Red anticollision light Reset.

BEFORE TAXIING

- *1. AC/DC power Check.
- *2. AVIONICS MASTER POWER ON.
- *3. EFIS POWER ON.
- *4. CABIN AIR/TEMP MODE and CABIN AIR/TEMP switches Set as desired.
- *5. BLEED AIR VALVES As required.
- *6. BRAKE DEICE switch As required.
 - 7. Avionics Check and set as required.
- *8. TCAS TEST and set.
 - 9. FLAPS Check.
- *10. Altimeters Set and Check.

***TAXIING**

- 1. Brakes Check.
- 2. Flight instruments Check.

ENGINE RUNUP

- 1. Parking brake As required.
- 2. Propeller feathering Check.
- ★ 3. AUTOFEATHER/AUTO IGN switch Check as required.

- ★ 4. Overspeed governors and rudder boost Check as required.
- ★ 5. Primary governors Check as required.
- ★ 6. Engine ice vanes Check.
 - 7. CONDITION levers HIGH IDLE.
 - 8. POWER levers IDLE.
- ★ 9. Anti-ice/deice systems Check.
- \star 10. Vacuum and pneumatic system Check.
- ★ 11. Automatic flight control system Check as required.
 - 12. Electric elevator trim Check.
- ★* 13. Pressurization Check and set.
 - 14. CONDITION levers As required.
 - 15. Ground Proximity Altitude Advisory System (GPAAS) Check.

***BEFORE TAKEOFF**

- 1. AUTOFEATHER switch ARM.
- 2. BLEED AIR VALVES As required.
- 3. **FUEL** panel Check fuel quantity and switches positions.
- 4. Flight and engine instruments Check.
- 5. CABIN ALT Set PA + 200 feet.
- 6. Annunciator panels Check.
- 7. PROP levers HIGH RPM.
- 8. FLAPS As required.
- 9. Trim Set.

- 10. Avionics Set.
- 11. Flight Controls Check.
- ★ 12. Departure briefing Complete.
 - 13. CABIN SIGNS switch As required.

*LINE UP

- 1. ICE & RAIN switches As required.
- 2. Altitude alerter Check.
- 3. Transponder/TCAS/Wx radar As required.
- 4. ENG AUTO IGN ARM.
- 5. Lights As required.
- 6. CONDITION levers HIGH IDLE.
- 7. POWER Stabilized 27% minimum.

AFTER TAKEOFF

- 1. GEAR UP.
- 2. FLAPS (105 KIAS) UP.
- 3. LANDING/TAXI lights OFF.
- 4. Climb power Set.

CLIMB

- 1. YAW DAMP As required.
- 2. Cabin pressurization Check.
- 3. AUTOFEATHER switch As required.
- 4. BRAKE DEICE switch As required.

- 5. WSHLD ANTI-ICE As required.
- 6. Wings and nacelles Check.
- 7. TCAS Set range.

CRUISE

- 1. POWER Set.
- 2. ICE & RAIN switches As required.
- 3. CABIN SIGNS switch As required.
- 4. Auxiliary fuel gauges Monitor.
- 5. Altimeters Check.
- 6. Engine instruments- Check.
- 7. RECOG lights As required.
- 8. TCAS Set for en route.

DESCENT - ARRIVAL

- 1. Cabin pressurization Set.
- 2. CABIN SIGNS switch As required.
- 3. ICE & RAIN switches As required.
- 4. WSHLD ANTI-ICE As required.
- 5. RECOG lights ON.
- 6. Radar altimeter As required.
- 7. Altimeters Set to current altimeter setting.
- 8. TCAS Set as required.
- ★ 9. Arrival briefing Complete.

DESCENT – MAXIMUM RATE (CLEAN)

- 1. Cabin pressurization Set.
- 2. CABIN SIGNS switch As required.
- 3. POWER levers IDLE.
- 4. **PROP** levers **HIGH RPM**.
- 5. **GEAR UP**.
- 6. FLAPS UP.
- 7. Airspeed V_{mo} maximum.
- 8. ICE & RAIN switches As required.
- 9. **RECOG** lights As required.

DESCENT – MAXIMUM RATE (LANDING CONFIGURATION)

- 1. Cabin pressurization Set.
- 2. CABIN SIGNS switch As required.
- 3. POWER levers IDLE.
- 4. PROP levers HIGH RPM.
- 5. FLAPS APPROACH.
- 6. GEAR DN.
- 7. Airspeed 181 KIAS maximum.
- 8. ICE & RAIN switches As required.
- 9. **RECOG** lights As required.

APPROACH

- 1. EHSI NAV SOURCE As required.
- 2. TCAS Set as required.

BEFORE LANDING

- 1. CABIN SIGNS switch BOTH.
- 2. AUTOFEATHER switch ARM.
- 3. BRAKE DEICE switch As required.
- 4. **PROP** levers As required.
- 5. FLAPS (below 199 KIAS) APPROACH.
- 6. GEAR (below 181 KIAS) DN, confirm.
- 7. LANDING LIGHTS As required.
- 8. CONDITION levers HIGH IDLE.
- 9. TCAS Set as required.

LANDING

- 1. AP & YD Disengaged.
- 2. GEAR DOWN lights Check/confirm.
- 3. PROP levers HIGH RPM.

TOUCH AND GO LANDING

- 1. PROP levers HIGH RPM.
- 2. FLAPS As required.
- 3. Trim Set.
- 4. Power stabilized Check 27% minimum.

5. Takeoff power - Set.

GO AROUND/MISSED APPROACH

- 1. Power As required.
- 2. GEAR UP.
- 3. FLAPS APPROACH.
- 4. FLAPS (105 KIAS) UP.
- 5. LANDING/TAXI LIGHTS OFF.
- 6. Climb power Set.
- 7. YAW DAMP As required.
- 8. BRAKE DEICE switch Off.

AFTER LANDING

- 1. CONDITION levers As required.
- 2. AUTO IGN Off.
- 3. ICE & RAIN switches Off.
- 4. FLAPS UP.
- 5. **XPNDR** As required.
- 6. Radar As required.
- 7. Lights As required.

ENGINE SHUTDOWN

- 1. BRAKE DEICE switch Off.
- 2. Parking brake Set.
- 3. LANDING/TAXI lights OFF.

- 4. EFIS POWER OFF.
- 5. INVERTERS OFF.
- 6. AUTOFEATHER switch OFF.
- 7. CABIN AIR/TEMP MODE switch OFF.
- 8. VENT and AFT VENT BLOWER switch AUTO.
- 9. BATT condition Check.
- 10. ITT/TGT Check.
- 11. CONDITION levers FUEL CUTOFF.
- 12. PROP levers FEATHER.
- 13. AVIONICS MASTER POWER OFF.
- 14. MASTER PANEL LIGHTS Off.
- 15. EXTERIOR LIGHTS OFF.
- 16. MASTER SWITCH Off.
- 17. Key lock switch OFF.
- 18. Oxygen system OFF.
- 19. Chocks As required.
- 20. Parking brake As required.
- 21. Control locks As required.

BEFORE LEAVING AIRCRAFT

- 1. Wheel chocks As required.
- 2. Parking brake As required.
- 3. Flight controls As required.
- 4. OVERHEAD FLOOD LIGHT OFF.

- 5. STANDBY PUMPS OFF.
- 6. MAP lights OFF.
- 7. Windows As required.
- 8. Emergency exit lock As required.
- 9. Galley power switches OFF.
- 10. Aft cabin light OFF.
- 11. Door light **OFF**.
- 12. Walk-around inspection Complete.
- 13. Aircraft forms Complete.
- 14. Aircraft secured Check.



EMERGENCY PROCEDURES

ENGINE MALFUNCTION

ENGINE MALFUNCTION BEFORE V (ABORT)

- 1. POWER levers IDLE.
- 2. Braking As required.

ENGINE MALFUNCTION AFTER V1

- 1. GEAR (positive climb) UP.
- 2. POWER As required.
- 3. FLAPS (105 KIAS) UP.

IF THE PROP DID NOT FEATHER, PERFORM STEP 4.

4. PROP (dead engine) - FEATHER.

ONCE THE PROP IS FEATHERED, PERFORM STEPS 5 THROUGH 8.

- O 5. <u>TCAS Set TA</u>.
 - 6. LANDING/TAXI LIGHTS OFF.
 - 7. BRAKE DEICE OFF.
 - 8. Engine cleanup Perform.

ENGINE MALFUNCTION DURING FLIGHT

- 1. Autopilot/yaw damp Disengage.
- 2. **POWER** As required.
- 3. Dead engine Identify.
- 4. PROP lever (dead engine) FEATHER.



- 5. GEAR As required.
- 6. FLAPS As required.
- O 7. <u>TCAS Set TA</u>.
 - 8. Power Set for single-engine cruise.
 - 9. Engine cleanup Perform.

ENGINE MALFUNCTION DURING FINAL APPROACH

- 1. POWER As required.
- 2. <u>GEAR DN</u>.

ENGINE MALFUNCTION (SECOND ENGINE)

- 1. <u>Airspeed As required</u>.
- 2. PROP lever As required.

ENGINE SHUTDOWN IN FLIGHT

- 1. POWER lever IDLE.
- 2. PROP lever FEATHER.
- 3. CONDITION lever FUEL CUTOFF.
- 4. Engine cleanup Perform.

ENGINE CLEANUP

- 1. CONDITION lever FUEL CUTOFF.
- 2. ENG AUTO IGN OFF.
- 3. AUTOFEATHER switch OFF.
- 4. GEN switch OFF.
- 5. **PROP SYNC** switch **OFF**.



ENGINE RESTART DURING FLIGHT (USING STARTER)

- 1. CABIN AIR/TEMP MODE switch OFF.
- 2. Electrical load Reduce to minimum.
- 3. FIRE PULL handle In.
- 4. POWER lever IDLE.
- 5. **PROP** lever **FEATHER**.
- 6. CONDITION lever FUEL CUTOFF.
- 7. TGT (operating engine) 700 °C or less.
- 8. Engine Start.
- 9. GEN switch RESET, then ON.
- 10. Engine cleanup Perform if engine restart is unsuccessful.
- 11. CABIN AIR/TEMP MODE switch As required.
- 12. Electrical equipment As required.
- 13. ENG AUTO IGN switch ARM.
- 14. **PROP SYNC** switch As required.
- 15. **POWER** As required.

ENGINE RESTART DURING FLIGHT (NOT USING STARTER)

- 1. CABIN AIR/TEMP MODE switch OFF.
- 2. Electrical load Reduce to minimum.
- 3. GEN switch (affected engine) OFF.
- 4. FIRE PULL handle In.
- 5. POWER lever IDLE.



- 6. PROP lever HIGH RPM.
- 7. CONDITION lever FUEL CUTOFF.
- 8. Airspeed 140 KIAS minimum.
- 9. Altitude Below 20,000 feet.
- 10. ENG AUTO IGN switch ARM.
- 11. CONDITION lever LOW IDLE.
- 12. TGT 1000° 5 seconds maximum.
- 13. Oil pressure Check.
- 14. GEN switch RESET, then ON.
- Engine cleanup Perform if engine restart is unsuccessful.
- 16. CABIN AIR/TEMP MODE switch As required.
- 17. Electrical equipment As required.
- 18. Propellers Synchronized.
- 19. POWER As required.

LOW OIL PRESSURE

- Torque 49% maximum. Oil pressure less than 105 psi below 21,000 feet or 85 psi 21,000 feet and above.
- 2. Oil pressure below 60 psi. Perform engine shutdown or land as soon as practicable using minimum power to ensure safe arrival.

CHIP DETECTOR WARNING LIGHT

If a **CHIP DET** warning light illuminates and safe single-engine flight can be maintained, perform engine shutdown.





DUCT OVERTEMP CAUTION LIGHT

- 1. CABIN AIR control In.
- 2. CABIN AIR/TEMP MODE switch AUTO.
- 3. CABIN AIR/TEMP switch Decrease.
- 4. VENT BLOWER switch HI.
- 5. CABIN AIR/TEMP MODE switch MAN COOL.
- 6. MANUAL TEMP switch DECREASE (hold).
- 7. LEFT BLEED AIR VALVE switch PNEU & ENVIRO OFF.
- Light still illuminated (30 seconds) LEFT BLEED AIR VALVE switch – OPEN.
- 9. RIGHT BLEED AIR VALVE switch PNEU & ENVIRO OFF.
- 10. Light still illuminated (30 seconds) **RIGHT BLEED AIR VALVE** switch – **OPEN**.

ICE VANE FAILURE

- 1. Airspeed 160 KIAS or below.
- 2. ICE VANE CONTR circuit breaker Pull.
- 3. ICE VANE Operate manually.
- 4. Airspeed Resume normal airspeed.

ENGINE BLEED AIR SYSTEM MALFUNCTION

BLEED AIR FAILURE LIGHT ILLUMINATED

1. BRAKE DEICE switch - Off.

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- 2. TGT and TORQUE Monitor (note readings).
- 3. BLEED AIR VALVE switch OFF.
- 4. Cabin pressurization Check.

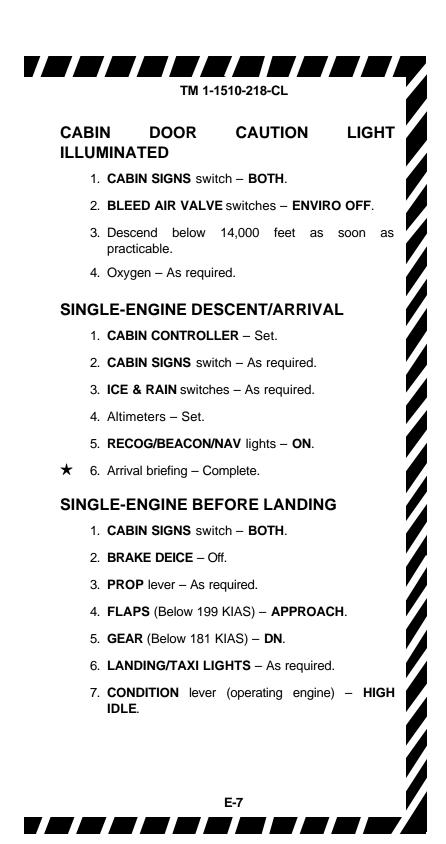
EXCESSIVE DIFFERENTIAL PRESSURE

- 1. Cabin pressurization controller Select higher setting.
- IF CONDITION PERSISTS:
- 2. Oxygen (crew and passengers) As required.
- LEFT BLEED AIR VALVE switch ENVIRO OFF (light illuminated).
- IF CONDITION STILL PERSISTS:
- RIGHT BLEED AIR VALVE switch ENVIRO OFF (light illuminated).
- 5. Descend As required.
- IF CONDITION STILL PERSISTS:
- 6. Oxygen masks 100% and on.
- 7. CABIN PRESS switch DUMP.
- BLEED AIR VALVE switches OPEN (if cabin heating is required).

LOSS OF PRESSURIZATION (ABOVE 10,000 FEET)

- 1. Crew oxygen masks 100% and on.
- Passenger oxygen ON and checked to ensure all passengers have oxygen masks on and are receiving supplemental oxygen if required.





SINGLE-ENGINE LANDING CHECK

- 1. AP/YD Disengaged.
- 2. GEAR DOWN lights Check.
- 3. PROP lever (live engine) HIGH RPM.

SINGLE-ENGINE GO-AROUND

- 1. POWER As required.
- 2. <u>GEAR UP</u>.
- 3. FLAPS APPROACH.
- 4. FLAPS (105 KIAS) UP.
- 5. LANDING/TAXI LIGHTS OFF.
- 6. Climb power Set.
- 7. YD As required.

PROPELLER FAILURE (OVER 2120 RPM)

- 1. POWER lever (affected engine) IDLE.
- 2. PROP lever FEATHER.
- 3. CONDITION lever As required.
- 4. Engine cleanup As required.

FIRE

ENGINE FIRE

ENGINE/NACELLE FIRE DURING START OR GROUND OPERATIONS

1. PROP levers - FEATHER.



- 2. CONDITION levers FUEL CUTOFF.
- 3. FIRE PULL handle Pull.
- 4. PUSH TO EXTINGUISH switch Push.
- 5. MASTER SWITCH OFF.

ENGINE FIRE IN FLIGHT (IDENTIFIED)

- 1. POWER lever IDLE.
- 2. <u>PROP lever FEATHER</u>.
- 3. CONDITION lever FUEL CUTOFF.
- 4. FIRE PULL handle Pull.
- 5. Fire extinguisher Actuate as required.
- 6. Engine cleanup Perform.

FUSELAGE FIRE

- 1. Fight the fire.
- 2. Land as soon as possible if fire continues.

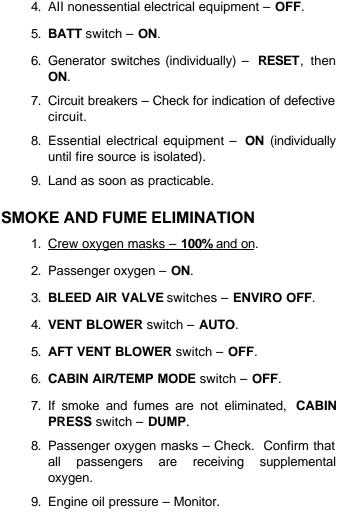
WING FIRE

- 1. Perform engine shutdown on affected side.
- 2. Land as soon as possible if fire continues.

ELECTRICAL FIRE

- 1. Crew oxygen masks As required.
- 2. Passenger oxygen As required.
- 3. **MASTER SWITCH OFF**. (Visual conditions only).





FUEL SYSTEM

FUEL PRESS WARNING LIGHT

1. STANDBY PUMP switch - ON.



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- 2. FUEL PRESS light out Check.
- FUEL PRESS light still on Record unboosted time.

NO FUEL TRANSFER CAUTION LIGHT ILLUMINATED

- 1. AUX TRANSFER switch (affected side) OVERRIDE.
- 2. Auxiliary fuel quantity Monitor.
- AUX TRANSFER switch (after respective auxiliary fuel has completely transferred) – AUTO.

NACELLE FUEL LEAK

- 1. Perform engine shutdown.
- 2. FIRE PULL handle Pull.
- 3. Land as soon as practicable.

FUEL CROSSFEED

- 1. AUX TRANSFER switches AUTO.
- 2. STANDBY PUMPS OFF.
- CROSSFEED switch As required.
- 4. **FUEL CROSSFEED** annunciator illuminated Check.

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- 5. FUEL PRESS light extinguished Check.
- 6. Fuel quantity Monitor.

NAC LOW LIGHT ILLUMINATED

- 1. Usable fuel remaining Confirm.
- 2. Land as soon as possible.

ELECTRICAL SYSTEM EMERGENCIES

DC GEN LIGHT ILLUMINATED

- 1. GEN switch OFF, RESET, then ON.
- IF THE GENERATOR DOES NOT RESET:
- 2. GEN switch (no reset) OFF.
- 3. Operating loadmeter 100% maximum.

BOTH DC GEN LIGHTS ILLUMINATED

- 1. All nonessential equipment Off.
- 2. Land as soon as practicable.

EXCESSIVE LOADMETER INDICATION (OVER 100%)

- 1. Loadmeter Monitor.
- 2. BATT switch OFF (monitor loadmeter).
- IF LOADMETER STILL INDICATES ABOVE 100%:
- 3. Nonessential electrical equipment off.
- IF LOADMETER INDICATES 100% OR BELOW:
- 4. BATT switch ON.

INVERTER LIGHT ILLUMINATED

1. Affected INVERTER switch - OFF.

INST AC LIGHT ILLUMINATED

- 1. N_1 and TGT indications Check.
- 2. Other engine instruments Monitor.

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CIRCUIT BREAKER TRIPPED

- 1. BUS FEEDER breaker tripped Do not reset.
- 2. Nonessential circuit Do not reset.
- 3. Essential circuit Reset once.

BATTERY CHARGE LIGHT ILLUMINATED

- 1. Loadmeter Check and note indication.
- 2. BATT switch OFF.
- Loadmeter Check. If loadmeter indicates less than 2.5% change (one needle width), turn BATT switch ON and monitor for increasing load. If load continues to increase, turn BATT switch OFF.
- BATT switch (landing gear/flap extension only) -ON.

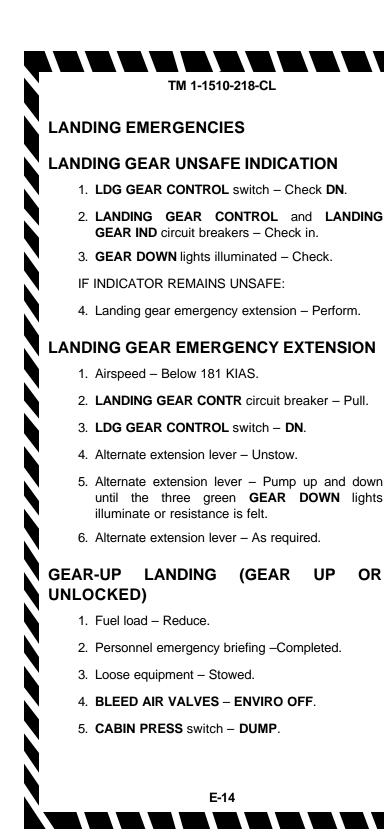
AVIONICS MASTER POWER SWITCH FAILURE

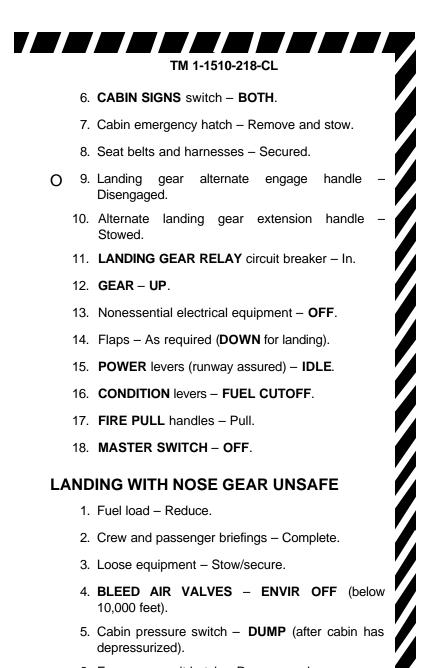
1. AVIONICS MASTER CONTR circuit breaker - Pull.

EMERGENCY DESCENT

- 1. POWER levers IDLE.
- 2. PROP levers HIGH RPM.
- 3. FLAPS APPROACH.
- 4. <u>GEAR DN</u>.
- 5. Airspeed 181 KIAS maximum.

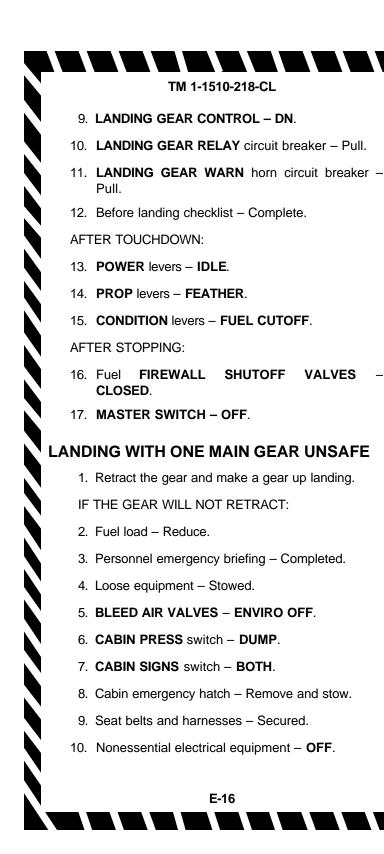
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- 6. Emergency exit hatch Remove and secure.
- 7. Seat belts and harnesses Fasten.
- 8. Extension handle Stow.

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VALVES

11. Touchdown - On safe main gear first.

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- 12. POWER levers IDLE.
- 13. CONDITION levers FUEL CUTOFF.
- 14. FIRE PULL handle Pull.
- 15. MASTER SWITCH OFF.

CRACKED WINDSHIELD

INTERNAL CRACK

- 1. Descend Below 25,000 feet.
- Cabin Pressure Reset pressure differential to 4 psi or less within 10 minutes.

CRACKED CABIN WINDOW

- 1. Crew oxygen masks **100%** and on, if above 10,000 feet.
- 2. CABIN SIGNS switch BOTH.
- 3. Passenger oxygen **ON** and checked, if above 10,000 feet.
- 4. Cabin pressure Depressurize.
- 5. Land as soon as practicable.

DITCHING

- 1. Radio calls/transponder As required.
- 2. Personnel emergency briefing As required.
- 3. BLEED AIR VALVES ENVIRO OFF/PNEU ONLY.
- 4. CABIN PRESS switch DUMP.
- 5. CABIN SIGNS switch BOTH.

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- 6. Cabin emergency hatch Remove and stow.
- 7. Seat belts and harnesses Secured.
- 8. GEAR UP.
- 9. FLAPS DN.
- 10. Nonessential electrical equipment OFF.
- 11. Approach Normal, power on.
- 12. Emergency lights As required.

FLIGHT CONTROLS MALFUNCTION

UNSCHEDULED RUDDER BOOST ACTIVATION

- 1. RUDDER BOOST OFF.
- IF CONDITION PERSISTS:
- 2. RUDDER BOOST circuit breaker Pull.
- 3. BLEED AIR VALVE OFF (Below 10,000 feet).
- 4. Rudder trim Adjust.

UNSCHEDULED ELECTRIC ELEVATOR TRIM

- 1. Control wheel disconnect switch Depress.
- 2. ELEV TRIM switch OFF.
- 3. ELEC TRIM circuit breaker OUT.



PERFORMANCE CHECKS

FUEL PUMPS/CROSSFEED OPERATION

- 1. FIRE PULL handles Pull.
- 2. STANDBY PUMP ON.
- 3. BATT switch ON (#1 and #2 FUEL PRESS lights illuminated.
- 4. Fuel FIRE PULL HANDLES IN.
- 5. **#1** and **#2 FUEL PRESS** annunciators Extinguished.
- 6. **STANDBY PUMPS** Off.
- 7. **#1** and **#2** FUEL PRESS annunciators illuminated.
- CROSSFEED flow alternately left and right (FUEL CROSSFEED annunciator illuminated, #1 and #2 FUEL PRESS annunciators extinguished).
- 9. CROSSFEED flow OFF.
- 10. AUX TRANSFER AUTO.

EFIS POWER AND INVERTERS

- 1. EFIS POWER Push ON.
- 2. Turn ON either INVERTER.
- 3. Ensure both pilots' EADI and EHSI are fully operational.
- 4. EFIS POWER and INVERTERS Off.

FIRE EXTINGUISHER PRESSURE

A gauge, calibrated in psi, is mounted on each supply cylinder for determining the level of charge and should be checked during preflight. Refer to Table P-1.

Table P-1. Engine Fire Extinguisher Gauge Pressure

TEMP °C	-40	-29	-18	-06	04	16	20	38	48
	190	220	250	290	340	390	455	525	605
PSI	to								
	240	275	315	365	420	480	550	635	730

CABIN DOOR

- 1. Open cabin door Check that **CABIN DOOR** annunciator is extinguished.
- Latch cabin door, but do not lock Check that CABIN DOOR annunciator illuminates.
- 3. **BATT** switch **ON**. Check that **CABIN DOOR** annunciator is still illuminated.
- Close and lock cabin door Check that CABIN DOOR annunciator is extinguished.
- 5. BATT switch OFF.

CREW/PASSENGER BRIEFING

- 1. Crew introduction.
- 2. Equipment.
 - a. Personnel to include ID tags.
 - b. Professional (medical equipment, etc.).
 - c. Survival.

- 3. Flight data.
 - a. Route.
 - b. Altitude.
 - c. Time en route.
 - d. Weather.
- 4. Normal Procedures.
 - a. Entry and exit of aircraft.
 - b. Seating and seat position.
 - c. Seat belts.
 - d. Movement in aircraft.
 - e. Internal communications.
 - f. Security of equipment.
 - g. Smoking.
 - h. Oxygen.
 - i. Refueling.
 - j. Weapons and prohibited items.
 - k. Protective masks.
 - I. Toilet.
- 5. Emergency Procedures.
 - a. Emergency exits.
 - b. Emergency equipment.
 - c. Emergency landing / ditching procedures.

AUTOFEATHER/AUTO IGN

 AUTO IGN switches – ARM. IGN ON annunciators illuminated.

- 2. **POWER** levers 22% torque. **IGN ON** annunciators extinguish.
- 3. **AUTOFEATHER** switch Hold to **TEST**. Both **AUTOFEATHER** annunciators illuminated.
- 4. **POWER** levers Retard individually.
 - Approximately 16% 21% torque, opposite
 AUTOFEATHER annunciator extinguishes,
 IGN ON annunciator illuminated.
 - Approximately 9% 14% torque, both AUTOFEATHER annunciators extinguished (prop begins to feather). Both IGN ON annunciators illuminated.

NOTE

AUTOFEATHER annunciators will illuminate and extinguish with each fluctuation of torque as the propeller attempts to feather.

- c. Return **POWER** lever to 22% torque.
- 5. Repeat procedure with other engine.
- POWER levers IDLE (Both AUTOFEATHER lights extinguished, props do not feather).
- 7. AUTOFEATHER switch ARM.
- 8. AUTO IGN switches Off.

OVERSPEED GOVERNORS AND RUDDER BOOST

- 1. RUDDER BOOST switch On.
- 2. **PROP** levers **HIGH RPM**.
- LEFT PROP GOV TEST switch Hold in TEST position.

- Left **POWER** lever Increase until propeller is stabilized at 1830 RPM – 1910 RPM. Continue to increase until rudder movement is noted. Observe ITT/TGT and torque limits and PROP remains stabilized at 1830 RPM – 1910 RPM.
- 5. **POWER** lever Retard to **IDLE**.
- 6. Repeat steps 3, 4, and 5 for the right engine.

PRIMARY GOVERNORS

- 1. POWER levers Set 1800 RPM.
- PROP levers Retard to FEATHER detent. Note propellers stabilize between 1600 and 1640 RPM.
- 3. **PROP** levers **HIGH RPM**. Note propellers return to 1800 RPM.

ENGINE ICE VANES

- 1. ICE VANES On/EXTENDED.
 - a. Both advisory lights illuminated.
 - b. Both bypass doors extended.
 - c. Maximum time for a and b is 15 seconds.

2. ICE VANES – Off/RETRACTED.

- a. Both advisory lights extinguish.
- b. Both bypass doors retracted.
- c. Maximum time for a and b is 15 seconds.

ANTI-ICE/DEICE SYSTEMS

- Prop deice Check. When MANUAL mode is selected, note rise on DC loadmeter. When AUTO mode is selected, monitor prop ammeter for the appropriate number of seconds and ensure the indicator remains in the normal operating range the entire time.
- Windshield heat Check. Note increases on the loadmeter and cycle through both normal and high settings.

NOTE

If windshield heat is needed prior to takeoff, use NORMAL setting for a minimum of 15 minutes prior to selecting HIGH to provide adequate preheating and minimize the effects of thermal shock. The windshield heat thermostat will invalidate the check in OAT above 20° to 30 °C.

- 3. All anti-ice/deice switches OFF.
- 4. Surface deice system Check.

VACUUM AND PNEUMATIC SYSTEM

1. LEFT BLEED AIR VALVE - OFF.

- a. Pneumatic and suction pressures remain normal.
- b. L BL AIR OFF annunciator illuminates
- c. Both **BL AIR FAIL** annunciators remain extinguished.

2. RIGHT BLEED AIR VALVE – OFF.

- a. Pneumatic and suction pressures read zero.
- b. Both **BL AIR OFF** and **BL AIR FAIL** annunciators illuminated.

3. LEFT BLEED AIR VALVE - ON/OPEN.

- a. Pneumatic and suction pressures return to normal.
- b. L BL AIR FAIL annunciator extinguished.

4. RIGHT BLEED AIR VALVE - ON/OPEN.

- a. R BL AIR OFF annunciator extinguished.
- b. Both **BL AIR FAIL** annunciators extinguished.

AUTOMATIC FLIGHT CONTROL SYSTEM

1. Altitude alert.

NOTE

Pause a few seconds after each step to allow time for the proper indications.

- a. Set alert controller more than 1000 feet above altitude indicated on pilot's altimeter. The pilot's altimeter alert light should be extinguished.
- b. Decrease the alert controller to within 1000 feet of the pilot's altimeter setting. The alert light should illuminate.
- c. Decrease the controller to less than 250 feet above the pilot's altimeter setting. The alert light should extinguish.
- d. Increase the controller to 300 ±50 feet above the pilot's altimeter indication and check that the alert light illuminates.
- e. Set the desired altitude.
- 2. Autopilot.
 - a. Autopilot controller **UP TRIM** and **DN TRIM** annunciators Check not illuminated.

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A steady illumination of UP TRIM or DN TRIM annunciator indicates that the automatic synchronization is not functioning and the autopilot should not be engaged.

- b. Turn knob Center.
- c. **ELEV TRIM** switch On.
- d. Control wheel Hold to mid travel.
- e. AP button Press. AP ENGAGE and YD ENGAGE annunciators on autopilot controller will flash. Servo clutches will engage. FD flag on ADI should be in view.
- f. Control movement Check.
 - Rudder pedals Overpower slowly. YD ENGAGE annunciator stops flashing.
 - (2) Control wheel Overpower slowly in both pitch and roll axis. AP ENGAGE annunciator stops flashing. FD flag on ADI retracts.

WARNING

If autopilot or yaw damper disengages during overpower test or if AP ENGAGE or YD ENGAGE annunciator continues to flash, the system is considered non-operative and should not be used. The elevator trim system must not be forced beyond the limits that are indicated on the elevator trim tab indicator.

g. Elevator trim follow-up - Check.

- Control wheel Hold aft of mid travel. Trim wheel should run nose down after approximately 3 seconds. Trim down annunciator should illuminate after approximately 8 seconds.
- Control wheel Hold forward of mid travel. Trim wheel should run nose up after approximately 3 seconds. trim uр annunciator should illuminate after approximately 8 seconds, and AP TRIM FAIL annunciator and MASTER WARNING lights should illuminate after approximately 15 seconds.
- h. AP/YD & TRIM DISC button Press through second level. Autopilot and yaw damper should disengage and ELEV TRIM OFF annunciator should illuminate. AP ENG and YD ENG annunciators on instrument panel should flash five times.
- i. ELEV TRIM switch OFF, then On. ELEV TRIM OFF annunciator should extinguish.
- j. **AP** Re-engage and overpower another time.
- k. Turn controller Check that control wheel follows in each applied direction and center.
- Pitch wheel Check that trim responds to pitch wheel movement. UP TRIM and DN TRIM annunciators may illuminate.
- m. Heading bug Center and engage HDG. Check that control follows a turn in each direction.
- n. Disengage AP by selecting GA. Check that AP disengages and FD commands 7° nose up, wings level attitude.

PRESSURIZATION

- 1. BLEED AIR VALVES Both ON/OPEN.
- CABIN ALT Set 500 feet lower than field pressure altitude.
- CABIN PRESS switch TEST. Cabin climb/descent gauge indicates a descent.
- CABIN PRESS switch Release. Cabin climb/descent gauge indicates a climb, then stabilizes at zero climb.
- 5. Altitude selector Set as required. Pressure altitude plus 200 feet.

DEPARTURE BRIEFING

- 1. ATC clearance Review.
 - a. Routing.
 - b. Initial altitude.
- 2. Departure procedure Review.
 - a. Named departure procedure.
 - b. Obstacle clearance departure procedure / noise abatement procedure.
 - c. VFR departure route.
- 3. Copilot's duties Review.
 - a. Adjust takeoff power.
 - b. Monitor engine instruments.
 - c. Power check at 65 knots.
 - d. Call out engine malfunctions.
 - e. Tune/identify all nav/comm radios.
 - f. Make all radio calls.

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- g. Adjust transponder and radar as required.
- h. Complete flight log during flight and note altitudes and headings.
- i. Note departure time.
- j. Retract gear and flaps as directed.
- 4. TOLD card Review.
 - a. Takeoff power.
 - b. V_1/V_r .
 - c. V₂ + 10 KIAS (climb to 1500 feet AGL).
 - d. V₂/V_{yse}.

ARRIVAL BRIEFING

- 1. Weather/altimeter setting.
- 2. Airfield/facilities Review.
 - a. Field elevation.
 - b. Runway length.
 - c. Runway condition.
- 3. Approach procedure Review.
 - a. Approach plan/profile.
 - b. Altitude restrictions/VDP.
 - c. Missed approach.
 - (1) Point.
 - (2) Time.
 - (3) Intentions.
 - d. Decision height or minimum descent altitude.
 - e. Lost communications.
- 4. Backup approach/frequencies.
- 5. Copilot's duties Review.
 - a. Nav/comm set-up.
 - b. Monitor altitude and airspeeds.
 - c. Monitor approach.
 - d. Call out visual/field in sight.
- 6. Landing performance data Review.
 - a. Approach speed.
 - b. Runway required.
- 7. Passenger briefing As required.

P-12/(P-12 blank)

By Order of the Secretary of the Army:

Official:

Joel B. Hubo

JOEL B. HUDSON Administrative Assistant to the Secretary of the Army 0120501

ERIC K. SHINSEKI General, United States Army Chief of Staff

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