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# NAVAL POSTGRADUATE SCHOOL Monterey, California 



## THESIS

## A METHODOLOGY FOR EVALUATING THE

 CAPABILITY OF THE BRADLEY 25 mm CANNON TO ENGAGE AND DEFEAT PIONEER CLASS UNMANNED AERIAL VEHICLES.by<br>Danny A. Wiley

June, 1996

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Unmanned Aerial Velicles (UAVs) represent a serious threat to forward deployed forces of the United States Army. The defense against such threats is currently provided primarily by the Bradley Stinger Fighting Vehicle (BSFV). The problem addressed is how to evaluate the effectiveness of the BSFV against a UAV. This thesis develops a computer simulation methodology for modeling the capability of a gun system to engage a UAV. Specifically, a review is made of the BSFV, BSFV 25 mm Ammunition, and UAVs. These reviews formed the basis for a computer simulation, coded in Common Lisp Object System (CLOS), modeling the characteristics of three objects: a Projectile, a Launcher and a UAV. Although assumptions were made to simplify the model, simulation runs demonstrated that the rate of fire and aiming system used for launching projectiles resulted in one or more hits in 125 out of 154 engagement sequences. These engagement sequences were against a UAV flying at constant speed and altitude in crossing and inbound/outbound flight profiles. While all data used in this simulation were unclassified, the methodology presented could be used for further ctassified study. potentratty producing a lowet cost means for detennining the effectiveness of air defense weapons against UAV threats.
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# A METHODOLOGY FOR EVALUATING THE CAPABILITY OF THE BRADLEY 25mm CANNON TO ENGAGE AND DEFEAT PIONEER CLASS UNMANNED AERIAL VEHICLES. 

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Submitted in partial fulfillment of the requirements for the degree of

# MASTER OF SCIENCE IN COMPUTER SCIENCE 

from the

## ABSTRACT

Unmanned Aerial Vehicles (UAVs) represent a serious threat to forward deployed forces of the United States Army. The defense against such threats is currently provided primarily by the Bradley Stinger Fighting Vehicle (BSFV). The problem addressed is how to evaluate the effectiveness of the BSFV against a UAV. This thesis develops a computer simulation methodology for modeling the capability of a gun system to engage a UAV. Specifically, a review is made of the BSFV, BSFV 25 mm Ammunition, and UAVs These reviews formed the basis for a computer simulation, coded in Common Lisp Object System (CLOS), modeling the characteristics of three objects: a Projectile, a Launcher and a UAV. Although assumptions were made to simplify the model, simulation runs demonstrated that the rate of fire and aiming system used for launching projectiles resulted in one or more hits in 125 out of 154 engagement sequences. These engagement sequences were against a UAV flying at constant speed and altitude in crossing and inbound/outbound flight profiles. While all data used in this simulation were unclassified, the methodology presented could be used for further classified study, potentially producing a lower cost means for determining the effectiveness of air defense weapons against UAV threats.

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## I. INTRODUCTION

## A. GOALS

The goal of this thesis is to develop a method to determine the ability of the Bradley Stinger Fighting Vehicle (BSFV) 25mm Cannon to counter the emerging threat of Unmanned Aerial Vehicles (UAVs). The intent is to simulate the BSFV engagement of a UAV, to analyze the results to understand this capability, and to provide insight for improving existing weapons, munitions and systems. This work develops an approach and is unclassified. Bradley effectiveness results contained herein do not represent actual operational characteristics, but are illustrative of an application of the simulation system described.

## B. BACKGROUND AND MOTIVATION

As the Army evolves toward the Twenty First Century, the need for knowledge about the capability of its weapons against emerging threats becomes more important Commanders must improve their ability to detect, locate, identify and engage targets at maximum range. UAVs will likely be a target on the battlefields of the Twenty First Century. Research today may provide commanders at all echelons with an awareness of the capabilities possessed by today's weapons and needed by tomorrow's weapons to effectively engage and defeat a hostile UAV.

A UAV can perform a wide variety of missions, but such vehicles are primarily used to gather information about a specific area of interest. As capabilities increase, the mission of the UAV will be similar to that of manned aerial vehicles. The relatively low cost of such unmanned vehicles, along with the ability to accomplish a mission without risking human life, are a valuable asset for military forces around the world. The increased use of the UAV by other countries leads to concerns about actions to counter this threat.

Defending against a UAV is the shared responsibility of several components on the battlefield. One such component is the Forward Area Air Defense (FAAD) weapons of an Army Division. The FAAD Weapons that engage an air target are: [Ref. 19]

- Line-of-Sight-Rear (LOS-R) FAAD Weapon. A system composed of a missile component and a machine gun that is Identification Friend or Foe (IFF)
capable. The Army's current system is the AVENGER. The AVENGER has an organic Forward Looking Infra-Red (FLIR) sensor.
- Man Portable Air Defense System (MANPADS). A shoulder fired system with an IFF capable missile. The Army's current system is the Stinger. It has no organic sensor capability.
- Bradley Fighting Vehicle (BSFV). A system composed of a Bradley Fighting Vehicle (BFV) with an externally mounted Stinger weapon system. The BSFV has a 25 mm cannon and Stinger missiles on-board with no organic sensor capability.

It is imperative to study the capability of these weapons to defend against a UAV. A FAAD weapon system is very likely to encounter a UAV flying at low altitude that is difficult to detect and engage. The potential widespread use of UAVs on the battlefields of tomorrow requires the study of the effectiveness of today's weapons against this threat. Such research will help planners design weapons to counter the future UAV threat. This thesis will focus on the capability of the BSFV 25 mm Cannon to engage and hit a UAV.

## C. RESEARCH QUESTIONS

This thesis will examine the following research areas:

- Review the physical capabilities to bring fire to bear on a target such as the cannon elevation limits and the operational flight characteristics of a UAV.
- Analyze the types of ammunition available.
- Determine the potential of a hit based on:
- the simulation of a projectile, constrained by the characteristics of the Bradley Cannon and ballistics data derived from unclassified specifications.
- the simulation of a UAV, derived from the unclassified specifications and operational characteristics of a Pioneer UAV.


## D. ORGANIZATION

Chapter II of this thesis provides a general overview of unmanned aerial vehicles and provides a detailed description of the Pioneer UAV. Chapter III provides a detailed description of the Bradley Fighting Vehicle and its role as an air defense weapon. Chapter III looks specifically at the 25 mm Cannon and its ammunition. Chapter IV outlines the UAV and Cannon model and provides the limitations, assumptions and profiles for the simulation. Chapter V details the results of the simulations and analyzes the data and provides an evaluation of the results. Chapter VI presents conclusions and recommendations for further research. The source code is provided in the Appendices A I.

## II. UNMANNED AERIAL VEHICLES

## A. BACKGROUND

Although UAVs were envisioned as far back as 1916, the first modern UAV was an unmanned $\mathrm{B}-17$ controlled from a second aircraft. The controlling aircraft accompanied the unmanned $\mathrm{B}-17$ to a target area where it was then remotely flown closer to the actual target. These unmanned B-17 aircraft were used extensively during the atomic bomb tests in the South Pacific to monitor radiation. [Ref. 13]

Following the Soviet interception of Francis Gary Powers' U-2 in 1960, the Air Force and other national agencies directed resources into UAV development programs. The AQM-34 was one such system. Variations of the AQM-34 flew more than 17,500 missions around the world from 1958 to 1975. [Ref. 13]

A highly successful system, known as Buffalo Hunter, flew more than 1,600 missions in Southeast Asia. It was a variation of the AQM-34 that usually operated at altitudes of more than 60,000 feet. The operational concept later evolved to include very low-level photo-reconnaissance missions over North Vietnam. Other missions included signal intelligence (SIGINT) and psychological operations missions flown by the 100th Strategic Reconnaissance Wing. The last mission of Buffalo Hunter took place over Saigon on April 30, 1975 during the final stages of the United States' evacuation. [Ref. 13]

Post-Vietnam era UAV developments were led primarily by the Central Intelligence Agency and the U.S. Army. The Field Artillery's Aquila target acquisition, designation and reconnaissance system (TADARS) began development in 1974, but was terminated in 1987 after extensive testing. In 1985 the U.S. Army Intelligence Center at Fort Huachuca, Arizona was designated as the UAV proponent, less Aquila, to develop a family of UAVs to provide organic, near real-time support to battlefield commanders. The requirements for deep and close UAVs were approved by the Training and Doctrine Command (TRADOC) in December 1988. [Ref. 13]

## B. GENERAL OVERVIEW

## 1. Capabilities and Value

Since the Vietnam conflict, the ability of UAVs to gather and return sensitive intelligence has improved as the UAV has grown smaller, lighter and quieter with increased operational range and flight endurance. [Ref. 7] Current UAV systems can provide real-time multi-spectral video to ground forces in less than a minute while the aircraft is being remotely piloted from either a permanent control station at sea or a portable station on the ground. The high resolution video platforms, which usually have day and night image capability, are gyro-stabilized. On average, today's UAVs can fly for up to six hours and have ranges of 100-300 miles from their control station. These aircraft can often fly further if control is transferred to a forward station. [Ref. 12]

The advantage these UAVs have over other aircraft with similar imaging capabilities is their relative low cost without risk of human life. Even though NATO peacekeepers have met little resistance (as of this publication) in Bosnia, bad weather can still pose a threat to manned aircraft [Ref. 18]. UAVs can also be used in advance of other air units to check terrain for possible unmarked hazards like low power lines

## 2. Operation

An internal pilot flies the UAV from the control station using gauges and on-board cameras; external pilots fly the aircraft by sight and are directly responsible for control of the plane's take-off and landing procedures. Once a mission is recorded, the plane is launched and taken to a set altitude and heading by the external pilot. The mission is then handed off to the internal pilot who controls the duration of the mission until return for recovery or landing. Most UAVs are outfitted with an autopilot system, navigational equipment, and communications capabilities that allow the plane to be flown in an autonomous mode with human monitoring. Once missions are underway, the person at the control station is able to modify parameters for future legs of the flight plan. Weather conditions, for example, often require a change in altitude to allow the point of interest to be viewed. [Ref. 12]

An example of the operation of a UAV for a military style deep fire mission would involve the use of two UAVs with ranges of 200 kilometers each. One UAV flies a
mission to locate and confirm a suspected missile launch site. The other UAV is used as an airborne communications relay station, facilitating the information flow from the UAV on the missile site to the planners in the rear area of the battlefield. This simple example illustrates the operation of the UAV in two roles: as a reconnaissance vehicle and as a communications relay station.

## C. PIONEER UNMANNED AERIAL VEHICLE

## 1. Description

The pioneer aircraft has a shoulder-wing, pod-and-twin-tailboom configuration, with fixed tricycle landing gear and an arrester hook. Payloads are mounted in a small slot located on the front belly of the aircraft. The wings, booms and tail unit are detachable. This simplifies dismantling and assembly in the field and facilitates the prompt dispatch of several vehicles at one time. [Ref. 7]

A Pioneer fitted with standard electronic equipment for a reconnaissance mission is depicted below.


Figure 1: Computer Generated Pioneer UAV Image
The Pioneer operates at ranges up to one hundred nautical miles and altitudes up to twelve thousand feet for as long as five hours. Pioneer carries an independently controlled high resolution daytime TV or a nighttime infrared camera which transmits real-
time imagery back to a ground or ship-based control station. The Pioneer can take off from a fixed airport runway on its own power or it can be launched from the ship's flight deck with a rocket assist. Similarly, the Pioneer lands at land based sites with runways or it is recovered by a fifty foot high net raised over a ship's flight deck. The Pioneer cruises at between sixty and one hundred knots.

Continuous communications between the ground control station and the UAV allow for constant control and monitoring of Pioneer's flight systems, navigation, and video payload. A single operator can fly and control a Pioneer. Once the craft is airborne, it is controlled by the control station. This station is equipped with high resolution color graphics displays. The control station operator can manipulate the cameras and other equipment on board the Pioneer. All imagery data can be frozen, stored, and recorded for future processing. [Ref. 7]

The control station supports graphical entry of a mission flight plan on a digitized color map display. During flight operations, the map display shows the UAV's position and the payload's optical axis. The control station can automatically transmit commands to the UAV to fly each leg of the mission flight plan. Another option is to transmit the entire flight plan to the UAV, which will then fly the plan autonomously. Control stations can be deployed in a manner to increase the operational range and flexibility of the Pioneer. [Ref. 7]

## 2. Operational Status

The Pioneer was used successfully in the Gulf War and continues to be used by the U.S. Navy, U.S. Army, and U.S. Marine forces [Ref. 7]. The Pioneer is currently used for surveillance, over-the-horizon targeting, naval gunfire spotting, overland reconnaissance, and real-time battle damage assessment.

## 3. Specifications

A Pioneer UAV system is composed of five air vehicles, nine payloads, one ground control station, one pilot control station, one launcher, and a recovery system. Table 1 contains details of the air vehicle specifications.

| Power Plant | 19.4 kW (26 hp) Sachs SF 3502 cylinder 2 stroke engine |
| :---: | :---: |
| Wing Span | 5.11 m |
| Length | 4.26 m |
| Max Payload | 45 kg |
| Max Speed | $176 \mathrm{~km} / \mathrm{h}$ |
| Cruising Speed | $120 \mathrm{~km} / \mathrm{h}$ |
| Loiter Speed | $111 \mathrm{~km} / \mathrm{h}$ |
| Max S/L Rate of Climb | $4.1 \mathrm{~m} / \mathrm{s}$ |
| Operating Height Range | 305-3660 m |
| Data Link Range | 185 km |
| Endurance | 5.5 hrs |

Table 1: Specifications of Pioneer UAV [Ref. 7]

## D. CURRENT UAV PROGRAMS

Congressional interest has focused on developing UAVs that have a common architecture and can interact on the battlefield. To ensure this coordinated effort, in 1988 Congress halted all service UAV funding and established the Joint Program Office for UAVs (JPO-UAV). This office was chartered to develop a Department of Defense (DoD) master plan for UAVs and establish a family of common interoperable UAVs. [Ref. 13]

Since 1989, this family of UAVs has evolved into the UAV-Short Range (UAVSR), UAV-Close Range (UAV-CR) and UAV-Endurance (UAV-E). Table 2 shows the operational characteristic of each type of UAV. [Ref. 13]

| Fielding Date | FY98 | FY95 | FY98 |
| :---: | :---: | :---: | :---: |
| Range | 50 km | 200 km | 500 nm |
| Flight Endurance | 3 hrs | 10 hrs | 24 hrs |
| Service Ceiling | 13,000 AGL | 15,000 AGL | 25,000 AGL |

Table 2: Specification Classes of Current UAV Programs [Ref. 13]

## III. BRADLEY FIGHTING VEHICLE

## A. GENERAL OVERVIEW

The Bradley Fighting Vehicle was designed to be a common vehicle for both the infantry and scout roles. The Bradley Fighting Vehicle is made of all-welded aluminum armor and is about seven meters long, three meters wide, and three meters high. The Bradley has two main components: the chassis and the turret. [Ref. 15]


Figure 2: Bradley Fighting Vehicle

The chassis is powered by a Cummins VTA-903T diesel engine with a Martin Marietta HMPT-500 hydromechanical transmission. The turret contains the main armament consisting of a McDonnell Douglas M242 25 mm Chain Gun. Additionally the turret contains a 7.62 mm M 240 C coaxial mounted machine gun to the right of the 25 mm cannon. [Ref. 14]

Since its initial fielding in 1983, the Bradley has taken on many different configurations and roles. Some of these are: Personnel Carrier, Cavalry Fighting Vehicle,
and Bradley Stinger Fighting Vehicle. Variants of the Bradley Fighting Vehicle are equipped with the Hughes TOW anti-tank missile and the Stinger anti-aircraft missile system. The chassis of this vehicle is also used for the Multiple Launch Rocket System (MLRS) and serves as an ammunition carrier. Currently contractors are building various Bradley based prototype systems to replace older systems mounted on the M113 chassis such as the Bradley Fire Support Team Vehicle.

## B. BRADLEY STINGER FIGHTING VEHICLE

## 1. Introduction

Following the withdrawal of the M163 20mm self-propelled Vulcan Air Defense system, the U.S. Army began using the Bradley Fighting Vehicle to transport Stinger missiles and the soldiers that employ them. This combination of the Bradley Fighting Vehicle with the Stinger teams was the initial Bradley Stinger Fighting Vehicle (BSFV). [Ref. 14]

## 2. Description

The current BSFV has a firing pod containing four Stinger missiles on the left side of the turret (see Figure 3). The main armament for the system is the 25 mm cannon and the Stinger missiles. This version of the Bradley has a five man crew consisting of a squad leader, a senior gunner, a driver and two Stinger gunners. The Stinger missiles can be fired from the pod or mounted on to grip stocks for dismounted shoulder firing. [Ref. 14]


Figure 3: Bradley Stinger Fighting Vehicle

## 3. Operational Status

The BSFV is currently fielded to all U.S. Army Air Defense Battalions assigned to Mechanized Divisions. The BSFV is the most forward deployed Air Defense weapon on the battlefield and performs the short range (less than five kilometers) Air Defense mission for the U.S. Army. Although the BSFV is not a combat proven Air Defense weapon, the Bradley's 25 mm cannon reportedly exceeded all expectations during the Persian Gulf War. During combat operations in the Persian Gulf, the Bradley maintained an operational readiness rate above ninety percent. The BSFV has performed extremely well in rigorous live fire tests and field testing at the National Training Center (NTC) at Fort Irwin, California. [Ref. 14]

## 4. Characteristics of the BSFV

The BSFV is a lethal weapons platform that enhances the firepower and survivability of air defense assets on the battlefield. Table 3 below outlines the characteristics of the BSFV.

|  |  |  |
| :---: | :---: | :---: |
| COMBAT WEIGHT | 23.5 tons |  |
| HEIGHT | 116 inches |  |
| WIDTH | 128 inches |  |
| GROUND CLEARANCE | 18 inches |  |
| ACCELERATION (0 TO 32 KMPH ) | 7.7 seconds |  |
| ROAD SPEED | 66 kmph |  |
| CROSS-COUNTRY SPEED | 48 kmph |  |
| CRUISING RANGE | 480 km |  |
| VERTICAL OBSTACLE | 36 inches |  |
| TRENCH | 100 inches |  |
| FORDING DEPTH | 3.5 feet |  |
| MAIN ARMAMENT | 25 mm (900 rounds) |  |
|  | Tow (5 rounds) |  |
|  | Stinger (6 rounds) |  |
| MAXIMUM EFFECTIVE RANGE | 25 mm ground | 3000 meters |
|  | 25 mm air | 2000 meters |
|  | Stinger | 4000 meters+ |
|  | Tow | 3700 meters |
|  | Coax 7.62 mm | 900 meters |

Table 3: General Characteristics of the BSFV [Ref. 15]

## 5. 25 mm Cannon

| Elevation | +60 degree to -10 degree |
| :---: | :---: |
| Traverse | 360 degree continuous |
| Sight | Thermal, Direct View Optics |
| Sight Magnification | 3 x and 12 x |
| Firing Rate - Low | $75-125$ rounds per minute |
| Firing Rate - High | 175-225 rounds per minute |

Table 4: General characteristics of the BSFV Gun [Ref. 15]
The 25 mm cannon subsystem is a weapon system designed to attack and defeat enemy armored vehicles and other targets, such as aircraft, using armor-piercing (AP) or high-explosive (HE) projectiles. The gun is an electrically powered, chain driven
automatic weapon. It is fed by a metallic link and has dual feed capability. The 25 mm ammunition cans hold up to seventy rounds of AP and two hundred and thirty rounds of HE ammunition. The 25 mm Cannon has three parts: the barrel assembly, feeder assembly, and receiver assembly. The gun has both electrical and manual fire control. [Ref. 16]

The major components and functions of the 25 mm cannon are:

- The Barrel Assembly
- gives directional control to the projectile and
- includes a muzzle brake that reduces blast flash and absorbs some of the recoil.
- The Feeder Assembly
- removes linked ammunition from feed chutes,
- strips rounds from links,
- places round into bolt face,
- removes spent cartridge case from bolt face,
- provides a means of selecting two types of ammunition,
- provides a means to manually operate the gun during power failure, and
- sends electrical signals to the turret controls indicating position of the bolt.
- The Receiver Assembly
- rams and fires the rounds,
- extracts and ejects spent cartridge cases or unfired rounds,
- suppresses the recoil force from barrel and breech, and
- contains a mechanical interlock mechanism that stops the gun if a round misfires or hangfires. [Ref. 15]
The 25 mm Cannon is normally operated in the electrical mode. The weapon control box allows for selection of the 25 mm Cannon, ammunition type (AP or HE), rate of fire, and arming of the 25 mm Cannon. Firing rate can be single shot, low rate (around 75 to 125 rounds per minute) or high rate (around 175 to 225 rounds per minute) Indicator lights on the weapon control box show which mode is selected. Weapon select will not function if the gun is out of sear. When AP or HE is selected, the feed select solenoid is enabled. Once the feed selection is made and the gun is armed, it can be fired using one of three triggers (gunner's, commander's hand station, or traverse hand wheel). When the trigger is pressed, the sear solenoid is energized releasing the sear pin from the
master link on the chain. The sear pin action energizes the electric motor which drives the track and bolt assembly, and feeder. The feeder places a live round in front of the bolt and the bolt moves forward and locks to the breech. When the gun is in breech locked position, the sear pin is engaged against the chain's safety link. When the round is fired, the bolt unlocks and the sear pin is released allowing the cycle to continue. The spent case is extracted by the rearward motion of the bolt. The rotor then turns to place the spent case into the eject chute while a new round is placed in front of the bolt. In single shot and low rate, the motor is turned off and the sear pin is engaged to contact the master link and stop the gun in the sear position. On the next forward motion of the bolt, the spent case is pulled out of the ejection chute by the ejector on the bolt carrier. [Ref. 16]

Ammunition for the 25 mm cannon is stored below the weapon in an ammunition can. The ammunition can has two sections for AP and HE ammunition. Each ammunition belt feeds through its own forwarder and chute. The forwarders are manually operated and used while loading ammunition. The ammunition can has two sensors that tell when the 25 mm cannon is low on ammunition. This leaves the end of the ammunition belt in a position that allows the ammunition to be easily attached to the old belt. [Ref. 16]

## 6. Ammunition Assessment

## a. Introduction

Two types of ammunition are used by the BSFV's 25 mm cannon. The ammunition characteristics are outlined in Table 5 below.

| M791 APDS-T | ARMOR PIERCING | BLACK TIP | NONE |
| :--- | :--- | :--- | :--- |
| M792 HEI-T | HIGH EXPLOSIVE | YELLOW | M758 |
|  | INCENDIARY TRACER |  |  |

Table 5: Two Types of Ammunition [Ref. 16]
Target identification is the critical element in ammunition selection. Both the M791 and M792 can be used against slow moving, fixed-wing aircraft and helicopters. Rounds should be fired in twenty to twenty-five round bursts with the high rate of fire selected. The M791 has a higher probability of hit than M792; however, the M792 has a higher
probability of kill given a hit. At ranges beyond twelve hundred meters, the M791 is more effective against helicopters. At ranges less than twelve hundred meters, the M792 is more effective against helicopters. Beyond two thousand meters, the 25 mm cannon loses its effectiveness to kill. [Ref. 16]

## b. M792 High-Explosive

The M792 projectile is a high-explosive incendiary tracer that is commonly referred to as a "heat" round. Because the projectile is explosive in nature, if it hits the target it causes the more damage to an air target than the M791 Armor-Piercing projectile Its characteristics are given in Table 6 below.

| Specified Muzzle Velocity | 1100 meters per second |
| :--- | :--- |
| Drop at $\mathbf{3 0 0}$ meters | -0.402 meters |
| Drop at $\mathbf{5 0 0}$ meters | -1.200 meters |
| Drop at 700 meters | -2.542 meters |
| Velocity dispersion | $+/-5$ meters per second |
| Angular dispersion | $+/-.89$ mils |

Table 6: M792 Projectile [Ref. 17]

## c. M791 Armor-Piercing

The M791 projectile is a armor piercing projectile that is commonly referred to as a "SABOT" round. It has a higher probability of hit against an air target than the M792 High-Explosive projectile. Its characteristics are given in Table 7 below.

| Muzzle Velocity | 1345 meters per second |
| :---: | :---: |
| Drop at 300 meters | -0.251 meters |
| Drop at 500 meters | -0.708 meters |
| Drop at 700 meters | -1.413 meters |

Table 7: M791 Projectile [Ref. 17]

## IV. SIMULATION MODEL

## A. INTRODUCTION

This chapter describes the UAV and projectile object models and the algorithms that simulate the gunner and determine when a projectile hits the UAV. The UAV flight profiles are introduced as are the object model limitations and assumptions. The models were implemented in Franz Common Lisp Object System (CLOS) using Allegro Common Windows for graphics display.

## B. GENERAL OVERVIEW

This simulation model uses data from unclassified sources and personal insight from military experience and interviews with operators of the UAV and Bradley Fighting Vehicle systems. In general, for this simulation to provide accurate results, the author must address:

- the limitations and assumptions about the environment, and
- the assumptions about the UAV and Projectile behavior

This model represents a methodology that provides insight into the potential effectiveness of a BSFV against an unmanned aerial craft.

## C. SCENARIO

The general approach in this model is for the gun to fire at a sustained rate of fire at an aim point in front of the aircraft. In this method of firing, adjustments are determined by firing at the target and correcting the aim based on observations from a second individual (the squad leader). The gunner will attempt to establish a volume of fire in front of the aircraft so that the aircraft flies into the projectiles. After the aircraft has flown through the wall of projectiles, the gunner will adjust the aim point to once again establish a volume of fire in front of the aircraft. This continues until the aircraft is either destroyed or out of range. This scenario is modeled through the simulation of a UAV in flight and projectiles fired from a launcher described in the following sections.

## D. CHARACTERISTICS OF UNMANNED AERIAL VEHICLE

The characteristics of the Pioneer UAV described in Chapter II and information obtained during conversations with experienced Pioneer UAV pilots are the basis of the UAV model in this simulation. A comparison of Tables 8 and 9 below with Table 1 shows that the simulated UAV is somewhat larger than the Pioneer UAV, although otherwise very similar, in order to represent the average size of UAVs of the same class as the Pioneer UAV.

The mission and flight profile of the UAV is comparable to that of a Pioneer UAV conducting reconnaissance against armored vehicles in the forward battle area. The modeled UAV flies at altitudes of 2500 or 3500 feet and at an operating speed of 65 knots. The altitude is selected to maintain line of sight between the aircraft and its controller. This line of sight restriction forces the aircraft to maintain a higher altitude at greater distances from the controller. The speed is a function of the UAV type, the payload, and the mission; in general, 65 knots is the operating speed selected for reconnaissance of the forward edge of the battlefield. [Ref. 18]

|  | 24 feet |
| :--- | :--- |
| Wing Span | 29 feet |
| Length | 74.80 mph or 65 knots ( 110 feet per second) |
| Operating Speed | 2500 or 3500 feet |
| Operating Altitude |  |

Table 8: Characteristics of Simulated UAV
The simulated UAV is composed of thirteen rectangular surfaces. Each surface is independently tested for penetration by a projectile. The UAV surfaces and their dimensions are listed in Table 9 below.

| Body | 2 feet | 14 feet |
| :--- | :--- | :--- |
| Belly and Top | 3 feet | 2 feet |
| Nose and Rear | 3 feet | 14 feet |
| Sides, Left and Right | 3 feet | 11 feet |
| Wings, Left and Right | 1 feet | 16 feet |
| Tail Extensions, Left and Right | 3 feet | 10 feet |
| Tail | 3 feet | 3 feet |
| Tail Fins, Left and Right |  |  |

Table 9: Simulated UAV Surfaces
The line drawings in Figure 4 below illustrate two aspects of the simulated UAV Any projectile that penetrates a surface is classified as "hitting" the UAV in that area.


Figure 4: Line Drawing of UAV
Appendix A contains the Lisp Code that generates the graphical display and models the flight of the UAV

## E. CHARACTERISTICS OF PROJECTILE AND THE LAUNCHER

The projectile was modeled after the M792 High-Explosive 25 mm round discussed in Chapter III. The muzzle velocity is 3608 feet per second and the drop characteristics of the round were derived using Heun integration with the data in Table 6. This method results in the drop values listed in Table 10 below. The dynamic model used to obtain these results assumes no aerodynamic lift acting on the projectile and drag proportional to the square of the velocity. In the absence of further information, and for simplicity, angular and velocity dispersion figures were modeled as uniformly and independently distributed random variables with zero mean and range equal to plus or minus the given dispersion figures.

| Velocity Dispersion | 16.4 feet per second |
| :--- | :--- |
| Angular Dispersion | 0.05006219 degrees |
| Average Muzzle Velocity | 3608 feet per second |
| Drop at $\mathbf{1 0 3 1}$ feet | 1.44 feet |
| Drop at $\mathbf{1 6 6 5}$ feet | 4.02 feet |
| Drop at $\mathbf{2 2 6 0}$ feet | 7.89 feet |
| Effective Range | 6100 feet |

Table 10: Simulated Projectile Characteristics
The simulated launcher characteristics are listed in Table 11 below. The simulated launcher has a single rate of fire while the BSFV has both a low and high rate of fire.

|  |  |
| :--- | :--- |
| Elevation | +60 degree to -10 degree continuous |
| Traverse | 360 degree continuous |
| Firing Rate - Burst | 300 rounds per minute (1 every .2 seconds) |

Table 11: Simulated Projectile Launcher Characteristics
Appendix A contains the Lisp code that models the projectile and launcher.

## F. DETERMINING A HIT

The algorithm that determines if a projectile has penetrated any of the simulated UAVs surfaces has four steps. Step 1 screens out all projectiles that are too distant from the UAV. This step is illustrated in Figure 5 below.


Projectiles ( $\mathbf{P}$ ) outside of circumference are not tested for hit.
P1 would not be tested while P2 would be tested for a hit.

Figure 5: Step 1 of Hit Algorithm
Step 1 tests if the projectile is within 225 feet of the center point of the UAV. This distance is chosen because of the projectile velocity and the time step used in the simulation. The minimum range from the UAV to the projectile is 2500 feet (equal to the minimum UAV altitude). Analysis of the projectile at 2500 feet using Heun integration indicates that the projectile travels less than 225 feet in a time step. (This follows because a projectile fired at zero degrees elevation travels 223 feet in the next $1 / 10^{\text {th }}$ second after reaching the 2500 foot minimum range for engagement.) Step 1 screens out computationally costly calculations for projectiles incapable of hitting the UAV in the current time step.

Step 2 determines if the projectile has crossed the plane that contains a given UAV surface. This step is illustrated in Figure 6 below.


Figure 6: Step 2 of Hit Algorithm
Step 2 determines if the projectile's previous position and current position lie on different sides of the plane which contains the UAV surface. Points p1, p2, p3 and p4 are arbitrary points in the plane containing the UAV surface. The normal vector for the plane is determined by computing the cross product of two vectors in the plane. The dot product of the normal of the plane and the vector formed by the previous projectile position and point pl is compared with the dot product of the normal of the plane and the vector formed by the current projectile position and the point pl. If the comparison
reveals that these two values have the same sign, then the projectile did not cross the plane


Figure 7: Step 3 of Hit Algorithm

Step 3 determines the $\mathrm{X}, \mathrm{Y}$ and Z coordinate of the bullet's impact on the plane by assuming a straight line path between the previous position and the current position and using the normal computed in Step 2. This step is illustrated Figure 7 above.

Step 3 requires numerous calculations which use the vector from the previous projectile position to its current position and the normal vector to solve for the point on the plane along the path of the projectile. This result will also be used in Step 4 to determine if the point of impact on the plane lies inside the boundaries of the UAV surface being tested for penetration.

Step 4 simply computes the vectors from the impact point of the projectile on the plane to each corner point of the UAV surface. If the sum of the angles between the vectors is equal to two times pi, then the projectile must be inside the UAV surface and is classified as a hit. If the sum is not equal to two times pi, then the projectile lies on the boundary of or outside of the UAV surface and is classified as a miss. This step is illustrated in Figure 8 below.

## STEP4

## UAVSURFACE



If angles between V1 and V2, V2 and V3, V 3 and V4, V 4 and V1 sum up to 360 degrees then the projectile is inside the boundaries of the UAV surface
A $n g l e s$ betw een vectors are computed as:
$\overrightarrow{\mathrm{V}}_{1} \bullet \overrightarrow{\mathrm{~V}}_{2}=\left|\overrightarrow{\mathrm{v}}_{1} \| \vec{v}_{2}\right| \cos \theta$
$\theta=a \cos \left|\frac{\vec{v}_{1} \bullet \vec{v}_{2}}{\left|\vec{v}_{1}\right| \quad\left|\vec{v}_{2}\right|}\right|$
Figure 8: Step 4 of Hit Algorithm

The formulas used in steps 1 to 4 are coded in Lisp in Appendix D.

## G. CHARACTERISTICS OF THE PERFECT GUNNER

The modeled perfect gunner provides insight into the potential effectiveness of a projectile against a UAV. The perfect gunner can successfully engage the target with the highest probability of hit and kill, given that the object to be shot at is in range and the gunner knows its speed, altitude, and has an operational gun. The perfect gunner takes aim with all data known about the target to be engaged and is limited solely by the characteristics of the projectile, its launcher, and the algorithms used to determine the aim point to fire upon.

Due to the difficulty associated with engaging aerial targets, a high volume of fire should be established in front of the target [Ref. 20]. The perfect gunner fires five projectiles per second at an aim point in front of the UAV. This aim point is determined by two algorithms:

1. an "Adjust Fire Elevation" algorithm that computes the elevation (vertical angle of fire) needed to hit the UAV at its current altitude, and
2. an "Adjust Fire Azimuth" algorithm that computes the azimuth (horizontal angle of fire) needed to ensure the wall of fire is in front of the UAV.

## 1. Adjust Fire Elevation

## a. 0 Degree Offset Adjust Fire Elevation

This algorithm has three steps and the code is found in Appendix E. Step 1 aims the gun at the current UAV position and computes the angle (Theta 1) as if the projectile travels on a straight path. Step 2 computes the angle (Theta 2) from the gun to the actual projectile location at the range to the target. This actual location is determined by firing a spotting round at the current UAV position and taking the coordinates when it passes the current UAV down range position. Step 3 computes the angle (Theta 3) to the future UAV position accounting for the time of flight of the projectile. This angle (Theta 3 ) is added to the difference between Theta 1 and Theta 2. The result is an elevation that takes into account the drop error of the spotting round and the time of flight. Test results indicated that this elevation angle should be further adjusted to determine the angle needed to ensure the projectile would be in front of the UAV. After preliminary testing of the
algorithm, a one degree (positive for inbound and negative for outbound) super elevation constant was chosen to ensure the projectile is in front of the UAV. Figure 9 below graphically depicts Step 1-3. Code for this algorithm is listed in Appendix D.


Figure 9: Adjust Fire Elevation

## b. 45 Degree Offset Adjust Fire Elevation

This algorithm has 2 steps and the code is found in Appendix E. Step 1 aims the gun at the future UAV position and computes the straight line angle (Theta 3 from Figure 9 above) for this shot. Step 2 fires a series of test projectiles at increasing increments ( $1 / 2$ degree per step) until a test projectile's old position and current position
lie on opposite sides of the Z plane of the UAV. This elevation is then used and the gunner establishes a wall of fire on the path of the UAV.

## 2. Adjust Fire Azimuth

The azimuth from the gun to the UAV is computed by determining the azimuth from gun to the future UAV position. This calculation uses the future position of the UAV computed in the Adjust Fire Algorithm. In the simple case of Zero Degree offset engagements the azimuth will be either 0 or 180 degrees. Figure 10 illustrates the adjust fire azimuth algorithm


Figure 10: Adjust Fire Azimuth

## H. FLIGHT PROFILES ANALYZED

Two flight profiles were considered in the tests using the perfect gun. These profiles ensure that at least one or more in-range engagements are possible. The following profiles are analyzed:

- 0 degree offset inbound, outbound UAV
- 45 degree offset crossing UAV

Figure 11 below illustrates graphically the three flight profiles at 2500 feet and 3500 feet and the engagement footprint (the area between maximum and minimum gun ranges)


Figure 11 - Flight Profiles and Engagement Footprint

The flight profiles were run first with the perfect gunner taking shots within the appropriate minimum and maximum ranges of the gun. The results are described in what follows to allow conclusions to be drawn about the effectiveness of the modeled projectile
and aiming method, and to evaluate the model in general. This is intended to provide insights into the potential effectiveness of the projectile against a UAV.

The flight profiles begin at the edge of the engagement footprint and are analyzed while inside the footprint. Table 12 below illustrates the ranges of the Gun, which has a 6100 foot effective projectile range and an elevation limit of 60 degrees for UAVs at 2500 and 3500 feet.

| $\%$ | I. \%. |  |
| :---: | :---: | :---: |
| 0 degree inbound, outbound |  |  |
| 2500 feet | 1443 feet | 5564 feet |
| 3500 feet | 2020 feet | 4996 feet |
| 45 degree crossing |  |  |
| 2500 feet | 1443 feet | 5564 feet |
| 3500 feet | 2020 feet | 4996 feet |

Table 12: Ranges of Gun for flight profiles
The flight profiles are analyzed until the UAV has flown from maximum range to minimum range for incoming UAVs and from minimum range to maximum range for outgoing UAVs. Crossing Shots are evaluated from acquisition at maximum range to loss of acquisition at maximum range (from left to right).

## I. LIMITATIONS AND ASSUMPTIONS

## - PROJECTILES:

- are only effected by gravity and head on drag forces;
- are ineffective beyond 6100 feet;
- fly a straight line between $1 / 10^{\text {th }}$ second updates;
- do not detonate upon impact of a UAV surface because of the soft nature of the UAVs surfaces;
- have a random muzzle velocity of $3608+/-16.4$ feet per second;
- have an angular dispersion of .05006219 degrees.
- UAV:
- flies at a constant altitude (2500 or 3500 feet) and
- flies at a constant speed ( 65 knots).
- GUNNER:
- detects all UAVs in the engagement footprint,
- can accurately determine the speed and altitude of the UAV,
- is oriented on the target line prior to the UAV crossing the engagement footprint,
- is only posed with one threat UAV at a time,
- is oriented on the target line for 0 degree incoming/outgoing shots and oriented near (within 223 feet) the target line for 45 degree offset simulations,
- has unlimited ammunition.


## J. SUMMARY

This chapter presents a complete simulation model with a scenario similar to that of a BSFV in the Air Defense role, a UAV with characteristics similar to that of a Pioneer class UAV, and a projectile and launcher modeled after the BSFV turret and the M792 25 mm Heat round. Algorithms developed to determine the penetration of any UAV surface by the projectile and to aim in a manner similar to that of doctrinal US Army firing techniques allow for testing of the Modeled Launcher and Projectile with aiming capabilities against the simulated UAV using the profiles described earlier. Chapter V reviews each profile, displays the results, and offers an analysis of each run.

## V. RESULTS

## A. 0 DEGREE OFFSET INBOUND/OUTBOUND

## 1. Introduction

The following tables show the results obtained using the simulation described in Chapter IV of a BSFV against the UAV flying the 0 degree offset inbound/outbound flight profile. The tables show the acquired range where the perfect gunner establishes his aim and commences firing, and the cease fire range where the perfect gunner stops engaging due to the UAV passing beyond the wall of fire. At each acquired range it is assumed that the target has no damage. The simulation results shown included randomly generated angular and velocity dispersions, so repeat runs will give different results. However, these results are typical of results obtained from other runs.

| \&.l. «»": | N/: | \& |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1443 | 1577 | 16 | 2 | 0 | 2 | 0 | 0 | 4 |
| 1577 | 1719 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 1719 | 1869 | 13 | 2 | 0 | 0 | 0 | 0 | 2 |
| 1869 | 2019 | 13 | 2 | 0 | 0 | 1 | 0 | 3 |
| 2019 | 2181 | 14 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2181 | 2349 | 15 | 2 | 1 | 0 | 0 | 0 | 3 |
| 2349 | 2528 | 16 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2528 | 2720 | 17 | 0 | 1 | 0 | 0 | 0 | 1 |
| 2720 | 2924 | 18 | 0 | 0 | 0 | 0 | 0 | 0 |
| 2924 | 3146 | 19 | 2 | 0 | 0 | 0 | 0 | 0 |
| 3146 | 3380 | 20 | 2 | 1 | 0 | 0 | 0 | 3 |
| 3380 | 3632 | 22 | 4 | 0 | 0 | 0 | 0 | 4 |
| 3632 | 3908 | 24 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3908 | 4196 | 25 | 0 | 1 | 0 | 0 | 0 | 1 |
| 4196 | 4508 | 27 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4508 | 4838 | 28 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4838 | 5192 | 30 | 0 | 1 | 0 | 2 | 0 | 3 |
| 5192 | Max | 33 | 0 | 2 | 0 | 0 | 0 | 2 |

Table 13: 2500 Foot Altitude, 0 Degree Offset Outbound Simulation

|  |  |  |  |  |  | 友 |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 5564 | 4867 | 63 | 0 | 0 | 0 | 1 | 0 | 0 |
| 4867 | 4310 | 47 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4310 | 3860 | 38 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3860 | 3482 | 32 | 4 | 0 | 1 | 0 | 0 | 5 |
| 3482 | 3158 | 28 | 0 | 1 | 1 | 0 | 0 | 2 |
| 3158 | 2876 | 24 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2876 | 2624 | 21 | 2 | 0 | 1 | 0 | 0 | 3 |
| 2624 | 2396 | 19 | 0 | 1 | 0 | 0 | 0 | 1 |
| 2396 | 2192 | 17 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2192 | 2000 | 16 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2000 | 1820 | 15 | 2 | 0 | 0 | 0 | 0 | 2 |
| 1820 | 1652 | 14 | 2 | 0 | 1 | 1 | 0 | 4 |
| 1652 | 1496 | 13 | 2 | 1 | 1 | 0 | 0 | 4 |
| 1496 | Min | 3 | 1 | 0 | 0 | 0 | 0 | 1 |

Table 14: 2500 Foot Altitude, 0 Degree Offset Inbound Simulation

|  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4996 | 4491 | 47 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4491 | 4053 | 37 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4053 | 3664 | 33 | 0 | 0 | 0 | 0 | 0 | 0 |
| 3664 | 3328 | 28 | 0 | 1 | 1 | 0 | 0 | 2 |
| 3328 | 3022 | 26 | 4 | 0 | 1 | 0 | 0 | 5 |
| 3022 | 2740 | 24 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2740 | 2482 | 22 | 2 | 1 | 0 | 0 | 0 | 3 |
| 2482 | 2248 | 20 | 2 | 0 | 0 | 0 | 0 | 2 |
| 2248 | 2020 | 19 | 2 | 1 | 0 | 0 | 0 | 3 |
| 2020 | Min | 0 |  |  |  |  |  |  |

Table 15: 3500 Foot Altitude, 0 Degree Offset Inbound Simulation


Table 16: 3500 Foot Altitude, 0 Degree Offset Outbound Simulation

## 2. Analysis

The tables clearly show that the simulated gun and projectile are capable of penetrating the simulated UAVs surfaces in most engagement sequences (only 9 of 54 engagement sequences resulted in 0 hits). Clearly the abundance of body hits are a result of firing on the target line and the relatively large number of body surfaces (top body, bottom body, left body, right body, nose and rear). Similarly, the lack of any success against the UAVs fins are a result of the fin not being exposed on an inbound/outbound simulation run. From this, it can be concluded that the rate of fire and aiming method coupled with the characteristics of the projectile are indeed adequate for engaging the UAV inside the range of the projectile.

## B. 45 DEGREE OFFSET CROSSING

## 1. Introduction

The following tables show the results obtained using the simulation described in Chapter IV of the BSFV against the UAV flying the 45 degree offset crossing flight profile. The tables show the acquired range where the perfect gunner establishes his aim
and commences firing, and the cease fire range where the perfect gunner stops engaging due to the UAV passing beyond the wall of fire. At each acquired range it is again assumed that the target has no damage. As in the 0 degree offset cases, these simulation results also included randomly generated angular and velocity dispersions, so repeat runs will give different, although similar, results.

|  |  |  |  | Kises | Yis. |  |  | $\sum_{i}$ | Kisk |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4330 | 4244 | 45 | 18 | 0 | 0 | 1 | 0 | 0 | 1 |
| 4244 | 4167 | 47 | 18 | 0 | 0 | 0 | 1 | 0 | 1 |
| 4167 | 4092 | 49 | 13 | 0 | 1 | 0 | 0 | 0 | 1 |
| 4092 | 4022 | 52 | 13 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4022 | 3955 | 54 | 13 | 0 | 0 | 0 | 1 | 0 | 1 |
| 3955 | 3895 | 56 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3895 | 3839 | 59 | 12 | 2 | 0 | 0 | 0 | 1 | 3 |
| 3839 | 3790 | 61 | 12 | 0 | 0 | 0 | 1 | 1 | 2 |
| 3790 | 3742 | 64 | 12 | 0 | 0 | 1 | 1 | 0 | 2 |
| 3742 | 3701 | 67 | 11 | 2 | 0 | 1 | 0 | 0 | 3 |
| 3701 | 3664 | 69 | 11 | 0 | 1 | 0 | 0 | 0 | 1 |
| 3664 | 3632 | 72 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3632 | 3605 | 75 | 11 | 0 | 0 | 1 | 0 | 0 | 1 |
| 3605 | 3582 | 77 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3582 | 3564 | 79 | 11 | 0 | 0 | 0 | 0 | 1 | 1 |
| 3564 | 3550 | 83 | 11 | 0 | 0 | 0 | 0 | 0 | 0 |
| 3550 | 3540 | 86 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3540 | 3536 | 89 | 11 | 0 | 0 | 0 | 0 | 1 | 1 |
| 3536 | 3536 | 92 | 12 | 0 | 0 | 0 | 1 | 0 | 1 |
| 3536 | 3540 | 95 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3540 | 3549 | 98 | 12 | 0 | 0 | 0 | 1 | 0 | 1 |
| 3549 | 3563 | 101 | 11 | 0 | 1 | 0 | 0 | 0 | 1 |
| 3563 | 3582 | 104 | 12 | 0 | 0 | 0 | 0 | 0 | 0 |
| 3582 | 3606 | 106 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3606 | 3634 | 109 | 12 | 2 | 1 | 0 | 0 | 0 | 3 |
| 3634 | 3667 | 112 | 12 | 0 | 0 | 1 | 0 | 0 | 1 |
| 3667 | 3704 | 113 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3704 | 3748 | 116 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3748 | 3796 | 119 | 12 | 2 | 0 | 0 | 0 | 0 | 2 |
| 3796 | 3846 | 121 | 12 | 0 | 0 | 0 | 0 | 0 | 0 |
| 3846 | 3902 | 123 | 12 | 0 | 1 | I | 0 | 0 | 2 |
| 3902 | 3962 | 126 | 12 | 2 | 0 | 1 | 0 | 1 | 4 |
| 3962 | 4029 | 128 | 13 | 2 | 0 | 0 | 1 | 0 | 3 |
| 4029 | 4100 | 130 | 12 | 2 | 0 | 0 | 0 | 0 | 0 |
| 4100 | 4178 | 132 | 13 | 0 | 0 | 0 | 0 | 1 | 1 |
| 4178 | 4260 | 133 | 13 | 0 | 0 | 0 | 0 | 1 | 1 |
| 4260 | 4349 | 135 | 14 | 0 | 0 | 0 | 0 | 1 | 1 |

Table 17: 2500 Foot Altitude, 45 Degree Offset Crossing Simulation

| 4730 | 4662 | 45 | 17 | 0 | 1 | 0 | 0 | 0 | 1 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4662 | 4600 | 47 | 12 | 0 | 0 | 0 | 1 | 0 | 1 |
| 4600 | 4543 | 50 | 12 | 0 | 0 | 1 | 0 | 0 | 1 |
| 4543 | 4489 | 53 | 12 | 0 | 1 | 0 | 0 | 0 | 1 |
| 4489 | 4441 | 55 | 11 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4441 | 4397 | 57 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4397 | 4356 | 60 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4356 | 4318 | 63 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4318 | 4285 | 65 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4285 | 4257 | 68 | 11 | 0 | 1 | 0 | 1 | 0 | 2 |
| 4257 | 4232 | 71 | 11 | 2 | 1 | 0 | 0 | 0 | 3 |
| 4232 | 4210 | 74 | 11 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4210 | 4192 | 77 | 11 | 0 | 1 | 0 | 0 | 0 | 1 |
| 4192 | 4169 | 80 | 11 | 0 | 1 | 0 | 0 | 0 | 1 |
| 4169 | 4163 | 83 | 11 | 0 | 0 | 0 | 0 | 1 | 1 |
| 4163 | 4160 | 86 | 11 | 0 | 1 | 0 | 0 | 0 | 1 |
| 4160 | 4162 | 90 | 11 | 0 | 0 | 1 | 0 | 0 | 1 |
| 4162 | 4167 | 93 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4167 | 4176 | 96 | 11 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4176 | 4189 | 99 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4189 | 4206 | 103 | 11 | 2 | 1 | 0 | 0 | 0 | 3 |
| 4206 | 4226 | 105 | 11 | 2 | 0 | 0 | 1 | 0 | 3 |
| 4226 | 4250 | 108 | 11 | 2 | 0 | 1. | 0 | 0 | 3 |
| 4250 | 4279 | 111 | 11 | 2 | 1 | 0 | 0 | 0 | 3 |
| 4279 | 4312 | 114 | 11 | 0 | 0 | 0 | 0 | 1 | 1 |
| 4312 | 4349 | 117 | 11 | 2 | 0 | 0 | 0 | 0 | 2 |
| 4349 | 4389 | 120 | 11 | 2 | 0 | 1 | 0 | 0 | 3 |
| 4389 | 4435 | 122 | 12 | 2 | 0 | 1 | 1 | 0 | 4 |
| 4435 | 4484 | 125 | 12 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4484 | 4358 | 127 | 12 | 0 | 0 | 0 | 1 | 1 | 2 |
| 4358 | 4597 | 129 | 13 | 0 | 0 | 1 | 0 | 0 | 1 |
| 4597 | 4660 | 131 | 13 | 0 | 0 | 0 | 0 | 0 | 0 |
| 4660 | 4727 | 135 | 13 | 0 | 1 | 0 | 0 | 0 | 1 |

Table 18: 3500 Foot Altitude, 45 Degree Offset Crossing Simulation

## 2. Analysis

The tables clearly show that the simulated gun and projectile are capable of penetrating the simulated UAVs surfaces in most engagement sequences (only 10 of 70 engagement sequences resulted in 0 hits). Clearly the success of engaging the UAV in all parts is due to the fact that as the UAV crosses, its wings, extensions, body and tail each are exposed to the wall of fire. Once again, the simulation establishes the capability of the projectile to successfully penetrate the UAV given the aiming method and rate of fire utilized

## VI. CONCLUSIONS AND RECOMMENDATIONS

## A. CONCLUSIONS

The goal of this thesis was to develop a method to determine the ability of the Bradley Stinger Fighting Vehicle (BSFV) 25 mm Cannon to counter the emerging threat of Unmanned Aerial Vehicles (UAVs). The modeled objects all inherit unclassified characteristics of the BSFV, one of its primary ammunition rounds, and a representative UAV. Although many assumptions were made to simplify the model, simulation runs clearly demonstrate that the rate of fire utilized in the simulation along with the aiming system of placing a wall of fire in front of the target allow the projectile to penetrate the surface of the UAV in 125 out of 154 independent runs of inbound, outbound and crossing patterns. From these tests, it would seem that the modeled BSFV, given assumptions made about the gunner's aiming method and acquisition capability, is in fact capable of successfully engaging the modeled Pioneer class UAV. This conclusion, however, is only relative to this model and may not be substantiated in actual physical experimentation. Clearly, the most questionable assumption in the simulation is that the gunner can acquire the UAV and orient the gun on the target line at maximum range. One critical, and perhaps most unrealistic, implication made by this assumption is the ability of the gunner to aim the system accurately in front of the UAVs flight path. However, in the author's opinion, it is not unrealistic to assume that today's research and development efforts may in fact allow that capability to become a reality in the near future through the development of appropriate training aids to the gunner coupled with advances in aiming and detection systems.

## B. RECOMMENDATIONS FOR FUTURE RESEARCH

There are many ways to improve upon this methodology or to build upon it in its current form. Much work remains to be accomplished regarding the number of engagements a system can expect to get with the current human acquisition capability and sensor acquisition. Additionally, the effects of nature on all facets of the model could be added. To do this it would be necessary to bring into bear the effects of weather, terrain, vision and human ability into the model. In its current state, the model can provide some
insight into the probability of hit and kill based on some analysis of the surface area vulnerabilities to the projectile in use and the results obtained in Chapter V. The source code and algorithms developed in this research could be improved upon to work in a more efficient and possibly more realistic manner, especially if used with classified data regarding the gun and target

Several simplifying assumptions were made during the design and implementation of the simulation in order to allow the author to complete and test a fully functional methodology. Each assumption should be examined and relaxed to better reflect a true representation of the modeled objects.

Lastly, it would be useful for this model to be ported to a desktop Personal Computer (PC) which would involve rewriting Appendix B (Camera Code) to conform to a PC platform.

## APPENDIX A: SCREEN IMAGES

The following screen images provide an example of the output generated during a simulation run with the UAV flying a crossing, inbound and outbound profile. The projectiles are represented by a horizontal line which allows for depth perception; the wider the line the closer the projectile is to the gun

## xtenn-ai4

Bullets fired: 22<br>Location of UAF: (-4171. 99990678092550 0.000 -3500.0 0.00 .0 3.14159265358999300)<br>Range Gun to Target: 5445. 69385417323330<br>Azinuth Gun to Target: 180. 0000021011048330<br>Computing angles for elevation/azinuth to lay wall of fire!<br>Bullets fired: 23<br>Location of vat: (-4447. 9996400703199500 0.080 - 3500.00 .00 .0 3.14159265358979300)<br>Range Gun to Target: 5659. 9205651409960<br>Azinuth Gun to Target: 180.000002101108330<br>Computing angles for elevation/azinuth to lay wall of fire!<br>Hit the vad at: *tail ${ }^{\text {t }}$<br>Bullets fired: 25<br>Location of vay: (-4741.999570316057d0 0.0d0 -3500.0 0.00 .0<br>3.14159265358979300)

## Graphical Display of UAV and Projectiles



UAV CROSSING INTO WALL OF FIRE

Graphical Display of UAV and Projectiles:
Graphical Display of UAV and Projectiles


## UAV INBOUND INTO A WALL OF FIRE

## Graphical Display of UAV and Projectil

 Graphical Display of UAV and Projectiles

## APPENDIX B: SOURCE CODE (UAV AND PROJECTILE FILES)

, File: pioneer-euler-angle-rigid-body.cl Franz Common LISP
; ** RIGID BODY CLASS DEFINITION **
; Defines a Pioneer UAV like rigid body
, Code written by R.B. McGhee, Naval Postgraduate School, \& modified by D. Wiley , mcghee@cs.nps.navy.mil
(defconstant *gravity* 32.2185)
(defclass rigid-body
()
((posture ;The vector (xe ye ze phi theta psi).
initform '(-200 00000 )
:initarg :posture
:accessor posture)
(posture-rate ;The vector (xe-dot ye-dot ze-dot phi-dot theta-dot psi-dot). :initarg posture-rate accessor posture-rate)
(velocity ;The six-vector ( $\mathrm{u} v \mathrm{wpq} \mathrm{r}$ ) in body coordinates.
: initform '(0 00000 )
:initarg :velocity
:accessor velocity)
(velocity-growth-rate ;The vector (u-dot v-dot w-dot p-dot q-dot r-dot). accessor velocity-growth-rate)
(forces-and-torques ;The vector (Fx Fy Fz L M N) in body coordinates.
initform (list $00(-$ gravity*) 000$)$
accessor forces-and-torques)
(moments-of-inertia ;The vector ( lx Iy Iz ) in principal axis coordinates. initform '(lll $\left.\begin{array}{lll}1 & 1 & 1\end{array}\right)$
initarg :moments-of-inertia
accessor moments-of-inertia)
(mass
initform 1
initarg :mass
accessor mass)
(node-list $\quad ;(\mathrm{x}$ y z l) in body coord for each node. Starts with ( 0001 ).

$$
\left(\begin{array}{llll}
2 & 1 & 0 & 1
\end{array}\right)(-1-1011)(-11001)(2-1201)(21201) \quad ; 10-14
$$

$$
(-1-12001)(-112001)(-17-5-31)(-17-531)(-20-5-31) ; 15-19
$$

$$
(-175-31)(-17531)(-205-31)(-20531)(-1401) \quad ; 25-29
$$

$$
(-15001)(-17401)(-17501)(-20-501)(-20501)) ; 30-34
$$

;Defines a "pioneer UAV" as default rigid body.
initarg :node-list
: accessor node-list)
(polygon-list
initform '( ; comments read as looking from the "top" at the "body" (1243) ;top body
(1265); front nose
(1375) ;side left_body
(2486) ;side right_body
(5687) ;bottom body
(9 1315 ll ) ; top left wing
(10 141612 );top right wing
(3487) ;rear body
(17 182019 ) ; left tail fin
(25 262827 ) ;right tail fin
(21 222423 ); bottom left extender
(29 303231 ) ;bottom right extender
(23 323433 ) ;bottom tail
)
initarg polygon-list
:accessor polygon-list)
(transformed-node-list ; x y z l) in earth coord for each node in node-list. : accessor transformed-node-list)
(H-matrix :initform (unit-matrix 4)
:accessor H-matrix)
(time-stamp accessor time-stamp)))
(defmethod initialize ((body rigid-body))
(setf (transformed-node-list body) (node-list body))
(setf (velocity-growth-rate body) (update-velocity-growth-rate body))
(setf (posture-rate body) (earth-velocity body))
(setf (time-stamp body) (get-internal-real-time)))
(defmethod update-posture ((body rigid-body) delta-t) ; Euler integration.
(setf (posture-rate body) (earth-velocity body))
(setf (posture body)
(vector-add (posture body) (scalar-multiply delta-t (posture-rate body)))))
(defmethod move-body ((body rigid-body) azimuth elevation roll x y z)
(setf (posture body) (list x y z roll elevation azimuth))
(setf (H-matrix body)
(homogeneous-transform azimuth elevation roll $x$ y $z$ ))
(transform-node-list body))
(defmethod update-rigid-body ((body rigid-body)) ;Euler integration. (let* ((delta-t (get-delta-t body)))
(update-posture body delta-t)
(setf (H-matrix body) (homogeneous-transform (sixth (posture body))
(fifth (posture body)) (fourth (posture body)) (first (posture body))
(second (posture body)) (third (posture body))))
(transform-node-list body)
(update-velocity body delta-t)
(update-velocity-growth-rate body)))
(defmethod get-delta-t ((body rigid-body)) 1)
(defmethod update-velocity ((body rigid-body) delta-t) ;Euler integration.
(setf (velocity body)
(vector-add (velocity body)
(scalar-multiply delta-t (velocity-growth-rate body)))))
(definethod earth-velocity ((body rigid-body))
(let* ((linear-velocity (firstn 3 (velocity body))) (rotational-velocity (cdddr (velocity body))) (posture (posture body)) (R-matrix (rotation-matrix (sixth posture) (fifth posture) (fourth posture))) (linear-earth-velocity (post-multiply R-matrix linear-velocity)) (T-matrix (body-rate-to-euler-rate-matrix (sixth posture) (fifth posture) (fourth posture))) (rotational-earth-velocity (post-multiply T-matrix rotational-velocity)))
(append linear-earth-velocity rotational-earth-velocity)))
(defmethod transform-node-list ((body rigid-body))
(setf (transformed-node-list body) (mapcar \#'(lambda (node-location)
(post-multiply (H-matrix body) node-location)) (node-list body))))
;
; ** RIGID BODY CLASS DEFINITION **
, Defines a bullet like rigid body
, Code written by D. A. Wiley, Naval Postgraduate School, , dawiley@cs.nps.navy.mil
(defclass bullet (rigid-body)
((forces-and-torques :initform '(0 000000 ) )
(old-posture ;The vector (xe ye ze phi theta psi) initform '(0 00000 ) initarg :old-posture accessor old-posture)
(time-of-flight
initform 0 :accessor time-of-flight)
(polygon-list :initform '((1 2)))))
(setf kl 0.0001)
(defmethod update-velocity-growth-rate ((body bullet))
(let ((u-dot (- (* $(-\mathrm{kl})$ ) first (velocity body)) (abs (first (velocity body))))
(* ( $\sin ($ fifth $($ posture body $))){ }^{*}$ gravity*) $)$ )
(w-dot (+ (* (-k1) (third (velocity body)) (abs (third (velocity body))))
(* ( $\cos ($ fifth (posture body))) *gravity*))))
(setf (velocity-growth-rate body) (list u-dot 0 w-dot 000 ))))
(defun distance-bullet-plane ( xp xb yp yb zp zb )
(sqrt (+(square(- xp xb))(square(- yp yb))(square(- zp zb)))))
(defun update_bullets (bullet-list)
(cond ((null bullet-list) nil)
(t (update-rigid-body (first bullet-list))
(if (relevent-bullet airplane-1 (first bullet-list))
(setf bullet-1 (cons (first bullet-list)
(update_bullets (rest bullet-list))))))))
(defun relevent-bullet (plane bullet)
(if (and (above-ground-bullet (third (posture bullet)))
(< (distance-gun-object bullet)
(distance-gun-object airplane-1)))t))
(defun above-ground-bullet (projectile)
(if (< projectile 1.0) t))
(defun distance-gun-object (object)
(let ((x (first (posture object)))(y (second (posture object)))
(z (third (posture object))))
(sqrt (+ (square $x)($ square $y)($ (square $z)))$ ))
(defmethod update-posture ((bullet bullet) delta-t) (setf (posture-rate bullet) (earth-velocity bullet)) (setf (old-posture bullet) (posture bullet)) (setf (posture bullet)
(vector-add (posture bullet) (scalar-multiply delta-t (posture-rate bullet)))))
, File: uav-components.cl Franz Common LISP
** The makeup of a UAV rigid body and functions for screen output, this file is used in conjuntion with the determine-hit.cl for testing independently the surface of a UAV **
; Code written by D. A. Wiley, Naval Postgraduate School, ; dawiley@cs.nps.navy.mil

```
(defconstant *top-body* '((-5 -1 0)(9-1 0)(9 1 0)(-5 1 0)))
(defconstant *nose* '((9-1 0)(9 1 0)(91 2)(9-1 2)))
(defconstant *left-body* '((9-1 2)(9-1 0)(-5 -1 0)(-5 -1 2)))
(defconstant *right-body* '((-5 1 2)(-5 1 0)(9 1 0)(9 1 2)))
(defconstant *bottom-body* '((9-1 2)(9 1 2)(-5 1 2)(-5 -1 2)))
(defconstant *left-wing* '((-1 -12 0)(2-12 0)(2-10)(-1 -1 0)))
(defconstant *right-wing* '((-1 1 0)(2 1 0)(2 12 0)(-1 12 0)))
(defconstant *left-tail-extender*'((-17 -5 0)(-1 -5 0)(-1 -4 0)(-17 -4 0)))
(defconstant *right-tail-extender* '((-17 4 0)(-1 4 0)(-1 5 0)(-17 5 0)))
(defconstant *left-tail-fin* '((-20 -5 3)(-17 -5 3)(-17 -5 -3)(-20 -5 -3)))
(defconstant *right-tail-fin* '((-20 5 3 3)(-17 5 3)(-17 5 -3)(-20 5 -3)))
(defconstant *tail* '((-20 -5 0)(-17 -5 0)(-17 5 0)(-20 5 0)))
(defconstant *rear* '((-5 -1 2)(-5 -1 0)(-5 1 0)(-5 1 2)))
```

```
(defconstant *uav-components* (list *top-body* *nose* *left-body*
*right-body* *bottom-body* *left-wing*
*right-wing* *left-tail-extender* *right-tail-extender*
*left-tail-fin* *right-tail-fin*
*tail* *rear*))
(defconstant *uav-components-strings* ("*top-body*" "*nose*""*left-body*""*right-body*"
    "*bottom-body*""*left-wing*""*right-wing*""*left-tail-extender*"
    "*right-tail-extender*""*left-tail-fin*""*right-tail-fin*""*tail*""*rear*"))
(setf *bullet-#* 0)
(defun provide-feedback (uav-part bullet-ID)
    (cond ((equal uav-part (nth 0*uav-components*))
    (format t "~%Hit the UAV at: ~A" (nth 0 *uav-components-strings*)))
    ((equal uav-part (nth 1 *uav-components*))
    (format t "~%Hit the UAV at: ~A" (nth l *uav-components-strings*)))
    ((equal uav-part (nth 2 *uav-components*))
    (format t "~%Hit the UAV at: ~A" (nth 2 *uav-components-strings*)))
    ((equal uav-part (nth 3 *uav-components*))
    (format t "~%Hit the UAV at: ~A" (nth 3 *uav-components-strings*)))
    ((equal uav-part (nth 4 *uav-components*))
    (format t "~%Hit the UAV at: ~A" (nth 4 *uav-components-strings*)))
    ((equal uav-part (nth 5 *uav-components*))
    (format t "~%Hit the UAV at: ~A" (nth 5 *uav-components-strings*)))
    ((equal uav-part (nth 6 *uav-components*))
    (format t "~%Hit the UAV at:: ~A" (nth 6 *uav-components-strings*))
```

((equal uav-part (nth 7 *uav-components*))
(format t "~\%Hit the UAV at: ~A" (nth 7 *uav-components-strings*))) ((equal uav-part (nth 8 *uav-components*))
(format t " $\sim \mathrm{H}$ Hit the UAV at: $\sim \mathrm{A}$ " (nth 8 *uav-components-strings*)))
((equal uav-part (nth 9 *uav-components*))
(format t " $\sim \%$ Hit the UAV at: $\sim \mathrm{A}$ " (nth 9 *uav-components-strings*)))
((equal uav-part (nth 10 *uav-components*))
(format t " $\sim \%$ Hit the UAV at: $\sim$ A" (nth 10 *uav-components-strings*)))
((equal uav-part (nth 11 *uav-components*))
(format t " $\sim \%$ Hit the UAV at: ~A" (nth 11 *uav-components-strings*)))
((equal uav-part (nth 12 *uav-components*))
(format t "~\%Hit the UAV at: ~A" (nth 12 *uav-components-strings*)))
( t (pprint "error in providefeedback function"))))

## APPENDIX C: SOURCE CODE (STROBE CAMERA FILE)

File: camera.cl
** CAMERA CLASS DEFINITION **
A Camera "takes a picture" of rigid-body class objects ; and displays the image. A sequence of images may be ; displayed by superimposing them or by first erasing the display , window and then creating and displaying the next image.
; Requires: rigid-body.cl
; by Shirley Isakari CS4314 Winter 1994 Final Project
; Modifications \& enhancements to Prof. McGhee's Strobe-Camera CLOS code
(require :xcw)
(use-package :cw) ; Note that this is required for use of mouse and color
; This forced renaming of some original functions, i.e.
; move and translate. Causes some problem when compiling.
(cw:initialize-common-windows)
(defclass strobe-camera (rigid-body)
((focal-length :accessor focal-length
:initform 150)
(posture
:accessor posture ; x y z phi theta psi
:initform (list 000000 ))
(camera-window
:accessor camera-window

```
initform (cw:make-window-stream :borders 5
```

:left 300
:bottom 300
:width $400 ; 1000$
:height $400 ; 900$
title "Graphical Display of UAV and Projectiles"
:background-color white
:foreground-color white :activate-p t))
(H-matrix
initform (homogeneous-transform 000000 ))
(inverse-H-matrix
accessor inverse-H-matrix
initform (inverse-H (homogeneous-transform 000000 )) )
(enlargement-factor
accessor enlargement-factor
:initform 100)))
(defun create-camera-1 ()
(setf camera-1 (make-instance 'strobe-camera))
(queue-mouse camera-1))
, *** Defined global color constants $* * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * ~+~$
To be used as the draw-color argument in take-picture and new-picture
; functionss (and also jack-picture, jack-video, jack-movie functions)
(defconstant *white* 0)
(defconstant *yellow* 1)
(defconstant *red* 2)
(defconstant *green* 3)
(defconstant *black* 4)
(defconstant *cyan* 5)
(defconstant *magenta* 6)
(defconstant *blue* 7)
; *** Draw picture functions(defmethod take-picture ((camera strobe-camera) (body rigid-body) draw-color)(let ((camera-space-node-list (mapcar \#'(lambda (node-location)(post-multiply (inverse-H-matrix camera) node-location))
(transformed-node-list body))))
(dolist (polygon (polygon-list body)) (clip-and-draw-polygon camera polygon camera-space-node-list draw-color)))) (defmethod erase-camera-window ((camera strobe-camera)) (cw:clear (camera-window camera)))
(defmethod erase-block ((camera strobe-camera) (body rigid-body)) (let ((center (perspective-transform camera (post-multiply (inverse-H-matrix camera) (first (transformed-node-list body)))))) (cw:draw-filled-rectangle (camera-window camera) (make-position :x (- (first center) 150)

$$
\text { y }(-(\text { second center }) 150))
$$

300300 :color white)))
(defmethod new-picture ((camera strobe-camera) (body rigid-body) draw-color) (erase-camera-window camera)
(take-picture camera body draw-color))
(defmethod clip-and-draw-polygon
((camera strobe-camera) polygon node-coord-list draw-color)
(do* ((initial-point (nth (first polygon) node-coord-list))
(from-point initial-point to-point)
(remaining-nodes (rest polygon) (rest remaining-nodes))
(to-point (nth (first remaining-nodes) node-coord-list)
(if (not (null (first remaining-nodes)))
(nth (first remaining-nodes) node-coord-list))))
((null to-point)
(draw-clipped-projection camera from-point initial-point draw-color))
(draw-clipped-projection camera from-point to-point draw-color)))
(defmethod draw-clipped-projection ((camera strobe-camera)
from-point to-point draw-color)
(cond ((and $(<=$ (first from-point) (focal-length camera))
(<= (first to-point) (focal-length camera))) nil)
$((<=$ (first from-point) (focal-length camera))
(draw-line-in-window camera
(perspective-transform camera (from-clip camera from-point to-point))
(perspective-transform camera to-point) draw-color))
((<= (first to-point) (focal-length camera))
(draw-line-in-window camera
(perspective-transform camera from-point) draw-color))
(t (draw-line-in-window camera (perspective-transform camera from-point) (perspective-transform camera to-point) draw-color))))
(defmethod from-clip ((camera strobe-camera) from-point to-point)
(let ((scale-factor (/ (- (focal-length camera) (first from-point))

$$
(-(\text { first to-point })(\text { first from-point })))))
$$

(list (+ (first from-point)
(* scale-factor (- (first to-point) (first from-point)))) (+ (second from-point) (* scale-factor (- (second to-point) (second from-point))))
(+ (third from-point)
(* scale-factor (- (third to-point) (third from-point)))) 1)))
(defmethod to-clip ((camera strobe-camera) from-point to-point)
(from-clip camera to-point from-point))
(defmethod draw-line-in-window ((camera strobe-camera) start end draw-color)
(cond ( $(=0$ draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start))
(cw:make-position :x (first end) :y (second end))
:brush-width 0 :color white))
$((=1$ draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start))
(cw:make-position :x (first end) :y (second end))
brush-width 0 :color yellow))
(( $=2$ draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start)) (cw:make-position :x (first end) :y (second end)) brush-width 0 :color red))
((= 3 draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start)) (cw:make-position :x (first end) :y (second end)) :brush-width 0 :color green))
((= 4 draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start))
(cw:make-position :x (first end) :y (second end)) brush-width 0 :color black))
( $(=5$ draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start))
(cw:make-position :x (first end) :y (second end)) :brush-width 0 :color cyan))
((= 6 draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start))
(cw:make-position :x (first end) :y (second end)) :brush-width 0 :color magenta))
((= 7 draw-color) (cw:draw-line (camera-window camera)
(cw:make-position :x (first start) :y (second start))
(cw:make-position :x (first end) :y (second end)) brush-width 0 :color blue))))
(defmethod perspective-transform ((camera strobe-camera) point-in-camera-space)
(let* ((enlargement-factor (enlargement-factor camera)) (focal-length (focal-length camera))
( x (first point-in-camera-space)) ; x axis is along optical axis ( y (second point-in-camera-space)) y is out right side of camera ( $z$ (third point-in-camera-space))) $; z$ is out bottom of camera (list (+ (round (* enlargement-factor (/ (* focal-length y) x))) 200) ;to right in camera window $(+200$ (round (* enlargement-factor (/ (* focal-length $(-z)) x))$
))))) ;up in camera window;500 500
; *** Position camera functions *
(defmethod move-camera ((camera strobe-camera) azimuth elevation roll x y z ) (setf (posture camera) (list x y z roll elevation azimuth)) (setf (H-matrix camera) (homogeneous-transform azimuth elevation roll x y z)) (setf (inverse-H-matrix camera) (inverse-H (H-matrix camera))))
(defmethod zoom-camera ((camera strobe-camera) zoom-amount) (setf (slot-value camera 'enlargement-factor)
(+ (slot-value camera 'enlargement-factor) zoom-amount)))
(defun deg-to-rad (angle) (* . 017453292519943295 angle))
(defconstant tilt-limit (* . 017453292519943295 89.9))
; ${ }^{* * *}$ Auxiliary functions
(defun kill ()
(cw:kill-common-windows))
(defun reset-windows ()
(kill)
(cw:initialize-common-windows))

## APPENDIX D: SOURCE CODE (PERFECT AUTOPILOT FILE)

File: pioneer-perfect-autopilot cl Franz Common LISP
; ** PERFECT AUTOPILOT CLASS DEFINITION **
; A Perfect Autopilot steers the body axes of the rigid-body to the
; desired orientation with no time delay.
, Requires: pioneer-euler-angle-rigid-body.cl
; Code written by R.B. McGhee, Naval Postgraduate School, \& modified by D Wiley , mcghee@cs.nps.navy.mil

水水 $* * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * * *$
(defclass perfect-autopilot ()
((vehicle-name
; This is the name of an instance of the rigid-body class :accessor vehicle-name)
(current-trajectory-segment
accessor current-trajectory-segment)
(current-time
:initform 0
accessor current-time)
(longitudinal-acceleration-gain accessor longitudinal-acceleration-gain)
(trajectory-segment-list
:accessor trajectory-segment-list)))
;This is a list of lists. Each list contains
;start-time and commanded speed,
;heading-rate, and depth. Last segment is end-time followed by nil.
(defmethod initialize-4 ((autopilot perfect-autopilot) vehicle gain trajectory) (setf (longitudinal-acceleration-gain autopilot) gain (trajectory-segment-list autopilot) (rest trajectory) (current-trajectory-segment autopilot) (first trajectory) (vehicle-name autopilot) vehicle))
(defmethod update-segment ((autopilot perfect-autopilot) time) (if (and (not (null (second (current-trajectory-segment autopilot)))) (>= time (caar (trajectory-segment-list autopilot)))) (setf (current-trajectory-segment autopilot) (pop (trajectory-segment-list autopilot)))))
(defmethod commanded-velocity ((autopilot perfect-autopilot) delta-t) (setf (current-time autopilot) (+ (current-time autopilot) delta-t)) (update-segment autopilot (current-time autopilot)) (if (second (current-trajectory-segment autopilot)) (list (+ (first (velocity (vehicle-name autopilot))) (* (longitudinal-acceleration autopilot) delta-t)) 00 (fourth (current-trajectory-segment autopilot)) (fifth (current-trajectory-segment autopilot)) (third (current-trajectory-segment autopilot))))) (defmethod longitudinal-acceleration ((autopilot perfect-autopilot)) (* (longitudinal-acceleration-gain autopilot) (- (second (current-trajectory-segment autopilot))
(first (velocity (vehicle-name autopilot))))))
(defmethod move-vehicle ((autopilot perfect-autopilot) delta-t) (setf (velocity (vehicle-name autopilot)) (commanded-velocity autopilot delta-t)) (when (second (velocity (vehicle-name autopilot))) (update-posture (vehicle-name autopilot) delta-t) (setf (H-matrix (vehicle-name autopilot)) (homogeneous-transform (sixth (posture (vehicle-name autopilot))) (fifth (posture (vehicle-name autopilot))) (fourth (posture (vehicle-name autopilot))) (first (posture (vehicle-name autopilot))) (second (posture (vehicle-name autopilot))) (third (posture (vehicle-name autopilot))))) (transform-node-list (vehicle-name autopilot))))
(defmethod accelerometer-output ((autopilot perfect-autopilot))
(let ((longitudinal-velocity (first (velocity (vehicle-name autopilot))))
(pitch-angle (fifth (posture (vehicle-name autopilot)))) (roll-angle (fourth (posture (vehicle-name autopilot)))) (pitch-rate (fifth (velocity (vehicle-name autopilot)))) (yaw-rate (sixth (velocity (vehicle-name autopilot))))) (list (+ (longitudinal-acceleration autopilot)

```
    (* *gravity* (sin pitch-angle)))
(- (* longitudinal-velocity yaw-rate)
    (* *gravity* (cos pitch-angle) (sin roll-angle)))
    (+ (- (* longitudinal-velocity pitch-rate))
        (- (* *gravity* (cos pitch-angle) (cos roll-angle)))))))
```

```
(defmethod mission-data ((autopilot perfect-autopilot))
    (append (IMU-data autopilot)
    (list (sixth (posture (vehicle-name autopilot))))))
(defmethod IMU-data ((autopilot perfect-autopilot))
    (cons (current-time autopilot)
            (append (accelerometer-output autopilot)
                (angular-rate-output autopilot))))
(defmethod angular-rate-output ((autopilot perfect-autopilot))
    (cons (fourth (velocity (vehicle-name autopilot)))
            (cons (fifth (velocity (vehicle-name autopilot)))
            (list (sixth (velocity (vehicle-name autopilot)))))))
(defun initialize-mission (* X** }\mp@subsup{\textrm{Y}}{}{*}*\mp@subsup{\textrm{Z}}{}{*}*\mp@subsup{\textrm{AZM}}{}{*}
    (setf airplane-l (make-instance 'rigid-body))
    (setf autopilot-1 (make-instance 'perfect-autopilot))
    (initialize-4 autopilot-1 airplane-1 1 *trajectory*)
    (move-vehicle autopilot-1 0)
    (setf camera-1 (make-instance 'strobe-camera))
    (move-camera camera-1 000000)
    (setf (first(posture airplane-1)) *X*)
    (setf (second(posture airplane-1))* Y*)
    (setf (third(posture airplane-1)) *Z*)
    (setf (sixth(posture airplane-1)) *AZM*)
    (new-picture camera-1 airplane-1 *black*)
    (setf *bullets-fired* 0)
```

(perfect-gunner-aim-shoot airplane-1 2) (dolist (bullet bullet-1) (take-picture camera-1 bullet *red*))) ;(queue-mouse))
(defun execute-mission () (do* ((firecontrol 0 (+ firecontrol 1))
(mission-data (list (mission-data autopilot-1))
(cons (mission-data autopilot-1) mission-data))
(new-node-list (move-vehicle autopilot-1 .1) (move-vehicle autopilot-1 .1)))
((not (second (velocity (vehicle-name autopilot-1))))
(setf *mission-data* (reverse mission-data)))
(update_bullets bullet-1)
(dolist (bullet bullet-1)
(if (close-to-hit (posture airplane-1)(posture bullet))
(test-for-hit bullet airplane-1)))
(if (and (> (first (posture airplane-1)) 1)
(< (second (posture airplane-1)) .1))
(gunner-fire-bullet-in *azimuth* *elevation* firecontrol))
(if (and (< (first (posture airplane-1)) 1)
(< (second (posture airplane-1)) .1))
(gunner-fire-bullet-out *azimuth* *elevation* firecontrol))
(if (>(second (posture airplane-1)) .1) (gunner-fire-bullet-offset
*azimuth* *elevation* firecontrol))
(new-picture camera-1 airplane-1 *black*)
(dolist (bullet bullet-1) (take-picture camera-1 bullet *red*))))
(setf *trajectory* '((060 000 ) (100 nil)))

## APPENDIX E: SOURCE CODE (DETERMINE A IIIT FILE)

; File: determine-hit.cl

Franz Common LISP

; ** Functions to test a projectile for penetration of a **
** UAV surface
; Code written by D. WILEY, Naval Postgraduate School, dawiley@cs.nps.navy.mil
(defconstant * distance* 223)
(defconstant *range-of-projectile* 6200)
(defun test-for-hit (bullet plane)
(let ((bullet-prev (convert-bullet (old-posture bullet)plane))
(bullet-curr (convert-bullet (posture bullet) plane)))
(cross-infinite-planep bullet-prev bullet-curr bullet)))
(defun convert-bullet (bullet plane)
(let ((bullet-pos (bullet-in-plane-coordinates bullet plane)))
(list (first bullet-pos)(second bullet-pos)(third bullet-pos))))
(defun bullet-in-plane-coordinates (bullet plane)
(let $((\mathrm{xb}$ (first bullet) $)(\mathrm{yb}$ (second bullet))
(zb (third bullet)))
(post-multiply (inverse-H (H-matrix plane))(list xb yb zb 1))))
;Tests to see if the old bullet position and its current bullet have crossed ;the geometric plane that the UAV component lies in, if that has happened ;it then calls a function to find the
; X Y and Z coordinates of the bullet on the geometric plane (defun cross-infinite-planep (bullet-prev bullet-curr bullet-ID) (dolist (plane *uav-components*)
(let* ((uav-component plane)
(point-on-geo-plane (first plane))
(vecl2 (vector-subtract (second plane)(first plane)))
(vecl4 (vector-subtract (fourth plane)(first plane)))
(normal-plane (cross-product vecl4 vec12))
(vecl-bulletcurr(vector-subtract bullet-curr (first plane)))
(vecl-bulletprev(vector-subtract bullet-prev (first plane)))
(result l (dot-product normal-plane vecl-bulletcurr))
(result2 (dot-product normal-plane vecl-bulletprev)))
(if (signs-samep result 1 result 2 ) $t$
(bullet-on-plane bullet-prev bullet-curr normal-plane point-on-geo-plane uav-component bullet-ID)))))
(defun signs-samep (x y)

$$
(\text { if }(\operatorname{or}(\operatorname{and}(>=x 0)(>=y 0))(\operatorname{and}(<=x 0)(<=y 0))) t))
$$

;Determines the X Y and Z coordinate of a bullet on a geometric plane ; given the previous position, new position and a point on the geometric plane.
(defun bullet-on-plane (bullet-prev bullet-curr normal point uav-component bullet-ID)
(let* ((bpath (vector-subtract bullet-curr bullet-prev))
(xpath (first bpath))(ypath (second bpath))(zpath (third bpath))
(a (first normal))(b (second normal))(c (third normal))
(x (first bullet-prev) $)(\mathrm{y}$ (second bullet-prev) $)(\mathrm{z}$ (third bullet-prev) $)$
(xPlane (first point))(yPlane (second point))(zPlane (third point))
$\left(\mathrm{d}\left(-\left(+(*\right.\right.\right.$ a xPlane $)\left({ }^{*}\right.$ b yPlane $)\left({ }^{*}\right.$ c zPlane $\left.\left.\left.)\right)\right)\right)$
$\left(\mathrm{k}\left(-\left(/\left(+\left({ }^{*}\right.\right.\right.\right.\right.$ a x $)\left({ }^{*}\right.$ b y $)\left({ }^{*}\right.$ c z $\left.) \mathrm{d}\right)\left(+\left(^{*}\right.\right.$ a xpath $)\left({ }^{*}\right.$ b ypath $)\left({ }^{*}\right.$ c zpath $\left.\left.\left.\left.)\right)\right)\right)\right)$
(planeX $\left(+\mathrm{x}\left({ }^{*} \mathrm{k}\right.\right.$ xpath $\left.)\right)$ )
(plane $\mathrm{Y}\left(+\mathrm{y}\left({ }^{*} \mathrm{k}\right.\right.$ ypath $\left.\left.)\right)\right)\left(\right.$ planeZ $\left(+\mathrm{z}\left({ }^{*} \mathrm{k}\right.\right.$ zpath $\left.\left.\left.)\right)\right)\right)$
(angle-sum-test (list planeX planeY planeZ) uav-component bullet-ID)))
,Tests a bullet to see if it lies om any part of a rectangle on a plane by ;summing the angle between vectors and checking for the sum to be close to 2 pi (defun angle-sum-test (bullet-on-plane UAV-surface bullet-ID) (let* ((p UAV-surface)(b bullet-on-plane)
(vl (vector-subtract (first $p) b))(v 2($ vector-subtract $(\operatorname{second} p) b))$
(v3 (vector-subtract (third p) b))(v4 (vector-subtract (fourth p) b))
(angle-sum (+ (angle-between-vectors vl v2)(angle-between-vectors v2 v3)
(angle-between-vectors $\vee 3 \mathrm{v} 4)($ angle-between-vectors v 4 v 1$)$ )) )
(if (and ( $>$ angle-sum 6.25)(< angle-sum 6.30)) ; threshold for around 2 pi (provide-feedback p bullet-ID))))
(defun angle-between-vectors (v1 v2)
(let ((result (/ (dot-product vl v2)
(*(vector-magnitude v 1$)($ vector-magnitude v 2$))$ )) )
(if $(>=$ result 1.0$)(\operatorname{acos} 1.0)(\operatorname{acos}$ result $)))$ )
(defun close-to-hit (plane bullet)
(let $((\mathrm{xp}($ first plane $))$ (yp (second plane) $)(\mathrm{zp}$ (third plane))
$(\mathrm{xb}($ first bullet $))(\mathrm{yb}($ second bullet $))(\mathrm{zb}($ third bullet $)))$
(if (< (distance-bullet-plane xp xb yp yb zp zb) *distance*) t)))
(defun distance-bullet-plane (xp xb yp yb zp zb)
(sqrt (+(square(-xp xb))(square(-yp yb))(square(-zp zb)))))
(defun update_bullets (bullet-list) (cond ((null bullet-list) nil)
(t (update-rigid-body (first bullet-list))
(if (relevent-bullet airplane-1 (first bullet-list)) (setf bullet-1 (cons (first bullet-list)
(update_bullets (rest bullet-list)))))))))
(defun relevent-bullet (plane bullet)
(if (and (above-ground-bullet (third (posture bullet)))
$(<($ distance-gun-object bullet $) *$ range-of-projectile* $)) \mathrm{t}))$
(defun above-ground-bullet (projectile)
(if (< projectile 1.0) t))
(defun distance-gun-object (object)
(let $((\mathrm{x}$ (first $($ posture object $)))(\mathrm{y}($ second $($ posture object $)))$
(z (third (posture object))))
$($ sqrt $(+($ square $x)($ square $y)($ square $z)))))$

## APPENDIX F: SOURCE CODE (ADJUST FIRE FILE)

; File: adjust-fire cl Franz Common LISP
;
; ${ }^{* *}$ Gunners elevation and azimuth computations ${ }^{* *}$
; A Perfect Gunner establishes a wall of fire in front
; of a UAV until the UAV is killed or past the wall
; of fire
; Code written by D. A. Wiley, Naval Postgraduate School,
; dawiley@cs.nps.navy.mil
(defconstant *super-elevation* .0175) ; approximately 1 degree .0175
(defconstant *max-elevation-of-gun* 1.047197)
(defconstant *specified-velocity* 3608 )
(defconstant * velocity-dispersion-feet* 16.4) ;+-5 meters per shot $=16.4 \mathrm{ft}$
(defconstant *elevation-step* .005)
(defconstant *lead-radians* .02)
(defconstant *angular-dispersion-radians* 0.00087375)
(defun gunner-fire-bullet-offset (azimuth elevation firecontrol) (if (or(= (length bullet-1) 0)
(>(azimuth-angle (first bullet-1))(- (azimuth-angle airplane-1)
*lead-radians*)))
; fire a bullet if there has not been one fired or
; your azimuth to bullet is > azimuth to plane
(cond ((evenp firecontrol)
(let ((x (first (posture camera-1)))

```
                    (y (second (posture camera-1)))
                    (z (third (posture camera-1)))
                    (roll (fourth (posture camera-1)))
                    (bullet (make-instance 'bullet))
                    (azmRandom (get-random-angle azimuth))
                    (elevRandom (get-random-angle elevation)))
                    (initialize bullet)
                    (move-body bullet azmRandom elevRandom roll x y z)
                    (move-camera camera-1 azimuth elevation roll x y z)
                    (setf (first(velocity bullet)) (get-random-velocity))
                    (push bullet bullet-1)
                    (setf *bullets-fired* (+ *bullets-fired* 1))
                    (if (>= (fifth (posture(first bullet-1)))
            *max-elevation-of-gun*)
            (pprint "At positive elevation limit")))
            t t))
                    (perfect-gunner-aim-shoot airplane-1 2)))
(defun gunner-fire-bullet-in (azimuth elevation firecontrol)
    (if (or(= (length bullet-1) 0)
    (>(fifth (posture (first bullet-1))) (- (shot-1-theta-1 airplane-1)
        *lead-radians*)))
    ;fire a bullet if there has not been one fired or if
    ;you are still in front of the target, *lead-radians*
    ;assures plane has passed through wall of fire
    (cond ((evenp firecontrol)
```

(let ((x (first (posture camera-1)))
(y (second (posture camera-1)))
(z (third (posture camera-1)))
(roll (fourth (posture camera-1)))
(bullet (make-instance 'bullet))
(azmRandom (get-random-angle azimuth))
(elevRandom (get-random-angle elevation)))
(initialize bullet)
(move-body bullet azmRandom elevRandom roll x y z)
(move-camera camera-1 azimuth elevation roll x y z )
(setf (first(velocity bullet)) (get-random-velocity))
(push bullet bullet-1)
(setf *bullets-fired* (+ *bullets-fired* 1))
(if (>=(fifth (posture(first bullet-1)))
*max-elevation-of-gun*)
(pprint "At positive elevation limit")))
t t))
(perfect-gunner-aim-shoot airplane-1 2)))
(defun gunner-fire-bullet-out (azimuth elevation firecontrol)
(if (or (= (length bullet-1) 0)
$(<($ fifth $($ posture $($ first bullet-1) $))(+$ (shot-1-theta-1 airplane-1)
*lead-radians*)))
;fire a bullet if there has not been one fired or if ;you are still in front of the target, *lead-radians* ; assures plane has passed through wall of fire
(cond ((evenp firecontrol)
(let ((x (first (posture camera-1)))
(y (second (posture camera-1)))
(z (third (posture camera-1)))
(roll (fourth (posture camera-1)))
(bullet (make-instance 'bullet))
(azmRandom (get-random-angle azimuth))
(elevRandom (get-random-angle elevation)))
(initialize bullet)
(move-body bullet azmRandom elevRandom roll x y z )
(move-camera camera-1 azimuth elevation roll x y z )
(setf (first(velocity bullet)) (get-random-velocity))
(push bullet bullet-1)
(setf *bullets-fired* (+ *bullets-fired* 1))
(if $(>=($ fifth $($ posture $($ first bullet-1) $))$
*max-elevation-of-gun*)
(pprint "At positive elevation limit")))
t t)
(perfect-gunner-aim-shoot airplane-1 2)))
(defun perfect-gunner-aim-shoot (UAV firecontrol)
(format t " $\sim \%$ Bullets fired: $\sim A^{\prime \prime}$ *bullets-fired*)
(format t " $\sim$ Location of UAV: $\sim \mathrm{A}^{\prime \prime}$ (posture airplane-1))
(format t " $\sim \%$ Range Gun to Target: $\sim A$ " (distance-gun-object airplane-1))
(format $t " \sim \%$ Azimuth Gun to Target: $\sim A "($ rad-to-deg (azimuth-angle UAV))) (setf *bullets-fired* 0 )

```
    (setf bullet-1 nil)
    (if (and (> (first (posture UAV)) I)(< (second (posture UAV)).1))
        (determine-angles-inbound UAV firecontrol))
    (if (and (< (first (posture UAV)) 1)(< (second (posture UAV)). 1))
        (determine-angles-outbound UAV firecontrol))
    (if (>(second (posture UAV)) .1)
        (determine-angles-offset UAV firecontrol)))
(defun determine-angles-offset (UAV firecontrol)
    (let* ((theta-1 (shot-1-theta-1 UAV))
        (theta-2 (shot-1-theta-2-inbound UAV theta-1))
        (theta-3 (shot-1-theta-3-inbound UAV))
        (azm (azimuth-angle (UAV-at-future-point UAV))))
    (setf *azimuth* azm)
    (format t "~%~%Computing angles for elevation/azimuth to lay wall of fire!")
    (setf *elevation* (determine-elevation-offset UAV))
    (gunner-fire-bullet-offset *azimuth* * elevation* firecontrol)))
(defun determine-elevation-offset (UAV)
(let ((new-UAV (UAV-at-future-point UAV)))
    (do ((elevation (get-elev-angle-to-UAV new-UAV)(+ elevation *elevation-step*))
        (close-flag nil))
        (close-flag elevation)
        (do ((test-bullet (make-test-bullet elevation)))
        ((or(not(relevent-bullet new-UAV test-bullet))
        close-flag
        (< (third (posture test-bullet))(third (posture new-UAV))))
```

nil)
(update-rigid-body test-bullet)
(if (bullet-path-thru-UAV-plane-p new-UAV test-bullet)
(setf close-flag 1))))))
(defun bullet-path-thru-UAV-plane-p (UAV bullet)
(let ((uav-Z (third (posture UAV)))
(prev-bullet-Z (third (old-posture bullet)))
(curr-bullet-Z (third (posture bullet))))
(if (and (> prev-bullet-Z uav-Z)(<curr-bullet-Z uav-Z))t)))
(defun make-test-bullet (elevation)
(let ((test-bullet (make-instance 'bullet)))
(initialize test-bullet)
(move-body test-bullet *azimuth* elevation 0000 )
(setf (first (velocity test-bullet)) *specified-velocity*)
test-bullet))
(defun UAV-at-future-point (UAV)
(let((new-UAV (make-instance 'rigid-body)))
(setf (first(posture new-UAV))
(- (first (posture UAV))100))
(setf (second(posture new-UAV)) (second(posture UAV)))
(setf (third(posture new-UAV))(third(posture UAV)))
(setf $(\operatorname{sixth}($ posture new-UAV $))(\operatorname{sixth}($ posture UAV $)))$

```
    new-UAV))
```

(defun determine-angles-inbound (UAV firecontrol)
(let* ((theta-1 (shot-1-theta-1 UAV))
(theta-2 (shot-1-theta-2-inbound UAV theta-1))
(theta-3 (shot-1-theta-3-inbound UAV))
(azm (azimuth-angle UAV)))
(setf *azimuth* azm)
(format t " $\sim \% \sim \%$ Computing angles for elevation/azimuth to lay wall of fire!")
$\left(\right.$ setf $*$ elevation* $\left(+(+ \text { theta- } 3(- \text { theta-l theta- } 2))^{*}\right.$ super-elevation*) $)$
(gunner-fire-bullet-in *azimuth* *elevation* firecontrol)))
(defun determine-angles-outbound (UAV firecontrol)
(let* ((theta-1 (shot-1-theta-1 UAV))
(theta-2 (shot-1-theta-2-outbound UAV theta-1))
(theta-3 (shot-l-theta-3-outbound UAV))
(azm (azimuth-angle UAV)))
(setf * azimuth* azm)
(format t "~\%Computing angles for elevation/azimuth to lay wall of fire!")
(setf *elevation* (- (+ theta-3 (- theta-1 theta-2)) *super-elevation*))
(gunner-fire-bullet-out *azimuth* *elevation* firecontrol)))
(defun shot-1-theta-1 (UAV)
(let ((X (first (posture UAV)))
(Y (second (posture UAV)))
( $\mathrm{Z}($ third (posture UAV) $))$ )
(elev-angle X Y Z ) ) )
(defun shot-1-theta-2-inbound (UAV theta-1)
(let* ((Bullet (actual-shots-location-inbound theta-l UAV))
(X (first (posture Bullet)))
(Y (second (posture UAV)))
(Z (third (posture Bullet))))
(elev-angle X Y Z)))
(defun shot-1-theta-2-outbound (UAV theta-1)
(let* ((Bullet (actual-shots-location-outbound theta-1 UAV))
(X (first (posture Bullet)))
( Y (second (posture UAV)))
(Z (third (posture Bullet))))
(elev-angle X Y Z)))
(defun actual-shots-location-inbound (elevation-angle UAV)
(setf *test-bullet* (make-instance 'bullet))
(initialize *test-bullet*)
(move-body *test-bullet* (azimuth-angle UAV) elevation-angle 0000 )
(fire-test-bullet UAV))
(defun actual-shots-location-outbound (elevation-angle UAV)
(setf *test-bullet* (make-instance 'bullet))
(initialize *test-bullet*)
(move-body *test-bullet* (azimuth-angle UAV) elevation-angle 0000 )
(fire-test-bullet UAV))
(defun fire-test-bullet (UAV)
(setf (time-of-flight *est-bullet*) 0)
(setf (first (velocity *test-bullet*)) *specified-velocity*)
(do ()
((and (> (abs (first (posture *test-bullet*)))
(abs (first (posture UAV))))
$(<($ third(posture *test-bullet*) $)($ third(posture airplane-1)))))
(update-rigid-body *test-bullet*)
(update-time-of-flight *test-bullet*))
*test-bullet*)
(defun shot-1-theta-3-inbound (UAV)
(let ((X (- (first (posture UAV)) (* (* (time-of-flight *test-bullet*) .1) 6)))
( Y (second (posture UAV)))
(Z (third (posture UAV))))
(elev-angle X Y Z) ))
(defun shot-1-theta-3-outbound (UAV)
(let ((X (+ (first (posture UAV)) (* (* (time-of-flight *test-bullet ${ }^{*}$ ) . 1) -6)))
(Y (second (posture UAV)))
(Z (third (posture UAV))))
(elev-angle X Y Z)))
(defun get-elev-angle-to-UAV (UAV)
(let ((X (first (posture UAV)))
(Y (second (posture UAV)))
(Z (third (posture UAV))))
(elev-angle X Y Z)))
(defun elev-angle (X Y Z)
(atan (abs Z) (sqrt(+ (* X X)(* Y Y)))))
(defun azimuth-angle (UAV)
(let ((X (first (posture UAV)))
(Y (second (posture UAV))))
(atan Y X)))
(defun update-time-of-flight (bullet)
(setf (time-of-flight bullet) (+ (time-of-flight bullet) $)$ ))
(defun update-time-of-flight (bullet)
(setf (time-of-flight bullet) (+ (time-of-flight bullet)l))
(defun get-random-velocity ()
$(+$ *specified-velocity* (**velocity-dispersion-feet* (-(random 2.0) 1))))
(defun get-random-angle (angle)
(+ angle (**angular-dispersion-radians* (- (random 2.0) 1))))

## APPENDIX G: SOURCE CODE (MOUSE HANDLER FILE)

; File: mouse cl Franz Common LISP
: Mouse handler to allow adjusting azimuth/elevation of the launcher, zooming ; of the camera and firing of a bullet
; Must be invoked with the call (queue-mouse) at start of simulation ; Code written by D. T. Davis, Naval Postgraduate School, \& modified by D. Wiley;
(defconstant * 1 degree* 0.017453 )
(defconstant *positive-elevation* 1.05)
(defconstant *negative-elevation* -0.158)
(defmethod queue-mouse ()
(cw:sun-enable-super-and-hyper)
(cw:modify-window-stream-method (camera-window camera-1) :left-button-down :after 'mouse-handler)
(cw:modify-window-stream-method (camera-window camera-1) :middle-button-down :after 'mouse-handler)
(cw:modify-window-stream-method (camera-window camera-1) :right-button-down :after 'mouse-handler))
(defun mouse-handler (wstream cw:mouse-state \&optional event)
(let ((button-state (cw:mouse-button-state)))
;(format t "Mouse Handler Invoked: ~a ~\%" button-state)
(case button-state
(128 (slew-left)) ; left click
(64 (fire-bullet)) ; middle click
(32 (slew-right)) ; right click
(136 (zoom-out-X2)) ; left click + alt
(40 (zoom-in-X2)) ; right-click + alt
(144 (slew-up)) ; left click + shift
(48 (slew-down))))) ; right click + shift
(defun slew-left ()
(let ((x (first (posture camera-1)))
(y (second (posture camera-1)))
(z (third (posture camera-1)))
(azimuth (- (sixth (posture camera-1)) ${ }^{*} 1$ degree*) $)$
(elevation (fifth (posture camera-1)))
(roll (fourth (posture camera-1))))
(move-camera camera-1 azimuth elevation roll $x$ y z)))
(defun slew-right ()
(let ((x (first (posture camera-1)))
(y (second (posture camera-1)))
(z (third (posture camera-1)))
(azimuth (+ (sixth (posture camera-1)) *1degree*))
(elevation (fifth (posture camera-1)))
(roll (fourth (posture camera-1))))
(move-camera camera-1 azimuth elevation roll x y z)))
(defun slew-up ()
(cond $((<$ (fifth (posture camera-1)) *positive-elevation*)
(let ((x (first (posture camera-1)))
(y (second (posture camera-1)))
(z (third (posture camera-1)))
(azimuth (sixth (posture camera-1)))
$\left(\right.$ elevation $\left(+\left(f i f t h(\right.\right.$ posture camera-1) $) * 1$ degree $\left.\left.^{*}\right)\right)$
(roll (fourth (posture camera-1))))
(move-camera camera-1 azimuth elevation roll x y z )))
(t (pprint "At positive elevation limit"))))
(defun slew-down ()
(cond $((>($ fifth $($ posture camera-1) $) *$ negative-elevation*)
(let ((x (first (posture camera-1)))
(y (second (posture camera-1)))
(z (third (posture camera-1)))
(azimuth (sixth (posture camera-1)))
(elevation $(-($ fifth $($ posture camera-1) $) * 1$ degree* $))$
(roll (fourth (posture camera-1))))
(move-camera camera-1 azimuth elevation roll x y z)))
(t (pprint "At negative elevation limit"))))
(defun zoom-in-X2 ()
(let ((x (focal-length camera-1)))
(if $(<=x$ l)
(setf (focal-length camera-1) 2)
( $\operatorname{setf}($ focal-length camera-1) (* x 2) )) )
(format t "Scope set to power: $\sim \mathrm{a} \sim \%$ " (focal-length camera-1)))
(defun zoom-out-X2 ()
(let $((x$ (focal-length camera-1) $))$
(if $(<=x 2)$
(setf (focal-length camera-1) 1)
$(\operatorname{setf}($ focal-length camera-1) $(/ \times 2))))$
(format t "Scope set to power: $\sim \mathbf{a} \sim \%$ " (focal-length camera-1)))
(defun fire-bullet ()
(let ((x (first (posture camera-1)))

```
(y (second (posture camera-1))) (z (third (posture camera-1)))
(azimuth (sixth (posture camera-1)))
(elevation (fifth (posture camera-1)))
(roll (fourth (posture camera-1)))
(bullet (make-instance 'bullet)))
(initialize bullet)
(move-body bullet azimuth elevation roll x y z)
(setf(velocity bullet) '(360800000))
(push bullet bullet-1)))
```


## APPENDIX H: SOURCE CODE (ROBOT KINEMATICS FILE)

; File: robot-kinematics.cl

Franz Common LISP

, Utility functions
; Code written by R.B. McGhee, Naval Postgraduate School, \& modified by D. Wiley ; mcghee@cs.nps.navy.mil
(defun transpose (matrix) ;A matrix is a list of row vectors. (cond ((null (cdr matrix)) (mapcar 'list (car matrix)))
(t (mapcar 'cons (car matrix) (transpose (cdr matrix) )) )) )
(defun dot-product (vector-1 vector-2) ;A vector is a list of numerical atoms.
(apply '+ (mapcar '* vector-1 vector- 2 ) $)$ )
(defun cross-product (v1 v2)
(cons $\left(-\left(^{*}(\operatorname{second} \vee 1)(\right.\right.$ third $\left.v 2)\right)(*(\operatorname{second} v 2)($ third $\left.\vee 1))\right)$
$\left(\operatorname{cons}\left(-(*(f i r s t \vee 2)(t h i r d \vee 1))\left({ }^{*}(\right.\right.\right.$ first $\vee 1)($ third $\left.\left.\vee 2)\right)\right)$
(list
$(-(*($ first $\vee 1)(\operatorname{second} \vee 2))(*($ first $\vee 2)(\operatorname{second} \vee 1)))))))$
(defun vector-magnitude (vector) (sqrt (dot-product vector vector)))
(defun square ( x ) (* x x ))
(defun post-multiply (matrix vector) (cond ((null (rest matrix)) (list (dot-product (first matrix) vector)))
(t (cons (dot-product (first matrix) vector)
(post-multiply (rest matrix) vector)))))
(defun pre-multiply (vector matrix)
(post-multiply (transpose matrix) vector))
(defun matrix-multiply ( $\mathrm{A} B$ ) ; A and B are conformable matrices. (cond ((null (cdr A)) (list (pre-multiply (car A) B)))
(t (cons (pre-multiply (car A) B) (matrix-multiply (cdr A) B)) )) )
(defun chain-multiply ( L ) ; L is a list of names of conformable matrices. (cond ((null (cddr L)) (matrix-multiply (eval (car L)) (eval (cadr L))))
(t (matrix-multiply (eval (car L)) (chain-multiply (cdr L))))))
(defun cycle-left (matrix) (mapcar 'row-cycle-left matrix))
(defun row-cycle-left (row) (append (cdr row) (list (car row))))
(defun cycle-up (matrix) (append (cdr matrix) (list (car matrix))))
(defun unit-vector (one-column length) ;Column count starts at 1
(do ((n length $(1-n))$
(vector nil (cons (cond $((=$ one-column n) 1$)(\mathrm{t} 0))$ vector $))$ )
((zerop $n)$ vector)))
(defun unit-matrix (size)
(do ((row-number size (1-row-number))
(I nil (cons (unit-vector row-number size) I)))
((zerop row-number) I)))
(defun concat-matrix ( $\mathrm{A} B$ ) ; A and B are matrices with equal number of rows (cond ((null A) B)
$(\mathrm{t}(\operatorname{cons}(\operatorname{append}(\operatorname{car} A)(\operatorname{car} B))($ concat-matrix $(\operatorname{cdr} A)(\operatorname{cdr} B))))))$ (defun augment (matrix) (concat-matrix matrix (unit-matrix (length matrix))))
(defun normalize-row (row) (scalar-multiply (/ 1.0 (car row)) row))
(defun scalar-multiply (scalar vector)
(cond ((null vector) nil)
(t (cons (* scalar (car vector))
(scalar-multiply scalar ( cdr vector) $)$ ))))
(defun solve-first-column (matrix) ;Reduces first column to ( $10 \ldots 0$ ) (do* ((remaining-row-list matrix (rest remaining-row-list))
(first-row (normalize-row (first matrix)))
(answer (list first-row)
(cons (vector-add (first remaining-row-list)
(scalar-multiply (- (caar remaining-row-list))
first-row))
answer)) )
((null (rest remaining-row-list)) (reverse answer))))
(defun vector-add (vector-1 vector-2) (mapcar '+ vector-1 vector-2))
(defun vector-subtract (vector-1 vector-2) (mapcar '- vector-1 vector-2))
(defun first-square (matrix) ;Returns leftmost square matrix from argument. (do ((size (length matrix))
(remainder matrix (rest remainder))
(answer nil (cons (firstn size (first remainder)) answer)))
((null remainder) (reverse answer))))
(defun firstn ( n list)
(cond ((zerop n) nil)
( $\mathrm{t}(\mathrm{cons}($ first list $)($ firstn $(1-\mathrm{n})($ rest list $))))))$
(defun max-car-firstn (n list)
(append (max-car-first (firstn n list)) (nthcdr n list)))
(defun matrix-inverse (M)
(do ((Ml (max-car-first (augment M))
(cond ((null M1) nil) ;Abort for singular matrix.
(t (max-car-firstn $n($ cycle-left (cycle-up M1) $))$ )))
$(\mathrm{n}(1-($ length M$))(1-\mathrm{n})))$
((or (minusp n) (null M1)) (cond ((null M1) nil) (t (first-square M1))))
(setq M1 (cond ((zerop (caar M1)) nil) (t (solve-first-column M1))))))
(defun max-car-first ( L ) ; L is a list of lists. This function finds list with
(cond ((null $(\mathrm{cdr} \mathrm{L})) \mathrm{L})$; largest car and moves it to head of list of lists.
$(\mathrm{t}($ if $(>($ abs $($ caar $L))($ abs $($ caar $($ max-car-first $(c d r L))))) L$
(append (max-car-first (cdr L)) (list (car L)))))))
(defun dh-matrix (cosrotate sinrotate costwist sintwist length translate)
(list (list cosrotate (- (* costwist sinrotate))
(* sintwist sinrotate) (* length cosrotate))
(list sinrotate (* costwist cosrotate)
(- (* sintwist cosrotate)) (* length sinrotate))
(list 0. sintwist costwist translate) (list 0.0.0.1.))) (defun homogeneous-transform (azimuth elevation roll x y z)
(let ((spsi (sin azimuth)) (cpsi (cos azimuth)) (sth (sin elevation))
$(\operatorname{cth}(\cos$ elevation $))($ sphi $(\sin$ roll $))($ cphi $(\cos$ roll $)))$
(list (list (* cpsi cth) (- (* cpsi sth sphi) (* spsi cphi)) .
$\left(+\left({ }^{*}\right.\right.$ cpsi sth cphi) (* spsi sphi)) x)
(list (* spsi cth) (+ (* cpsi cphi) (* spsi sth sphi))
(- (* spsi sth cphi) (* cpsi sphi)) y)
(list (-sth) (* cth sphi) (* cth cphi) z)
(list 0.0.0.1.))))
(defun inverse-H $(\mathrm{H}) \quad ; \mathrm{H}$ is a $4 \times 4$ homogeneous transformation matrix (let* ((minus-P (list (- (fourth (first H)))
(- (fourth (second H$)$ ))
(- (fourth (third H)))))
(inverse-R (transpose (first-square (reverse (rest (reverse H)))))) (inverse-P (post-multiply inverse-R minus-P)))
(append (concat-matrix inverse-R (transpose (list inverse-P)))
(list (list 0001 )))))
(defun rotation-matrix (azimuth elevation roll)
(let ((spsi (sin azimuth)) (cpsi (cos azimuth)) (sth (sin elevation))
(cth (cos elevation)) (sphi (sin roll)) (cphi (cos roll)))
(list (list (* cpsi cth) (- (* cpsi sth sphi) (* spsi cphi))
(+ (* cpsi sth cphi) (* spsi sphi)))
(list (* spsi cth) (+ (* cpsi cphi) (* spsi sth sphi))
(- (* spsi sth cphi) (* cpsi sphi)))
(list (- sth) (* cth sphi) (* cth cphii)))))
(defun body-rate-to-euler-rate-matrix (azimuth elevation roll) (let ((sth (sin elevation)) (cth (cos elevation)) (tth (tan elevation))
(sphi (sin roll)) (cphi (cos roll)))
(list (list 1 (* thh sphi) (* tth cphi))
(list 0 cphi (- sphi))
(list $0(/$ sphi cth) (/ cphi cth)))))
(defun rad-to-deg (angle) (*57.2957795130823 angle)) (defun deg-to-rad (angle) (*. 017453292519943295 angle))

## APPENDIX I: SOURCE CODE (COMPILE AND LOAD FILES)

; File: compile-files cl Franz Common LISP
; Defines basic function to run simulation and compiles all ; files used in the simulation.
; To run simulation, inside the CLOS interpreter do the following
; > (load "compile-files")
; > (load "load-files")
; $>$ (test X Y Z) ; where X Y Z are the coordinates of the UAV
; Z is negative for altitudes above ground
Code written by D. A. Wiley, Naval Postgraduate School,
; dawiley@cs.nps.navy.mil
(compile-file "strobe-camera.cl")
(compile-file "robot-kinematics.cl")
(compile-file "pioneer-euler-angle-rigid-body.cl")
(compile-file "pioneer-perfect-autopilot.cl")
(compile-file "bullet.cl")
(compile-file "mouse.cl")
(compile-file "determine-hit.cl")
(compile-file "uav-components.cl")
(compile-file "adjust-fire.cl")
;UAV will fly from north to south along X axis (defun test (X Y Z)
(initialize-mission X Y Z 3.141592653589793d0)
File: Load-files.cl Franz Common LISP
Loads compiled files; Code written by D. A. Wiley, Naval Postgraduate School,; dawiley@cs.nps.navy.mil
(load "strobe-camera.fasl")
(load "robot-kinematics.fasl")
(load "pioneer-euler-angle-rigid-body.fasl")(load "pioneer-perfect-autopilot.fasl")
(load "bullet.fasl")
(load "mouse.fasl")
(load "determine-hit.fasl")
(load "uav-components.fasl")
(load "adjust-fire.fasl")

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