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## Retaining-Walls for Earth.

INCLUDING

THE THEORY OF EARTH-PRESSURE<br>AS DEvELOPED FROM THE ELLIPSE OF STRESS.

WITH

AN APPENDIX PRESENTING THE THEORY OF PROF. WEYRAUCH.

BY
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Second zedition, Rebised and zenlarged.

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## PREFACE.

The first edition of this work was based upon the theory advanced by Prof. Weyrauch in 1878, but owing to the length of the demonstrations used by him, it was thought advisable to present different and shorter demonstrations in this edition. To show that the new demonstrations give identical results with those obtained by Prof. Weyrauch, his demonstrations have been given in an appendix as they appeared in the first edition.

The new demonstrations are based upon the theory first advanced by Prof. Rankine in 1858. Those readers who are familiar with Rankine's Ellipse of Stress can omit pages 27 to 35, inclusive, in following the demonstrations.

An attempt has been made to present the theory in a shape easily followed by those who have only a knowledge of algebra, geometry, and trigonometry; whenever calculus has been resorted to, the work has been simplified as much as possible. For convenience in practice, the formulas have been arranged in a condensed shape in Part I, and are followed by numerous examples illustrating their application.

The values of various coefficients have been computed and tabulated and will be found to very materially decrease the labor of substitution in the formulas.

It is hoped that the introduction of a brief treatment of the supporting power of earth in the case of foundations, as well as the formula for determining the breadth of the base of a retaining-wall, will prove acceptable.

For valuable help in the verification of proofs of formulas, and the critical reading of the whole text, I acknowledge the kind assistance of Prof. Thos. Gray.
M. A. H.

Terre Haute, Ind., March, 1891.

## NOMENCLATURE.

$\phi=$ the angle of repose, or the maximum angle which any force acting upon any plane within the mass of earth can make with the normal to the plane.
$\epsilon=$ the angle made by the surface of the earth with the horizontal; $\epsilon$ is positive when measured above and negative when measured below the horizontal.
$\alpha=$ the angle which the back of the wall makes with the vertical passing through the heel of the wall; $\alpha$ is positive when measured on the left and negative when measured on the right of the vertical.
$\delta=$ the angle which the direction of the resultant earthpressure makes with the horizontal.
$\phi^{\prime}=$ the angle of friction between the wall and its foundation.
$\phi^{\prime \prime}=$ the angle of friction between the back of the wall and the earth.
$H=$ the vertical height of the wall in feet.
$h=$ the depth of earth in feet which is equivalent to a given load placed upon the surface of the earth.
$B^{\prime}=$ the width in feet of the top of the wall.
$B=$ the width in feet of the base of the wall.
$Q=$ the distance in feet from the toe of the wall to the point where $R$ cuts the base.
$P=$ the resultant earth-pressure in pounds against a vertical wall.
$E=$ the resultant earth-pressure in pounds against any wall.
$R=$ the resultant pressure in pounds on the base of the wall.
$G=$ the total weight in pounds of material in the wall. $\gamma=$ the weight in pounds of a cubic foot of earth.
$W=$ the weight in pounds of a cubic foot of wall.
$p=$ the intensity of the pressure in pounds on the base of the wall at the toe.
$p^{\prime}=$ the intensity of the pressure in pounds on the base of the wall at the heel.
$p_{0}=$ the average intensity of the pressure in pounds on the base of the wall.
$x=H \tan \alpha$.

## RETAINING-WALLS FOR EARTH.

## FORMULAS FOR EARTH-PRESSURE.

In the following formulas $\alpha$ and $\epsilon$ are considered as positive, and the wall is assumed to be one foot long.

Case I. General case of inclined earth-surface and inslined back of wall.

$$
\begin{align*}
E= & \frac{H^{2} \gamma}{2} \frac{\cos (\epsilon-\alpha)}{\cos ^{2} \alpha \cos \epsilon} \times \\
& \sqrt{\sin ^{2} \alpha+\cos ^{2}(\epsilon-\alpha)\left\{\begin{array}{l}
\left.\frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}\right\}^{2}
\end{array}\right.} ;  \tag{1}\\
& +2 \sin \epsilon \sin \alpha \cos (\epsilon-\alpha)\left\{\begin{array}{c}
\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi} \\
\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}
\end{array}\right\}
\end{align*}
$$

or

$$
\begin{aligned}
E & =\frac{H^{2} \gamma}{2}(B) \sqrt{(C)+(D) A^{2}+(E) A .} . \\
\tan \delta & =\frac{\sin \alpha \cos \epsilon+\sin \epsilon \cos (\epsilon-\alpha) A}{\cos \epsilon \cos (\epsilon-\alpha) A} ; . \\
\tan \delta & =\frac{\sin \alpha}{\cos (\epsilon-\alpha) A}+\tan \epsilon, \quad . \quad . \quad .\left(1^{\prime} a\right)
\end{aligned}
$$

where

$$
\begin{equation*}
A=\cos \epsilon \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}} \tag{d}
\end{equation*}
$$

Case II. Surface of earth inclined and $\alpha=0$.
$E=P=\frac{H^{2} \gamma}{2}\left\{\cos \epsilon \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}=A\right\}$.
From Diagram I the values of $A$ can be found for all values of $\phi$ from $0^{\circ}$ to $90^{\circ}$ and of $\epsilon$ from $0^{\circ}$ to $90^{\circ}$, varying by $5^{\circ}$.

$$
\begin{equation*}
\delta=\epsilon ; \tag{2a}
\end{equation*}
$$

or for all vertical walls the direction of the earth-pressure is parallel to the surface of the earth.

Case III. The surface of the earth parallel to the surface of repose.

$$
\epsilon=\phi
$$

$$
E=\frac{H^{2} \gamma}{2} \frac{\cos (\phi-\alpha)}{\cos ^{2} \alpha \cos \phi} \sqrt{\sin ^{2} \alpha+\cos ^{2}(\phi-\alpha)} \begin{align*}
& +2 \sin \alpha \sin \phi \cos (\phi-\alpha) \tag{3}
\end{align*}
$$

$$
\begin{equation*}
\tan \delta=\frac{\sin \alpha+\sin \phi \cos (\phi-\alpha)}{\cos \phi \cos (\phi-\alpha)} \tag{3a}
\end{equation*}
$$

Case IV. The surface of the earth parallel to the surface of repose and the back of the wall vertical.

$$
\begin{gather*}
\epsilon=\phi \quad \text { and } \quad \alpha=0 . \\
E=\frac{H^{2} \gamma}{z} \cos \phi .  \tag{4}\\
\delta=\phi . \tag{4a}
\end{gather*}
$$

## FORMULAS FOR EARTH-PRESSURE.

Case V. The surface of the earlh horizontal.

$$
\begin{gather*}
\epsilon=0 . \\
E=\frac{H^{2} \gamma}{2} \sqrt{\tan ^{2} \alpha+\tan ^{4}\left(45^{\circ}-\frac{\phi}{2}\right)} .  \tag{5}\\
\tan \delta=\frac{\tan \alpha}{\tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right)} . \tag{5a}
\end{gather*}
$$

Case VI. The surface of the earth horizontal and the back of the wall vertical.

$$
\begin{align*}
& \epsilon=0 \text { and } \alpha=0 \\
& E=\frac{H^{2} \gamma}{2} \tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right) . \quad . \quad . \quad .  \tag{6}\\
& \delta=0 . \quad . \quad . \quad . \quad . \quad . \quad . \quad . \quad . \quad . \tag{6a}
\end{align*}
$$

Case VII. Fluid pressure.

$$
\begin{align*}
& \epsilon=\phi=0 \\
& E=\frac{H^{2} \gamma}{2 \cos \alpha} \cdot . \quad . \quad . \quad . \quad .  \tag{7}\\
& \delta=\alpha, . . . . . \tag{}
\end{align*}
$$

Graphical Constructions for determining the Thrust of Earth.

The following constructions are perfectly general, and apply to any plane within a mass of earth. When applied
for determining the thrust of earth against a retaining-wall, $\alpha$ and $\epsilon$ are taken as positive.

> * Construction (a).

Let $B E$ represent the surface of the earth and $B A$ the back of the wall. Draw $A F$ parallel to $B E$, and at any point $D$ in $A F$ lay off $D F$ equal to the vertical $D E$. Draw


Fig. 1.
FG horizontal, and FH, making the angle $\phi$ with $D F$. With any point $J$ in $D F$ describe the are $K I$ tangent to $H F^{\prime}$ at $I$ cutting $F G$ at $K$, and draw $G L$ parallel to $K J$; with $L$ as a centre and $L F$ as radius, describe the circumference $F Q O N$ cutting $A D$ at $N$. Through $N$ draw $N O$

[^0]parallel to $A B$ cutting the circumference $F Q O N$ at $O$; at $A$ draw $A C$ ecual to $O G$ and normal to $A B$; the area of the triangle $A B C$ multiplied by $\gamma$ will be the thrust of the earth on the wall.

To determine the direction of the thrust $E$, prolong $O G$ to $Q$; then $Q N$ will be the direction of the thrust.

This thrust acts on the wall at ${ }_{3}^{2} A B$ below $B$.

* Construction (b).

Let $B Q$ represent the surface of the earth, and $B A$ the back of the wall. Draw $A D$ parallel to $B Q$, and at any


Fig. 2.
point $D$ in $A D$ draw the vertical $D G$ equal to the normal $D Q$; draw $D M$ making the angle $\phi$ with the normal $D Q$.

* This construction follows directly from Rankine's Ellipse of Stress. See Rankine's Applied Mechanics.

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At any point $J$ in $D Q$ as a centre, describe the arc $I K$ tangent to $D M$ cutting $D G$ at $K$, and draw $G L$ parallel to $J K$. Bisect the angle $Q L G$, and at $A$ draw $A P$ parallel to $L R$. At $A$ draw $A N$ nornial to $A B$ and equal to $D L$; with $N$ as a centre and $A N$ as radius, describe an arc $A P$ cutting $A P$ at $I^{\prime}$; connect $P$ and $N$, and make $N O$ equal to $L G$; with $A$ as a centre and $A O$ as a radius, describe the arc $O C$ cutting $A N$ at $C$; then the area of the triangle $A B C$ multiplied by $\gamma$ will be the thrust against the wall. The direction of this thrust is parallel to $A O$ and it is applied at $\frac{2}{3} A B$ below $B$.

The constructions ( $a$ ) and (b) give identical results in every case.

## Trapezoidal and Triangular Walls.

Formulas for the width of the base of trapezoidal walls under the condition that the resultant $R$ cuts the base at a point distant from the toe of the wall equal to one third the width of the base, or $Q=\frac{1}{3} B$.

Case I. The general case in which the back of the watl is inclined, and E makes an angle with the horizontal.

$$
\begin{align*}
B^{2}+B & \left(\frac{4 E}{H W} \sin \delta+B^{\prime}-x\right) \\
& =\frac{2 E}{H W}(H \cos \delta+x \sin \delta)+2 B^{\prime} x+B^{\prime 2} \tag{8}
\end{align*}
$$

Case II. The back of the wall vertical.

$$
x=0
$$

$$
\begin{equation*}
B^{2}+B\left(\frac{4 E}{H W} \sin \delta+B^{\prime}\right)=\frac{2 E}{W} \cos \delta+B^{\prime 2} \tag{9}
\end{equation*}
$$

Case III. The back of the wall vertical and the thrust normal to the wall.

$$
x=0 \quad \text { and } \quad \delta=0
$$

$$
\begin{equation*}
B^{2}+B B^{\prime}=\frac{2 E}{W}+B^{\prime 2} \tag{10}
\end{equation*}
$$



Fig. 3.
If $B=B^{\prime}$ and $x=0$, the section of the wall is a rectangle, and (9) becomes

$$
\begin{equation*}
B^{2}+B \frac{4 E}{H W} \sin \delta=\frac{2 E}{W} \cos \delta, . \tag{9a}
\end{equation*}
$$

and (10) becomes

$$
B=\sqrt{\frac{2 E}{W}} \cdot \quad . \quad . \quad . \quad .(10 a)
$$

Formulas for the width of the base of triangular walls under the condition that the resultant $R$ cuts the base at a point distant from the toe of the wall equal to one third the width of the base, or $Q=\frac{1}{3} B$.

Case I. The general case in which the back of the wall is inclined, and $E$ makes an angle with the horizontal.
$B^{2}+B\left(\frac{4 E}{H W} \sin \delta-x\right)=\frac{2 E}{H W}(H \cos \delta+x \sin \delta)$.
Case II. The back of the wall vertical.

$$
\begin{gather*}
\alpha=0 \\
B^{2}+B\left(\frac{4 E}{H W} \sin \delta\right)=\frac{2 E}{W} \cos \delta \tag{12}
\end{gather*}
$$

Case III. The back of the wall vertical, and the thrust normal to the wall.

$$
\begin{gather*}
x=0 \quad \text { and } \quad \delta=0 . \\
B=\sqrt{\frac{2 E}{W}} . \tag{13}
\end{gather*}
$$

The above formulas do not contain the condition that $R$ shall not make an angle greater than $\phi^{\prime}$ with the normal to the base of the wall.

From Fig. 3,

$$
\begin{equation*}
\tan \phi^{\prime} \geqq \frac{E \cos \delta}{G+E \sin \delta}=\tan L J K, . \tag{14}
\end{equation*}
$$

which expresses the condition under which the wall will not slide,

## Depth of Foundations.

Case I. When the intensity of the pressure on the earth is uniform.

Letting $x^{\prime}$ equal the depth of the foundation below the surface,

$$
\begin{equation*}
x^{\prime}=\frac{p_{0}(1-\sin \phi)^{2}}{(1+\sin \phi)^{2} \gamma-W(1-\sin \phi)^{2}}, . \tag{15}
\end{equation*}
$$

when the weight of the foundation is included; and

$$
\begin{equation*}
x^{\prime}=\left\{\frac{1-\sin \phi}{1+\sin \phi}\right\}^{2} \frac{p_{0}}{\gamma}, \ldots . \tag{16}
\end{equation*}
$$

when the weight of the foundation is not included.
$x^{\prime}$ is the minimum deptis to which the foundation must be extended for equilibrium. The actual depth should be based upon the minimum value which $\phi$ is likely to have under any condition of the earth.

Case II. When the intensity of the pressure on the earth is uniformly varying.

$$
\begin{equation*}
x^{\prime}=\frac{p_{0}}{\gamma} \frac{(1-\sin }{1+\sin ^{2}} \frac{\phi)^{2}}{\phi}, . \tag{19}
\end{equation*}
$$

where $x^{\prime}$ is the minimum depth to which the foundation must be extended for equilibrium;

$$
\begin{equation*}
x_{0}=\frac{1}{3} \frac{\sin \phi}{1+\sin ^{2} \phi}, \tag{20}
\end{equation*}
$$

where $x_{0}$ is the maximum distance from the centre of the base of the foundation to the point where the resultant pressure cuts the base of the foundation.

## Abutting Power of Earth.

$$
\begin{equation*}
P=\frac{\left(x^{\prime}\right)^{2} \gamma}{2} \frac{1+\sin \phi}{1-\sin \phi} \tag{21}
\end{equation*}
$$

where $P$ represents the maximum resultant pressure which horizontal earth can resist, when $P$ is applied against : vertical plane of the depth $x^{\prime}$.

## APPLICATIONS.

The determination of the earth-pressure by the preceding formulas and graphical constructions is a very simple operation when the angle $\phi$ has been determined or assumed. 'That care and judgment be used in assuming the value of $\phi$ is very important, since a change of a few degrees in the value of $\phi$ sometimes causes a large change in the value of $E$. An inspection of Diagram I shows that the value of the coefficient $A$ increases very rapidly as $\phi$ decreases.

When the earth to be retained contains springs, the bank must be thoroughly drained if it is to be retained by an economical tight wall; if it is not drained, the angle $\phi$ will be likely to become very small as the earth becomes wet.

When the location of the earth to be retained is subjected to jurs, the value of $\phi$ will be decreased.

Hence, in assuming the value of $\phi$, the engineer must be sure that the value assumed will be the least value which, in his judgment, it is likely to have.

In constructing the wall the judgment and authority of the engineer must again be exercised in order that the wall be constructed as designed.

In all cases, to insure perfect drainage between the back
of the wall and the earth, numerous " weep-holes" should be provided in the body of the wall, or proper arrangements made to carry away the water at the base of the wall. To facilitate drainage, the backing resting against the wall should be sand or gravel.

In no case should water be permitted to get under the foundation of the wall, neither should the earth in front of the wall be allowed to become wet.

In cold localities the back of the wall near the top should have a large batter to prevent the frost from moving the top courses of stone. As a guard against sliding, the courses of the wall should have very rough beds. The strength of a wall is increased the nearer it approaches a monolith.

Care should be taken to have the foundation broad and deep enough to prevent sliding and upheaving of the earth in front. In clay thie fcundation should be deep, while in sand or gravel it may be broad and shallow.

The following examples illustrate the application of the formulas:

Ex. 1. Design a trapezoidal wall of sandstone, weighing 150 lbs . per cubic foot, having a width of 3 ft . on top, a height of 30 ft ., and the baek inclining forward $5^{\circ}$, to retain a bank of sand sloping upward at an angle of $20^{\circ}$.

## Data.

$\gamma-100 \mathrm{lbs} ., W=150 \mathrm{lbs} . ; \epsilon=20^{\circ}, \phi=39^{\circ}, \alpha=5^{\circ} ;$ $H=30 \mathrm{ft} ., B^{\prime}=3 \mathrm{ft} ., x=2.63 \mathrm{ft}$.
$1^{\circ}$. Graphical determination of the values of $E$ and $\delta$.
The graphical solution of the problem is shown in Fig. 4, where $E$ is found to equal 15,000 pounds. $\delta$ lies between $35^{\circ}$ and $36^{\circ}$.
$2^{\circ}$. Algebraic determination of $E$ and $\delta$.

$$
E=\frac{H^{2} \gamma}{2}(B) \sqrt{(C)+(D) A^{2}+(E) A} .
$$



Fig. 4.
Substituting the values of $B, C, D$, and $E$ as given in the tables, and that of $A$ as given by Diagram I , this becomes $E=\frac{900 \times 100}{2}(1.036) \times$

$$
\sqrt{(0.008)+(1.057)(0264)^{2}}+(0.061) 0.264,
$$

$E=45,000(1.036) \sqrt{0.09}=14,500 \mathrm{lbs}$.

$$
\tan \delta=\frac{\sin \alpha}{\cos (\epsilon-\alpha) A}+\tan \epsilon
$$

$$
\begin{aligned}
& \tan \delta=\frac{0.08 \%}{0.966(0.264)}+0.364 \\
& \tan \delta=0 . \% 05=\tan 35^{\circ} 11^{\prime}, \text { about. }
\end{aligned}
$$

$3^{\circ}$. Alyebraic determination of the value of $B$ under the (1ssumption that $Q=\frac{1}{3} B$.

$$
\begin{aligned}
& B^{2}+B\left\{\frac{4 E}{H W} \sin \delta+B^{\prime}-x\right\} \\
& \quad=\frac{2 E}{H W}\{H \cos \delta+x \sin \delta\}+2 B^{\prime} x+B^{\prime 2} \\
& E^{2}+B\left\{\frac{4 \times 14500}{30 \times 150} 0.576+3-2.63\right\} \\
& =\frac{2 \times 14500}{30 \times 150}\{30 \times 0.817+2.63 \times 0.576\}+6 \times 2.63+9
\end{aligned}
$$

$$
\begin{gathered}
B^{2}+7.79 B=1 \% 2.53 \\
B=-3.89 \pm \sqrt{172.53+\overline{3.9}^{2}}
\end{gathered}
$$

$$
\therefore B=13.69-3.89=9.80 \mathrm{ft} . ;
$$

or, practically, 10 feet is the required width of the base.
$4^{\circ}$. To determine if the wall will slide on a foundation of sandstone.

From (14),

$$
\tan \phi^{\prime} \geqq \frac{E \cos \delta}{G^{\prime}+E \sin \delta^{\circ}}
$$

Taking $B=10 \mathrm{ft} ., G=\frac{10+3}{2} 30 \times 150=29250 \mathrm{lbs}$.
$\delta=35^{\circ} 11^{\prime}, \cos \delta=0.817$, and $\sin \delta=0.5 \% 6$, then

$$
\frac{E \cos \delta}{G+E \sin \delta}=\frac{14500 \times 0.81 \%}{29250+14500 \times 0.5 \tau 6}=0.315 .
$$

From 'Table II, the value of $\tan \phi^{\prime}$ for masonry is 0.6 to 0.7 ; hence there is no danger of the wall sliding on the foundation.
$5^{\circ}$. To determine the minimum depth to which the foundation must extend consistent with the stability of the earth.
First determine the maximum value of $x_{0}$. From (20),

$$
x_{0}=\frac{1}{3} \frac{\sin \phi}{1+\sin ^{2} \phi},
$$

where $\phi$ must be assumed at its minimum value. Assume that the minimum value of $\phi$ in this case is $30^{\circ}$; then

$$
x_{0}=\frac{1}{3} \frac{0.5 \% 7}{1.333}=0.133
$$

showing that the resultant must cut the base of the fomdation within 0.133 feet of its centre. The resultant cuts the base of the wall $1.6 \%$ feet from the centre of its base; hence the width of the foundation must be increased.

Assuming that the depth to which the foundation extends is 4 feet, and that it is vertical in the rear; then the direction of the resultant pressure (not including the additional weight of the foundation) will cut the base of the foundation 7.93 feet from the rear or heel. The required width of the base of the foundation is $(\% .93-0.13) 2=$ 15.6 ; say, 16 feet.

The value of $\nu_{0}$ can now be found, which corresponds to the assumed value of $x^{\prime}=4$ feet.

From (19),

$$
\begin{aligned}
& p_{0}=x^{\prime} \gamma \frac{1+\sin ^{2} \phi}{(1-\sin \phi)^{2}} \\
& p_{0}=400 \frac{1.333}{0.1 \% 9}=2960 \mathrm{lbs}
\end{aligned}
$$

The average intensity of the pressure on the base of the foundation due to the resultant $R$ is

$$
\frac{29250+14500 \sin \delta}{16}=2350 \mathrm{lbs}
$$

The foundation adds an intensity equal to $4 \times 150=600$ pounds approximately; hence the actual value of $p_{0}=2350$ $+600=2950$ pounds ; therefore, if the foundation has a depth of 4 feet and a base of 16 feet, the wall will not sink nor the earth in front of the wall heave, until $\phi$ becomes less than $30^{\circ}$.
$6^{\circ}$. To determine if the wall and foundation will stide on the earth.

This is resisted in two ways-by the friction between the masonry and the earth, and by a prism of earth in front of the wall.

The horizontal force tending to make the wall slide equals $E \sin \delta$, or $14500.0 .5 \% 6=8352$ pounds. The horizontal force tending to make the foundation slide equals the resultant earth-pressure on the rear face of the foundation, which is vertical and 4 feet in height. From (6),

$$
E=\left\{\frac{(30+4)^{2}}{2}-\frac{30^{2}}{2}\right\} \gamma \tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right)
$$

or $\quad E=12800 \times 0.226=2893$.

Then the total horizontal force tending to make the wall slide is

$$
8352+2893=11245 \mathrm{lbs} .
$$

From Table II the tangent of the angle of friction between masonry and moist clay is 0.33 , which evidently is much smaller than the tangent of the actual angle of friction between masonry and dry earth. Assume this tangent to be 0.500 .

The total vertical pressure upon the base of the foundation is $3 \sim 600$ pounds, hence the ability to resist sliding is $3 \% 600(0.5)=18800$ pounds, which is much larger than 11245; hence there is no danger of the wall slipping, even if the earth in front of the wall does not act.

Ex. 2. Design a trapezoidal wall of sandstone weighing 150 lbs . per cubic foot, having a width of 3 ft . on top, a height of 30 ft ., and the back inclining backward $15^{\circ}$, to retain a bank of sand sloping upward at an angle of $30^{\circ}$.

## Data.

$\gamma=100 \mathrm{lbs} ., W=150 \mathrm{lbs} . ; \epsilon=30^{\circ}, \phi=33^{\circ}, \alpha=-15^{\circ}$;
$H=30 \mathrm{ft} ., B^{\prime}=3 \mathrm{ft} ., x=8 \mathrm{ft}$.
$1^{\circ}$. Graphical determination of the values of $E$ and $\delta$.
In Fig. 5, let $E G$ represent the surface of the earth, and $A B$ the back of the wall. Draw $A F^{r}$ parallel to $B G$, and from any point $D^{\prime}$ in $A F$ lay off $D^{\prime} F$ equal to the vertical $D^{\prime} G$, and draw $F L$ horizontal; lay off the angle $I F D^{\prime}=\phi$ $=33^{\circ}$, and locate the point $M$ in $D^{\prime} F$ so that if an arc be described with $M$ as a centre and $L M$ as a radius the are will be tangent to $I F$; then with $M$ as a centre and $M F$ as a radius, describe the circumference $F H J$ and draw $J H$
parallel to $A B$; at $A$ draw $A L$ perpendicular to $A B$ and equal to $H I$. Then

$$
\frac{(A B)(A L)}{2} \gamma=\frac{(30.9)(9.6)}{2} 100=14800=E .
$$

To determine $\delta$, prolong $H I$ to $K$ and draw $K J$. Then the angle which this line makes with the horizontal is equal to $\delta$, which is $6^{\circ}$ to $7^{\circ}$ in this case.


Fig. 5.
$2^{\circ}$. Algebraic determination of $E$ and $\delta$.
Substituting in (1) and remembering that $\alpha$ is negative,

$$
E=45000(0,8 \% 5) \sqrt{0.067+0.183-0.111}=14600 \mathrm{lbs} .
$$

From ( $1^{\prime}($ ),

$$
\tan \delta=\frac{-0.259}{0.707(0.524)}+.577=-0.123=\tan \left(-7^{\circ}\right) .
$$

$3^{\circ}$. Algebraic determination of the value of $B$ under the assumption that $Q=\frac{1}{3} B$.

Substituting the proper values in (11) and remembering that $\alpha$ is negative,

$$
B=-4.7 \pm \sqrt{ } 163.44+(4.7)^{2}=9.0 \mathrm{ft} .
$$

The foundation can be designed in the manner outlined in Ex. 1.

Ex. 3. Determine the dimensions of a brick wall having a vertical back to retain a bank of sand sloping upward at an angle of $20^{\circ} . \quad \phi=30^{\circ}, H=20^{\prime}, B^{\prime}=2^{\prime}$, $\gamma=100$.
$1^{\circ}$. Algebraic determination of $E$ and $\delta$.
Since $\alpha=0$,

$$
\begin{array}{r}
E=\frac{H^{2} \gamma}{2} A \ldots  \tag{2}\\
E=\frac{400 \times 100}{2} 0.424=8480 ; \text { say, } 8500 \mathrm{lbs} .
\end{array}
$$

The value of $A$ is readily found from Diagram I.

$$
\delta=\epsilon=20^{\circ}, \quad \text { since } \quad \alpha=0 .
$$

2. Algebraic determination of the value of $B$ under the condition that $Q=\frac{1}{3} B$.

$$
\begin{equation*}
B^{2}+B\left\{\frac{4 E}{H W} \sin \delta+B^{\prime}\right\}=\frac{2 E}{W} \cos \delta+B^{\prime 2} . \tag{9}
\end{equation*}
$$

From Table I, $W=125 \mathrm{lbs}$. Then

$$
B^{2}+B\left\{\frac{4 \times 8500}{20 \times 125} 0.342+2\right\}=\frac{2 \times 8500}{125} 0.940+4,
$$

or

$$
B^{2}+6.65 B=131.84 .
$$

$$
B=-3.30 \pm \sqrt{131.84+\overline{3.36}^{2}},
$$

and

$$
B=-3.36+11.96=7.60 \mathrm{ft} .
$$

Ex. 4. Determine the value of $B$ in Ex. 3 under the assumption that $\epsilon=0$ (horizontal earth-surface).

$$
\begin{equation*}
E=\frac{H^{2} \gamma}{2}\left\{\tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right)=\frac{1-\sin \phi}{1+\sin \phi}\right\}, . \tag{6}
\end{equation*}
$$

or $E=20000(0.333)=6660$, say $6 \sim 00 \mathrm{lbs}$.
Since $\alpha=0$, and $\epsilon=0, \delta=0$,

$$
\begin{aligned}
& B^{2}+B B^{\prime}=\frac{2 E}{W}+B^{\prime 2} ; \\
& B^{2}+2 B=111.2 \\
& B=-1 \pm \sqrt{111.2+1}
\end{aligned}
$$

and

$$
B=-1+10.59=9.6 \mathrm{ft} .
$$

Ex. 5. Determine the value of $B$ in Ex. 3, under the assumption that $\epsilon=\phi=30^{\circ}$.

$$
E=\frac{H^{2} \gamma}{2} \cos \phi=20000(0.866)=17320 \mathrm{lbs} .
$$

From (9),

$$
B^{2}+B\left\{\frac{4 \times 17320}{20 \times 12 \overline{0}} 0.5+2\right\}=\frac{2 \times 17320}{125} 0.866+4
$$

$$
\begin{aligned}
& B^{2}+15.86 B=244.05 \\
& B=-7.93+\sqrt{244.05+7.93^{2}} .
\end{aligned}
$$

and $\quad B=-\% .93+1 \% .52=9.6 \mathrm{ft}$.
Ex. 6. Determine the resultant pressure against the back of a wall when the surface of the earth carries a load equivalent to 5 feet in depth of sand.

$$
H=30 \mathrm{ft} ., \alpha=10^{\circ}, \phi=30^{\circ}, \epsilon=0 \text {, and } \gamma=100
$$

lbs.


Fig. 6.
Graphical solution of the problem.-In Fig. 6, let BS represent the surface of the earth, and $B A$ the back of the wall.

Make $S T=5$, and draw $H T$ and $B H$. Draw $A R$ parallel to $B S$, parallel to $H T$, and make $L R$ equal to $L T$; lay off the angle $L R P$ equal to $30^{\circ}$; with $O$ as a centre
draw an arc passing through $L$ tangent to $P R$, and then with $O R$ as a radius describe the circumference of the circle $R Q M$, and at $M$ draw $M N$ parallel to $A H$; at $A$ and normal to $A H$ draw $A C$ equal to $N L$. Then

$$
\frac{A C+B V}{2} B A \cdot \gamma=E
$$

The direction of $E$ will be parallel to $Q M$.
To determine the point of application of $E$, find the centre of gravity $E^{\prime \prime}$ of $A B V C$, and draw $E^{\prime} D$ parallel to $A C$, then $D$ will be the point of application of $E$.
$E^{\prime \prime}$ can be found as follows: Produce $A C$ and $B V$, make $A I=C K=B V, B G=V F=A C$, and join $F$ and $I$ and $G$ and $K$. 'Then $E$ ', the intersection of $F l$ and $G K$, will be the centre of gravity of $A B V C, B D$ can be found from the formula

$$
B D \cos 10^{\circ}=\frac{1}{3} \frac{(T L)^{3}-3(T L)(T S)^{2}+2\left(T S^{\prime}\right)^{3}}{(T L)^{3}-\left(T S^{3}\right)^{3}}
$$

See (30) of Appendix.
Ex. \%. Determine graphically the value of $E$ when $\epsilon=0$ and $\alpha=0, \phi, \gamma$, and $H$ being given.

In Fig. $z$ let $B F$ represent the surface of the earth, and $A B$ the back of the wall. Draw $A L$ parallel to $B F$ and make $I L=I F$; lay off the angle $G L H=\phi$, and at any point $K$ in $L H$ draw $M K$ perpendicular to $H L$, and lay off $M O=M K$; draw $M J$ parallel to $O I$. Then will the are $I N$, described with $J$ as a centre and $I J$ as a radius, pass through $I$ and be tangent to $G L$; with $J$ as a centre and $J L$ as radius describe the circumference $L H$; at $A$ lay off $A C=H I$ and normal to $A B$. Then

$$
\frac{A C \times A B}{2} \gamma=E .
$$

$E$ is parallel to $B F$ and applied at $D, A D$ being equal to ${ }_{3}^{\frac{1}{3}} A B$.


Fig. \%.
Ex. 8. Determine the earth-thrust on the profile shown in Fig. 8, $H, \gamma, \phi$, and $\epsilon$ being given.

Graphical solution of the problem.-Let $B C D E A$ represent the given profile, and let the surface of the earth be horizontal. Prolong $B C$ until it intersects $S A$ in $S$; draw $S R$ normal to $B C S$ and equal to the intensity of the earth-pressure at $S$; connect $B$ and $R$. Then from the middle point of $B C$ draw $G F$ parallel to $S R$; the distance ${ }^{*}$ $G F$ multiplied by $\gamma$ will be the average intensity of the earth-pressure on $B C$. In a similar manner the average intensities on $C D, D E$, and $E A$ can be found, and hence the total pressures on each determined. The points of application of these resultant pressures, $E_{1}, E_{2}, E_{3}$, and $E_{4}$,
can be found by the method used in Ex. 6 for finding the centre of gravity of a trapezoid. The directions of


Fig. 8.
$E_{1}, E_{2}, E_{3}$, and $E_{4}$ are found from the construction on the right.

Ex. 9. Determine the thrust of the earth against a vertical wall when $\epsilon$ is negative.

For the explanation of this construction, see Part II, page 47 , Fig. $8 a$.

Ex. 10. From the following data determine $E, \delta$, and $Q$ :
$\epsilon=0, \phi=38^{\circ}, \alpha=10^{\circ} 23^{\prime} ; \gamma=90 \mathrm{lbs} ., W=1 \% \mathrm{lbs} . ;$

$$
H=15 \mathrm{ft} ., \quad B=6 \mathrm{ft} ., \quad B^{\prime}=2 \mathrm{ft} .
$$

Ans. $E=3037^{\prime} \mathrm{lbs} ., \delta=27^{\circ} 13^{\prime}, Q=2.2 \mathrm{ft}$.
Ex. 11. Determine the dimensions of a trapezoidal wall built of dry, rough granite, having a vertical back and being 20 feet high, to safely retain the side of a sand cut,
the surface of the sand being level with the top of the wall. $W=165$ lbs., $\gamma=100$ lbs., $\phi=33^{\circ} 40^{\prime}, H=20$ ft., $B^{\prime}=2 \mathrm{ft}$.

Ans. $E=5 \% 34 \mathrm{lbs} ., \delta=0, B=8 \mathrm{ft}$., and $Q=2.8 \mathrm{ft}$., about.
Ex. 12. The same as Ex. 11, with $\alpha=8^{\circ}$ instead of $\alpha=0$. Ans. $E=6330 \mathrm{lbs} ., B=8 \mathrm{ft}$, and $Q=2.7 \mathrm{ft}$.


Fig. $8 a$.
Ex. 13. What must be the dimensions of a rubble wall of large blocks of limestone, laid dry, to retain a sand filling which supports two lines of standard-gauge railroad track? (Assume the depth of sand to produce a pressure on the earth equal to that produced by the railroad and trains as 4 feet.)
$H=15 \mathrm{ft} ., \alpha=8^{\circ}, \phi=33^{\circ} 40^{\prime}, \gamma=100 \mathrm{lbs} ., W=170$ lbs., $B^{\prime}=3.5 \mathrm{ft}$.

Ans. $E=5 \gamma 60 \mathrm{lbs} ., \delta=18^{\circ} \mathrm{r}^{\prime}, B=8 \mathrm{ft} ., Q=2 . \% \mathrm{ft}$.
Ex. 14. Determine $E, \delta, B$, and $Q$, when $W=170$ lbs., $\gamma=100 \mathrm{lbs} ., \alpha=8^{\circ}, \epsilon=\phi=33^{\circ} 40^{\prime}, H=20 \mathrm{ft}$., $B^{\prime}=2 \mathrm{ft}$.
$A » s . E=21760 \mathrm{lbs} ., \delta=32^{\circ} 25^{\prime}, B=9 \mathrm{ft} ., Q=3 \mathrm{ft}$.

* Ex. 15. A wall 9 ft . high faces the steepest declivity of earth at a slope of $20^{\circ}$ to the horizon; weight of earth 130 lbs. per cubic foot, angle of repose $30^{\circ}$. Determine $E$ when $\alpha=0$.

$$
\text { Ans. } E=2187 \mathrm{lbs}
$$

* Ex. 16. $\epsilon=33^{\circ} 42^{\prime}, \phi=36^{\circ}, H=3$ ft., $\gamma=120$ lbs., $\alpha=0$. Determine $E$.

Ans. $E=278$ lbs.

* Ex. 1\%. $\phi=25^{\circ}, \epsilon=0, \alpha=0, H=4 \mathrm{ft} ., \gamma=120$ lbs., $E=$ ?

$$
\text { Ans. } E=390 \text { lbs. }
$$

* Ex. 18. $\phi=38^{\circ}, \epsilon=0, \alpha=0, H=3 \mathrm{ft} ., \gamma=94$ lbs., $E=$ ?

$$
\text { Ans. } E=100.5 \mathrm{lbs} .
$$

* Ex. 19. A ditch 6 feet deep is cut with vertical faces in clay. These are shored up with boards, a strut being put across from board to board 2 feet from bottom, at intervals of 5 feet apart. The coefficient of friction of the moist clay is $0.28 \%$, and its weight 120 lbs. per cubic foot. Find the thrust on a strut, also find the greatest thrust which might be put upon the struts before the adjoining earth would heave up.

$$
\text { Ans. } E=1230 \mathrm{lbs} .
$$

Thrust per strut $=6128 \mathrm{lbs}$. Greatest thrust $=19029 \mathrm{lbs}$.

* Ex. 20. A wall 10 ft . high, 2 ft . thick, and weighing 144 lbs. per cubic ft., is founded in earth weighing 112 lbs. per cubic ft., and whose angle of repose is $32^{\circ}$. Find the least depth of the foundation.

$$
\text { Ans. } x^{\prime}=1.21 \mathrm{ft} . \quad 10-1.21=8.79 \mathrm{ft} .=\text { amount of }
$$ wall above the ground.

* Ex. 21. An iron column is to bear a weight of 20 tons ( 2240 lbs. $=$ one ton); the foundation is a stone 3 ft . square on bed, sunk in earth weighing 120 lbs . per cu. ft.; angle of repose $27^{\circ}$. Find the least depth to which it must be sunk for equilibrium.

$$
\text { Ans. } x^{\prime}=6 \mathrm{ft} .
$$

* Ex. 22. A brick wall, allowing for openings, weighs 42000 lbs. per rood of 36 sq . ft. (on an average one brick and a half), and stands 45 ft . above the ground; the foundation is to widen to four bricks at the bottom. Find depth of foundation in clay weighing 130 lbs. per cu. ft. (angle of repose $27^{\circ}$ ), neglecting weight of unknown foundation.

$$
\text { Ans. } x^{\prime}=1.7 \mathrm{ft}
$$

* Alexander's Applied Mechanics.


## PART II.

## THE THEORY OF EARTH-PRESSURE AND THE STABILI'TY OF RETAINING-WALLS.

Preliminary Principles.-Before demonstrating the general formula for the thrust of earth against a wall, it will be necessary to establish the relations between the stresses in an unconfined and homogeneous granular mass.

* In Fig. 1 let $A B C^{\prime}$ be any small prism within a granu-


Fig. 1.
lar mass which is in equilibrium un er the action of the three stresses $P, Q$, and $R$, having the intensities $p, q$, and $r$ respectively.

[^1]Let $\theta$ represent the angle of inclination of the plane $C B$ with $A B$, and the angle at $A$ be a right angle.

The planes $A B$ and $A C$ are called planes of principal stress, and $P$ and $Q$ are called principal stresses.

Case I. If the principal stresses are of the same kind and their intensitues the same, then will the resultant stress on any third plane be normal to that plane and its intensity be equal to that of either principal stress.
In Fig. 1, for convenience, let $A B=1$, then $A C=\tan \theta$, and $C B=\frac{1}{\cos \theta}$. Hence
$P=p, Q=q \tan \theta=p \tan \theta$, since $p=q$, and $R=\frac{r}{\cos \theta}$.
Since $P, Q$, and $R$ are in equilibrium, they will form a closed triangle, as shown on the right in Fig. 1. Hence

$$
R^{2}=P^{2}+Q^{2},
$$

or

$$
\frac{r^{2}}{\cos ^{2} \theta}=p^{2}+p^{2} \tan ^{2} \theta=p^{2}\left(1+\tan ^{2} \theta\right) ;
$$

$$
\therefore r=p=q .
$$

Also, $R \cos F D E=P$,
or

$$
\frac{r}{\cos \theta} \cos F D E=p ; \quad \text { but } r=p
$$

Hence

$$
\cos \theta=\cos F D E=\cos H D G ;
$$

$\therefore H D G=\theta$ and $\quad R$ is normal to $C B$.

Case II. If the principal stresses are not of the same kind but their intensities the same, then will the resultant make the angle $\theta$ with the direction of the principal stress, but on the opposite side from that on which the resultant in Case I lies, and its intensity be equal to that of either principal stress.

The demonstration of Case I proves this principle if Fig. 1 is replaced by Fig. 2.


Fig. 2.
Case III. Given the principal stresses of the same kind but having unequal intensitues, to determine the intensity and direction of the resultant stress on any third plane.

Let $P$ and $Q$ be compressive and the intensity $p>$ the intensity $q$.

The following identities can be written:

$$
p=\frac{1}{2}(p+q)+\frac{1}{2}(p-q),
$$

and

$$
q=\frac{1}{2}(p+q)-\frac{1}{2}(p-q) ;
$$

or the resultant intensity on the plane $C^{C} B$ may be considered as being the resultant of two intensities, one being the intensity of the resultant stress caused by two like principal stresses having the same intensity $\frac{1}{2}(p+q)$, and the other the intensity of the resultant stress caused by two unlike principal stresses having the same intensity $\frac{1}{2}(p-q)$.


Fig. 3.
The intensity of the resultant stress caused by the first two principal stresses will be, by Case $\mathrm{I}, \frac{1}{2}(p+q)$, and the direction of the resultant will be normal to the plane $C B$. By Case II the resultant of the second pair of principal stresses will make the angle $\theta$ with the direction of $P$, and its intensity will be $\frac{1}{2}(p-q)$; then the resultant intensity can be found as follows:

In Fig. 3 draw $M D$ normal to $B C$, and make $L D=$ $\frac{1}{2}(p+q)$; with $L$ as a centre and $L D$ as radius, describe an arc cutting $F D$ at $F$. Then the angle $L F D=L D F=\theta$. Lay off $L G=\frac{1}{2}(p-q)$, and draw $G D$, which is the result-
ant intensity, and the intensity of the resultant stress on $C D$ caused by the two principal stresses $P$ and $Q$. $G D$ also represents the direction of the resultant stress $R$.

Since the intensities of the principal stresses remain constant, $\frac{1}{2}(p+q)$ and $\frac{1}{2}(p-q)$ will remain the same for any inclination of the plane $C B$; hence the intensity $r$ of the resultant depends upon the angle $\theta$ when $p$ and $q$ are given.

From Fig. 3,

$$
\begin{gathered}
G L \cos 2 \theta=L M \text { and } G L \sin 2 \theta=G M \\
D M=D L+L M=\frac{1}{2}(p+q)+\frac{1}{2}(p-q) \cos 2 \theta \\
\overline{G D}^{2}=r^{2}=\overline{G M}^{2}+\overline{D M}^{2}
\end{gathered}
$$

or

$$
\begin{equation*}
r=\sqrt{p^{2} \cos ^{2} \theta+q^{2} \sin ^{2} \theta} \tag{a}
\end{equation*}
$$

which is the general expression for the intensity of the resultant stress of a pair of principal stresses.

As the angle $\theta$ changes, the angle $\beta$ will also change, and it will have its maximum value when the angle $L G D=90^{\circ}$. This is easily proven as follows:

With $L$ as centre and $G L$ as radius describe an arc; then $\beta$ will have its maximum value when the line $D G$ is tangent to the are; but when $D G$ is tangent to the arc the angle $L G D$ is a right angle, since $L G$ is the radius of the are.

$$
\begin{equation*}
\sin \max \beta=\frac{p-q}{p+q} \tag{b}
\end{equation*}
$$

from which the following can be easily obtained:

$$
\begin{equation*}
\frac{p}{q}=\frac{1-\sin \max \beta}{1+\sin \max \beta} \tag{c}
\end{equation*}
$$

which expresses the limiting ratio of the intensities of the principal stresses consistent with equilibrium, $p$ being greater than $q$.

Case IV. Given the intensity and direction of the resultant stress on any plane, and the value of $\max \beta$, to determine the intensities and directions of the principal stresses.


Fig. 4.
Let $A D$ represent the given plane and $G D$ the direction and intensity of the resultant stress at the point $D$.

Draw $D L$ normal to $A D$, and draw $D I$, making the angle $\max \beta$ with $L D$. At any point $J$ in $D L$ describe an are tangent to $D I$, cutting $G D$ in $K$ and draw $G L$ parallel to $K J$; with $L$ as a centree and $L G$ as radıus describe
a circumference. This circumference will pass through $G$ and be tangent to $D I$; hence $\frac{G L}{D L}=\sin \max \beta$.

Since $\sin \max \beta=\frac{p-q}{p+q}$, and $G L$ and $L D$ are components of $r$,

$$
G L=\frac{1}{2}(p-q) \quad \text { and } \quad D L=\frac{1}{2}(p+q) ;
$$

then $N D=N L+L D=\frac{1}{2}(p-q)+\frac{1}{2}(p+q)=p$,
and $M D=L D-L M=\frac{1}{2}(p+q)-\frac{1}{2}(p-q)=q$,
which completely determines the intensities of the principal stresses.

According to Case III, the direction of the greater principal stress bisects the angle between the prolongation of $L M$ and the line $G L$; hence $R L$ represents the direction of the greater principal stress, and that of the other is at right angles to $R L$.
The above intensities and directions being determined, the intensity of the resultant stress on any other plane passing through $D$ is easily determined as follows:

Let $D Y$ represent any plane passing through $D$, draw $D L^{\prime}$ normal to $M Y$ and equal to $\frac{1}{2}(p+q)$. Draw $R^{\prime} D$ parallel to $R L$, and with $L^{\prime}$ as a centre and $L^{\prime} D$ as radius describe an arc cutting $R^{\prime} D$ at $O$, and make $L^{\prime} G^{\prime}=\frac{1}{2}(p-q)$; then $G^{\prime} D=r^{\prime}=$ the intensity of the resultant stress on $D Y$.

It is clear that if the value of max $\beta$ can be obtained for a mass of earth that the construction of Fig. 3 can be employed in determining the intensity of the earth-pressure at any point in diny:plane within the mass.

It has been established by experiment that if a body be placed upon a plane, that (as the plane is made to incline to the horizontal) at some angle of inclination the body will commence to slide down the plane, and that this angle depends largely upon the character of the surfaces in contact.


Fig. 5.
In Fig. 5 let $A B$ represent a plane inclined at the angle $\phi$ with the horizontal, and $C$ any mass just on the point of sliding down the plane. Let $E C$ represent the weight of the mass $C$, and $E D$ and $D C$ the components respectively parallel and normal to the plane $A B$. Then $D E$ is the force required to just keep the mass $C$ from sliding down the plane, assuming the plane to be perfectly smooth, or if the plane is rough this force represents the effect of friction.

$$
\frac{D E}{D C}=\tan \phi,
$$

or when the mass $C$ is about to slide, the resultant pressure $E C$ on $A B$ makes the angle $\phi$ with the normal to the
plane, the angle $\phi$ being the inclination of the plane $A B$, and is called the angle of friction.

In the case of earth, considered as a dry granular mass, the inclination of the steepest plane upon which earth will not slide is called the angle of repose, and the plane the surface of repose.

From the above, then, it follows that in a mass of earth the resultant pressure on any plane cannot make an angle with the normal to that plane which is greater than the angle of repose $\phi$; therefore the construction of Case IV applies to earth when $\max \beta$ is replaced by $\phi$. The values of $\phi$ for earth under various conditions are given in Table 1 I.

The preceding principles will now be applied in determining the thrust of earth against a retaining-wall.

## EARTH-PRESSURE.

In order that the formulas may not become too complex for practical use, it will be assumed that the earth is a homogeneous granular mass without cohesion. The surface of the earth will be considered to be a plane, and the length of the mass measured normally to the page as unity.

* Given the intensity and direction of the resultant stress at any point in any plane parallel to the surface of the earth, the inclination of the surface of the earth with the horizontal, and the angle of repose, io determine the intensity and direction of the resultant stress on a vertical plane passing through the same point.

[^2]

In Fig. 6 let $B Q$ represent the surface of the earth, and $D$ any point in the plane $A D$ parallel to $B Q$; draw $D Q$ normal to $A D$, and make the vertical $G D$ equal to $Q D$; then $G D \cdot \gamma$ is the intensity of the resultant pressure at $D$. Draw $D M$, making the angle $\phi$ with $L D$, and with $L$ as centre describe an arc tangent to $D M$ and passing through $G$; then by Case IV $L G \cdot \gamma=\frac{1}{2}(p-q), L D \cdot \gamma=\frac{1}{2}(p+q)$,


Fig. 6.
and $R L$ bisecting the angle $Q L G$ is the direction of the greater principal stress. To determine the intensity and direction of the resultant stress at $D$ on a vertical plane, proceed according to Case IV. Draw $R^{\prime} D$ parallel to $R L$ and $D L^{\prime}=D L$ normal to $D G$. With $L^{\prime}$ as a centre and $L^{\prime} D$ as radius describe an arc cutting $R^{\prime} D$ at $R^{\prime \prime}$, and make
$L^{\prime} G^{\prime}=L G$; then $D G^{\prime}$ represents the direction of the resultant stress, and $D G^{\prime} \cdot \gamma$ the intensity of the resultant.

In Fig. 6 the angle $R^{\prime} D L^{\prime}=D R^{\prime \prime} L^{\prime}=90^{\circ}-\omega+\theta^{\prime}$.
$\therefore G^{\prime} L^{\prime} D=2 \omega-2 \mathcal{G}^{\prime}$. But $2 \theta^{\prime}=\omega+\epsilon$; hence $G^{\prime} L^{\prime} D$ $=\omega-\epsilon$.

Draw $L Y=L G$; then the angle $D L Y=\omega-\epsilon . \therefore$ Since $L D=D L^{\prime}$ and $L Y=L G=L^{\prime} G^{\prime}$, the triangle $G^{\prime} L^{\prime} D$ equals the triangle $L Y D$ and the angle $G^{\prime} D L^{\prime}=\epsilon$; or the direction of the resultant earth-pressure against a vertical plane is parallel to the surface of the earth.

From Fig. 6,

$$
\begin{aligned}
& \frac{1}{2}(p-q) \cos \omega=G X \cdot \gamma, \\
& \frac{1}{2}(p-q) \sin \omega=L X \cdot \gamma, \\
& \frac{1}{2}(p+q) \cos \epsilon=D X \cdot \gamma .
\end{aligned}
$$

Now

$$
D Y=D G^{\prime}=D G-2 G X,
$$

or

$$
\begin{aligned}
D G^{\prime} \cdot \gamma & =D G \cdot \gamma-(p-q) \cos \omega \\
& =\frac{1}{2}(p+q) \cos \epsilon-\frac{1}{2}(p-q) \cos \omega, \\
\frac{1}{2}(p+q) & : \sin \omega \quad: \quad \frac{1}{2}(p-q): \sin \epsilon,
\end{aligned}
$$

and

$$
\sin \omega=\frac{p+q}{p-q} \sin \epsilon,
$$

or
$\overline{\cos \omega}=\sqrt{1-\left(\frac{p+q}{p-q}\right)^{2} \sin ^{2} \epsilon}=\sqrt{\frac{\left(p-q^{2}\right)-(p+q)^{2} \sin ^{2} \epsilon}{(p-q)^{2}}}$,
and since

$$
\frac{1}{2}(p+q) \sin \phi=\frac{1}{2}(p-q),
$$

$$
\cos \omega=\frac{1}{\sin \phi} \sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}
$$

Substituting this value for $\cos \omega$ in the equation for $D G^{\prime} \cdot \gamma$, it becomes
$D G^{\prime} \cdot \gamma=\frac{1}{2}(p+q) \cos \epsilon-\frac{1}{2}(p-q) \frac{1}{\sin \phi} \sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}$,
or since

$$
\frac{1}{\sin \phi}=\frac{p+q}{p-q}
$$

$$
D G^{\prime} \cdot \gamma=\frac{1}{2}(p+q)\left\{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}\right\} .
$$

In a similar manner,

$$
D G \cdot \gamma=\frac{1}{2}(p+q)\left\{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}\right\}
$$

and

$$
\frac{D G^{\prime}}{D G}=\frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}
$$

hence

$$
D G^{\prime}=D G \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}
$$

Let $x=$ the vertical distance between the two planes $B Q$ and $A D$, then

$$
D G=D Q=x \cos \epsilon
$$

$\therefore D G^{\prime} \cdot \gamma=(x) \gamma \cos \epsilon \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}$,
which is the expression for the intensity of the resultant earth-pressure on a vertical plane at any depth $x$ below the surface.

Let

$$
\begin{equation*}
* A=\cos \epsilon \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}} \tag{d}
\end{equation*}
$$

[^3]The average intensity of the resultant earth-pressure on a vertical plane of the length $x$ will be

$$
\left(\frac{x}{2}\right) \gamma A
$$

and hence the total pressure will be

$$
P=\frac{x^{2}}{2} \gamma A . . \circ \cdot \cdot \cdot \cdot .(e)
$$

Since the intensities of the pressures are uniformly varying from the surface, and increasing as $x$ increases, the application of the resultant thrust will be at a depth of $\frac{2}{3} x$ below the surface.

Considering the earth as an unconfined mass, the above formula is perfectly general and can be applied under all conditions, including the case when $\epsilon$ is negative.

The resultant stress on any plane as $A B$, Fig. 6, can be found by applying the principles of Case IV. Draw $P A$. parallel to $R L$, make $A N=L D$ and $N O=L G$; then $A O$ represents the direction of the resultant pressure on $A B$. Make $A C=A O$; then the area of the triangle $A B C$ multiplied by $\gamma$ is the total pressure on the plane $A B$, and this pressure is applied at $\frac{2}{3} A B$ below $B$.

In unconfined earth this construction is perfectly general and applies to any plane. It also applies equally well to curved profiles. An example illustrating the application of the method will be given in the applications. See pages 22 and 23 .

The following graphical construction, Fig. 7, is more convenient than that of Fig. 6.

As before, let $B E$ represent the surface of the earth, and
$A D$ a plane parallel to the surface. At any point $D$ in this plane, draw $D E$ vertical and make $D F=D E$; draw $F G$ horizontal and make the angle $H F D=\phi$.
With $L$ as a centre, describe an arc passing through $G$ and tangent to $M F$; then with $L$ as a centre and $L F$ as


Fig. 7.
radius, describe the circumference $F O N$, cutting $A D$ at $N$; through $N$ draw $N O$ parallel to $A B$, then draw $A C$ normal to $A B$ and equal to $O G$. The area of the triangle $A B C$ multiplied by $\gamma$ will be the total earth-pressure on $A B$. To determine the direction of the thrust prolong $O G$ to $Q$, then $Q N$ is the direction of the thrust.

That this construction is equivalent to that of Fig. 6 is
proved as follows. The triangle $G L F$ of Fig. 7 equals the triangle $G L D$ of Fig. 6.
$\therefore G L \cdot \gamma=\frac{1}{2}(p-q)$ and $L F \cdot \gamma=L O \cdot \gamma=\frac{1}{2}(p+q)$.
In Fig. 6, the angle $N A P=N P A=90^{\circ}-\frac{1}{2}(\omega-\epsilon)-\alpha$.

$$
\therefore O N A=\omega-\epsilon+2 \alpha
$$

In Fig. '7, the angle $O L N=2 \epsilon-2 \alpha$. But $G L N=\omega+\epsilon$.

$$
\therefore G L O=\omega-\epsilon+2 \alpha,
$$

and $G O$ of Fig. 7 equals $A O$ of Fig. 6.
In Fig. '7, the angle $Q N O=90^{\circ}-\beta^{\prime}$.
In Fig. 6, the angle $O A B=90^{\circ}-\beta^{\prime}$.
Therefore the direction of the thrust is the same in both constructions.
The two constructions given above are all that is required to determine the thrust of earth upon any plane within the mass of earth, as one can be used as a check upon the other; but as a formula is often very convenient, a general formula will now be deduced which will enable one to determine the values of $E$ and $\delta$ for any plane within a mass of earth.

General Formula for the Thrust of Earth.
In Fig. 8, let $B Q$ represent the surface of the earth and $A B$ any plane upon which the earth-pressure is desired.
Draw $A D$ parallel to $B Q$ and let the vertical distance $Q D=F A=x$,

From ( $e$ ) the earth-pressure upon $F A$ is parallel to the surface and equal to

$$
P=\frac{x^{2}}{2} \gamma A
$$



Fig. 8.
But $A F=x=H(1+\tan \alpha \tan \epsilon)=H \frac{\cos (\epsilon-\alpha)}{\cos \alpha \cos \epsilon} ;$

$$
\therefore P=\frac{H^{2} \gamma}{2} \frac{\cos ^{2}(\epsilon-\alpha)}{\cos ^{2} \alpha \cos ^{2} \epsilon} A . \quad . \quad . \quad .(f)
$$

Now the thrust $P$ combined with the weight of the prism $A B F$ must produce the resultant pressure upon $A B$.

Then from Fig. 8,
$V=\frac{H^{2} \gamma}{2} \tan \alpha(1+\tan \alpha \tan \epsilon)$

$$
=\frac{H^{2} \gamma}{2} \frac{\sin \alpha \cos (\epsilon-\alpha)}{\cos ^{2} \alpha \cos \epsilon}
$$

$E=\sqrt{(V+P \sin \epsilon)^{2}+(P \cos \epsilon)^{2}}=\sqrt{V^{2}+P^{2}+2 V P \sin \epsilon .}$
Substituting $(f)$ and $(g)$ in this it becomes

$$
\begin{aligned}
E= & \frac{H^{2} \gamma}{2} \frac{\cos (\epsilon-\alpha)}{\cos ^{2} \alpha \cos \epsilon} \times \\
& \sqrt{\sin ^{2} \alpha+2 \sin \alpha \sin \epsilon \cos (\epsilon-\alpha) \frac{A}{\cos \epsilon}+\cos ^{2}(\epsilon-\alpha) \frac{A^{2}}{\cos ^{2} \epsilon}},
\end{aligned}
$$

which becomes, by replacing $A$ by its value from ( $d$ ),
$E=\frac{H^{2} \gamma}{2} \frac{\cos (\epsilon-\alpha)}{\cos ^{2} \alpha \cos \epsilon} \times$

$$
\begin{gather*}
+\begin{array}{c}
+\sin ^{2} \alpha \\
+2 \sin \alpha \sin \epsilon \cos (\epsilon-\alpha) \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}} \\
+\cos ^{2}(\epsilon-\alpha)\left\{\frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}\right\}^{2}
\end{array} \tag{1}
\end{gather*}
$$

which is the general equation for the thrust of earth upon any plane within the mass.

To determine the direction of the thrust of the earth, let $\delta$ be the angle which the direction of the thrust makes with the horizontal; then, from Fig. 8,

$$
\tan \delta=\frac{V}{P \cos \epsilon}+\tan \epsilon
$$

Substituting the values of $V$ and $P$ given above, this becomes

$$
\begin{equation*}
\tan \delta=\frac{\sin \alpha \cos \epsilon+\sin \epsilon \cos (\epsilon-\alpha) A}{\cos \epsilon \cos (\epsilon-\alpha) A} \tag{1a}
\end{equation*}
$$

where

$$
\begin{equation*}
A=\cos \epsilon \frac{\cos \epsilon-\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}}{\cos \epsilon+\sqrt{\cos ^{2} \epsilon-\cos ^{2} \phi}} \tag{d}
\end{equation*}
$$

Equations (1) and (1a) are readily reduced to more simple forms for special cases. These forms will be found in Part I.

The Plane of Rupture.-Although it is not necessary to know the position of the plane of rupture in order to determine the thrust of the earth, yet it may be of interest to know its position, which can be easily determined as follows:

The plane of rupture will be back of the wall and pass through the heel of the wall. The resultant earth-pressure will make the angle $\phi$ with the normal to this plane. Now the tangent of the angle which the direction of the resultant earth-pressure on any plane makes with the horizontal is determined from the formula

$$
\tan \delta=\frac{\sin \alpha}{\cos (\epsilon-\alpha) A}+\tan \epsilon
$$

If $\omega$ represents the angle which the plane of rupture makes with the vertical passing through the heel of the wall, $\alpha=\omega$ and $\delta=\phi+\omega$.

$$
\tan (\phi+\omega)=\frac{\sin \omega}{\cos (\epsilon-\omega) A}+\tan \epsilon
$$

from which the value of $\omega$ can be determined for any case.

For the case where $\epsilon=\phi, \epsilon$ being positive with respect to the wall and negative with respect to the plane of rupture, the above equation becomes

$$
\tan (\phi+\omega)=\frac{\sin \omega}{\cos (\phi+\omega) \cos \phi}-\tan \phi,
$$

which is satisfied when $\omega=90^{\circ}-\phi$.
For the case where $\epsilon=0$,

$$
\tan (\phi+\omega)=\frac{\sin \omega}{\cos \omega \tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right)},
$$

which is satisfied when $\omega=45^{\circ}-\frac{\phi}{2}$.
Reliability of the Preceding Theory.-The preceding theory is based upon the assumptions that the earth is a homogeneous mass and without cohesion, and the formulas are deduced under the assumption that the surface of the earth is a plane.
All writers on the subject have considered the earth as a homogeneous mass and, with a few exceptions, without cohesion.
Old and recent experiments indicate that cohesion has very little effect upon the pressure of the earth, which explains why it has not been considered by most writers.
The assumption of a plane earth-surface is necessary whenever practical formulas and direct graphical constructions for obtaining the thrust of the earth are obtained. General formulas can be deduced for any character of surface, but they are too complex for practical use. Those graphical constructions which do not require a plane earth-
surface are not direct in their solution of the problem, but require a series of trials to obtain the maximum thrust.

If the earth-surface is not a plane, one can be assumed which will give the thrust of the earth sufficiently exact for all practical purposes.

For unconfined earth no exceptions can be taken to the preceding theory, the assumptions upon which it is based being accepted, and for confined earth the theory must be true when the direction of the principal stress passing through the heel of the wall lies entirely within the carth.

For all cases in which $\alpha$ and $\epsilon$ are positive the theories of Rankine, Winkler, Weyrauch, and Mohr agree and give identical results with the preceding theory, as they should, being founded upon the same assumptions.

When $\alpha$ is negative Weyrauch does not consider his theory reliable, and his equations lead to indeterminate results.

Winkler and Mohr consider their theories reliable whenever the direction of the principal stress passing through the heel of the wall lies entirely within the earth.

Rankine's method of considering the case where $\alpha$ is negative is equivalent to assuming that the introduction of a wall does not affect the stresses within the mass.

It may be concluded that the preceding theory is perfectly exact when $\alpha$ and $\epsilon$ are positive; and when $\alpha$ or $\epsilon$ is negative that the stresses obtained will be the maximum which under any circumstances can exist.

For the case where $\epsilon$ is negative the stress obtained (which represents the maximum thrust the wall can have against the earth and have equilibrium) will be considerably larger than the actual stress (when a wall is introduced), depending upon the magnitude of $\epsilon$. For small values of $\epsilon$ the results will be practically correct. For large values of $\epsilon$
the following method can be employed in determining the thrust of the earth. The method depends upon the assumption that the pressure of the earth is normal to the back of the wall. This may or may not be the case, but it appears to be the most consistent assumption to make for this rare and not important case.


Fig. 8a.

* In Fig. $8 a$, let $A B$ be the back of the wall and $B f$ the surface of the earth. Make $B a=a b=b c=c d=$ etc. Some prism $B A a$ or $B A b$ or $B A c$, etc., will produce the maximum thrust on the wall; and when this maximum thrust is produced, the resultant pressure on the plane $A a$

[^4]or $A b$ or $A c$, etc., will make the angle $\phi$ with the normal to the plane.

On the vertical line $A d^{\prime}$ lay off $A a^{\prime}=a^{\prime} b^{\prime}=b^{\prime} c^{\prime}$, etc., and draw $A a^{\prime \prime}$ making the angle $\phi$ with the normal to $A a, A b^{\prime \prime}$ making the angle $\phi$ with the normal to $A b$, etc.; then draw $a^{\prime} a^{\prime \prime}, b^{\prime} b^{\prime \prime}$, etc., perpendicular to $A B$, and draw a curve through $A a^{\prime \prime}, b^{\prime \prime}, c^{\prime \prime}$, etc. Then there will be a maximum distance parallel to $a^{\prime} a^{\prime \prime}$ between $A l^{\prime}$ and this curve which will be proportional to the thrust of the earth against $A B$. This maximum distance multiplied by the altitude $A c \div 2$ and the product by $\gamma$, the weight of a cubic foot of earth, will be the pressure of the earth.

This method is perfectly general and can be applied in any case.

If the earth-pressure is assumed to have the direction given by the formulas of the preceding theory, the construction will give the same value of $E$, the pressure of the earth.

Some writers assume that the direction of $E$ makes the angle $\phi^{\prime \prime}=\phi$ with the normal to the back of the wall in all cases. This assumption cannot be correct until the wall commences to tip forward, and then it is doubtful that such is the case unless the earth and wall are perfectly dry.

To be on the side of safety in every case, it is better to take the direction of $E$ as given by the above theory.

The construction of Fig. $8 a$ will give the maximum thrust for any assumed direction for any case.

## Trapezoidal Walls.

It will be assumed that the direction and magnitude of the earth-pressure is known, that the position and extent of the back of the wall and the width of the top are given,
to determine the width of the base for stability against overturning, sliding, and crushing of the material.


Fig. 9.
Stability against Overturning.-Let $A B C D$, Fig. 9, represent a section of a trapezoidal wall, $T R$ the direction of the earth-thrust, $J G$ the vertical passing through the centre of gravity of the wall, and $J O$ the direction of the resultant pressure on the base $A D$ caused by $E$ and $G$.

As long as $R$ cuts the base $A D$, the wall will be stable against overturning. When $R$ takes the direction $J Q$, the wall may be said to be on the point of overturning; then the factor of safety against overturning is $\frac{Q N}{O N}$, where $O N$ is the actual value of $E$, and $Q N$ the value of $E$ required to make the resultant $R$ pass through $D$.

Stability against Sliding.-Since the wall will not slide
along the surface $D A$ until the resultant $R$ makes an angle with the normal to $D A$ greater than the angle of friction $\phi^{\prime}$, the factor of safety against sliding can be obtained as follows: Draw $J P$ making the angle $J M U=\phi^{\prime}$; then the factor of safety against sliding is $\frac{P N}{O N}$, where $P N$ is the force required in the direction of $E$ to make $R$ make the angle $\phi^{\prime}$ with the normal to $A D$, and $O N$ the actual value of $E$.

Stability against the Crushing of the Material.-In ordinary practice walls for retaining earth are not of sufficient height to canse very large pressures at their bases, but it is necessary to consider the subject on account of the tendency of the bed-joints to open under certain conditions.


Fig. 10.
Let $A B$, Fig. 10, represent any bed-joint in the wall, $P$ the vertical resultant pressure upon the joint, and $x_{0}$ the distance of the point of application from the centre of the joint.

The intensity of $P$ can be considered as composed of a uniform intensity $p_{0}=\frac{P}{B}$, and a uniformly varying intensity $p_{0}{ }^{\prime}$, so that $p_{x}=p_{0}+p_{0}{ }_{0}$. Let a equal the tangent of the angle $C D E$, then $p_{0}^{\prime}=a x$ and $p_{x}=p_{0}+a x$.

The pressure upon a surface $(d x)$-the joint being considered unity in the dimension normal to the page-is

$$
p_{x} d x=p_{0} d x+a x d x
$$

and the moment of this about $D B$ is

$$
\left(p_{0} d x+a x d x\right) x
$$

The algebraic sum of these moments for values of $x$ between the limits $\pm \frac{B}{2}$ must equal $P x_{0}$, or

$$
\left.P x_{0}=\int_{-\frac{1}{2} B}^{+\frac{1}{v} B} p_{0} x d x+a x^{2} d x\right)
$$

Integrating,

$$
a=\frac{12 x_{0} P}{B^{3}}=\frac{12 x_{0} p_{0}}{B^{2}}
$$

and

$$
p_{x}=\frac{B^{2}+12 x x_{0}}{B^{2}} p_{0}
$$

or

$$
p=\left\{1+\frac{6 x_{\mathrm{c}}}{B}\right\} \frac{P}{B}
$$

and if $x_{0}$ be replaced by $\frac{1}{2} B-Q$, where $Q$ is the distance from $A$ to the point where $P$ cuts the base, (Fig. 11,)

$$
p=2\left(B-\frac{3 Q}{B}\right) \frac{P}{B}
$$

and

$$
p^{\prime}=2\left(1-B+\frac{3 Q}{B}\right) \frac{P}{B}
$$

If $Q=\frac{1}{3} B$,

$$
p^{\prime}=0 \quad \text { and } \quad p=2 p_{0}
$$

from which it is seen that when $R$ cuts the base outside the middle third, the joint will have a tendency to open at points which are at a maximum distance from $R$ where it cuts the base.

Therefore in no case should the resultant pressure be permitted to cut the base outside the middle third. This makes it unnecessary to consider the stability against overturning.


Fig. 11.

Then in designing a wall the following conditions must exist for stability:
I. The resultant $R$ must cut the base for stability against overturning.
II. The resultant $R$ must not make an angle with the normal to the base of the wall greater than the angle of friction $\phi^{\prime}$ 。
III. The resultant $R$ must not cut the base outside of the middle third, in order that there may be no tendency for the bed-joints to open.

The above three conditions apply to any bed-joint of the wall; but if they are satisfied at the base and the wall has the section shown in Fig. 11, it will not be necessary to consider any joints above the base unless the character of the stone or the bonding is different.

Determination of the width of the base of a retainingwall under the condition that $R$ cuts the base at a point $\frac{1}{3} B$ from the toe of the wall.

Let $H, B^{\prime}, x, \delta$, and $E$ be given to determine $B$.
From Fig. 11,

$$
\begin{aligned}
& K F=\frac{x}{3} \sin \delta+\frac{H}{3} \cos \delta-\frac{2 B}{3} \sin \delta \\
& H D=\frac{2 B^{2}+2 B B^{\prime}-B x-2 B^{\prime} x-B^{\prime 2}}{3\left(B+B^{\prime}\right)} \\
& H F=H D-\frac{B}{3}=\frac{B^{2}+B B^{\prime}-B x-2 B^{\prime} x-B^{\prime 2}}{3\left(B+B^{\prime}\right)}
\end{aligned}
$$

For equilibrium

$$
E(K F)=G(H F)=\frac{B+B^{\prime}}{2} H W\left(H F^{\prime}\right)
$$

Substituting the values of $K F$ and $H F$ in the above and reducing, it becomes

$$
\begin{align*}
& B^{2}+B\left(\frac{4 E}{H W} \sin \delta+B^{\prime}-x\right) \\
& \quad=\frac{2 E}{H W}(H \cos \delta+x \sin \delta)+2 B^{\prime} x+B^{\prime 2} \tag{8}
\end{align*}
$$

which is the general equation for the width of the base of a trapezoidal wall.

For a rectangular wall $B^{\prime}=B$.
For a triangular wall $B^{\prime}=0$.
For a wall with a vertical front $B^{\prime}+x=B$ or $B^{\prime}=B-x$.

For a wall with a vertical back $x=0$.
Equation (8) is easily transformed to satisfy the requirements of special cases.

The width of the base can be found graphically by assuming a value for $B$ and finding the value of $Q$; if it is less than $\frac{1}{3} B$ another value of $B$ must be assumed, and so on until $Q$ is equal to or greater than $\frac{1}{3} B$.

Depth of Foundations.-Given the angle of repose $\phi$ of any earth, to determine the depth to which it is necessary to sink a foundation to support a given load. The surface of the earth is assumed to be horizontal.


Case I. When the intensity of the pressure on the base of the foundation is uniform.

In Fig. 12, let $p_{0}$ represent the intensity of the pressure on the base of the foundation.

Now when the masonry is about to sink (see Eq. (c)),

$$
\frac{p_{0}}{q}=\frac{1+\sin \phi}{1-\sin \phi} \quad \text { or } \quad q=p_{0} \frac{1-\sin \phi}{1+\sin \phi}
$$

If $x^{\prime}$ represents the depth to which the foundation extends below the surface of the earth and $\gamma$ the weight of a cubic foot of earth, then $\gamma x^{\prime}$ equals the vertical intensity of the earth-pressure on a plane at the depth of the lowest point of the foundation.

When the wall is on the point of sinking, the earth must be on the point of rising, or

$$
\frac{q}{\gamma x^{\prime}}=\frac{1+\sin \phi}{1-\sin \phi}
$$

or

$$
\begin{equation*}
p_{0}=\gamma x^{\prime}\left\{\frac{1+\sin \phi}{1-\sin \phi}\right\}^{2} \cdot \ldots . \tag{15}
\end{equation*}
$$

In any case $p_{0}$ must not have a greater value than that obtained from (15)-

$$
\begin{equation*}
x^{\prime}=\frac{p_{0}}{\gamma}\left\{\frac{1-\sin \phi}{1+\sin \phi}\right\}^{2}=\frac{p_{0}}{\gamma} \tan ^{4}\left(45^{\circ}-\frac{\phi}{2}\right) \tag{16}
\end{equation*}
$$

The value of $x^{\prime}$ as obtained from (16) is the least allowable value consistent with equilibrium. Since $x^{\prime}$ is a function of $\tan ^{4}\left(45^{\circ}-\frac{\phi}{2}\right)$, care must be taken that $\phi$ is assumed at its least value. As $\phi$ becomes smaller the value of $x^{\prime}$ increases rapidly.

Case II. When the intensity of the pressure on the base is uniformly varying.

Let $p$ represent the maximum intensity of the pressure on the earth and $p^{\prime}$ the minimum intensity; then for
equilibrium $p$ must not exceed the value obtained from the following equation:

$$
\begin{equation*}
p=x^{\prime} \gamma\left\{\frac{1+\sin \phi}{1-\sin \phi}\right\}^{2} \cdot \ldots . \tag{17}
\end{equation*}
$$

Also, $p^{\prime}$ must never be less than $x^{\prime} \gamma$; then
$p_{0}=\frac{p+p^{\prime}}{2}=\frac{x^{\prime} \gamma}{2}\left\{1+\left(\frac{1+\sin \phi}{1-\sin \phi}\right)^{2}\right\}=x^{\prime} \gamma \frac{1+\sin ^{2} \phi}{(1-\sin \phi)^{2}}$,
which expresses the maximum value which $p_{0}$ can have for the equilibrium of the earth. Solving (18) for $x^{\prime}$,

$$
\begin{equation*}
x^{\prime}=\frac{p_{0}}{\gamma} \frac{(1-\sin \phi)^{2}}{1+\sin ^{2} \phi} \tag{19}
\end{equation*}
$$

which is the minimum value $x^{\prime}$ can have for the equilibrium of the earth.

In order that $p^{\prime}$ may never be less than $x^{\prime} \gamma$ the resultant pressure on the base of the foundation must cut the base within a certain distance of the centre of the base. If $x_{0}$ equal this distance, then (see page 51 )

$$
p^{\prime}=\left(1-\frac{6 x_{0}}{B}\right) p_{0}=x^{\prime} \gamma
$$

Substituting the value of $p_{0}$ from (18) and solving for $x_{0}$,

$$
\begin{equation*}
x_{0}=\frac{1}{3} \frac{\sin \phi}{1+\sin ^{2} \phi}, \quad . \quad . \quad . \tag{20}
\end{equation*}
$$

which is the maximum value $x_{0}$ can have, consistent with the stability of the earth.

Abutting Power of Earth.-Let the surface of the earth be horizontal and the body pushing the earth have a verti-
cal face; then at the depth $x^{\prime}$ the maximum horizontal pressure per unit of area is (see Case I above)

$$
q=x^{\prime} \gamma \frac{1+\sin \phi}{1-\sin \frac{\phi}{\phi}},
$$

and since $q$ varies directly as $x^{\prime}$, the total thrust $P$ which the earth is capable of resisting is

$$
\begin{equation*}
P=\frac{\left(x^{\prime}\right)^{2} \gamma}{2} \frac{1+\sin \phi}{1-\sin \phi} \tag{21}
\end{equation*}
$$

## APPENDIX.

## WEYRAUCH'S <br> THEORY OF THE RETAINING-WALL.*

In the following the earth is supposed without cohesion, and its pressure is determined independently of any arbitrary assumptions as to direction of the earth-pressure, and with sole reference to the three necessary conditions of equilibrium. The single and only supposition, then, is as follows: That the forces upon any imaginary plane-section through the mass of earth have the same direction.

This assumption lies at the foundation of all theories of earth-pressure against retaining-walls. For those cases, therefore, to which the following discussion does not apply no complete or satisfactory theory is yet possible. In what follows, the ordinary assumption as to the direction of the earth-pressure will be proved to be incorrect, except for special cases.

[^5]
## I.

## GENERAL RELATIONS.

Let the surface of the earth have any form, and the wall $A B$, Fig. 1, have any inclination. The earth-pressure makes any angle, $\delta$, with the normal to the wall.

Suppose through the point $A$ the plane $A C$. Then the weight $G$ of the prism $A B C$ is held in equilibrium by the


Fig. 1.
reaction of the wall, $E$, and by the resultant, $R$, of all the forces acting upon $A C$.

Now decompose $E, G$, and $R$ into components parallel and normal to $A C$; then for every unit in length of the wall, denoting by $e, g$, and $r$ the lever-arms of $E, G$, and $R$ respectively with reference to $A$, the sum of the forces parallel to $A C=0$, or

$$
\begin{equation*}
P-P_{1}-P_{2}=0 ; \cdot \text { • • . . } \tag{1}
\end{equation*}
$$

the sum of the forces perpendicular to $A C=0$, or

$$
\begin{equation*}
Q+Q_{1}-Q_{2}=0 ; \cdot . . . . \tag{2}
\end{equation*}
$$

the sum of moments about $A=0$, or

$$
\begin{equation*}
G g+E e-R r=0 . \quad \text {. . . . } \tag{3}
\end{equation*}
$$

Equation (3) was first introduced by Prof. Weyrauch.
Further, according to the theory of friction, if $\varphi$ is the coefficient of friction for earth on earth,

$$
\begin{equation*}
\frac{P_{2}}{Q_{2}} \leq \tan \varphi \text { or } \frac{P-P_{1}}{Q+Q_{1}} \leq \tan \varphi . \tag{4}
\end{equation*}
$$

If now there is any plane for which

$$
\begin{equation*}
P-P_{1}=\left(Q+Q_{1}\right) \tan \varphi \tag{5}
\end{equation*}
$$

this plane $A C$ will be a plane of equilibrium, and $\frac{P-P_{1}}{Q+Q_{1}}$ will be a maximum, or

$$
\begin{equation*}
\frac{d\left(\frac{P-P_{1}}{Q+Q_{1}}\right)}{d \omega}=0 . . \tag{6}
\end{equation*}
$$

This plane is designated as the "surface of rupture."
From Fig. 1, for every position of $A C$,

$$
\begin{array}{ll}
P=G \cos \omega, & Q=G \sin \omega \\
P_{1}=E \sin (\omega+\alpha+\delta), & Q_{1}=E \cos (\omega+\alpha+\delta) .
\end{array}
$$

Substituting the above values of $P, P_{1}, Q$, and $Q_{1}$ in equation (5), it becomes

$$
\begin{aligned}
& G \cos \omega-E \sin (\omega+\alpha+\delta) \\
& \quad=[G \sin \omega+E \cos (\omega+\alpha+\delta)] \tan \varphi ;
\end{aligned}
$$

and when $\omega$ refers to the surface of rupture, the earthpressure upon $A B$ becomes

$$
E=\frac{\cos \omega-\sin \omega \tan \varphi}{\sin (\omega+\alpha+\delta)+\cos (\omega+\alpha+\delta) \tan \varphi} G
$$

Substituting the value of $\tan \varphi$ or $\frac{\sin \varphi}{\cos \varphi}$, this becomes

$$
E=\frac{\cos \varphi \cos \omega-\sin \omega \sin \varphi}{\sin (\omega+\alpha+\delta)} \frac{\cos \varphi+\cos (\omega+\alpha+\delta) \sin \varphi}{} G,
$$

which becomes

$$
\begin{equation*}
E=\frac{\cos (\varphi+\omega)}{\sin (\varphi+\omega+\alpha+\delta)} G \tag{7}
\end{equation*}
$$

In order to refer to the surface of rupture, the following relation must exist :

$$
\begin{equation*}
\frac{d\left(\frac{G \cos \omega-E \sin (\omega+\alpha+\delta)}{G \sin \omega+E \cos (\omega+\alpha+\delta)}\right)}{d \omega}=0 . \tag{7a}
\end{equation*}
$$

Performing the differentiation indicated in the equation ( ${ }^{*} a$ ), considering $G$ and $\omega$ as the variables, it becomes

$$
\begin{align*}
& +[d G \cos \omega-\sin \omega d \omega G-E \cos (\omega+a+\delta) d \omega][G \sin \omega+E \cos (\omega+a+\delta)] \\
& \frac{-[d G \sin \omega+\cos \omega d \omega G-E \sin (\omega+a+\delta) d \omega][G \cos \omega-E \sin (\omega+a+\delta)]}{\left[[\xi \sin \omega+E \cos (\omega+a+\delta)]^{2} d \omega\right.}= \\
& =0 ; \text {. . . . . . . . . . . . . . . . . . . . . (7b) }
\end{align*}
$$

dividing by $d \omega$, this becomes

$$
\begin{align*}
& +\left[\frac{d G \cos \omega}{d \omega}-[G \sin \omega+E \cos (\omega+\alpha+\delta)]\right][G \sin \omega+E \cos (\omega+a+\delta)] \\
& \frac{-\left[\frac{d G \sin \omega}{d \omega}+[G \cos \omega-E \sin (\omega+\alpha+\delta)]\right][G \cos \omega-E \sin (\omega+a+\delta)]}{[G \sin \omega+E \cos (\omega+a+\delta)]^{2}}= \\
& =0, \ldots . . . . . . . . . . . . . . . . . . . . . . . . . . \tag{7c}
\end{align*}
$$

or
$+\frac{d G \cos \omega}{d \omega}[G \sin \omega+E \cos (\omega+a+\delta)]-[G \sin \omega+E \cos (\omega+a+\delta)]^{2}$
$-\frac{d G \sin \omega}{d \omega}[G \cos \omega-E \sin (\omega+a+\delta)]-[G \cos \omega-E \sin (\omega+a+\delta)]^{2}$
$=0$. . . . . . . . . . . . . . . . . . . . . . . . . . . . (7d
Now, since
$\cos \omega \cos (\omega+\alpha+\delta)+\sin \omega \sin (\omega+\alpha+\delta)=\cos (\alpha+\delta)$ and $\sin ^{2} \omega+\cos ^{2} \omega=1$,
by clearing of fractions this becomes
$-\frac{E d G \cos (\alpha+\delta)}{d \omega}+G^{2}-2 G E \sin (\alpha+\delta)+E^{2}=0$.

Now since $d G=\frac{1}{2} k . d \omega . k \gamma$, equation ( $7 e$ ) reduces to

$$
G^{2}-2 G E \sin (\alpha+\delta)-\frac{E k^{2} \gamma \cos (\alpha+\delta)}{2}+E^{2}=0,(\gamma f)
$$

which becomes, after dividing by $G E$,

$$
\begin{equation*}
\frac{G}{E}-2 \sin (\alpha+\delta)-\frac{k^{2} \gamma \cos (\alpha+\delta)}{2 G}+\frac{E}{G}=0 \tag{8}
\end{equation*}
$$

Substituting the value of $\frac{E}{\bar{G}}$ from equation (7), transposing and multiplying by two, equation (8) reduces to

$$
\frac{2 \sin (\phi+a+\omega+\delta)}{\cos (\phi+\omega)}-4 \sin (\alpha+\delta)+\frac{2 \cos (\phi+\omega)}{\sin (\phi+\omega+a+\delta)}=\frac{k^{2} \gamma \cos (a+\delta)}{G},(8 a)
$$

whence

$$
\begin{equation*}
G=\frac{k^{2} \gamma \cos (\alpha+\delta)}{\frac{2 \sin (\phi+\omega+\alpha+\delta)}{\cos (\phi}-4 \sin (\alpha+\delta)+\frac{2 \cos (\phi+\omega)}{\sin (\phi+\omega+a+\delta)}}, \tag{8b}
\end{equation*}
$$

which reduces to

$$
\begin{equation*}
G=\frac{\cos (\phi+\omega) \sin (\phi+\omega+a+\delta) \cos (a+\delta) k^{2} \gamma}{2\left[\sin ^{2}(\phi+\omega+a+\delta)-2 \sin (a+\delta) \cos (\phi+\omega) \sin (\phi+\omega+a+\delta)+\cos ^{2}(\phi+\omega)\right]} \tag{8c}
\end{equation*}
$$

Since

$$
\begin{aligned}
\sin (\varphi+\omega+\alpha+\delta) & =\sin (\varphi+\omega) \cos (\alpha+\delta) \\
& +\cos (\varphi+\omega) \sin (\alpha+\delta)
\end{aligned}
$$

the parenthetical portion of the denominator becomes

$$
\begin{aligned}
\sin ^{2}(\varphi+ & +\cos ^{2}(\alpha+\delta) \\
& +2 \sin (\alpha+\delta) \cos (\varphi+\omega) \sin (\varphi+\omega) \cos (\alpha+\delta) \\
& +\cos ^{2}(\varphi+\omega) \sin ^{2}(\alpha+\delta) \\
& -2 \sin (\alpha+\delta) \cos (\phi+\omega) \sin (\varphi+\omega) \cos (\alpha+\delta) \\
& -2 \sin (\alpha+\delta) \cos (\phi+\omega) \cos (\varphi+\omega) \sin (\alpha+\delta) \\
& +\cos ^{2}(\phi+\omega)
\end{aligned}
$$

or

$$
\begin{aligned}
\sin ^{2}(\phi+\omega) \cos ^{2} & (\alpha+\delta) \\
& -2 \sin ^{2}(\alpha+\delta) \cos ^{2}(\varphi+\omega) \\
& +\sin ^{2}(\alpha+\delta) \cos ^{2}(\phi+\omega)+\cos ^{2}(\varphi+\omega)
\end{aligned}
$$

or

$$
\begin{aligned}
\sin ^{2}(\varphi+\omega) \cos ^{2}(\alpha+\delta) & -\cos ^{2}(\varphi+\omega) \sin ^{2}(\alpha+\delta) \\
& +\cos ^{2}(\varphi+\omega),
\end{aligned}
$$

or $\sin ^{2}(\varphi+\omega) \cos ^{2}(\alpha+\delta)+\cos ^{2}(\varphi+\omega)\left[1-\sin ^{2}(\alpha+\delta)\right]$,
or $\sin ^{2}(\varphi+\omega) \cos ^{2}(\alpha+\delta)+\cos ^{2}(\phi+\omega) \cos ^{2}(\alpha+\delta)$,
or

$$
\cos ^{2}(\alpha+\delta)\left[\sin ^{2}(\phi+\omega)+\cos ^{2}(\varphi+\omega)\right]
$$

which equals $\cos ^{2}(\alpha+\delta)$, and equation (8c) becomes, after dividing by $\cos (\alpha+\delta)$ and factoring,

$$
\begin{equation*}
G=\frac{\cos (\varphi+\omega) \sin (\varphi+\omega+\alpha+\delta)}{\cos (\alpha+\delta)} \cdot \frac{k^{2} \gamma}{2}=\text { Function } \gamma, \tag{9}
\end{equation*}
$$

from which

$$
\sin (\varphi+\omega+\alpha+\delta)=\frac{2 G}{k^{2} \gamma} \cdot \frac{\cos (\alpha+\delta)}{\cos (\phi+\omega)},
$$

which being substituted in equation (7) gives

$$
\begin{equation*}
E=\frac{G \cos (\phi+\omega)}{\frac{2 G \cos (\alpha+\delta)}{k^{2} \gamma \cos (\phi+\omega)}}=\frac{\cos ^{2}(\phi+\omega)}{\cos (\alpha+\delta)} \cdot \frac{k^{2} \gamma}{2} \tag{10}
\end{equation*}
$$



Fig. 2.
And, since the sum of the horizontal components of $E, G$, and $R$ must be equal to 0 , or Fig. 2,
and

$$
E \cos (\alpha+\delta)=R \cos (\omega+\varphi)
$$

$$
R=E \frac{\cos (\alpha+\delta)}{\cos (\omega+\varphi)}
$$

which becomes, after substituting the value of $E$ from equation (10),

$$
\begin{equation*}
R=\cos (\varphi+\omega) \frac{\hbar^{2} \gamma}{2} . \tag{11}
\end{equation*}
$$

Let $A D$, Fig. 2, be the natural slope of the ground. From $C$ let fall the perpendicular $C H$, and draw $C J$, making the angle $(\alpha+\delta)$ with CH ; then

$$
C H=k \cos (\varphi+\omega), \quad A J=\frac{\sin (\varphi+\omega+\alpha+\delta)}{\cos (\alpha+\delta)} k
$$



Fig. 2.
The expression for $A J$ is obtained in the following manner (Fig. 2):

$$
\begin{gathered}
C H=k \cos (\phi+\omega), \quad A H=k \sin (\phi+\omega) \\
H J:(' H:: \sin (\alpha+\delta): \cos (\alpha+\delta)
\end{gathered}
$$

and $H J=\frac{C H \sin (\alpha+\delta)}{\cos (\alpha+\delta)}=\frac{\cos \{\varphi+\omega) \sin (\alpha+\delta)}{\cos (\alpha+\delta)} k$,

$$
\begin{aligned}
& I I I+H J=A J \\
& \quad=\frac{\sin (\phi+\omega) \cos (\alpha+\delta)+\cos (\varphi+\omega) \sin (\alpha+\delta)}{\cos (\alpha+\delta)} k
\end{aligned}
$$

which reduces to

$$
A J=\frac{\sin (\varphi+\omega+\alpha+\delta)}{\cos (\alpha+\delta)} k ;
$$

and hence, according to equation (9),

$$
\begin{equation*}
G=\text { Func. } \gamma=\gamma \Delta A C J . \quad . \quad . \quad . \tag{12}
\end{equation*}
$$

Also, if $A K$ is perpendicular to $C J$,

$$
\frac{C H}{A K}=\frac{k \cos (\varphi+\omega)}{k \sin (\varphi+\omega+\alpha+\delta)}=\frac{E}{G}
$$

and if $J L$ is made equal to $J C$, then, since the perpendicular from $L$ upon $C J$ is equal to $C H$,

$$
\frac{\Delta C J L}{\Delta C J A}=\frac{C H}{A K}=\frac{E}{G},
$$

or

$$
\begin{equation*}
E=\gamma \Delta C J L \tag{13}
\end{equation*}
$$

If, finally, $\quad A M=A C$,

$$
\Delta A C M=\frac{A M \cdot C H}{2}=\frac{1}{2} k^{2} \cos (\varphi+\omega)
$$

or

$$
\begin{equation*}
R=\gamma \triangle A C M \tag{14}
\end{equation*}
$$

All these geometrical results may be summed up as follows:

Draw from the highest point $C$ of the surface of rupture a line $U J$, which makes with the normal $C H$ to the natural slope the angle $\alpha+\delta$, or the angle which the earthpressure makes with the horizontal ; then the $\triangle A C J$ is
equal in area to the $\triangle A B C$, the prism of rupture. Then lay off $J L=J C$ and $A M=A C$ and draw $C L$ and $C M$; then for every unit in length of the wall the following relations exist :
$\left.\begin{array}{ll}\text { Weight of prism of rupture, } & G=\gamma \Delta C A J ; \\ \text { Earth-pressure upon wall, } & E=\gamma \Delta C J L ; \\ \text { Reaction of the surface of rupture, } & R=\gamma \Delta C A M .\end{array}\right\}(14 a)$
The first two relations were first made known by Rebhahn in $18 \% 1$, for $\delta=0$ or $\varphi$.

Since, now, $G: E: R=A J: J C: C A, . \quad . \quad$
it can be asserted that-
The weight of the prism of rupture and the reactions of the wall and of the surface of rupture are to each other as the three sides of the $\triangle A C J$.

Thus far no assumption whatever has been made as to the value of the angle $\delta$. This is determined by equation (3), which, in all theories following Coulomb's method, does not occur.

## II.

## PLANE EARTH-SURFACE INCLINEB

Adopt in this case the notation of Fig. 3, an? let $E$ be first determined for any value of $\delta$.


Fig. 3.
If $A C$ is the surface of rupture, then $\triangle A B C=\triangle A C J$; or, since

$$
\frac{A B}{A \bar{C}}=\frac{\sin \mathrm{II}}{\sin \mathrm{III}}, \quad A B=A C \frac{\sin \mathrm{II}}{\sin \mathrm{III}}
$$

In like manner, $\quad A J=A C \frac{\sin V}{\sin V I}$.
But since $\quad \triangle A B C=\triangle A C J$, $A B . A C \sin I=A J . A C \sin I V ;$.
or $\quad \frac{\sin I \sin I I}{\sin I I I}=\frac{\sin I V \sin V}{\sin V I} ; ~ . ~ . ~$
or, finally,
$\sin (\alpha+\omega) \cos (\varepsilon+\omega) \cos (\alpha+\delta)$

$$
\begin{equation*}
=\sin (\phi+\omega+\alpha+\delta) \cos (\varphi+\omega) \cos (\alpha-\varepsilon) \tag{16b}
\end{equation*}
$$

Further, from Fig. 3, if $B N$ is perpendicular to $A D$,

$$
\Delta A D B=2 \Delta A J C+\triangle J D C
$$

or

$$
A D . B N=2 A J . C H+J D . C H
$$

and since

$$
\frac{B N}{U H}=\frac{B O}{C J}=\frac{O D}{J D}
$$

and

$$
\begin{aligned}
A D . O D & =J D(A J+A D) \\
A D(A D-A O) & =(A D-A J)(A J+A D)
\end{aligned}
$$

whence

$$
A O \cdot A J=A J . A D
$$



Fig. ${ }^{\prime}$.
Upon this relation rests the well-known construction of Poncelet for the earth-pressure。 Draw (Fig. 3') $B N$ perpendicular to the natural slope $A D$; draw $B O$, making the same angle with $B N$ that $E$ makes with the horizontal, and
then determine the point $J$ so that equation (17) is fulfilled, that is, make $A J$ a mean proportional between $A O$ and $A D$; then draw $J C$ parallel to $O B$. Thus the surface of rupture $A C$ is found, and use can now be made of the relations already deduced in I.

In order to determine $J(A, O$, and $D$ being given), there are several methods, one of which is indicated in the figure. In all these constructions $\delta$ is assumed.

Now from equation (13), $\quad E=\frac{1}{2} \gamma \overline{J C}^{2} \cos (\alpha+\delta)$, but

$$
\frac{C J}{B O}=\frac{A D-A J}{A D-A O}=\frac{A D-\sqrt{A D \cdot A O}}{A D}=\frac{1-\sqrt{\frac{A O}{A D}}}{1-\frac{A O}{A D}}
$$

Let $n=\sqrt{\frac{A O}{A D}}$, then $\quad C J=\frac{1-n}{1-n^{2}} B O=\frac{B O}{1+n}$.
From Fig. 3,

$$
\frac{A O}{A B}=\frac{\sin (\varphi+\delta)}{\cos (\alpha+\delta)}, \quad \frac{A B}{A D}=\frac{\sin (\varphi-\varepsilon)}{\cos (\alpha-\varepsilon)}
$$

and the multiplication of these equations gives

$$
\begin{equation*}
n=\sqrt{\frac{\sin (\varphi+\delta) \sin (\phi-\varepsilon)}{\cos (\alpha+\delta) \cos (\alpha-\varepsilon)}} \tag{18}
\end{equation*}
$$

If $A B=l, \quad B O=\frac{\cos (\varphi-\alpha)}{\cos (\alpha+\delta)} l ;$
and by substitution of $B O$ and $n$ in the value for $C J$, and of $C J$ in that for $E$,

$$
\begin{equation*}
E=\left[\frac{\cos (\phi-a)}{n+1}\right]^{2} \frac{l^{2} \gamma}{2 \cos (\alpha+\delta)}=\left[\frac{\cos (\phi-a)}{(n+1) \cos a}\right]^{2} \frac{h^{2} \gamma}{2 \cos (\alpha+\delta)^{\prime}} \ldots \tag{19}
\end{equation*}
$$

For the special case of the earth-surface parallel to the angle of repose, $\varepsilon=\varphi, n=0$, and

$$
E=\frac{\cos ^{2}(\varphi-\alpha)}{\cos (\alpha+\delta)} \frac{l^{2} \gamma}{2}=\left[\frac{\cos (\varphi-\alpha)}{\cos \alpha}\right]^{2} \frac{h^{2} \gamma}{2 \cos (\alpha+\delta)} \cdot(20)
$$

These formulæ hold good for any valne of $\delta$. But the angle $\delta$ is determined by equation (3). In order to insert $e$ and $r$ in this formula, the points of application of $E$ and $R$ must be known. The angles $\delta$ and ware connected by the relations in (16b), in which there are no other unknown quantities. Since now $\delta$, according to the single assumption of Prof. Weyranch's theory, is independent of the height, so also is $\omega$, and then for variable 7 equations (19) and (11) become

$$
\begin{aligned}
E & =C l^{2}, & R & =C_{1} k^{2} \\
d E & =2 C l d l, & d R & =2 C_{1} k d k
\end{aligned}
$$

Let $x$ and $z$ equal the distance of the point of application of $E$ and $R$ from $A$, respectively. Now considering the top as the origin or centre of moments,

$$
E(l-x)=2 C \int_{0}^{l} l^{2} d l, \quad R(k-z)=2 C_{1} \int_{0}^{k} k^{2} d k
$$

and therefore $\quad x=\frac{1}{3} l$ and $z=\frac{1}{3} k$.
Now $G$ must act through the centre of gravity of the $\triangle A B C$, and it has been already proved that the points
of application of $E$ and $R$ are at distances $\frac{1}{3} l$ and $\frac{1}{3} k$ respectively above $A$; hence (Fig. $3^{\prime}$ ) $a h=e d$ and $h f=g=$ $b d-a h=\frac{1}{3} k \sin \omega-\frac{1}{3} l \sin \alpha$.

Substituting these values in equation (3) and referring to equation (15),

$$
A B(C J \cos \delta-A J \sin \alpha)=A C(A C \cos \phi-A J \sin \omega), \ldots . \quad(22)
$$

or
$\sin I I(\sin I V \cos \delta-\sin V \sin \alpha)=\sin I I I(\sin V I \cos \phi-\sin V \sin \omega),(22 a)$ $\cos (\epsilon+\omega)[\cos (\phi+\omega) \cos \delta-\sin (\phi+\omega+a+\delta) \sin a]$ $=\cos (\alpha-\epsilon)[\cos (\alpha+\delta) \cos \phi-\sin (\phi+\omega+\alpha+\delta) \sin \omega]$. . . (22b)

By means of the two equations (16b) and (22b) the two unknown quantities $\delta$ and $\omega$ are completely determined. As soon as these are known, $E$ can be found from equation (19) or (20). Also by the relations in equations (16) and (22), or (16a) and (22b), the surface of rupture and direction of the earth-pressure may be determined, and can therefore be found by a graphical construction.

## III.

## HORIZONTAL EARTH-SURFACE.

For this most important practical case it is simply necessary to make $\varepsilon=0$ in equation (19). The proper values of $\delta$ and $\omega$ in this case are found from (16b) and (22b).

Making $\varepsilon=0$ in equation (22b), it becomes
$\cos \omega[\cos (\phi+\omega) \cos \delta-\sin (\varphi+\omega+\alpha+\delta) \sin \alpha]$ $-\cos \alpha[\cos (\alpha+\delta) \cos \varphi-\sin (\phi+\omega+\alpha+\delta) \sin \omega]=0$.

Since

$$
\begin{aligned}
& \sin (\varphi+\omega+\alpha+\delta)=\sin (\varphi+\omega) \cos (\alpha+\delta) \\
& +\cos (\varphi+\omega) \sin (\alpha+\delta), \\
& \cos (\alpha+\delta)=\cos \alpha \cos \delta-\sin \alpha \sin \delta, \\
& \text { and } \\
& \sin (\alpha+\delta)=\sin \alpha \cos \delta+\cos \alpha \sin \delta,
\end{aligned}
$$ the above expression becomes

$\cos \omega \cos \delta \cos (\varphi+\omega)$
$-\cos \omega \sin \alpha \cos \alpha \cos \delta \sin (\varphi+\omega)$

$$
+\cos \omega^{2} \sin ^{2} \alpha \sin \delta \sin (\varphi+\omega)
$$

$-\cos \omega \sin \alpha \cos \alpha \sin \delta \cos (\varphi+\omega)$

$$
-\cos \omega \sin ^{2} \alpha \cos \delta \cos (\varphi+\omega)
$$

$-\cos \alpha \cos \varphi \cos (\alpha+\delta)$
$+\cos ^{2} \alpha \sin \omega \cos \delta \sin (\varphi+\omega)$
$-\cos \alpha \sin \omega \sin \alpha \sin \delta \sin (\varphi+\omega)$
$+\cos ^{2} \alpha \sin \omega \sin \delta \cos (\varphi+\omega)$
$+\cos \alpha \sin \omega \sin \alpha \cos \delta \cos (\varphi+\omega)]$
which reduces to
$\cos \omega \cos (\varphi+\omega) \cos \delta$
$-\sin \alpha \cos \alpha[\sin (\varphi+\omega) \cos \omega-\cos (\varphi+\omega) \sin \omega] \cos \delta$
$-\sin \alpha \cos \alpha[\cos (\phi+\omega) \cos \omega+\sin (\varphi+\omega) \sin \omega] \sin \delta$ $+\left[\sin ^{2} \alpha \sin (\varphi+\omega) \cos \omega+\cos ^{2} \alpha \cos (\varphi+\omega) \sin \omega\right] \sin \delta$ $+\left[\cos ^{2} \alpha \sin (\varphi+\omega) \sin \omega-\sin ^{2} \alpha \cos (\varphi+\omega) \cos \omega\right] \cos \delta$
$-\cos ^{2} \alpha \cos \varphi \cos \delta+\sin \alpha \cos \alpha \cos \varphi \sin \delta$
= 0. . . . . . . . . . . . . . . . . (22c)
The expression in the first parenthesis is equal to $\sin \varphi$, in the second to $\cos \varphi$. If in the third $\cos ^{2} \alpha=1-\sin ^{2} \alpha$, and in the fourth $\sin ^{2} \alpha=1-\cos ^{2} \alpha$, equation (22c) becomes
$+\cos \omega \cos (\varphi+\omega) \cos \delta-\sin \alpha \cos \alpha \cos \delta \sin \varphi$ $-\sin \alpha \cos \alpha \sin \delta \cos \phi$
$+\sin \delta \sin ^{2} \alpha \sin (\varphi+\omega) \cos \omega+\sin \delta \sin \omega \cos (\varphi+\omega)$ $\left.-\sin ^{2} \alpha \sin \omega \sin \delta \cos (\varphi+\omega)\right\}=0$.
$+\cos \delta \cos ^{2} \alpha \sin (\varphi+\omega) \sin \omega-\cos \delta \cos \omega \cos (\phi+\omega)$ $+\cos ^{2} \alpha \cos \delta \cos \omega \cos (\varphi+\omega)$
$-\cos ^{2} \alpha \cos \varphi \cos \delta+\sin \alpha \cos \alpha \cos \varphi \sin \delta$
Reducing and dividing by $\cos \delta$,
$-\sin \alpha \cos \alpha \sin \varphi+\sin ^{2} \alpha \cos \omega \sin (\varphi+\omega) \tan \delta$ $+\sin \omega \cos (\varphi+\omega) \tan \delta$
$-\sin ^{2} \alpha \sin \omega \cos (\varphi+\omega) \tan \delta$ $=0$ 。 $+\cos ^{2} \alpha \sin \omega \sin (\varphi+\omega)$
$+\cos ^{2} \alpha \cos \omega \cos (\varphi+\omega)-\cos ^{2} \alpha \cos \varphi$

## Since

$$
\cos \omega \sin (\varphi+\omega)-\sin \omega \cos (\varphi+\omega)=\sin \varphi
$$

and

$$
\sin \omega \sin (\varphi+\omega)+\cos \omega \cos (\varphi+\omega)=\cos \varphi
$$

this reduces to
$-\sin \alpha \cos \alpha \sin \varphi+\sin ^{2} \alpha \sin \varphi \tan \delta$ $+\sin \omega \cos (\varphi+\omega) \tan \delta=0$;
and since

$$
\cos (\varphi+\omega) \sin \omega=\frac{1}{2} \sin (2 \omega+\varphi)-\frac{1}{2} \sin \varphi
$$

this becomes

$$
\tan \delta=\frac{2 \sin \alpha \cos \alpha \sin \varphi}{2 \sin ^{2} \alpha \sin \varphi+\sin (2 \omega+\varphi)-\sin \varphi}
$$

and since

$$
\sin \alpha \cos \alpha=\frac{1}{2} \sin 2 \alpha \quad \text { and } \quad 1-2 \sin ^{2} \alpha=\cos 2 \alpha
$$

this reduces to

$$
\begin{equation*}
\tan \delta=\frac{\sin \varphi \sin }{\sin (2 \omega+\varphi)-} \frac{\alpha}{\sin \varphi \cos } \overline{2 \alpha} \tag{23}
\end{equation*}
$$

This equation, therefore, expresses the condition that the "sum of the moments of $E, G$, and $R$ is zero."

Substituting $\frac{\sin \delta}{\cos \delta}$ for $\tan \delta$ in equation (23), clearing of fractions and factoring,
$\sin \delta \sin (2 \omega+\varphi)-\sin \delta \sin \varphi \cos 2 \alpha=\sin \varphi \cos \delta \sin 2 \alpha$,
or
$\sin \delta \sin (2 \omega+\varphi)=\sin \varphi \cos \delta \sin 2 \alpha+\sin \varphi \sin \delta \cos 2 \alpha$.

Since $\quad \cos \delta \sin 2 \alpha+\sin \delta \cos 2 \alpha=\sin (2 \alpha+\delta)$,
this becomes

$$
\begin{equation*}
\sin \delta \sin (2 \omega+\varphi)=\sin \varphi \sin (2 \alpha+\delta) \tag{24}
\end{equation*}
$$

In order to determine $\omega$ it is only necessary to make $\varepsilon=0$ in equation (16b) express $\sin (\varphi+\omega+\alpha+\delta)$ in terms of $\sin$ and $\cos (\varphi+\omega)$ and $(\alpha+\delta)$, and then the sin and $\cos$ of $(\alpha+\delta)$ in terms of the sin and $\cos$ of $\alpha$ and $\delta$. Making $\varepsilon=0$ in equation (16b), it becomes

$$
\begin{align*}
\sin (\alpha & +\omega) \cos (\alpha+\delta) \cos \omega \\
& =\sin (\varphi+\omega+\alpha+\delta)[\cos (\varphi+\omega) \cos \alpha] \tag{24a}
\end{align*}
$$

Since
$\sin (\varphi+\omega+\alpha+\delta)=\sin (\varphi+\omega) \cos (\alpha+\delta)$

$$
+\cos (\varphi+\omega) \sin (\alpha+\delta)
$$

$$
\sin (\alpha+\delta)=\sin \alpha \cos \delta+\cos \alpha \sin \delta
$$

$$
\cos (\alpha+\delta)=\cos \alpha \cos \delta-\sin \alpha \sin \delta
$$

hence

$$
\sin (\varphi+\omega+\alpha+\delta)=\sin (\varphi+\omega) \cos \alpha \cos \delta
$$

$$
-\sin (\varphi+\omega) \sin \alpha \sin \delta
$$

$$
+\cos (\phi+\omega) \sin \alpha \cos \delta
$$

$+\cos (\varphi+\omega) \cos \alpha \sin \delta$,
and equation (24a) reduces to
$\cos \omega \sin (\alpha+\omega) \cos \alpha \cos \delta$
$-\cos \omega \sin (\alpha+\omega) \sin \alpha \sin \delta$
$-\cos ^{2} \alpha \cos (\varphi+\omega) \sin (\varphi+\omega) \cos \delta$
$+\cos \alpha \cos (\varphi+\omega) \sin (\varphi+\omega) \sin \alpha \sin \delta\}=0$. $(24 b)$
$-\cos \alpha \cos ^{2}(\varphi+\omega) \sin \alpha \cos \delta$
$-\cos ^{2} \alpha \cos ^{2}(\varphi+\omega) \sin \delta$

Dividing by $\cos \delta$,
$\cos \alpha \cos \omega \sin (\alpha+\omega)$
$-\cos \omega \sin \alpha \sin (\alpha+\omega) \tan \delta$
$-\cos ^{2} \alpha \cos (\varphi+\omega) \sin (\varphi+\omega)$
$+\cos \alpha \sin \alpha \cos (\varphi+\omega) \sin (\varphi+\omega) \tan \delta\}=0$. (24c)
$-\cos \alpha \sin \alpha \cos ^{2}(\varphi+\omega)$
$-\cos ^{2} \alpha \cos ^{2}(\varphi+\omega) \tan \delta$
Since
$\cos \alpha \cos \omega \sin (\alpha+\omega)$ equals, by expanding $\sin (\alpha+\omega)$, $\sin \alpha \cos \alpha \cos ^{2} \omega+\sin \omega \cos \omega \cos ^{2} \alpha$, and likewise
$-\cos \omega \sin \alpha \sin (\alpha+\omega) \tan \delta=-\cos ^{2} \omega \sin ^{2} \alpha \tan \delta$
$-\cos \alpha \sin \alpha \cos \omega \sin \omega \tan \delta$,
equation (24c) becomes
$-\sin \alpha \cos \alpha\left[\cos ^{2}(\varphi+\omega)-\cos ^{2} \omega\right]$
$-\cos ^{2} \alpha[\sin (\phi+\omega) \cos (\varphi+\omega)-\sin \omega \cos \omega]$
$\left.-\left[\cos ^{2} \alpha \cos ^{2}(\varphi+\omega)+\sin ^{2} \alpha \cos ^{2} \omega\right] \tan \delta\right\}=0 .(24 d)$
$+\sin \alpha \cos \alpha[\sin (\varphi+\omega) \cos (\varphi+\omega)$
$-\sin \omega \cos \omega] \tan \delta$

Now

$$
\cos ^{2}(\varphi+\omega)-\cos ^{2} \omega=\frac{\cos 2(\varphi+\omega)-\cos 2 \omega}{2}
$$

which equals

$$
\begin{gathered}
\frac{2 \sin \frac{1}{2}[2 \omega-(2 \varphi+2 \omega)] \sin \frac{1}{2}[2 \omega+(2 \varphi+2 \omega)]}{2} \\
=\frac{2 \sin (-\varphi) \sin (2 \omega+\varphi)}{2}, \\
-\sin (2 \omega+\varphi) \sin \varphi,
\end{gathered}
$$

or
and

$$
\begin{aligned}
\sin (\varphi+\omega) \cos (\varphi+\omega) & -\sin \omega \cos \omega \\
& =\frac{1}{2} \sin 2(\varphi+\omega)-\frac{1}{2} \sin 2 \omega ;
\end{aligned}
$$

also,

$$
\sin \alpha \cos \alpha=\frac{\sin 2 \alpha}{2}, \text { and } \cos ^{2} \alpha=\frac{\cos 2 \alpha}{2}-+\frac{1}{2}
$$

Hence, after multiplying by 2 , equation (24d) reduces to

$$
\left.\begin{array}{l}
\quad \sin 2 \alpha \sin (2 \omega+\varphi) \sin \varphi \\
\quad-\cos 2 \alpha \frac{1}{2} \sin 2(\varphi+\omega)+\cos 2 \alpha \frac{1}{2} \sin 2 \omega \\
-\frac{1}{2} \sin 2(\varphi+\omega)+\frac{1}{2} \sin 2 \omega \\
-\tan \delta \cos 2 \alpha \cos ^{2}(\varphi+\omega)-\cos ^{2}(\varphi+\omega) \tan \delta \\
-2 \tan \delta \sin ^{2} \alpha \cos ^{2} \omega \\
\quad \quad+\sin 2 \alpha \sin (\varphi+\omega) \cos (\varphi+\omega) \tan \delta \\
-\sin 2 \alpha \sin \omega \cos \omega \tan \delta
\end{array}\right\}=0 .(24 e)
$$

Now
$-2 \tan \delta \sin ^{2} \alpha \cos ^{2} \omega=\left[\right.$ since $\left.\sin ^{2} \alpha=1-\cos ^{2} \alpha\right]$

$$
-\left[\cos ^{2} \omega-\cos ^{2} \alpha \cos ^{2} \omega\right] 2 \tan \delta,
$$

which equals

$$
-\underline{2 \cos ^{2} \omega \tan \delta}+2 \tan \delta \cos ^{2} \alpha \cos ^{2} \omega .
$$

Also,

$$
\begin{aligned}
& -\frac{\cos 2 \alpha \sin 2(\phi+\omega)}{2}+\frac{\cos 2 \alpha \sin 2 \omega}{2} \\
& =-\cos 2 \alpha\left[\frac{\sin 2(\phi+\omega)-\sin 2 \omega}{2}\right] \\
& =-\frac{\cos 2 \alpha[2 \sin \varphi \cos (2 \omega+\varphi)]}{2} \frac{\cos 2 \alpha \cos (2 \omega+\varphi) \sin \varphi,}{} \\
& =-\frac{\cos }{2}
\end{aligned}
$$

and

$$
\begin{aligned}
& -\frac{\sin 2(\phi+\omega)}{2}+\frac{\sin 2 \omega}{2}=-\frac{\sin 2(\phi+\omega)-\sin 2 \omega}{2} \\
= & -\frac{2 \sin \frac{1}{2}(2 \phi+2 \omega-2 \omega)}{2} \frac{\cos \frac{1}{2}(2 \varphi+2 \omega+2 \omega)}{} \\
= & -\frac{\sin \varphi \cos (2 \omega+\varphi),}{}
\end{aligned}
$$

and
$-\tan \delta \cos 2 \alpha \cos ^{2}(\varphi+\omega)+2 \tan \delta \cos ^{2} \alpha \cos ^{2} \omega$
$=\left(\right.$ by making $\left.\cos ^{2} \alpha=\frac{\cos 2 \alpha}{2}+\frac{1}{2}\right)$
$-\tan \delta \cos 2 \alpha\left[\cos ^{2}(\varphi+\omega)-\cos ^{2} \omega\right]+\tan \delta \cos ^{2} \omega$,
or $\quad \tan \delta \cos 2 \alpha \sin (2 \omega+\varphi) \sin \varphi+\tan \delta \cos ^{2} \omega$,

Also, $\quad$|  | $-\cos ^{2}(\varphi+\omega) \tan \delta+\tan \delta \cos ^{2} \omega$ |
| ---: | :--- |
|  | $=-\tan \delta\left[\cos ^{2}(\varphi+\omega)-\cos ^{2} \omega\right]$ |
|  | $=\underline{\sin \varphi \sin (2 \omega+\varphi) \tan \delta .}$ |

Also,
$\tan \delta \sin 2 \alpha \sin (\varphi+\omega) \cos (\varphi+\omega)$
$-\sin 2 \alpha \sin \omega \cos \omega \tan \delta$
$=\tan \delta \sin 2 \alpha[\sin (\varphi+\omega) \cos (\varphi+\omega)-\sin \omega \cos \omega]$
$=\tan \delta \sin 2 \alpha\left[\frac{\sin 2(\varphi+\omega)-\sin 2 \omega}{2}\right]$
$=\tan \delta \sin 2 \alpha \sin \varphi \cos (2 \omega+\phi) ;$
and hence equation (24e) becomes

$$
\begin{array}{r}
+\sin \varphi[\sin (2 \omega+\varphi) \sin 2 \alpha-\cos (2 \omega+\varphi) \cos 2 \alpha] \\
-\sin \varphi \cos (2 \omega+\varphi) \tag{24f}
\end{array}
$$

$+\sin \varphi[\sin (2 \omega+\varphi) \cos 2 \alpha$

$$
+\cos (2 \omega+\phi) \sin 2 \alpha] \tan \delta
$$

$+\sin \varphi[\sin (2 \omega+\varphi) \tan \delta]-2 \cos ^{2} \omega \tan \delta \quad$
and
$\tan \delta=\frac{\sin \phi[\sin (2 \omega+\phi) \sin 2 \alpha-\cos (2 \omega+\phi) \cos 2 \alpha]-\sin \phi \cos (2 \omega+\phi)}{2 \cos ^{2} \omega-\sin \frac{1}{\phi[\sin (2 \omega+\phi) \cos 2 \alpha+\cos (2 \omega+\phi) \sin 2 \alpha]-\sin \phi \sin (2 \omega+\phi)} .}$
By making $\sin 2 \alpha=2 \sin \alpha \cos \alpha$ and $\cos 2 \alpha=1-2 \sin ^{2} \alpha$ in the numerator, and $\cos 2 \alpha=2 \cos \alpha \cos \alpha-1$ and $\sin$ $2 \alpha=2 \sin \alpha \cos \alpha$ in the denominator, this becomes
$\tan \delta=$
$\sin \phi\left[\sin (2 \omega+\phi) 2 \sin a \cos \alpha-\cos (2 \omega+\phi)+\cos (2 \omega+\phi) 2 \sin ^{2} a\right]-\sin \phi \cos (2 \omega+\phi)$ $2 \cos ^{2} \omega-\sin \phi\left[\sin (2 \omega+\phi) 2 \cos ^{2} \alpha-\sin (2 \omega+\phi)+\cos (2 \omega+\phi) 2 \sin \alpha \cos \alpha\right]-\sin \phi \sin (2 \omega+\phi){ }^{\prime}$
or
$\tan \delta=\frac{2 \sin \phi \sin \alpha}{2 \cos ^{2} \omega-2 \sin (2 \omega+\phi)} \frac{\cos \alpha+\cos }{-2 \sin \alpha[\sin (2 \omega+\phi) \sin \alpha]-2 \sin \phi \cos (2 \omega+\phi)}$,
which reduces to
$\tan \delta=$
$\frac{\sin \varphi \sin \alpha \sin (2 \omega+\phi+\alpha)-\sin \varphi \cos (2 \omega+\phi)}{\cos ^{2} \omega-\sin \varphi \cos \alpha \sin (2 \omega+\varphi+\alpha)}$.
Equating this value of $\tan \delta$ with that in equation (23),

$$
\begin{gathered}
\frac{\sin \varphi \sin \alpha \sin (2 \omega+\phi+\alpha)-\sin \varphi \cos (2 \omega+\phi)}{\cos ^{2} \omega-\sin \varphi \cos \alpha \sin (2 \omega+\varphi+\alpha)} \\
\quad=\frac{\sin \varphi \sin 2 \alpha}{\sin } \frac{\alpha \omega+\phi)-\sin }{\varphi \cos 2 \alpha} .
\end{gathered}
$$

Dividing by $\sin \varphi$, clearing of fractions and dividing by $\sin \alpha$, also transposing, this becomes
$\left.\begin{array}{l}\sin (2 \omega+\phi+\alpha) \sin (2 \omega+\phi) \\ -\sin (2 \omega+\phi+\alpha) \sin \varphi \cos 2 \alpha-\frac{\sin 2 \alpha}{\sin \alpha} \cos ^{2} \omega \\ +\frac{\sin 2 \alpha}{\sin \alpha} \cos \alpha \sin (2 \omega+\varphi+\alpha) \sin \varphi \\ -\frac{\cos (2 \omega+\phi)[\sin (2 \omega+\phi)-\sin \varphi \cos 2 \alpha]}{\sin \alpha}\end{array}\right\}=0$,
or
$\left.\begin{array}{l}\sin (2 \omega+\varphi+\alpha) \sin (2 \omega+\phi) \\ -\sin \varphi \cos 2 \alpha \sin (2 \omega+\phi+\alpha)-2 \cos \alpha \cos ^{2} \omega \\ +\sin \varphi 2 \cos ^{2} \alpha \sin (2 \omega+\phi+\alpha) \\ -\frac{\cos (2 \omega+\varphi)\lceil\sin (2 \omega+\phi)-\sin \varphi \cos 2 \alpha\rceil}{\sin \alpha}\end{array}\right\}=0$.
Since $2 \cos ^{2} \alpha-\cos 2 \alpha=1$,
this becomes

$$
\begin{aligned}
\sin (2 \omega+\varphi+\alpha)[\sin (2 \omega+\varphi) & +\sin \varphi] \\
& -2 \cos \alpha \cos ^{2} \omega-D=0
\end{aligned}
$$

in which

$$
D=\frac{\cos (2 \omega+\varphi)[\sin (2 \omega+\varphi)-\sin \varphi \cos 2 \alpha]}{\sin \alpha},
$$

or
$\sin (2 \omega+\varphi+\alpha)[2 \sin (\omega+\varphi) \cos \omega]-2 \cos \alpha \cos ^{2} \omega-D=0$, or
$\sin (2 \omega+\varphi+\alpha) \sin (\omega+\varphi)-\cos \alpha \cos \omega-\frac{D}{2 \cos \omega}=0 .(25)$
The formulæ for $\omega, \delta$, and $E$ can now be found in the simplest manner. Equation (25) is satisfied for $2 \omega+\varphi=90^{\circ}$. Hence,

$$
\begin{equation*}
\omega=45^{\circ}-\frac{\varphi}{2} . \tag{26}
\end{equation*}
$$

Substituting this value in equation (23), it becomes

$$
\begin{align*}
\tan \delta & =\frac{\sin \varphi \sin 2 \alpha}{\sin (90-\phi+\varphi)-\sin \varphi \cos 2 \alpha} \\
& =\frac{\sin \varphi \sin 2 \alpha}{1-\sin \frac{2 \alpha}{\varphi} \frac{\cos 2 \alpha}{}, \quad . \quad . \quad .} \tag{27}
\end{align*}
$$

or the equivalent, but more convenient expression for calculation,

$$
\begin{equation*}
\tan (\delta+\alpha)=\frac{\tan \alpha}{\tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right)} \tag{28}
\end{equation*}
$$

If, finally, $\omega=45^{\circ}-\frac{\varphi}{2}$ in equation (10), it becomes, re. membering that $k^{2}=\frac{h^{2}}{\cos ^{2} \omega}$,

$$
\begin{aligned}
E & =\frac{\cos ^{2}\left(\varphi+45^{\circ}-\frac{\varphi}{2}\right)}{\cos (\alpha+\delta)} \cdot \frac{h^{2} \gamma}{2 \cos ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)} \\
& =\frac{\cos ^{2}\left(45^{\circ}+\frac{\varphi}{2}\right)}{\cos ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)} \cdot \frac{\hbar^{2} \gamma}{2 \cos (\alpha+\delta)} \\
& =\frac{\sin ^{2}\left[90^{\circ}-\left(45^{\circ}+\frac{\varphi}{2}\right)\right]}{\cos ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)} \cdot \frac{\hbar^{2} \gamma}{2 \cos (\alpha+\delta)}
\end{aligned}
$$

hence $\quad E=\tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right) \frac{\hbar^{2} \gamma}{2 \cos (\alpha+\delta)}$,
or, from equation (28),

$$
\begin{equation*}
E=\frac{\tan \alpha}{\sin (\alpha+\delta)} \frac{h^{2} \gamma}{2} \tag{29a}
\end{equation*}
$$

This last expression, however, when $\alpha=0$ takes the indeterminate form $\frac{0}{0}$.

The earth-pressure upon a portion of the wall reaching from the depth $h_{0}$ to the depth $H=h_{0}+h_{1}$ may be found
from equation (29) by substituting $H^{2}-h_{0}{ }^{2}$ in place of $h^{2}$, as is evident from the following:

Suppose the wall to have a height $H$, then $E_{0}=C_{0} \frac{H^{2}}{2} \gamma$, and likewise for a height $h_{\text {。 }}$
$E_{1}=C_{0} \frac{h_{0}{ }^{2}}{2} \gamma \therefore E=E_{0}-E_{1}=C_{0} \frac{H^{2}-h_{0}{ }^{2}}{2} \gamma,$.
$C_{0}$ representing the constant quantity.
From equation (296) $E=C\left(H^{2}-h_{0}{ }^{2}\right)$; hence $d E=$ $2 \mathrm{CHdH}-2 \mathrm{Ch}_{0} d h_{0}$. Now let $x$ equal the distance of the centre of pressure below the top of the wall, then

$$
E x=2 C \int_{0}^{H} H^{2} d H-2 C \int_{0}^{h} h_{0}^{2} d h
$$

or

$$
C\left(H^{2}-h_{0}{ }^{2}\right) x=\frac{2}{3} C H^{3}-\frac{2}{3} C h_{0}{ }^{3},
$$

or

$$
x=\frac{2}{3} \frac{H^{3}-h_{0}{ }^{3}}{H^{2}-h_{0}^{2}} ;
$$

and if $y=$ the distance from bottom,

$$
\begin{equation*}
y=\frac{1}{3} \frac{H^{3}-3 H h_{0}{ }^{2}+2 h_{0}{ }^{3}}{H^{2}-h_{0}{ }^{2}} \tag{30}
\end{equation*}
$$

Equation (30) holds good when the earth-surface is loaded and the loading is equal to a distributed load of the height $\hbar_{0}$. Still, even then, $\hbar_{0}$ is often so small that $\frac{h}{3}$ can be substituted for it just as for unloaded earth-surface.

In all cases $\delta$ is determined by equation (28).

Instead of using equations (28) and (29), the following simple construction can be used :


Fig. 4.
Draw (Fig. 4) $A C$ and $A D$ vertically and horizontally, each equal to $h$, also $D F$ making the angle $F D G=45^{\circ}-\frac{\varphi}{2}$ with the horizontal. Through the points $D$ and $F$ describe a circle whose centre lies in $A D$. Then draw $G H$ parallel to $A B$, and through $A$ the straight line $H J$. Then $J G$ is the direction of the earth-pressure upon the wall $A B$. If $A K$ is made perpendicular to $A B$, and equal to $A H$, then the $\triangle A B K$ gives the intensity and distribution of the earth-pressure, or

$$
E=\gamma \Delta A B K
$$

The proof of this construction is as follows: Conceive, in Fig. 4, JD and $F G$ drawn, then

$$
\tan A H G=\frac{A P_{0}}{P H}=\frac{A G \cos \alpha}{H G-\left[A G \sin \alpha=P^{\prime} G\right]} ;
$$

in which $A P$ represents the perpendicular let fall from $A$ upon $G H$.
but

$$
A G: A F:: A F: A D=h
$$

therefore $\quad A G=\frac{\overline{A F}^{2}}{h}=\hbar \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)$.
Now

$$
\begin{aligned}
H G=G D \sin \alpha & =(A G+A D) \sin \alpha \\
& =h \sin \alpha+h \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right) \sin \alpha
\end{aligned}
$$

$\tan A H G=$
$\frac{h \tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right) \cos \alpha}{h \sin \alpha+h \tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right) \sin \alpha-h \tan ^{2}\left(45^{\circ}-\frac{\phi}{2}\right) \sin \alpha} ;$
therefore
$\tan A H G=\frac{\cos \alpha}{\sin \alpha} \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)=\cot \alpha \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)$.
From Fig. 4, $\angle G D J=<A H G,<G D J+<J G D=90^{\circ}$, and therefore
$\tan J G D=\cot A H G=\tan \alpha \cot ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)=\tan (\alpha+\delta)$,
or $<J G D$ is the angle of the earth-pressure to the horizon.
Since, now, $\quad<A H G=90^{\circ}-\alpha-\delta$,
$A H=\frac{\cos \alpha}{\cos (\alpha+\delta)} A G=\hbar \dot{\tan }{ }^{2}\left(45^{\circ}-\frac{\varphi}{2}\right) \frac{\cos \alpha}{\cos (\alpha+\delta)}$, and

$$
\frac{1}{2} A H \cdot A B=\tan ^{2}\left(45^{\circ}-\frac{\phi}{z}\right) \frac{h^{2}}{2 \cos (\alpha+\delta)}=\frac{E}{\gamma}
$$

For a vertical wall the construction becomes much simpler. Draw, in Fig. 5, $A D=h$ horizontally, then $D F$ making the angle $45^{\circ}-\frac{\varphi}{2}$ with $A D$. Draw through $D$ and $F$ a circle with centre in $D A$ and continue it around to $K$.


Fig. 5.
then the $\triangle A B K$ gives the intensity and distribution of the earth-pressure, while in direction it is horizontal.

Hence

$$
E=\gamma \Delta A B K .
$$

The proof is as follows (Fig. 5):

$$
\begin{gathered}
A K=\frac{A F^{2}}{A D}=\frac{h^{2} \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)}{h}=h \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right) \\
\frac{1}{2} A B \cdot A K=\frac{h^{2}}{2} \tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)=\frac{E}{\gamma} .
\end{gathered}
$$

As $\alpha=0$, equation (28) gives $\tan \delta=0 ; . \therefore \delta=0$ and $E$ act normal to the surface of the wall.


Fig. 6.
Finally, in Fig. 6 is the construction for loaded earthsurface. The point of application of the earth-pressure is always found by drawing through the centre of gravity of $\triangle A B K$ a parallel to $A K$ and producing it to meet the wall. The proof for this construction is the same as that for Fig. 4.

## IV.

## EARTH SURFACE PARALLEL TO SURFACE OF REPOSE.

$$
\varepsilon=\mathscr{q}
$$

For this case,

$$
\begin{equation*}
E=\frac{\cos ^{2}(\varphi-\alpha)}{\cos (\alpha+\delta)} \frac{l^{2} \gamma}{2}=\left[\frac{\cos (\varphi-\alpha)}{\cos \alpha}\right]^{2} \frac{\hbar^{2} \gamma}{2 \cos (\alpha+\delta)} \tag{20}
\end{equation*}
$$

a formula which holds good for all values of $\delta$, and which for $\delta=0$ or $\varphi$ gives results usually accepted in previous theories of retaining-walls. In order to find the proper values of $\delta$ and $\omega$, equations (16b) and (22b) must be used.

In equation (22b) replace $\sin (\varphi+\omega+\alpha+\delta)$ by sin $(\varphi+\omega+\alpha) \cos \delta+\cos (\varphi+\omega+\alpha) \sin \delta$, and making $\varepsilon=\varphi$ it becomes

$$
\begin{aligned}
& \left.\begin{array}{l}
+\cos (\varphi+\omega) \cos (\varphi+\omega) \cos \delta \\
\quad-\cos (\varphi+\omega) \sin (\phi+\omega+\alpha) \cos \delta \sin \alpha \\
-\cos (\varphi+\omega) \cos (\varphi+\omega+\alpha) \sin \delta \sin \alpha
\end{array}\right\}= \\
& =\left\{\begin{array}{l}
+\cos (\alpha-\phi) \cos (\alpha+\delta) \cos \varphi \\
-\cos (\alpha-\phi) \sin (\varphi+\omega+\alpha) \sin \omega \cos \delta \\
-\cos (\alpha-\phi) \cos (\varphi+\omega+\alpha) \sin \delta \sin \omega ;
\end{array}\right.
\end{aligned}
$$

dividing by $\cos \delta$ and transposing,

$$
\left.\begin{array}{l}
-\frac{\cos (\alpha-\varphi) \cos (\alpha+\delta) \cos \varphi}{\cos \delta} \\
+\cos (\alpha-\phi) \sin (\varphi+\omega+\alpha) \sin \omega \\
+\cos (\phi+\omega) \cos (\phi+\omega) \\
-\cos (\varphi+\omega) \sin (\phi+\omega+\alpha) \sin \alpha
\end{array}\right\}=
$$

Since

$$
\begin{aligned}
& -\frac{\cos (\alpha-\phi) \cos (\alpha+\delta) \cos \phi}{\cos \delta}=-\frac{\cos (\alpha-\phi) \cos \phi(\cos \alpha \cos \delta-\sin \alpha \sin \delta)}{\cos \delta} \frac{\cos }{} \\
& =-\cos (\alpha-\phi) \cos \phi \cos \alpha+\cos (\alpha-\phi) \sin a \frac{\sin \delta}{\cos \delta} \cos \phi,
\end{aligned}
$$

the above expression reduces to
$\tan \delta=$
$\cos \alpha \cos (\alpha-\phi) \cos \phi-\cos \alpha \cos (\phi+\omega) \cos (\phi+\omega+\alpha)-\cos (\alpha-\phi) \sin \omega \sin (\phi+\omega+\alpha)$ $\sin a \cos (\alpha-\phi) \cos \phi-\sin \alpha \cos (\phi+\omega) \cos (\phi+\omega+\alpha)+\cos (\alpha-\phi) \sin \omega \cos (\phi+\omega+\alpha)$
and this equation fulfils the condition that the sum of the moments of $G, E$, and $R$ shall be zero.

If equation (16b) is treated in a like manner, the resulting equation will fulfil the condition that the sum of the forces parallel to the surface of rupture shall equal zero. Making $\varepsilon=\varphi$ in equation (16b), it reduces to

$$
\begin{aligned}
& \sin (\alpha+\omega) \cos (\phi+\omega) \cos (\alpha+\delta) \\
& \quad-\sin (\varphi+\alpha+\omega+\delta) \cos (\varphi+\omega) \cos (\alpha-\phi)=0 \\
& \quad 3
\end{aligned}
$$

or
$\sin (\alpha+\omega) \cos (\alpha+\delta)-\sin (\varphi+\omega+\alpha) \cos (\alpha-\varphi) \cos \delta$ $-\cos (\varphi+\omega+\alpha) \cos (\alpha-\varphi) \sin \delta=0$,
or

$$
\frac{\sin (\alpha+\omega) \cos \alpha \cos \delta}{\cos \delta}-\frac{\sin (\alpha+\omega) \sin \alpha \sin \delta}{\cos \delta}-
$$

$\sin (\phi+\omega+\alpha) \cos (\alpha-\varphi)-\frac{\cos (\phi+\omega+\alpha) \cos (\alpha-\phi) \sin \delta}{\cos \delta}=0 ;$
therefore
$\tan \delta=\frac{\cos \alpha \sin (\alpha+\omega)-\sin (\phi+\omega+\alpha) \cos (\alpha-\phi)}{\sin (\alpha+\omega) \sin \alpha+\cos (\phi+\omega+\alpha) \cos (\alpha-\phi)}$.
Setting both values of $\tan \delta$ equal to each other and clearing of fractions, the following expression is obtained:

$$
\begin{aligned}
& +\cos \alpha \cos \varphi \sin \alpha \sin (\omega+\alpha) \cos (\alpha-\varphi) \\
& -\cos \alpha \sin \alpha \sin (\omega+\alpha) \cos (\omega+\varphi) \cos (\omega+\varphi+\alpha) \\
& -\sin \omega \sin \alpha \sin (\omega+\alpha) \cos (\alpha-\varphi) \sin (\varphi+\omega+\alpha) \\
& +\cos \alpha \cos \varphi \cos (\alpha-\varphi) \cos (\varphi+\omega+\alpha) \cos (\alpha-\phi) \\
& -\cos \alpha \cos (\varphi+\omega) \cos ^{2}(\varphi+\omega+\alpha) \cos (\alpha-\varphi) \\
& -\sin \omega \cos ^{2}(\alpha-\varphi) \sin (\varphi+\omega+\alpha) \cos (\varphi+\omega+\alpha)
\end{aligned}
$$

for the first member of the equation, and
$+\cos \alpha \cos \varphi \sin \alpha \sin (\omega+\alpha) \cos (\alpha+\varphi)$
$-\sin \alpha \cos \alpha \sin (\omega+\alpha) \cos (\omega+-) \cos (\varphi+\omega+\alpha)$
$+\sin \omega \cos \alpha \sin (\omega+\alpha) \cos (\alpha-\varphi) \cos (\varphi+\omega+\alpha)$
$-\sin \alpha \cos \varphi \cos ^{2}(\alpha-\phi) \sin (\varphi+\omega+\alpha)$
$+\sin \alpha \cos (\phi+\omega) \cos (\phi+\omega+\alpha) \cos (\alpha-\phi) \sin (\phi+\omega+\alpha)$
$-\sin \omega \cos ^{2}(\alpha-\varphi) \cos (\varphi+\omega+\alpha) \sin (\varphi+\omega+\alpha)$
for the second member.

The first terms, second terms, and sixth terms cancel. Divide the equation by $\cos (\alpha-\phi)$. Terms number 3 combined give

$$
-\sin \omega \sin (\omega+\alpha)[\sin \alpha \sin (\phi+\omega+\alpha)+\cos \alpha \cos (\phi+\omega+\alpha)],
$$

which becomes

$$
-\sin \omega \sin (\omega+\alpha) \cos (\varphi+\omega)
$$

Terms number 5 combined give

$$
-\cos (\phi+\omega) \cos (\phi+\omega+a)[\cos a \cos (\phi+\omega+a)+\sin a \sin (\phi+\omega+\alpha)],
$$

which becomes

$$
-\cos (\varphi+\omega+\alpha) \cos (\varphi+\omega) \cos (\varphi+\omega)
$$

Terms number 4 combined give
$+\cos \varphi \cos (\alpha-\varphi)[\cos \alpha \cos (\varphi+\omega+\alpha)+\sin \alpha \sin (\varphi+\omega+\alpha)]$, which becomes

$$
+\cos \varphi \cos (\alpha-\varphi) \cos (\varphi+\omega)
$$

and hence, after dividing by $\cos (\varphi+\omega)$, the equation above reduces to
$\cos (\alpha-\varphi) \cos \varphi-\cos (\varphi+\omega+\alpha) \cos (\varphi+\omega)-\sin (\omega+\alpha) \sin \omega=0,(31)$ and this equation is fulfilled for

$$
\begin{equation*}
\omega=90^{\circ}-\varphi \tag{32}
\end{equation*}
$$

In order to find that value of $\delta$ which satisfies all conditions of equilibrium, substitute the above value of $\omega$ in the first expression for $\tan \delta$ and obtain $\frac{0}{0}$. If, according to
the method for discussing indeterminate fractions, the first differentials of the numerator and denominator and their ratio are found, and $\omega$ made equal to $90^{\circ}-\phi$, the value of $\tan \delta$ will be found.

The differential of the numerator is
$d[-\cos \alpha \cos (\varphi+\omega) \cos (\varphi+\omega+\alpha)-\cos (\alpha-\varphi) \sin \omega \sin (\varphi+\omega+\alpha)]$, which equals

$$
\left\{\begin{array}{l}
+\cos \alpha \cos (\varphi+\omega+\alpha) \sin (\phi+\omega) \\
+\cos \alpha \cos (\phi+\omega) \sin (\phi+\omega+\alpha) \\
-\cos (\alpha-\phi) \sin (\phi+\omega+\alpha) \cos \omega \\
-\cos (\alpha-\varphi) \sin \omega \cos (\varphi+\omega+\alpha)
\end{array}\right\} d \omega .
$$

Substituting for $\omega, 90^{\circ}-\varphi$, this becomes

$$
\left\{\begin{array}{l}
+\cos \alpha \cos \left(\phi+90^{\circ}-\phi+\alpha\right) \sin \left(\phi+90^{\circ}-\phi\right) \\
+\cos \alpha \cos \left(\phi+90^{\circ}-\phi\right) \sin \left(\phi+90^{\circ}-\phi+\alpha\right) \\
-\cos (\alpha-\phi) \sin \left(\phi+90^{\circ}-\phi+\alpha\right) \cos \left(90^{\circ}-\phi\right) \\
+\cos (\alpha-\phi) \sin \left(90^{\circ}-\varphi\right) \cos \left(\phi+90^{\circ}-\phi+\alpha\right)
\end{array}\right\} / l \omega .
$$

As the second term reduces to zero, this becomes $[\cos \alpha \sin \alpha-\cos (\alpha-\varphi) \cos \alpha \sin \varphi+\cos (\alpha-\varphi) \cos \varphi \sin \alpha] d \omega$, or
$\left[\frac{\sin 2 \alpha}{2}-\cos (\alpha-\varphi)(\cos \alpha \sin \varphi-\cos \varphi \sin \alpha)\right] d \omega$, or

$$
\begin{aligned}
& {\left[\frac{\sin 2 \alpha}{2}-\cos (\alpha-\varphi) \sin (\varphi-\alpha)\right] d \omega } \\
= & {\left[\frac{\sin 2 \alpha}{2}+\frac{\sin 2(\phi-\alpha)}{2}\right] d \omega, }
\end{aligned}
$$

or

$$
\left[\frac{2 \sin \frac{1}{2}(2 \varphi-2 \alpha+2 \alpha) \cos \frac{1}{2}(2 \varphi-2 \alpha-2 \alpha)}{2}\right] d \omega
$$

which equals $\quad \sin \varphi \cos (\varphi-2 \alpha) d \omega$.
The differential of the denominator is

$$
\left\{\begin{array}{l}
+\sin \alpha \cos (\varphi+\omega+\alpha) \sin (\varphi+\omega) \\
+\sin \alpha \cos (\phi+\omega) \sin (\phi+\omega+\alpha) \\
+\cos (\alpha-\phi) \cos (\varphi+\omega+\alpha) \cos \omega \\
+\cos (\alpha-\varphi) \sin \omega \sin (\phi+\omega+\alpha)
\end{array}\right\} d \omega
$$

Substituting $90^{\circ}-\Phi$ for $\omega$, and this becomes
$[\sin \alpha \sin \alpha+\cos (\alpha-\varphi) \sin \alpha \sin \varphi+\cos (\alpha-\varphi) \cos \varphi \cos \alpha] d \omega$, or
$\left[\sin ^{2} \alpha+\cos (\alpha-\varphi)(\sin \varphi \sin \alpha+\cos \varphi \cos \alpha)\right] d \omega$,
or

$$
\begin{aligned}
{\left[1-\cos ^{2} \alpha\right.} & +\cos (\alpha-\phi) \cos (\alpha-\varphi)] d \omega \\
& =\left[1-\frac{\cos 2 \alpha}{2}-\frac{1}{2}+\frac{\cos 2(\alpha-\phi)}{2}+\frac{1}{2}\right] d \omega
\end{aligned}
$$

or

$$
[1-\sin \varphi \sin (\varphi-2 \alpha)] d \omega
$$

therefore

$$
\begin{equation*}
\tan \delta=\frac{\sin \varphi \cos (\varphi-2 \alpha)}{1-\sin \varphi \sin (\varphi-2 \alpha)} \tag{33}
\end{equation*}
$$

To find an expression for the $\sin \delta$, clear equation (33)
of fractions and deduce $\tan \delta-\tan \delta \sin \varphi \sin (\varphi-2 \alpha)$ $=\sin \varphi \cos (\varphi-2 \alpha)$. Multiplying by $\cos \delta$, $\sin \delta-\sin \delta \sin \varphi \sin (\varphi-2 \alpha)=\sin \varphi \cos (\varphi-2 \alpha) \cos \delta$, or
$\sin \delta=\sin \varphi[\sin \delta \sin (\varphi-2 \alpha)+\cos (\varphi-2 \alpha) \cos \delta] ;$
therefore

$$
\begin{equation*}
\sin \delta=\sin \varphi \cos (2 \alpha-\varphi+\delta) \tag{34}
\end{equation*}
$$

from which the results of III. can be deduced.
If the earth-surface is parallel to the surface of repose, or makes the angle $\varphi$ with the horizontal, then, under the assumption of a plane surface of rupture, $\delta=\varphi$ only when the wall is vertical (make $\alpha=0$ in equation (33), then $\tan \delta=\tan \varphi ; . \therefore \delta=\phi)$, and $\delta=0$ only when the angle of the wall with the vertical $\alpha=45^{\circ}+\frac{\varphi}{2}$.

As it is often more convenient in determining the direction of the earth-pressure to know the angle $(\alpha+\delta)$ of $E$ with the horizon, $\tan (\alpha+\delta)$ may be expressed in terms of $\tan \alpha$ and $\tan \delta$, remembering that

$$
\cos \alpha-\sin \varphi \sin (\varphi-\alpha)=\cos \varphi \cos (\varphi-\alpha)
$$

and hence

$$
\begin{equation*}
\tan (\alpha+\delta)=\frac{\sin \alpha+\sin \varphi \cos (\varphi-\alpha)}{\cos \varphi \cos (\varphi-\alpha)} \tag{34a}
\end{equation*}
$$

With reference to a limited portion of wall which does
not reach as far as the surface, and with reference to loaded earth-surface, the same remarks hold good as in III.

Instead of formulæ (20) and (33) or (34), the following construction may be used:

Draw through $A$, Fig. 7, a parallel to the earth-surface,


Fig. 7.
and with $A C$ as a radius describe the circle $A D G$. Draw $D F$ horizontal and $G H$ parallel to $A B$, and then the straight line lIFJ. Then the direction of the earth-pressure is $G J$; and if $A K$ is made perpendicular to $A B$ and equal to $H F, E=\gamma \triangle A B K$, and the triangle gives the distribution of the pressure. The point of application is found by drawing through the centre of gravity of the triangle a perpendicular to $A B$.

The proof of this construction is as follows :
Conceive $H D$ drawn, and its intersection with $G J$ to be at $L$. Then from the notation of Fig. 3, where $\varepsilon=\varphi$,

$$
F D=A D \cos \varphi, \quad H D=2 A D \cos (\varphi-\alpha)
$$

Since, no:s, $<J L D=<J H D+\varphi-\alpha$, by expressing $\tan J L D$ by tan of $J H D$ and $\varphi-\alpha$, after reducing,

$$
\tan J L D=\frac{\cos \varphi \sin (2 \alpha-\phi)+\sin 2(\varphi-\alpha)}{1+\cos 2(\varphi-\alpha)-\cos \varphi \cos (2 \alpha-\phi)}
$$

or

Since $H D$ is perpendicular to $A B$, the earth-pressure has the direction GJ. Further,

$$
H F=\frac{F D \sin \alpha}{\sin (\alpha+\delta-\varphi)}=\frac{\sin \alpha \cos \varphi}{\sin (\alpha+\delta-\varphi)} A D
$$

$A D=\frac{l \cos (\phi-\alpha)}{\cos \varphi}$, or, with reference to the value of $F D$. $\triangle A B K=\frac{\cos (\phi-\alpha) \sin \alpha}{\sin (\alpha+\delta-\phi)} \overline{2}$, and since from equation (34) $\sin (\alpha+\delta-\varphi) \cos (\phi-\alpha)=\sin \alpha \cos (\alpha+\delta)$,

$$
\Delta A B K=\frac{\cos ^{2}(\varphi-\alpha)}{\cos (\alpha+\delta)} \frac{l^{2}}{2}=\frac{E}{\gamma}
$$

## RECAPITULATION OF FORMULA.

Inclined earth-surface, plane:

$$
\begin{equation*}
n=\sqrt{\frac{\sin (\psi+\delta) \sin (\psi-\varepsilon)}{\cos (\alpha+\delta) \cos (\alpha-\varepsilon)}} . \tag{18}
\end{equation*}
$$

The tan $\delta$ deduced from formulæ (22b) and (16乙):

$$
\tan \delta=\frac{\sin (2 \alpha-\varepsilon)-K \sin 2(\alpha-\varepsilon)}{K-\cos (2 \alpha-\varepsilon)+K \cos 2(\alpha-\varepsilon)},
$$

in which

$$
\begin{align*}
K & =\frac{\cos \varepsilon-\sqrt{\cos ^{2} \varepsilon-\cos ^{2} \varphi}}{\cos ^{2} \varphi} \\
E & =\left[\frac{\cos (\phi-\alpha)}{(n+1)} \frac{\cos \alpha}{\cos }\right]^{2} \frac{\lambda^{2} \gamma}{2 \cos (\alpha+\delta)} \tag{19}
\end{align*}
$$

Earth-surface parallel to natural slope :

$$
\begin{align*}
& \varepsilon=\phi ; \\
& E=\left[\frac{\cos (\varphi-\alpha)}{\cos \alpha}\right]^{2} \frac{l^{2} \gamma}{2 \cos (\alpha+\delta)} ;  \tag{20}\\
& \omega=90^{\circ}-\varphi ; \cdot \ldots \cdot .  \tag{32}\\
& \tan (\alpha+\delta)=\frac{\sin \alpha+\sin \varphi \cos (\varphi-\alpha)}{\cos \varphi \cos (\varphi-\alpha)} ; . \\
& \tan \delta=\frac{\sin \varphi \cos (\varphi-2 \alpha)}{1-\sin \varphi \sin (\varphi-2 \alpha)} . \tag{33}
\end{align*}
$$

Horizontal earth-surface:

$$
\begin{align*}
\omega & =45^{\circ}-\frac{\varphi}{2} ; \ldots \ldots  \tag{26}\\
\tan \delta & =\frac{\sin \varphi \sin 2 \alpha}{1-\sin \varphi \cos 2 \alpha} ; \ldots  \tag{27}\\
\tan (\alpha+\delta) & =\frac{\tan \alpha}{\tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right)} ; \ldots  \tag{28}\\
E & =\tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right) \frac{l^{2} \gamma}{2 \cos (\alpha+\delta)} ;  \tag{29}\\
E & =\frac{\tan \alpha}{\sin (\alpha+\delta)} \cdot \frac{l^{2} \gamma}{2} . \ldots .
\end{align*}
$$

If $\alpha=0$, then $\delta=0$, and

$$
\begin{equation*}
E=\tan ^{2}\left(45^{\circ}-\frac{\varphi}{2}\right) \frac{h^{2} \gamma}{2} . \tag{29d}
\end{equation*}
$$

If $\alpha=\left(45^{\circ}-\frac{\varphi}{2}\right)=\omega$, then $\delta=\varphi$, and

$$
E=\frac{\tan \left(45^{\circ}-\frac{\varphi}{2}\right) \quad h^{2} \gamma}{\sin \left(45^{\circ}+\frac{\varphi}{2}\right)} \quad \stackrel{z}{2} .
$$

If the surface is loaded, substitute $H^{2}+h^{\prime 2}$ for $h^{2}$, or consider $h$ to be the height of the earth increased by the height of an amount of earth weighing as much as the applied load.

## NOMENCLATURE.

Height of wall ..... $H$
Thickness at base ..... b
Thickness at top ..... $b^{\prime}$
Batter in inches per foot of $H$ on front face ..... $d$
Weight per cubic foot ..... W
Total weight of wall ..... G
Angle of repose of earth ..... $\varphi$
Angle made by surface of rupture with vertical ..... $\omega$
Weight of cubic foot of earth. ..... $\gamma$
Total thrust of earth against wall ..... $E$
Angle made with the horizontal by the surface of the earth ..... $\varepsilon$
Angle made by rear face of wall with the ver- tical. ..... $\alpha$
Angle made with normal by $E$. ..... $\delta$
Dist. of point where the resultant pressure cuts the base from the front edge of the wall.. $q$
The resultant pressure due to $E$ and $G \ldots \ldots R$

## NOTE.

For the trauslation of Prof. Weyrauch's paper the writer is indebted to the labor of Prof. A. J. Du Bois, of the Sheffield Scientific School, Yale College, who had copies printed by the electric-pen process. However, only the leading equations of Prof. Weyrauch were given ; hence a great deal of labor has been devoted to expanding, verifying, and filling in the intermediate steps of the work, and this nucleus of the mathematical part alone has grown to about double the original quantity. М. А. Н.

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| Belidor, | Levi, | Rebhann, |
| Blaveau, | de Köszegh Martony, Rondelet, |  |
| Bullet, | Maschek, | Saint-Guilhem, |
| Considère, | Mayniel, | Saint-Venant, |
| Coulomb, | Mohr, | Sallonnier, |
| Couplet, | Montlong, | Scheffler, |
| Culmann, | Moseley, | Trincaux, |
| Français, | Navier, | Vauban, |
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## DIAGRAM I.



## TABLES.

Table $I$ contains the crushing-strengths and the average weights of stone likely to be used in the construction of retaining-walls and foundations; also the average weights of different earths.

Table II contains the coefficients of friction, limiting angles of friction, and the reciprocals of the coefficients of friction for various substances.

Tables III, IV, and $V$ contain the values of the coefficients [see equation $\left(1^{\prime}\right)$ ] $(B),(C),(D)$ and $(E)$, where

$$
(B)=\frac{\cos (\epsilon-\alpha)}{\cos ^{2} \alpha \cos \epsilon}, \quad(C)=\sin ^{2} \alpha, \quad(D)=\left\{\frac{\cos (\epsilon-\alpha)}{\cos \epsilon}\right\}^{2}
$$

and $\quad(E)=2 \sin \alpha \sin \epsilon \frac{\cos (\epsilon-\alpha)}{\cos \epsilon}$.
The tables were computed with a Thacher calculating instrument and checked by means of diagrams. It is believed that they are correct to the second place of decimals; an error in the third place of decimals does not affect the results for practical purposes.

Table VI contains the natural sines, cosines and tangents.

## TABLE I.

VALUES OF $W$.


VALUES OF $\gamma$.


TABLE II.

* ANGLES AND COEFFICIENT'S OF FRICTION.

|  | $\boldsymbol{t a n} \phi$. | $\phi$ | $\frac{1}{\tan \phi}$ |
| :---: | :---: | :---: | :---: |
| Dry masonry and brickwork | 0.6 to 0.7 | $31^{\circ}$ to $35^{\circ}$ | 1.67 to 1.43 |
| Masonry and brickwork with damp mortar. ...... | 0.74 | $36 \frac{1}{2}^{\circ}$ | 1.35 |
| Timber on stone. . | about 0.4 | $22^{\text {c }}$ | 2.5 |
| Iron on stone | 0.7 to 0.3 | $35^{\circ}$ to $162^{\circ}$ | 1.43 to 3.33 |
| Timber on timber. | 0.5 " 0.2 | $26 \frac{1}{2}^{\circ}$ " $11 \frac{1}{3}^{\circ}$ | 2 " 5 |
| Timber on metals. | 0.6 " 0.2 | $31^{\text {² }}$ " $11 \frac{1}{3}{ }^{\circ}$ | 1.67 " 5 |
| Metals on metals | 0.25 " 0.15 | $14^{\circ}$ " $8 \frac{1}{2}^{\circ}$ | 4 " 6.67 |
| Masonry on dry clay...... ${ }_{\text {" }}$ | $\begin{aligned} & 0.51 \\ & 0.33 \end{aligned}$ | $17^{18}{ }^{10^{\circ}}$ | 1.96 3. |
| Earth on earth . . | 0.25 to 1.0 | $14^{\circ}$ to $45^{\circ}$ | 4 to 1 |
| Earth on earth, dry sand, clay, and mixed earth.... | 0.38 " 0.75 | $21^{\circ}$ " $37^{\circ}$ | 2.63 " 1.33 |
| Earth on earth, damp clay . | 1.0 | $45^{\circ}$ | 1 |
| Earth on earth, wet clay. | 0.31 | $17^{\circ}$ | 3.23 |
| Earth on earth, shingle and gravei. | 0.81 | $39^{\circ}$ to $48^{\circ}$ | 1.23 to 0.9 |

* From Rankine's Applied Mechanics.

TABLE III.

| $\epsilon$ | $\alpha=5^{\circ}$ | $a=6{ }^{\circ}$ | $\alpha=\imath^{\circ}$ | $a=8^{\circ}$ | $a=y^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (B) | (B) | (B) | (B) | (B) |
| 0 | 1.004 | 1.005 | 1.007 | 1.010 | 1.012 |
| 5 | 1.012 | 1.015 | 1.018 | 1.022 | 1.026 |
| 10 | 1.019 | 1.024 | 1.029 | 1.035 | 1.040 |
| 15 | 1.027 | 1.034 | 1.041 | 1.048 | 1.055 |
| 20 | 1.036 | 1.044 | 1.052 | 1.062 | 1.071 |
| 25 | 1.045 | 1.055 | 1.065 | 1.076 | 1.088 |
| 30 | 1.055 | 1.066 | 1.079 | 1.09\% | 1.105 |
| 35 | 1.065 | 1.079 | 1.094 | 1.109 | 1.124 |
| 40 | 1.078 | 1.094 | 1.111 | 1.129 | 1.147 |
| 45 | 1.093 | 1111 | 1.131 | 1.152 | 1.173 |
|  | (C) | (C) | ( ${ }^{\prime}$ ) | (C) | (C) |
|  | 0.008 | 0.011 | 0.015 | 0.019 | 0.024 |

TABLE IV.

| $\epsilon$ | $a=5^{\circ}$ | $a=6{ }^{\circ}$ | $a=\hat{r}^{\circ}$ | $a=8^{\circ}$ | $a=9{ }^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (D) | (D) | (D) | (D) | (D) |
| 0 | 0.992 | 0.989 | 0.985 | 0.981 | 0.976 |
| 5 | 1.008 | 1.008 | 1.006 | 1.005 | 1.003 |
| 10 | 1.023 | 1.026 | 1.028 | 1.030 | 1.031 |
| 15 | 1.040 | 1.146 | 1.051 | 1.056 | 1.060 |
| 20 | $1.05 \%$ | 1.066 | 1.075 | 1.084 | 1.092 |
| 25 | 1.075 | 1.089 | 1.102 | 1.114 | 1.125 |
| 30 | 1.096 | 1.113 | 1.130 | 1.147 | 1.163 |
| 35 | 1.118 | 1.140 | 1.164 | 1.183 | 1.204 |
| 40 | 1.144 | 1.172 | 1.199 | 1.226 | 1.253 |
| 45 | 1.174 | 1.208 | 1.242 | 1.276 | 1.309 |

TABLE V.

| $\epsilon$ | $a=5^{\circ}$ | $\alpha=6^{\circ}$ | $a=\%^{\circ}$ | $a=8^{\circ}$ | $\alpha=9{ }^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | ( $E$ ) | (E) | (E) | (E) | ( $E$ ) |
| 0 | 0 | 0 | 0 | 0 | 0 |
| 5 | 0.015 | 0.018 | 0.021 | 0.024 | 0.027 |
| 10 | 0.031 | 0.037 | 0.043 | 0.049 | 0.055 |
| 15 | 0.046 | 0.055 | 0.065 | 0.074 | 0.083 |
| 20 | 0.061 | 0.074 | 0.086 | 0.099 | $0.11{ }^{\text {² }}$ |
| 25 | 0.076 | 0.092 | 0.108 | 0.124 | 0.140 |
| 30 | 0.091 | 0.110 | 0.130 | 0.149 | 0.169 |
| 35 | 0.106 | 0.128 | 0.151 | 0.174 | 0.197 |
| 40 | 0.120 | 0.145 | 0.172 | 0.198 | 0.225 |
| 45 | 0.134 | 0.162 | 0.192 | 0.222 | 0.253 |

TABLE III-Continued.

| $\epsilon$ | $a=10^{\circ}$ | $a=11^{\circ}$ | $\alpha=12^{\circ}$ | $a=13^{\circ}$ | $a=14^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (B) | (B) | (B) | (B) | (B) |
| 0 | 1.015 | 1.019 | 1.022 | 1.026 | 1.031 |
| 5 | 1.031 | 1.037 | 1.041 | 1.047 | 1.053 |
| 10 | 1.046 | 1.055 | 1.061 | 1.068 | 1.076 |
| 15 | 1.063 | 1.073 | 1.081 | 1.090 | 1.100 |
| 20 | 1.081 | 1.092 | 1.103 | 1.112 | 1.12.) |
| 25 | 1.099 | 1.11\% | 1.124 | 1.136 | 1.150 |
| 30 | 1.119 | 1.135 | 1.151 | 1.163 | 1.179 |
| 35 | 1.141 | 1.159 | 1.175 | 1.195 | 1.211 |
| 40 | 1.166 | 1.186 | 1.205 | 1.225 | 1.245 |
| 45 | 1.195 | 1.218 | 1.240 | 1.263 | 1.288 |
|  | (C) | (C) | (C) | (C) | (C) |
|  | 0.030 | 0.036 | 0.043 | 0.0 .51 | 0.029 |

TABLE IV-Continued.

| $e$ | $a=10^{\circ}$ | $\alpha=11^{\circ}$ | $a=12^{\circ}$ | $a=13{ }^{\circ}$ | $a=14^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (D) | (D) | (D) | (D) | (D) |
| 0 | 0.970 | 0.964 | 0.957 | 0.950 | 0.943 |
| 5 | 1.000 | 0.997 | 0.993 | 0.988 | 0.983 |
| 10 | 1.031 | 1.031 | 1.030 | 1.028 | 1.026 |
| 15 | 1.064 | 1.067 | 1.069 | 1.061 | 1.072 |
| 20 | 1.099 | 1.105 | 1.110 | 1.116 | 1.121 |
| 25 | 1.136 | 1.147 | 1.156 | 1.165 | 1.173 |
| 30 | 1.178 | 1.194 | 1.204 | 1.220 | 1.232 |
| 35 | 1.224 | 1.244 | 1.262 | 1.281 | 1.300 |
| 40 | 1.291 | 1.304 | 1.328 | 1.353 | .1.377 |
| 45 | 1.342 | 1.375 | 1.407 | 1.438 | 1.469 |

TABLE V-Continued.

| ¢ | $a=10^{\circ}$ | $a=11^{\circ}$ | $a=1 \geqslant{ }^{\circ}$ | $a=133^{\circ}$ | $a=14{ }^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (E) | ( $L^{\prime}$ ) | (E) | (E) | (E) |
| 0 | 0 | 0 | 0 | 0 | 0 |
| 5 | 0.030 | 0.032 | 0.036 | 0.039 | 0.042 |
| 10 | 0061 | 0.067 | 0.073 | 0.079 | 0.085 |
| 15 | 0.093 | 0.102 | 0.111 | 0.119 | 0.130 |
| 20 | 0.124 | 0.137 | 0.150 | 0.163 | 0.175 |
| 25 | 0.156 | 0.113 | 0.189 | 0.205 | 0.221 |
| 30 | 0.188 | 0.208 | 0.216 | 0.248 | 0.269 |
| 35 | 0.220 | 0.244 | ${ }^{0} 268$ | 0.292 | 0.316 |
| 40 | 0.252 | 0.280 | 0.308 | 0.336 | 0.365 |
| 45 | 0.284 | 0.316 | 0.349 | 0.382 | 0.415 |

TABLE III-Continued.

| - | $a=15^{\circ}$ | $a=16^{\circ}$ | $a=1 \%^{\circ}$ | $a=18{ }^{\circ}$ | $a=20^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (B) | (B) | (B) | (B) | (B) |
| 0 | 1.1035 | 1.0411 | 1.048 | 1.051 | 1.062 |
| 5 | 1.059 | 1.066 | 1.076 | 1.081 | 1.098 |
| 10 | 1.084 | 1.093 | 1.104 | 1.112 | 1.132 |
| 15 | 1.110 | 1.120 | 1.134 | 1.138 | 1.168 |
| 20 | 1.135 | 1.149 | 1.165 | 1.177 | 1.218 |
| 25 | 1.165 | 1.179 | 1.197 | 1.211 | 1.245 |
| 30 | 1.195 | 1.212 | 1.233 | 1.248 | 1.288 |
| 35 | 1.229 | $1 . \because 49$ | 1.272 | 1.291 | 1.339 |
| 40 | 1.268 | 1.2 ! 1 | 1.317 | 1.340 | 1.389 |
| 45 | 1.313 | 1.338 | 1.369 | 1.393 | 1.451 |
|  | (C) | (C) | (C) | (C) | (C) |
|  | 0.06i\% | 0.0 .6 | 0086 | 0.095 | 0118 |

TABLE IV-Continued.

| e | $a=15^{\circ}$ | $a=16^{\circ}$ | $a=17^{\circ}$ | $a=18^{\circ}$ | $a=20^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | ( ${ }^{\text {d }}$ | (D) | ( $\overline{\text { J }}$ ) | (D) | (D) |
| 0 | 0933 | 0.924 | 0.915 | 0.905 | 0.883 |
| 5 | 0.977 | 0.971 | 0.964 | 0957 | 0.940 |
| 10 | 1.023 | 1.018 | 1.016 | 1.011 | 1.000 |
| 15 | 1.072 | 1.073 | 1.071 | 1069 | 1.068 |
| 20 | 1.124 | 1.127 | 1.129 | 1.131 | 1.132 |
| 25 | 1.181 | 1.188 | 1.194 | 1.200 | 1.208 |
| 30 | 1.244 | 1.256 | 1.266 | 1.276 | 1.293 |
| 35 | 1.316 | 1.332 | 1.348 | 1.363 | 1.390 |
| 40 | 1.100 | 1.422 | 1.444 | 1.465 | 1.505 |
| 45 | 1.500 | 1.530 | 1.559 | 1.588 | 1.643 |

TABLE V-Continued.

| $\epsilon$ | $a=15{ }^{\circ}$ | $a=16{ }^{\circ}$ | $a=1 \%^{\circ}$ | $a=18^{\circ}$ | $a=20^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | (E) | (E) | (E) | (E) | (E) |
| 0 | , | 0 | 0 | 0 | 0 |
| 5 | $0.04 \%$ | 0.047 | 0.0.0 | 0.053 | 0.058 |
| 10 | 0.091 | 0097 | 0.102 | 0.108 | 0.119 |
| 15 | 0.139 | 0.148 | 0.157 | 0.165 | 0.183 |
| 20 | 0.188 | 0.200 | 0.213 | 0.225 | 0.249 |
| 25 | 0.2:38 | 0.254 | 0.270 | 0.177 | 0.318 |
| 30 | 0.289 | 0.319 | $0.3 \cdot 9$ | 0.349 | 0.389 |
| 35 | 0341 | 0.36\% | 0.390 | 0.414 | 0.463 |
| 40 | 0.394 | 0.42:3 | 0.452 | 0.481 | 0.539 |
| 45 | 0.448 | 0.48\% | 0.516 | 0.551 | 0.620 |

## Table VI.

## NATURAL SINES, COSINES, TANGENTS AND COTANGENTS.



|  |  |  |  |  |  |  |  |  | $9^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine | Co | Sine | Cosin | Sine | Cosin | Sine | Cosin | - | Cosin |  |
| 0 | . 08 | . 99 | . 10453 |  | . 12 |  | . 1 |  | . 15643 | 69 | 60 |
| 1 |  | . 996 |  |  |  |  |  |  | -15672 |  | 59 |
| $\stackrel{2}{3}$ | . 088774. | . 996 | . 1 |  | . 1 |  |  |  | ${ }^{15701}$ |  | 58 |
| 4 | . $08881{ }^{\text {. }}$ |  | . 1 | . 99 | . 12 |  |  |  | ${ }^{15730}$ | . 988751 | 57 |
| 5 |  |  | . 105 | . 99 |  |  | . 14061 |  | . 15787 |  |  |
| 6 | . 08889 |  | 10 | . 994 | 12 |  | . 14090 | . 990 | . 15816 | . 98741 |  |
| 7 | 8918 | . 996 | . 106 | . 99431 | . 12389 |  | . 14119 | . 98 | . 15845 | . 98737 | 53 |
|  |  |  |  | . 994 |  |  | 14148 | . 989 | . 15 | . 98 | 52 |
| 9 |  |  |  |  |  |  |  |  |  |  | 51 |
| 10 |  |  |  |  |  |  |  |  |  |  | 50 |
| 11 | . 09 | . 99 |  | . | . 12504 | . 99 | . 14234 | . 98982 | . 15959 |  | 49 |
| 12 | . 09 |  | . 1 |  |  |  |  |  |  |  | 48 |
|  | . 0909 | . 99 |  | . 9941 | . 125 |  | . 14292 | . 989 | . 16017 |  | 47 |
| 14 | . 09121 | . 99 | . 108 | . 994 | . 12591 | . 9 | . 14320 |  | - 16046 |  | 46 |
| 15 | . 09 | . 995 | -108 | . 99 | - 12620 |  | . 14349 |  | . $160 \% 4$ |  | 45 |
| 16 | . 091 | . 995 | . 10916 |  | . 12649 |  |  | . 98 | . 16103 |  | 44 |
| 18 |  |  |  |  |  |  |  |  |  |  | 43 |
| 19 | .092 | $995 \%$ | . 11002 |  | . 12 | . 99 |  | . 989 | . 1618 |  | 41 |
| 20 | . 092 |  |  |  |  | . 991 |  | . 989 | 162 |  | 40 |
| 21 | . 093 |  | . 11060 | . 99 | . 1 | . 99 | . 14522 | . 98 | . 16246 | . 98671 | 39 |
| 22 | . 093 |  |  |  |  |  | . 1 |  | . 16 |  | 38 |
| 23 | . 0938 |  | , |  |  |  | . 14 |  |  |  | 37 |
| 24 | . 0941 | . 995 | . 111 | . 99 | . 128 | . 99 | . 14 |  |  |  | 36 |
|  | . 094 |  | .111 | . 993 | . 129 |  | . 146 |  | . 16361 |  | 85 |
| 26 | . 094 |  | . 11 | . 993 | . 1293 | . 99 | . 146 |  | . 16390 |  | 84 |
| 27 | . 09 |  | . 112 |  | . 129 |  | . 140 |  | . 16419 |  | 33 |
| 28 |  |  | . 11263 | . 993 | . 129 |  |  |  |  |  | 32 |
| 20 | . 09 |  | . 11291 |  | . 130 |  |  |  | . 16 |  | 81 |
| 30 | . 095 | . 995 | 11320 |  |  |  |  |  |  |  | 30 |
| 31 |  |  | 113 |  | . 130 |  | . 14 |  | . 16533 |  | 29 |
| 32 | . 09 |  |  |  |  |  |  |  |  |  | 28 |
| 33 | .0967 | . 995 | 11 |  |  |  | . 148 |  | - 165 |  | 27 |
| 34 |  |  | . 11 | - 993 |  |  | . 148 |  |  |  | 2 |
|  | . 097 |  |  |  |  |  |  |  | . 16 |  | 25 |
| 36 | . 097 | . 995 | . 114 | . 993 | . 132 | . 9912 | . 149 | . 988 | . 16677 |  | 2 |
| 37 |  |  |  |  |  |  |  |  | . 167 |  |  |
| 38 | . 08 | . 93 | 115 | . 993 |  |  | . 150 |  | . 167 |  |  |
| 49 |  |  |  |  |  |  |  |  |  |  | 20 |
|  |  |  |  |  |  |  |  |  |  |  |  |
| 42 | . 09 |  | . 116 | 93 | . 133 |  |  |  | -168 |  |  |
| 43 | . 09961 | . 9950 | . 110 | . 99314 | . 13 |  | . 151 | . | 168 |  | 17 |
| 44 | . 099 |  | . 11 | . 99 | . 1 |  | . 151 |  | . 16 |  | 16 |
|  | . 10 | . 994 |  |  |  |  |  |  |  |  |  |
| 46 | - 10 | . 99494 | . 11 | . 9933 | . 135 |  | . 15 |  | . 16 |  | 14 |
|  | . 1007 | . 99491 |  | -933 |  |  |  |  | -170 |  | 13 |
| 48 | . 10106 | . 9948 | . 118 | . 99297 | . 13572 | . 990 |  |  | . 170 |  | 12 |
| 4 | . 10135 | . 99485 | . 118 | . 99293 | . 1360 | . 990 | . 15327 | . 98818 | . 17050 | . 885 | 11 |
| 50 |  |  |  |  |  |  |  |  |  |  | 10 |
| 51 | . 101 | .99479 | . 119 |  | . 138 |  |  |  |  |  |  |
| 5 | . 1022 | . 99476 | . 119 | .992 | . 1368 | . 990 | . 154 | . 988 | . 17136 | . |  |
| 53 | . 10250 | -9\%10 | -120 | .992 | . 13110 | . 9903 | - | . 9880 | - | . |  |
| 54 | . 1023 | . 994 | . 120 | . 992 | . 13744 | . 990 | . 154 | . $98 \%$ | . 1719 | 98 |  |
| 55 | - 1030 | . 99467 | . 1204 | - |  | . 990 | - | . 987 | 1122 | . 985 |  |
| 56 | . 1033 | . 9946 | . 1207 | . 992 | . 138 | . 99 | . 155 | . 98 | 1725 | . 9850 |  |
| 57 | 1 | . 9946 | . 121 | .992 |  | . 990 | . 155 | . 987 | 172 | . 984 |  |
|  | . 1039 | . 9945 | . 1212 | . 99 | . 13 | . 990 | . 155 | . 98 | 17 |  |  |
| 59 | . 1042 |  |  | . 99 | . 13 | . 990 | . 156 | . 987 | . 1733 |  |  |
| 60 | . 10453 | . 99452 | . 1218 | . 99255 |  | . 9902 |  |  | . 173 | . 98481 | 0 |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |


|  |  |  | $11^{\circ}$ |  | $12^{\circ}$ |  | $13^{\circ}$ |  | 14* |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine | Co | Sine | Cosin | Sine | Cosi | Sine | Cosin | Sine | Cosin |  |
| 0 | 17 | . 98 | . 19081 | . 98 | 20 | . 97815 | . 22495 | . 97437 | . 24192 | . 97030 | 30 |
| 1 | . 17393 | . 9847 | 19109 | . 98157 |  | . 97809 | . 22523 | . 97430 | . 24220 | . 97023 | 59 |
| , | .17428 | . 98471 | . 19138 | . 98152 | . 20848 | .97803 | . 22552 | . 97424 | . 24249 | . 97015 | 58 |
| 3 | 17451 | . 98466 | :19167 | . 98146 | . 20877 | . 97797 | . 22580 | . 97417 | . 24277 | . 97008 | 57 |
| 4 | . 17479 | . 98461 | . 19195 | . 98140 | . 20905 | . 97791 | . 22608 | . 97411 | . 24305 | . 97001 | 56 |
| 5 | . 17508 | . 98455 | . 19224 | . 98135 | . 20933 | . 97784 | . 22637 | . 97404 | . 24333 | . 96994 | 55 |
| 6 | . 17537 | . 98450 | . 19252 | . 98129 | . 20962 | . 97778 | . 22665 | d77398 | . 24362 | . 96987 | 54 |
|  | . 17565 | . 98145 | . 19281 | . 98124 | . 20990 | . 97778 | . 22693 | . 97391 | . 24390 | . 96980 | 53 |
| 8 | . 17594 | . 98440 | . 19309 | . 98118 | . 21019 | . 97766 | . 22722 | . 97384 | . 24418 | . 96973 | 52 |
| 9 | . 17623 | . 98435 | . 19338 | . 98112 | . 21047 | . 97760 | . 22750 | . 97378 | . 24446 | . 96966 | 51 |
| 10 | . 17651 | . 98430 | 366 | . 98107 | 21076 | . 97754 | . 23778 | . 97371 | . 24474 | . 96959 | 50 |
| 11 | . 17680 | . 98425 | . 19395 | . 98101 | . 21104 | . 97748 | . 22807 | . 97365 | . 24503 | . 06952 | 49 |
| 12 | . 17708 | . 98420 | . 19423 | . 93096 | . 21132 | . 97742 | . 22835 | . 97358 | . 24531 | . 96945 | 48 |
| 13 | . 17737 | . 98414 | 19452 | . 98090 | . 21161 | . 97735 | . 22863 | . 97351 | . 24559 | . 96937 | 47 |
| 14 | . 17766 | . 98409 | . 19481 | . 98084 | . 21189 | . 97789 | . 22892 | . 97345 | . 24587 | . 96930 | 46 |
| 15 | . 17794 | . 98404 | . 19509 | . 98079 | . 21218 | . 97783 | . 229930 | . 97338 | . 24615 | . 96923 | 45 |
| 16 | . 17823 | . 98399 | 19538 | . 98073 | . 21246 | . 97717 | . 22948 | . 97331 | . 24644 | . 96916 | 44 |
| 17 | . 17852 | . 98394 | . 19566 | . 98067 | . 21275 | . 97711 | . 22977 | . 97325 | . 24672 | . 96909 | 43 |
| 18 | . 17880 | . 98389 | . 19595 | . 98061 | .21303 | . 97705 | . 23005 | . 97318 | . 24700 | . 96902 | 42 |
| 19 | . 17909 | . 08383 | . 19623 | . 98056 | . 21331 | . 97698 | . 23033 | . 97311 | . 24728 | . 96894 | 41 |
| 20 | . 17937 | . 98378 | 19652 | . 98050 | . 21360 | . 97692 | . 23062 | . 97304 | . 24756 | . 98887 | 40 |
| 21 | . 17966 | . 983 | . 19680 | . 98044 | . 21388 | .97686 | . 23090 | . 97298 | . 24784 | . 96880 | 39 |
| 22 | . 17995 | . 98368 | . 19709 | . 98039 | . 21417 | . 97680 | . 23118 | . 97291 | . 24813 | . 96873 | 38 |
| 23 | . 18023 | . 98362 | . 19737 | . 93033 | . 21445 | . 97673 | . 23146 | . 97284 | . 24841 | . 96866 | 37 |
| 24 | . 18052 | . 98357 | . 19766 | . 98027 | . 21474 | . 97667 | . 23175 | . $972 \%$ | . 24869 | . 96858 | 36 |
| 25 | . 18081 | . 98352 | . 19794 | . 98021 | . 21502 | . 97661 | . 23203 | . 97271 | . 24897 | . 96851 | 35 |
| 26 | . 18109 | . 98347 | . 19823 | . 98016 | . 21530 | . 97655 | . 23231 | . 97264 | . 24925 | . 96844 | 34 |
| 27 | . 18138 | . 98341 | . 19851 | . 98010 | . 21509 | . 97648 | . 23260 | . 97257 | . 24954 | . 96837 | 33 |
| 23 | . 18166 | . 98336 | . 19880 | . 98004 | . 21587 | . 97642 | . 23288 | . 97751 | . 24982 | . 96829 | 32 |
| 23 | . 18195 | . 98331 | . 19908 | . 97998 | . 21616 | . 97636 | . 23316 | . 97244 | . 25010 | . 96822 | 31 |
| 30 | . 18 | . 98325 | . 19937 | . 97992 | . 21644 | . 97630 | . 23345 | . 97237 | . 25038 | . 96815 | 30 |
| 31 | . 18 | . 98320 | . 19965 | . 97987 | . 21672 | . 97623 | . 23373 | . 97230 | . 25066 | . 96807 | 29 |
| 32 | . 18281 | . 98315 | . 19994 | . 97981 | . 21701 | . 97617 | . 23401 | . 97223 | . 25094 | . 96800 | 28 |
| 33 | . 18309 | . 98310 | . 20022 | . 97975 | . 21729 | . 97611 | . 23423 | . 97217 | . 25122 | . 96793 | 27 |
| 34 | . 18338 | . 98304 | . 20051 | . 97969 | . 2175 | . 97604 | . 23458 | . 97210 | . 25151 | . 96786 | 26 |
| 35 | . 18367 | . 98299 | - 20079 | . 97963 | . 21786 | . 37598 | . 23486 | . 97203 | . 25179 | . 96778 | 25 |
| 36 | . 18395 | . 98294 | - 20108 | . 97958 | . 21814 | . 97592 | . 23514 | . 97196 | . 25207 | . 96771 | 24 |
| 37 | . 18424 | . 98288 | $\therefore 20136$ | . 97952 | '. 21843 | . 97585 | . 23542 | . 97189 | . 25235 | . 96764 | 23 |
| 38 | . 18452 | . 98283 | :20165 | . 97946 | ?. 21871 | . 97579 | . 23571 | . 97182 | . 25263 | . 96756 | 22 |
| 39 | . 18481 | . 98277 | '. 20193 | . 97940 | $? 21899$ | . 97573 | .23599 | . 97176 | . 25291 | . 96749 | 21 |
| 40 | . 18509 | . 98272 | :20223 | . 97934 | . 21928 | . 97566 | . 23677 |  | . 25320 | . 96742 | 20 |
| 41 | . 18538 | . 98267 | -20250 | . 97928 | . 21956 | . 97560 | - 23656 | . 97162 | . 25348 | . 96734 | 19 |
| 42 | . 18567 | . 98261 | - 20279 | . 97922 | \% 21985 | . 97553 | :. 23684 | . 97155 | . 25376 | . 96727 | 18 |
| 43 | . 18595 | . 98556 | $\therefore 20307$ | : 97916 | -22013 | . 97547 | . 23712 | . 97148 | . 25404 | . 96719 | 17 |
| 44 | . 18624 | . 98250 | $\because 20336$ | . 97910 | -22041 | . 97541 | . 23740 | . 97141 | . 25432 | . 96712 | 16 |
| 45 | . 18652 | . 98245 | :20364 | . 97905 | . 22070 | : 97534 | . 23769 | . 97134 | . 25460 | . 96705 | 15 |
| 46 | . 18681 | . 98240 | $\because 20393$ | . 97899 | . 22098 | . 97528 | $\therefore 23797$ | . 97127 | . 25488 | . 96697 | 14 |
| 47 | . 18710 | . 98234 | :20421 | . 97893 | . 22126 | :97521 | . 23825 | . 97120 | . 25516 | . 96690 | 13 |
| 48 | . 18738 | . 98229 | $\bigcirc 20450$ | . 97887 | $\because 22155$ | :97515 | . 23853 | . 97113 | . 25545 | . 96682 | 12 |
| 49 | . 18767 | . 98223 | $\bigcirc 20478$ | . 97881 | $\because 22183$ | $\because 97508$ | ? 23882 | . 97106 | . 25573 | . 96675 | 11 |
| 50 | . 18795 | . 98818 | Y.20507 | :97875 | $\therefore 22212$ | 197502 | $\because 23910$ | . 97100 | . 25601 | . 96667 | 10 |
| 51 | . 18824 | . 98212 | . 20535 | . 97869 | \%.22240 | . 97496 | . 23938 | . 97093 | . 25629 | . 96660 | 9 |
| 52 | . 18852 | . 98207 | . 20563 | $\vdots 97863$ | $\because 22268$ | $\therefore 97489$ | $\therefore 23966$ | . 97086 | . 25657 | . 96653 | 8 |
| 53 | . 18881 | . 98201 | . 20592 | :97857 | -22297 | . 97483 | $\therefore 23995$ | . 97079 | . 25685 | . 96645 | 8 |
| 54 | . 18910 | . 98196 | .20620 | 97851 | -22325 | $\vdots 97476$ | . 24023 | . 97072 | . 25713 | . 96638 | 6 |
| 55 | . 18938 | . 98190 | 20549 | ${ }^{9} 97845$ | -22353 | . 97470 | . 24051 | . 97065 | . 25741 | . 96630 | 4 |
| 56 | . 18967 | . 98185 | -20677 | 97839 | \%22382 | $\therefore 97463$ | . 24079 | . 97058 | . 25769 | . 96623 | 4 |
| 58 | . 18995 | . 98179 | \% 20706 | $!97833$ | *22410 | $: 97457$ | . 24108 | . 97051 | . 25798 | . 96615 | \% |
| 58 | . 19024 | . 98174 | 1.20734 | ${ }_{7} 97827$ | R22438 | 97450 | . 24136 | . 97044 | . 25882 | . 96608 | 2 |
| 59 | . 19052 | . 98168 | \% 20763 | .97821 | . 22467 |  | 64 |  | 854 | 0 | 1 |
| 60 | . 19081 |  | 20\%91 | $\div 97815$ | $\underline{22495}$ | !9 | . |  |  |  | 0 |
| - | Cosin | Sine | osi | Sine | Cosin | Sine | Cos | Sine | Cosin | Sine |  |
|  |  |  |  |  |  |  |  |  |  |  |  |


|  | $15^{\circ}$ |  | 16 |  | $17{ }^{\circ}$ |  | $18^{\circ}$ |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine | Cosin | Sine | osin | Sine | Cosi | Sin | Cosin | ${ }^{\circ}$ |  |  |
| 0 | . 25882 | . 965 | . 27564 | . 96126 | . 292 | . 95 | . 30902 | - | 557 | 52 | $\overline{60}$ |
|  | 25910 | . 965 | . 27592 | . 961 | . 292 |  | . 30929 | . 95097 | . 32584 | 42 | 59 |
|  | . 25938 | . 906 | . 276 | . 96 | . 29293 | . 95 | . 30957 | . 95088 | . 32812 |  | 58 |
| 3 | . 259 |  |  |  |  |  | . 31098 | ${ }^{950} 950$ | . 322667 |  | 5 |
| $4$ | . 2602 | 965 | . 27 | . 960 | . 29376 | . 95 | . 31040 | 950 | . 32694 |  | 55 |
|  | . 26050 | . 965 | . 27731 | . 96078 | . 29404 | . 955 | . 310 | 950 | . 32722 | 94 | 5 |
|  | . 2607 | . 96540 | . 27 | -96070 | . 29 | . 955 | . 31095 | . 9504 | . 82749 | 94 | 3 |
|  | . 26107 | 96532 | . 27 | . 96062 | . 29460 | . 95556 | . 3112 | . 95 | ${ }^{.32777}$ | 94 | 52 |
| 9 | . 26135 | 9652 | . 278 | . 960 | . 29 |  | . 311178 |  |  |  | 50 |
| 10 |  |  |  |  |  |  |  |  |  |  |  |
| 11 | . 2611 | . 965 | . 27 | . 96 |  | $95$ | $.31206$ |  | $.32859$ |  | 43 |
| 12 | . 26 |  |  |  | $\|.29571\| .$ | $.95$ | . 31261 | 94988 | . 82914 |  | 47 |
| 14 |  | 96486 | . 279 | . 960 |  | . 95511 | . 31289 | . 94979 | . 82942 | 94 | 46 |
| 15 | . 2630 | . 96479 | . 279 | . 960 | . 296 | . 955 | . 31316 |  |  | 04 | 45 |
| 16 | . 26331 | . 96471 | . 28011 | . 95 |  | . 9544 | ${ }^{.31344}$ | .949 | . 323997 | 94399 94390 | 44 |
| 17 | . 263 |  |  |  | . 29773 |  | . 3131399 |  | . 33051 | ${ }_{94}^{943}$ | 42 |
| 19 | 26 |  | . 2809 | . 959 | . 297 | 954 | . 31427 | . 949 | . 33079 | 94370 |  |
| 20 | . 264 |  | . 28123 | . 959 | . 29793 | . 95 |  | . 949 | . 83106 | 94361 | 40 |
| 21 | . 26 |  | . 28 | . 95 | . 29 | . 95 | . 31482 | . 949 | . 83134 | . 94351 | 39 |
| 22 |  |  | . 28178 | . 959 | . 298 | 95 | . 31 |  | . 33161 |  | 38 |
| 23 | . 265 | 96 | . 28206 | . 959 | . 298 | . 95 | . 81 | . 94 | . 83189 | 9 | 37 |
| 24 | . 265 | 96 | . 28234 |  | . 2990 | . 954 | . 31 |  | . 33216 |  | 5 |
| 2 | . 2658 | 96 | . 28262 |  | . 299 | . 9541 | . 31 | . 94 | . 832 | . 94313 | 85 |
| 26 | . 266 |  | . 28290 | . 958 | . 29960 | . 954 | . 316 |  | 3 |  | 3 |
| 27 | . 266 |  | . 28318 | . 95 | . 2998 |  | . 316 |  | -83328 |  | 33 |
| 28 | . 266 |  | . 28346 |  | . 30015 | . 953 | . 31 | . 94 | . 83326 |  | 32 |
| $\stackrel{29}{30}$ | 2 |  | . 28374 | . 95 | . 30043 |  |  |  |  |  | 31 |
|  |  |  |  |  |  |  |  |  |  |  | 29 |
| 31 |  |  |  |  |  |  |  |  |  |  | 28 |
| 33 |  |  | . 2 |  |  |  |  |  | . 33 |  | 27 |
| 34 |  |  | . 28513 | . 95 |  |  | . 318 | . 947 | 334 | 94225 | 26 |
| 35 | . 2686 | 963 | . 28541 | . 958 | . 302 | . 95 | . 318 | . 94 | . 33 |  | 25 |
| 36 | . 26 |  | . 28569 |  |  | . 953 | . 31 |  | . 335 |  | 24 |
| 37 | . 2692 |  | . 285 | . 958 |  | . 953 | . 81 |  | . 3357 | . 94196 | 23 |
|  |  |  |  | . 9581 |  |  | . 819 |  | . 8360 |  | 22 |
| 39 | . 26 |  | . 28655 |  |  |  |  |  | . 836 | 9 | 21 |
| 40 | . 2700 |  | . 2868 |  |  |  |  |  |  |  | 20 |
| 41 | . 2 | . 96 | . 2 | . 957 |  |  | . 320 |  | . 33682 |  | 19 |
| 42 | . 2 |  |  | -951 |  |  | . 22 |  | . 33710 | 94 | 13 |
| 43 | . 2708 | 962 | . 28764 | . 957 | . 30 | . 952 | . 320 |  |  |  | 17 |
| 44 | . 2711 |  | . 287 | . 957 | . 30 | . 95 | . 82116 |  | . 83 | - | 16 |
| 45 | . 271 |  | .288 | . 957 |  | . 952 | . 3214 |  |  | 94118 | 15 |
| 46 | . 21 |  | . 28847 | . 957 | . 305 | . 95 | . 82171 |  | . 83819 | . | 14 |
| 47 |  |  | . 28875 | . 9577 | . 30 |  | . 32199 | . 94 |  |  | 1 |
| 48 | .272 |  | . 28903 | . 957 |  | . 952 | . 82 |  | . 83884 | . 9 | 12 |
| 49 | . 27256 | . 20 | . 28931 | . 95724 | . 80597 | . 95204 | . 822 | . 946 | - 83901 | 94 | 11 |
| 50 | . 27284 | . 962 | . 28959 | . 957 |  | . 9519 | . 32 | . 94 | 3929 | . 94 | 10 |
|  | . 27312 | . 96198 | . 2898 | . 95 | . 306 | . 95186 | . 3230 | . 946 | . |  | 9 |
| 52 | . 2734 | . 961 | . 2901 | . 956 | . 306 | . 951 | . 32337 | . 946 | . 83983 |  |  |
| 53 | . 27368 | . 9618 | . 2904 | . 956 | . 8070 | . 951 | . 82364 | .946 | . 84011 | d |  |
| 54 | . 27396 | . 9617 | . 2907 | . 95 | . 30 | . 951 | . 8239 | . 946 |  | 940 | 6 |
| 55 | . | . 961 | . 29093 | . 95 | . 30763 | . 95 | . 3241 | 94 | -84 | . 9 | 5 |
|  | . 2745 | . 861 | . 29126 |  | . 30791 | . 95 | . 3244 | 945 | . 844 | 94 | 4 |
|  | . 27480 | . 9615 | . 29154 | . 956 | . 3081 | 95 | . 82474 | 93 | . 84120 | - | 3 |
|  | . 27508 | . 9614 | . 2918 | . 95 | . 3084 | . 951 | . 32502 |  | . 84147 | . 83 | 2 |
| 59 60 |  |  |  |  |  |  | . 32529 |  |  |  | 1 |
|  |  | ne |  |  | Cosin |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |


|  | $20^{\circ}$ |  | $21^{\circ}$ |  | $22^{\circ}$ |  | $23^{\circ}$ |  | $24^{\circ}$ |  | , |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine | Cosin | Sine | Cosin | Sine | Cosin | Sine | Cosin | Sine | Cosin |  |
| 0 | . 34202 | . 93969 | . 358 | . 93358 | . 37461 | . 92718 | . 39073 | . 92050 | . 40674 | . 91355 | $\overline{60}$ |
| 1 | . 34229 | . 93959 | . 35864 | . 93348 | . 37488 | . 92707 | . 39100 | . 92039 | . 40700 | . 91343 | 59 |
| 2 | . 34257 | . 93949 | . 35891 | . 93337 | . 3 \%515 | . 92697 | . 39127 | . 92028 | . 40727 | . 91331 | 58 |
| 3 | . 34284 | . 93939 | . 35918 | . 93327 | .37542 | . 92686 | . 39153 | . 92016 | . 40753 | . 91319 | 57 |
| 4 | . 34311 | . 93929 | . 35945 | . 93316 | . 37569 | . 92675 | . 39180 | . 92005 | . 40780 | . 91307 | 56 |
| 5 | .34339 | . 93919 | . 35973 | . 93306 | . 37595 | . 92664 | . 39207 | . 91994 | . 40806 | . 91295 | 55 |
| 6 | . 34366 | . 93909 | . 36000 | . 93295 | . 37622 | . 92653 | . 39234 | . 91982 | . 40833 | . 91283 | 54 |
| 7 | . 34393 | . 93899 | . 36027 | . 93285 | . 37649 | . 92642 | . 39260 | . 91971 | . 40860 | . 91272 | 53 |
| 8 | . 34421 | . 93889 | . 36054 | . 93274 | . 37676 | . 92631 | . 39287 | . 91959 | . 40886 | . 91260 | 52 |
| 9 | . 34448 | . 93879 | . 36081 | . 93264 | . 37703 | . 92620 | . 39314 | . 91948 | . 40913 | . 91248 | 51 |
| 10 | . 34475 | . 93869 | . 36108 | . 93253 | . 37730 | . 92609 | . 39341 | . 91936 | . 40939 | . 91236 | 50 |
| 11 | . 34503 | . 93859 | . 36135 | . 93243 | . 37757 | . 92598 | . 39367 | . 91925 |  | . 91224 | 49 |
| 12 | . 34530 | . 93849 | . 36162 | . 93232 | . 37784 | . 92587 | . 39394 | . 91914 | . 40992 | . 91212 | 48 |
| 13 | . 34557 | . 93839 | . 36190 | . 93222 | . 37811 | . 92576 | . 39421 | . 91902 | . 41019 | . 91200 | 47 |
| 11 | . 34581 | . 93889 | . 36217 | . 93211 | . 37838 | . 92565 | . 39448 | . 91891 | . 41045 | . 91188 | 46 |
| 15 | . 34612 | . 93819 | . 36244 | . 93201 | . 37865 | . 92554 | . 39474 | . 91879 | . 41072 | . 91176 | 45 |
| 16 | . 34639 | . 93809 | . 36271 | . 93190 | . 37892 | . 92543 | . 39501 | . 91868 | . 41098 | . 21164 | 44 |
| 17 | . 34666 | . 93799 | . 36298 | . 93180 | . 37919 | . 92532 | . 39528 | . 91856 | . 41125 | . 91152 | 43 |
| 18 | . 34694 | . 93789 | . 36325 | . 93169 | . 37946 | . 92521 | . 39555 | . 91845 | . 41151 | . 91140 | 42 |
| 19 | . 34721 | . 93779 | . 36352 | . 93159 | . 37973 | . 92510 | . 39581 | . 91833 | . 41178 | . 91128 | 41 |
| 20 | . 34748 | . 93769 | . 36379 | . 93148 | . 37999 | . 92499 | . 39608 | . 91822 | . 41204 | . 91116 | 40 |
| 21 | . 34775 | . 93759 | . 36406 | 93137 | . 38026 | . 92488 | . 39635 | . 91810 | . 41231 | . 91104 | 39 |
| 22 | . 34803 | . 93748 | . 36434 | . 93127 | . 38053 | . 92477 | . 39661 | . 91799 | . 41257 | . 91092 | 38 |
| 23 | . 31830 | . 93738 | . 36461 | . 93116 | . 38080 | . 92466 | . 39688 | . 91787 | . 41284 | . 91080 | 37 |
| 24 | . 34857 | . 93728 | . 36488 | . 93106 | . 38107 | . 92455 | . 39715 | . 91775 | . 41310 | . 9106 | 36 |
| 25 | . 34884 | . 93718 | . 36515 | . 93095 | . 38134 | . 92444 | . 39741 | . 91764 | . 41337 | . 91056 | 35 |
| 26 | . 34912 | . 93708 | . 36542 | . 93084 | . 38161 | . 92432 | . 39768 | . 91752 | . 41363 | . 91044 | 34 |
| 27 | . 34939 | . 93698 | . 36569 | . 93074 | . 38188 | . 92421 | . 39795 | . 91741 | .41390 | . 91032 | 33 |
| 28 | . 34966 | . 93688 | . 36596 | . 93063 | . 38215 | . 92410 | . 39822 | . 91729 | . 41416 | . 91020 | 32 |
| 29 | . 34993 | . 93677 | . 36623 | . 93052 | .28241 | . 92399 | . 39848 | . 91718 | . 41443 | . 91008 | 31 |
| 30 | . 35021 | . 93667 | . 36650 | . 93042 | . 38268 | . 92388 | . 39875 | . 91706 | . 41469 | . 90996 | 30 |
| 31 | . 35 | . 93657 | . 36677 | . 93031 | . 38295 | . 92377 | . 39902 | . 91694 | . 41496 | . 90984 | 29 |
| 32 | . 35075 | . 93647 | . 36704 | . 93320 | . 38322 | . 92366 | . 30928 | . 91683 | . 41522 | . 90972 | 28 |
| 33 | . 35102 | . 93637 | . 36731 | . 93010 | . 38349 | . 92355 | . 39955 | . $916 \% 1$ | .41549 | . 90960 | 27 |
| 34 | . 35130 | . 93626 | . 36758 | . 92999 | . $383 \% 6$ | . 92343 | . 39982 | . 91660 | 41575 | . 90948 | 26 |
| 35 | . 35157 | . 93616 | . 36785 | . 92988 | . 38403 | . 92332 | . 40008 | . 91648 | . 41602 | . 90936 | 25 |
| 36 | . 35184 | . 93606 | . 36812 | . 92978 | . 38430 | . 92321 | . 40035 | . 91636 | . 41628 | . 90924 | 24 |
| 37 | . 35211 | . 93596 | . 36839 | . 92967 | . 38456 | . 92310 | . 40062 | . 91625 | . 41655 | . 90911 | 23 |
| 38 | . 35239 | . 93585 | . 36867 | . 92956 | . 38483 | . 92209 | . 40088 | . 91613 | . 41681 | . 90899 | 22 |
| 39 | . 35266 | . 93575 | . 36894 | . 92945 | . 38510 | . 92287 | . 40115 | . 91601 | . $4170{ }_{6}^{\prime \prime}$ | . 90887 | 21 |
| 40 | . 35293 | . 93565 | . 36921 | . 92935 | . 38537 | . 92276 | . 40141 | . 91590 | . 41734 | . 90875 | 20 |
| 41 | . 35320 | . 93555 | . 36948 | . 92924 | . 38564 | . 92265 | . 40168 | . 91578 | . 41760 | . 90863 | 13 |
| 42 | . 35347 | . .93544 | . 36975 | . 92913 | . 38501 | . 92254 | . 40195 | . 91566 | . 41787 | . 90851 | 18 |
| 43 | . 35375 | . 93534 | . 37002 | . 92902 | . 38617 | . 92243 | .40221 | . 91555 | . 41813 | . 90839 | 17 |
| 44 | . 35402 | . 93524 | . 27029 | . 92892 | . 38644 | . 92231 | . 40248 | . 91543 | . 41840 | . 90826 | 16 |
| 45 | . 35429 | . 93514 | . 37056 | . 98881 | . 38071 | . 92220 | .40275 | . 91531 | 41866 | . 90814 | 15 |
| 46 | . 35456 | . 93503 | . 37083 | . 92870 | . 38693 | . 92209 | . 40301 | . 91519 | . 41892 | . 90802 | 14 |
| 47 | . 35484 | . 93493 | . 37110 | . 92859 | . 38725 | . 92198 | . 40323 | . 91508 | . 41919 | . 90790 | 13 |
| 48 | . 35511 | . 93483 | . 37137 | . 92849 | . 38752 | . 92186 | . 40355 | . 91496 | . 41945 | . 90778 | 12 |
| 49 | . 35538 | . 93472 | . 37164 | . 98838 | . 38778 | . 92175 | . 40381 | . 91484 | . 41972 | . 90766 | 11 |
| 50 | . 35565 | . 93462 | . 37191 | . 98827 | . 38805 | . 92164 | . 40408 | . 91472 | . 41998 | . 90753 | 10 |
| 51 | . 35592 | . 93452 | . 37218 | . 98816 | . 38832 | . 92152 | . 40434 | . 91461 | . 42024 | . $90 \% 41$ | 9 |
| 52 | . 35619 | . 93441 | . 37245 | . 92805 | . 38859 | . 92141 | . 40461 | . 91449 | . 42051 | . $90 \% \% 9$ | 8 |
| 53 | . 35647 | . 93431 | . 37272 | . 92794 | . 38886 | . 92130 | . 40488 | . 91437 | . $420 \% 7$ | . 90717 | 7 |
| 54 | . 35674 | . 93420 | . 37299 | . 92784 | . 38912 | . 92119 | . 40514 | . 91425 | . 42104 | . 90704 |  |
| 55 | . 35701 | . 93410 | . 37326 | . 92773 | . 38939 | . 92107 | . 40541 | . 91414 | . 42130 | . 90692 | 5 |
| 56 | . 35728 | . 93400 | .37353 | . 92762 | . 38966 | . 92096 | . 40567 | . 91402 | . 42156 | . 90680 | 4 |
| 57 | . 35755 | . 93389 | . 37380 | . 92751 | . 38993 | . 92085 | . 40594 | . 91390 | . 42183 | . 90668 | 3 |
| 58 | . 35782 | . 93379 | . 37407 | . 92740 | . 39020 | . 92073 | . 40621 | . 91378 | . 42209 | . 90655 | 2 |
| 59 | . 35810 | . 93368 | . 37434 | . 92729 | . 39046 | . 92062 | . 40647 | . 91366 | . 42235 | . 90643 | 1 |
| 60 | . 35837 | . 93358 | 61 | 8 | 9973 | . 92050 | 67 | . 91355 | 2262 | 1 | 0 |
|  | Cosin | Sine | C | Sine | Cosin | Sine | Sin | Sine | OSi | Sine |  |
|  |  |  |  |  |  |  |  |  |  |  |  |


|  |  |  | $26^{\circ}$ |  |  |  | 28 |  | 29 |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  |  |  |  |  |  |
| 0 |  |  |  |  | . 45399 |  |  |  | . 48481 |  |  |
|  |  |  |  |  |  |  |  |  | 06 |  | 59 |
| $\stackrel{2}{3}$ |  |  |  |  |  |  |  |  |  |  |  |
| 3 | . 423 | . 90 | . 43 |  | $\begin{array}{\|l} .45477 \\ .4503 \end{array}$ |  | . 47024 | .88254 | . 485583 | 81406 |  |
|  |  |  |  |  |  |  | . 47076 |  | . |  |  |
|  | . 424 | . 905 |  |  | . 455 | . 890 | . 47101 |  | 4863 |  |  |
|  | 4244 | 9054 | . 44 | . 89790 | . 455 |  | . 471 | . 88199 | . 48659 |  |  |
|  | . 424 | . 90 | . 44 |  | . 456 |  | 47 | . 88185 | . 4868 |  |  |
| 10 |  |  |  |  |  |  |  |  | . 48710 | . 87 |  |
| 10 |  |  |  |  |  |  |  |  |  |  |  |
| 11 | . 42 | . 90 | . 4 | . 89 | . 45684 | . 88 | . 47 | . 88144 | 48761 |  |  |
| 12 |  |  | . 44 | . 80 | . 45710 |  |  |  |  |  |  |
| 13 | .4260 | . 30 |  | . 897 | . 45736 |  |  | . 88117 | . 488 |  |  |
| 14 | . 4263 | . 904 | . 44 | . 89700 | . 45762 |  | . 473 | . 88103 | . 48 |  |  |
| 15 | . 4265 | . 904 | . 44 | :896 | . 457 |  | . 47 | . 88089 | . 48862 |  |  |
| 16 | .42683 | . 9043 | . 44 | . 89 | . 458 |  | 47 |  |  |  |  |
| 17 | .427 | . 904 | . 44 |  | . 458 |  | . 473 |  | . 4891 |  |  |
| 18 |  | 904 | . 443 |  | :45 |  | . 47 |  | . 489 |  |  |
| 19 | . 42 | . 90 | . 443 |  | :4581 | . 8 | . 474 | . 8803 | . 48964 | . 87 |  |
| 20 |  |  |  |  |  |  | . 47 | . 88 |  |  |  |
| 21 | . 428 | . 90 | . 4 |  |  | . 88 | . 47 | . 88 | . 49014 |  | 9 |
| 29 | . 428 |  | . 41 | . 89 | . 450 |  | -4751 |  |  |  |  |
| 23 |  |  |  |  | . 459 |  |  | . 879 | . 490 |  |  |
| 24 | . 42 | . 9033 | . 41 |  | -46020 |  | . 47 | . 879 | 490 | 87 |  |
| 25 | .429 |  |  |  |  |  | . 47 |  | . 4911 |  |  |
| 26 | . 429 | . 903 |  |  |  |  | . 476 |  | . 4914 |  |  |
| 27 | . 429 |  | . 44 |  | . 460 | . 887 | . 476 |  | 491 |  |  |
| 23 |  |  |  |  | . 4012 |  | . 47 |  | 4 |  |  |
| 29 | . 43 |  | . 44 | . 895 | . 4614 | . 887 | . 476 | . 878 | . 492 |  |  |
| 30 | . 4305 |  |  |  |  | . 887 | . 477 |  | 492 |  |  |
| 31 | . 43 |  |  | . 89 |  |  |  |  |  |  |  |
| 39 |  |  |  |  |  |  |  |  |  |  |  |
| 83 | . 431 | . 90 |  |  |  |  |  |  | . 493 |  |  |
| 34 |  |  |  |  |  |  |  |  |  |  |  |
|  | . 4318 | . 9019 |  |  |  |  |  |  | . 4936 |  |  |
| 20 |  | . 9018 |  | . 89 |  |  |  | . 877 | . 49394 |  |  |
| 37 |  |  |  |  |  |  |  | . 877 | . 49419 |  |  |
| 38 | . 43 | . 9015 |  |  | . 46 |  |  |  | . 494 |  |  |
| 39 |  | . 015 |  |  | . 4640 |  | . 479 |  | . 43 |  |  |
| 40 | . 43 |  |  |  |  |  |  |  |  |  |  |
| 41 |  |  |  |  |  |  |  |  | . |  |  |
| 42 |  |  |  |  | . 464 |  |  |  | . 49 |  |  |
| 43 | . 4339 |  |  |  |  |  |  |  | . 49 |  |  |
| 4 | . 43 | . 0008 | . 44 |  |  |  | . 480 |  | . 495 |  |  |
|  | . 43 | . 9000 |  |  |  | . 8 |  |  | . |  |  |
| 46 |  | . 900 |  |  |  |  | . 481 |  | . 4964 | 8 |  |
|  |  | . 0003 |  |  |  | . 884 | . |  | . 430 | d |  |
| 48 | . 435 | . 9003 | . 450 | .892 | . 466 | . 8845 | . 4817 | .8763 | . 4969 | 86777 | 12 |
| 43 | . 43 | . 9001 | . 45 | . 8924 | . 4 | . 8844 | . 4820 | . 876 | . 497 | . 8676 |  |
| 50 |  |  |  |  |  |  |  |  |  |  |  |
|  | . 43 | . 8999 | . 45 |  |  |  | . 482 | . 87 | . 4 P7 |  |  |
| 52 | . 4362 | . 8998 | . 451 |  | . 4674 |  |  |  | . 497 |  |  |
| 53 | . 436 | - | . 45 | . 8911 | . 46 | . 88 | . 880 | . 8 | . 498 | . 867694 |  |
| 54 | . 436 | . 899 | . 45 | . 8918 | . 46 |  | . 4832 |  | . 498 | . 8669 |  |
| 55 | . 437 | . 8993 |  | . 891 | . 46819 | . 8836 | . 48 |  | .4987 | . 86 |  |
|  | . 43 | . 8993 |  | . 891 | . 4684 | . 883 | . 483 | . 875 | . 498 |  |  |
| 57 | . 43 | . 899 |  | . 89 | . 46 | . 883 | . 4840 | . 875 | . 4392 |  |  |
| 58 |  |  |  | . 891 | -4888 |  | . 48 | . 874 | . |  |  |
| 59 |  | . 8999 |  |  | . 46921 | . 88 | . 484 |  | 99 | . 86617 |  |
| 60 |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |


|  | $30^{\circ}$ |  | $31^{\circ}$ |  | $32^{\circ}$ |  | $33^{\circ}$ |  | $34^{\circ}$ |  | 1 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine | Cosin | Sine | Cosin | Sine | Cosin | Sine | Cosin | Sine | Cosin |  |
| 0 | . 50000 | . 86603 | . 51504 | . 85717 | . 52992 | . 84805 | . 54464 | . 83867 | 55919 | 82904 | 60 |
| 1 | . 50025 | . 86588 | . 51529 | . 85702 | . 53017 | . 84789 | . 54488 | . 838851 | . 55943 | $.82887$ | 59 |
| 2 | . 50050 | . 86573 | . 51554 | . 85687 | . 53041 | . 84774 | . 54513 | . 83835 | . 55968 | . 828871 | 58 |
| 3 | . 50076 | . 865559 | . 51579 | . 85677 | . 53066 | . 84759 | . 54537 | . 83819 | . 55992 | . 82855 | 57 |
| 4 | . 50101 | . 86544 | . 51604 | . 85657 | . 53091 | . 84743 | . 54561 | .8-804 | . 56016 | . 88839 | 56 |
| 5 | . 50126 | . 86530 | . 51628 | . 85642 | . 53115 | . 84728 | . 54586 | . 83788 | . 56040 | . 82822 | 55 |
| 6 | . 50151 | . 86515 | . 51653 | . 85627 | . 53140 | . 84712 | . 54610 | . 83772 | . 56064 | . 82806 | 54 |
| 1 | . 50176 | . 86501 | . 51678 | . 85612 | . 53164 | . 84697 | . 54635 | . 837756 | . 56088 | . 88790 | 53 |
| 8 | . 50201 | . 86486 | . 51703 | . 85597 | . 53189 | . 84681 | . 54659 | . 83740 | . 56112 | . 82773 | 52 |
| 9 | . 50227 | . 86471 | . 51728 | . 85582 | . 53214 | . 84666 | . 54683 | .83724 | . 56136 | . 82757 | 51 |
| 10 | . 50222 | . 86457 | . 51753 | . 85567 | . 53238 | . 84650 | . 54708 | . 83748 | . 56160 | . 82741 | 50 |
| 11 | . 50277 | . 86442 | . 51 | . 35551 | . 53263 | . 84635 | . 54732 | . 83692 | . 56184 | .82724 | 49 |
| 12 | . 50302 | . 86427 | . 51803 | . 85536 | . 53288 | . 84619 | . 54756 | . $836 \% 6$ | . 56208 | .82\%08 | 48 |
| 13 | . 50327 | . 86413 | . 51828 | . 85521 | . 53312 | . 84604 | . 54781 | . 83660 | . 56232 | . 82692 | 47 |
| 14 | . 50352 | . 86398 | . 51852 | . 85506 | . 53337 | . 84588 | . 54805 | . 83615 | . 56256 | . 82675 | 46 |
| 15 | . 50377 | . 86384 | . 51877 | . 85491 | . 53361 | . 84573 | . 54829 | . 83629 | . 56280 | . 82659 | 45 |
| 16 | . 50403 | . 86369 | . 51902 | . 85476 | . 53386 | . 84557 | . 54854 | . 83613 | . 56305 | . 82643 | 44 |
| 17 | . 50428 | . 86354 | . 51927 | . 85461 | . 53411 | . 84542 | . 54878 | . 83597 | . 56329 | . 82626 | 43 |
| 18 | . 50453 | . 86340 | . 51952 | . 85446 | . 53435 | . 84526 | . 54902 | . 83581 | . 56353 | . 82610 | 42 |
| 19 | . 50478 | . 86325 | . 51977 | . 85431 | . 53460 | . 84511 | . 54927 | . 83565 | . 56377 | . 82593 | 41 |
| 20 | . 50503 | . 86310 | . 52002 | . 85416 | . 53484 | . 84495 | . 54951 | . 83549 | . 56401 | . 82577 | 40 |
| 21 | . 50528 | 86295 | . 52026 | . 85401 | . 53509 | . 84480 | . 54975 | . 83533 | . 56425 | . 82561 | 39 |
| 22 | . 50553 | . 86281 | . 52051 | . 85335 | . 53534 | . 84464 | . 54999 | .83517 | . 56449 | . 82544 | 38 |
| 23 | . $505 \% 8$ | . 86266 | . 52076 | . 85370 | . 53558 | . 84448 | . 55021 | . 83501 | . 56473 | . 82528 | 37 |
| 24 | . 50603 | . 86251 | . 52101 | . 85355 | . 53583 | . 84433 | . 55043 | . 83485 | . 56497 | . 82511 | 36 |
| 25 | . 50628 | . 86237 | . 52126 | . 85340 | . 53607 | . 84417 | . $550 \% 2$ | . 83469 | . 56521 | . 82495 | 35 |
| 26 | . 50654 | . 86222 | . 52151 | . 85325 | . 53632 | . 84402 | . 55097 | . 83453 | . 56545 | . 82478 | 34 |
| 27 | . 50679 | . 86207 | . 52175 | . 85310 | . 53656 | . 84386 | . 55121 | . 83437 | . 56569 | . 82462 | 33 |
| 28 | . 50704 | . 86192 | . 52200 | . 85294 | . 53681 | . 84370 | . 55145 | . 83421 | . 56593 | . 82446 | 32 |
| 29 | . 50729 | . 86178 | . 52225 | . 85879 | . 53705 | . 84355 | . 55169 | . 83105 | . 56617 | . 82429 | 31 |
| 30 | . 50754 | . 86163 | . 52250 |  |  | . 84339 | . 55194 | . 83389 | . 56641 | . 82413 | 30 |
| 31 | . 50779 | . 86148 | . 52275 | . 85249 | . 53754 | . 84324 | . 55218 | . 83373 | . 56665 | . 82396 | 29 |
| 82 | . 50804 | . 86133 | . 52293 | . 85234 | . 53779 | . 84308 | . 55912 | . 83356 | . 56689 | . 82380 | 28 |
| 33 | . 50829 | . 86119 | . 52324 | . 85218 | . 53804 | . 84292 | . 55266 | . 83340 | . 50713 | . 82363 | 27 |
| 34 | . 50854 | . 86104 | . 52349 | . 85203 | . 53828 | . 84277 | . 55201 | . 83324 | . $56 \sim 36$ | . 82347 | 26 |
| 35 | . 50879 | . 86089 | . 52374 | . 85188 | . 53853 | . 84261 | . 55315 | . 83308 | . 56760 | . 82330 | 25 |
| 36 | . 50904 | . 86074 | . 52399 | . 85173 | . 53877 | . 84245 | . 55339 | . 83292 | . 56784 | . 82314 | 4 |
| 37 | . 50929 | . 86059 | . 52423 | . 85157 | . 53902 | . 84230 | . 55363 | . $83 \times 76$ | . 56808 | . 82297 | 23 |
| 38 | . 50954 | . 86045 | . 52448 | . 85142 | .53926 | . 84214 | . 55388 | . 83260 | . 56832 | . 82281 | 22 |
| 39 | . 50979 | . 86030 | . 52473 | . 85127 | . 03951 | . 84198 | . 55412 | . 83244 | . 56856 | . 82264 | 21 |
| 40 | . 51004 | . 86015 | . 52498 | . 85112 | . 53975 | . 84182 | . 55436 | . 83228 | $.56880$ | . 82248 | 20 |
| 41 | . 51029 | . 86000 | .525\%2 | . 85096 | . 54000 | . 84167 | . 55460 | . 83212 | . 56904 | . 82231 | 19 |
| 42 | . 51054 | . 855985 | . 52547 | . 85081 | . 54024 | . 84151 | . 55481 | . 83195 | . 56928 | . 82214 | 18 |
| 43 | . 51079 | . 85970 | . 52572 | . 85066 | . 54049 | . 84135 | . 55509 | . 83179 | . 56952 | . 82198 | 17 |
| 44 | . 51104 | . 85956 | . 52597 | . 85051 | . $540 \% 3$ | . 84120 | . 55533 | . 83163 | . 56976 | . 82181 | 16 |
| 45 | . 51129 | . 85941 | . 52621 | . 85035 | . 54097 | . 84104 | . 55557 | . 83147 | . 57000 | . 82165 | 15 |
| 46 | . 51154 | . 85926 | . 52646 | . 85020 | . 54122 | . 84088 | . 55581 | . 83131 | . 57024 | . 82148 | 14 |
| 47 | . 51179 | . 85911 | . 52671 | . 85005 | . 54146 | . 84072 | . 55605 | . 83115 | . 57047 | . 82132 | 13 |
| 48 | . 51204 | . 85896 | . 52695 | . 84989 | . 54171 | . 84057 | . 55630 | . 83098 | . 57071 | . 82115 | 12 |
| 49 | . 51229 | . 85881 | .52720 | . 84974 | . 54195 | . 84041 | . 55654 | . 83082 | . 57095 | . 82098 | 11 |
| 50 | . 51254 | . 85866 | . 52745 | . 84959 | . | . 84025 | - 56 | . 83066 | . 57119 | . 82082 | 10 |
| 51 | .51279 | . 85851 | . 5276 | . 84943 | . 54244 | . 84009 | . 55702 | . 83050 | . 57143 | . 82065 | 9 |
| 52 | . 51304 | . 85836 | . 52794 | . 84928 | . 54269 | . 83994 | . $55 \% 2$ | . 83034 | . 57167 | . 82048 | 8 |
| 53 | . 51329 | . 85821 | . 52819 | . 81913 | . 54293 | . 83978 | . 55750 | . 833017 | . 57191 | . 82032 | 7 |
| 54 | . 51354 | . 85806 | . 52844 | . 84897 | . 54317 | .83962 | . 55775 | . 83001 | . 57215 | . 82015 | 6 |
| 55 | . 51379 | . 85792 | . 52869 | . 81882 | . 54342 | . 83946 | . 55799 | . 89985 | . 57238 | . 81999 | 5 |
| 56 | . 51404 | . 85777 | . 52893 | . 84866 | . 54366 | . 83930 | . 55823 | . 82969 | . 57268 | . 81982 | 4 |
| 57 | . 51429 | . 85762 | . 52918 | . 84851 | . 54391 | . 83915 | . 55847 | . 82953 | . 57286 | . 81965 | 3 |
| 58 | . 51454 | . 85747 | . 52943 | . 84836 | . 54415 | . 83899 | . 55871 | . 82936 | . 57310 | . 81949 | 2 |
| 59 | . 51479 | . 85732 | . 52967 | . 84820 | . 54440 | . 83883 | . 55895 | . 82920 | . 5733 | . 81932 | 1 |
| 60 | . 51 |  | . 52992 |  |  |  |  | 8000 | 5r0 |  | 0 |
|  | Cosin | Sine | Co | Sine | Cos | Sine | Cos | Sine | Cosi | Sine |  |
|  |  | $9^{\circ}$ |  |  |  |  |  |  |  |  |  |


|  |  |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine |  |  |  |  |  | Sine |  | Sine |  |  |
| 0 |  |  |  |  |  |  | . 61566 |  | . 62932 |  | 30 |
|  | . 57381 |  |  |  | . 60 |  |  |  | 2955 |  |  |
| 2 | . 57405 |  |  |  |  |  |  |  |  |  |  |
|  | . 57429 | 818 |  |  | .60251 |  |  |  | . 63000 |  |  |
|  | . 57453 | . 81848 |  |  | . 602 | . 79793 |  | . 78729 | . 63022 | . 77 | 6 |
|  | . 57477 | 81832 |  |  | . 602 |  | . 61681 | . 78711 | . 63045 |  |  |
|  | . 57501 | 81815 |  | . 80799 | . 603 | . 7975 | . 61704 | . 78694 | . 63068 | 77 |  |
|  | . 57524 | .81798 | . 58943 |  | . 60334 | 79 | . 61726 | -78676 | . 63090 |  |  |
|  | . 57548 | . 817 | .589 | . 80 | . 603 | .792 | . 61749 | .786 | . 63113 |  |  |
|  | . 57572 | . 8176 |  |  | . 60390 | . 7970 | . 6172 | . 78640 | . 631 | . 7 |  |
| 10 | . 57596 |  |  |  |  |  |  |  |  |  |  |
| 11 | . 5 | . 81 | . 5 | . 8 | . 60 | . 7 | . 61 | 78 | . 63180 |  |  |
| 12 |  |  |  |  |  |  |  |  |  |  |  |
| 13 |  | . 816 | . 590 | . 800 |  |  | . 61 |  | . 632 |  |  |
| 14 | . 5768 | . 810 | . 591 |  | . 605 | 796 | . 61 |  | . 63248 |  |  |
| 15 | . 57 |  | . 59131 | . 80644 | . 605 | . 796 | . 61 |  | . 63271 |  |  |
| 16 | . 5 | . 8164 | . 59154 |  | . 605 | 795 | 61 | 785 |  | 77 |  |
| 17 | . 5 |  | . 59178 | . 80610 | . 605 | . 795 | . 61 | 78496 | . 63 | . 77 |  |
| 18 |  | 816 | . 59201 | . 80593 | . 6059 | . 795 | . 61 | . 78478 |  | . 77 |  |
| 19 | . 57810 |  | . 5 | . 805 |  | . 795 | . 6 | . 78 | . 63331 | . 77 |  |
| 20 |  |  |  |  |  |  |  |  |  |  |  |
| 21 | . 5 | . 81 | . 59272 | . 80 | . 60 | . 79 | 62 | 78424 | . 63406 | . 7 |  |
| 2 | . 578 |  |  |  |  |  |  |  |  | . 77 |  |
| 23 |  |  |  | . 805 | . 607 |  |  |  | . 63451 |  |  |
| 2 | . 5792 | . 81 | . 59342 | . 8048 |  | . 7944 | 62 |  | . 6347 | . 77 |  |
|  | . 5 |  | . 59365 |  |  | . 79 |  |  |  |  |  |
| 26 | . 57 |  | . 593 |  |  | . 794 | . 62 |  | . 6351 |  |  |
|  | . 5799 | . 81 | . 59412 |  |  | . 793 |  | 78315 | 63 | 77 |  |
| 28 |  |  | . 594 |  |  | . 79 |  | 7829 |  |  |  |
| 23 | . 58047 |  | 594 | . 8 |  | . 79 |  | 78279 | , | 771 |  |
| 20 | . 58 | . 8 |  | . 803 |  |  |  |  |  |  |  |
| 31 | . 5 |  | . 595 | . 80 | . 608 | . 793 | . 62 | 78 | . 63 |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
| 33 | . 58 | . 81361 | . 595 | . 803 |  |  |  | 781 | . 636 |  |  |
|  |  |  | . 595 |  |  | . 792 |  |  |  |  |  |
|  | . 58 |  | . 59599 |  |  | . 792 |  |  |  |  |  |
| 36 | . 58 | . 8131 | . 590 | . 80 | . 610 | . 7922 |  | 781 | . 637 | . 77 |  |
|  |  | . 8 | . 596 |  |  |  | . 62 | 781 | . 637 |  |  |
|  | . 58260 | . 81 | . 596 | . 802 | . 610 | 791 |  | 781 | . 637 |  |  |
|  | 58283 | . 81209 | . 59693 | . 80 |  | . 791 |  | . 7809 | . 638 | 76 |  |
| 40 |  |  |  |  |  |  |  |  |  |  |  |
| 41 | . 58 | . 81 | . 597 |  | . 611 | . 791 | . 625 | . 780 | . 638 |  |  |
| 42 |  |  | . 59 | . 80 | . 61 | . 79 |  | . 78 | - |  |  |
| 43 |  | . 811191 |  |  |  |  |  | 780 | . 63 |  |  |
| 44 |  |  |  |  | . 6 |  |  | . 780 | . 63822 |  |  |
| 48 |  |  |  | . 801 |  |  |  | 77 | ${ }^{6}$ |  |  |
| 46 |  |  | . 5988 | . 801 | . 612 | . 790 | . 626 | . 77970 | . 639 | 768 |  |
|  | . 5 | . 81123 |  |  |  |  |  | . 77 | - | . 768 |  |
| 48 |  | 81106 | . 59902 | . 800 | . 6129 | . 790 | . 626 | . 77934 | . 6401 |  | 12 |
| 49 | . 58519 | 81089 | 992 | . 800 | . 6131 | .789 | . 626 | . 77916 | . 6403 | 768 |  |
| 50 |  | . 81 |  |  |  |  |  | . 77897 |  |  |  |
| 51 |  | . 810 | . | . 80 | . 613 | . 78 | . 62 | . 77 | . |  |  |
| 52 | . 58590 | . 81038 | . 599 |  | . 61 |  | . 6275 | . 778 | . 641 |  |  |
|  | . 58 | . 810 | - 60 | - | . 61 | . 8 | . 62 | . 77843 | . 64123 |  |  |
| 54 |  | . 810 | . 600 | . 79 | . 614 |  | . 62 | . 77824 | . 641 | . |  |
|  | . 58 | . 809 | . | T99 | . 14 | . 788 | . 6281 | . 7780 | . 641 |  |  |
|  |  |  | . 6008 | 799 | . 614 |  | . 628 | . 77788 | 6419 | ( |  |
|  | . 587808 | . 809 | . 6011 | 799 |  |  |  | 77 | . 6421 | T6 |  |
|  | . 58731 |  | .6013 | . 798 | . 615 |  | . | . 77751 | . 642 |  |  |
|  | . 58755 | . |  | 79 |  | . 788 |  |  | . 642 |  |  |
| 60 |  |  |  |  |  |  |  |  | .642 | \% |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |


|  | $40^{\circ}$ |  | $41^{\circ}$ |  | $42^{\circ}$ |  | $43^{\circ}$ |  | $44^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Sine | Cosin | Sine | Cosin | Sine | Cosin | Sin $\theta$ | Cosin | Sine | Cosin |  |
| 0 | . 64279 | . 76604 | . 65606 | . 75471 | . 66913 | . 74314 | . 68200 | . 73135 | . 69466 | . 71934 | 60 |
| 1 | . 64301 | . 76586 | . 65628 | . 75452 | . 66935 | . 74295 | . 68221 | . 73116 | .69487 | . 71914 | 59 |
| 2 | . 64323 | . 76567 | . 65650 | . 75433 | . 66956 | - 74276 | . 68242 | .73096 | . 69508 | . 718 | 58 |
| 3 | . 64346 | . 76548 | . 65677 | . 775414 | . 66978 | . 74256 | . 68264 | . 73006 | .69529 | . 718 | 57 |
| 4 | . 64368 | . 76530 | . 65694 | . 775395 | . 66999 | . 74237 | . 68285 | . 73056 | . 69549 | . 718 | 56 |
| 5 | . 64397 | . 76511 | . 65716 | . 75375 | . 67021 | . 74217 | . 68306 | .73036 | . 69570 | - $\% 18$ | 55 |
| 6 | . 64412 | . 76492 | . 65738 | . 75356 | . 67043 | . 74198 | . 68327 | . 73016 | . 69591 | . 7181 | 54 |
| 7 | . 64435 | . 76473 | . 65759 | . 75337 | . 67064 | . 74178 | . 68349 | . 72996 | . 69612 | . 7179 | 53 |
| 8 | . 64457 | . 76455 | . 65 \%81 | . 75318 | . 67080 | . 74159 | . 68370 | . 72976 | . 69633 | . $717 \%$ | 52 |
|  | . 64479 | . 76433 | . 65803 | . 75299 | . 67107 | . 74139 | . 68391 | . 72957 | . 69654 | . 7175 | 51 |
| 10 |  | . 76417 | . 65825 | . 75280 | . 67129 | . 74120 | . 68412 | . 72937 | . 69675 | . 71 | 50 |
| 11 | . 64524 | . 76398 | . 65847 | . 75261 | . 67151 | . 74100 | . 68434 | . 72917 | . 69696 | . 71711 | 49 |
| 12 | . 64546 | . 76330 | . 65353 | . 75211 | . 67172 | . 74080 | . 68455 | . 72827 | . 69717 | . 7169 | 48 |
| 13 | . 64538 | . 76351 | . 65391 | . 7529 | . 67194 | . 74061 | . 68476 | . $728 \% 7$ | .69737 | . 71671 | 47 |
| 14 | . 64590 | . 76342 | . 65913 | . 75203 | . 67215 | . 74041 | . 68497 | . 72857 | . 69758 | . 71650 | 46 |
| 15 | . 64612 | . 76393 | . 65935 | . 75184 | . 67237 | . 74022 | . 68518 | . 72827 | . 69779 | . 7163 | 45 |
| 16 | . 64635 | . 76304 | . 65953 | . 75165 | . 67258 | . 74002 | . 68539 | . 72817 | . 69800 | . 7161 | 44 |
| 17 | . 64657 | . 76233 | . 65978 | . 75146 | . 67280 | . 73983 | . 68561 | . 72797 | . 69821 | . 715 | 43 |
| 18 | . 64679 | . 76267 | . 63009 | . 75123 | . 67301 | . 73963 | . 68582 | . $727 \%$ | . 69842 | . 71569 | 42 |
| 19 | . 64701 | . 76248 | .63022 | .75107 | . 67323 | . 73944 | . 68603 | . 72757 | . 69862 | . 71549 | 41 |
| 20 | . 64723 | . 76229 | . 66044 | . 75088 | . 67344 | . 73924 | . 68624 | . 72737 | . 69883 | .7152 | 40 |
| 21 | . 64746 | . 76210 | . 66066 | . 75069 | . 67366 | . 73904 | . 68645 | .72717 | . 69904 | . 71508 | 39 |
| 22 | . 647 | . 76192 | . 63033 | . 75050 | . 67307 | . 78885 | . 68666 | . 72697 | . 69925 | . 714 | 33 |
| 23 | . 64790 | . 76173 | . 63109 | . 75030 | . 67409 | . 73805 | . 68688 | . 72677 | . 69946 | . 71 | 37 |
| 24 | . 64812 | . 76154 | . 63131 | . 75011 | . 67433 | . 73846 | . 68709 | . 72657 | . 69966 | . 714 | 30 |
| 25 | . 64834 | . 76135 | . 66153 | . 74992 | .67452 | . 73820 | . 68730 | . 72637 | . 69987 | . 7142 | 35 |
| 26 | . 64856 | . 76116 | . 63175 | . 74973 | . 67473 | . 73803 | . 68751 | . 72617 | . 70008 | . 7140 | 31 |
| 27 | . 64878 | . 76097 | . 66197 | . 74953 | . 67495 | . 73787 | . $687{ }^{2}$ | . 72597 | . 70029 | . 7138 | 33 |
| 28 | . 64901 | . 76078 | . 66218 | . 74934 | . 67516 | . 73767 | . 68793 | . 72577 | . 70049 | . 7136 | 32 |
| 29 | . 64923 | . 76059 | . 68240 | . 74915 | . 67538 | . 73747 | . 68814 | . 72557 | . 70070 | . 71345 | 31 |
| 30 | . 64945 | . 76041 | . 66262 | . 74896 | 67559 | . 73728 | . 68835 | . 72537 | . 70091 | . 71325 | 30 |
| 31 | . 64967 | . 76022 | . 66284 | . 74876 | . 67580 | . 73708 | . 68857 | . 72517 | . 70112 | . 71305 | 29 |
| 32 | . 64989 | . 76003 | . 63306 | . $\% 4857$ | . 67632 | . 73683 | . $688 \% 8$ | . 72497 | 70132 | . 7128 | 23 |
| 33 | . 65011 | . 75984 | . 68327 | . 74838 | . 67623 | . 73669 | . 68899 | . $724 \% 7$ | 70153 | . 71264 | 27 |
| 34 | . 65033 | . 75965 | . 63349 | . 74818 | . 67645 | . 73649 | . 68920 | . 72457 | . 70174 | . 71243 | 26 |
| $3 \pm$ | . 65055 | . 75943 | . 63371 | . 74799 | . 67666 | .73623 | . 68941 | . 72437 | \%0195 | . 7122 | 35 |
| 36 | . 65077 | . 75927 | . 63393 | . 74783 | . 67633 | . 73610 | . $6895 \%$ | . 72417 | 70215 | . 71203 | 24 |
| 37 | . 65100 | . 75903 | . 66414 | . 74760 | . 67709 | . 73590 | . 68983 | . 72397 | . 70236 | . 71182 | 23 |
| 88 | . 65122 | . 75889 | . 66438 | . 71741 | . 67730 | . 73570 | . 69004 | . 72377 | 70257 | . 71162 | 23 |
| 39 | . 65144 | . 75870 | . 63453 | .74722 | . 67102 | . 73551 | . 69025 | . 77357 | . 70277 | . 71141 | 21 |
| 40 | . 65166 | . 75851 | - | . 74703 | - | . 73531 | . 69046 | . 78337 | 0298 | 71121 | 20 |
| 41 | . 65188 | .75832 | . 66501 | . 74683 | . 67795 | . 73511 | . 69067 | . 72317 | . 70319 | . 71100 | 19 |
| 42 | . 65210 | . 75313 | . 66593 | . 74664 | . 67816 | . 73491 | . 63088 | .72397 | . 70339 | . 7108 | 18 |
| 43 | . 65332 | . $75 \% 94$ | . 66545 | . 74644 | . 67837 | . $731 \% 2$ | . 69109 | . 72277 | . 70360 | . 7105 | 17 |
| 44 | . 63524 | . 75775 | . 66563 | . 74625 | . 67859 | . 73452 | . 69130 | . 72257 | . 70381 | . 7103 | 16 |
| 45 | . 65276 | . 75756 | . 66588 | . 74606 | . 67880 | . 73432 | . 69151 | . 72236 | . 70401 | . 7101 | 15 |
| 46 | . 65298 | . 75738 | . 66610 | . 74586 | . 67901 | . 73113 | . 69172 | . 72216 | . 70422 | . 7099 | 14 |
| 47 | . 65320 | . 75719 | . 66632 | . 74567 | . 67923 | . 73393 | . 69193 | . 72196 | . 70443 | . 7097 | 13 |
| 48 | . 65342 | . 75700 | . 66653 | . 74548 | . 67944 | . 73373 | . 69214 | . 72176 | . 70463 | . 7095 | 12 |
| 49 | . 65364 | . 75680 | . 66675 | . 745 5 | . 67965 | . 73353 | . 69235 | . 72156 | . 70484 | . 70937 | 11 |
| 50 | . 653 | . 75661 | . 66697 | . 74509 | . 67987 | . 7 | . 69256 | .6130 | . 70505 | . 70916 | 10 |
| 51 | . 65408 | . 75642 | . 66718 | . 74489 | . 68008 | . 73314 | .69277 | . 72116 | . 70525 | . 70896 | 9 |
| 52 | . 65430 | . 75623 | . 66740 | . 74470 | . 68029 | . 73294 | . 69298 | . 72095 | . 70546 | . 70875 | 8 |
| 53 | . 65452 | . 75604 | . 66762 | . 74451 | . 68051 | . 73274 | . 69319 | . 72075 | . 70567 | . 70855 | 7 |
| 54 | . 65474 | . 75585 | . 66783 | . 74431 | . 68072 | . 73254 | . 69340 | . 72055 | . 70587 | . 70834 | 6 |
| 55 | . 65496 | . 75566 | . 66805 | . 74412 | . 68093 | . 73234 | . 69361 | . 72035 | . 70608 | . 70813 | 5 |
| 56 | . 65518 | . 75547 | . 66827 | . 74392 | . 68115 | . 73215 | . 69382 | . 72015 | . 70628 | . 70793 | 4 |
| 57 | . 65540 | . 75528 | . 66848 | . 74373 | . 68136 | . 73195 | . 69403 | .71995 | . 70649 | . 70772 | 3 |
| 58 | . 65562 | . 75509 | . 66870 | . 74353 | . 68157 | . 73175 | . 69424 | . 71974 | . 70670 | . 70752 | 2 |
| 59 | . 65584 | . 75490 | . 66891 | . 74334 | . 68179 | . 73155 | . 69445 | . 71954 | . 70690 | . 70731 | 1 |
| 60 | . 65606 | . 75471 | . 66913 |  | 68200 | . 73135 |  |  |  |  | 0 |
|  | Cusin | Sine | Cos | Sine | Co | Sine | Cosin | Sine | Cosin | Sin |  |
|  |  |  |  | $8^{\circ}$ |  |  |  |  | 45 |  |  |


|  | $0^{\circ}$ |  | $1{ }^{\circ}$ |  |  |  | $8^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tan | Cotang | Tang | Cotang | Tan | Cotang |  |
|  | . 00000 | Infinite. | . 017 | 57.2900 | . 03492 | 28.6363 | . 05241 | 19.0811 | 60 |
|  | . 00029 | 3437.75 | . 017 | 56.3506 | . 03 | 28.3994 | . 05 | 18.9755 | 59 |
|  | . 00058 | 1718.87 | . 0180 | 55.4415 | . 03550 | 28.1664 | . 05299 | 18.8711 |  |
|  | . 00087 | 1145.92 | . 01833 | 54.5613 | . 03579 | 27.9372 | . 05338 |  |  |
|  | . 00116 | 859.436 | . 01862 | 53.7086 | . 03609 | 27.7117 | . 05357 | 18.6656 |  |
|  | . 00145 | 687.549 | . 01891 | 52.8821 | . 0363 | 27.4899 | . 05387 | 18.5645 |  |
|  | . 00175 | 572.957 | . 01920 | 52.0807 | . 0366 | 27.2715 | . 05416 | 18.4645 |  |
|  | .00204 | 491.106 | . 01949 | 51.3032 | .0369 | 27.0566 | . 054 | 18.3655 |  |
|  | .00233 | 429.718 | . 01978 | 50.5485 | .0372 | 26.8450 | . 054 | 18.2677 |  |
|  | . 002 | 381.971 34.784 | . 0200 | 49.8157 49.1039 | .03\% | ${ }_{26.4316}^{26.6367}$ | . 055 | 18.1208 |  |
| 0 |  |  | 203 |  |  | 26. | 05 |  |  |
| 11 | . 003 | 312. | . | 48.4 | . 03812 | 26.2 | . 05 | 17.9802 |  |
|  | . 00349 | 236.478 | . 0209 | 47.7 | . 03842 | 26.0307 | . 05591 | 17.8863 |  |
| 13 | .00378 | 204.441 | . 02124 | 47.0853 | . 03871 | 25.8318 | . 05620 | 17.7934 |  |
|  | . 00407 | 245.552 | . 02153 | 46.4189 | . 03900 | 25.6418 | . 05649 | 17.7015 |  |
|  | . 00436 | 229.182 | . 02182 | 45.8294 | . 03929 | 25.4517 | .056\%8 | 17.6106 |  |
|  | . 00465 | 214.858 | . 02211 | 45.2261 | . 0395 | 25.2644 | . 057 | 17.5205 |  |
|  | . 00495 | 202.219 | . 02240 | 44.6386 | . 0398 | 25.0798 | .05\% | 17.4314 |  |
|  | . 0052 | 190.984 | . 02269 | 44.0661 | . 0401 | 24.8978 | . 057 | 17.3432 |  |
| 19 | .0055 | 180.932 | .02298 | 43.5081 | . 0404 | 24.7185 | .0579 | 17.2 |  |
| 20 |  |  |  | 42. |  | 24. | 58 |  |  |
| 21 | . 006 | 163 | . 02357 | 42. | 41 | 24. |  |  |  |
|  | 00640 | 156.259 | . | 41.91 | . 0413 | 24.19 | .0308 | . |  |
| 23 | 00669 | 149.465 | . 0241 | 41.4106 | . 0416 | 24.0263 | . 0591 | 16.9150 | 37 |
|  | . 0009 | 143.237 | . 0244 | 40.91\%4 | . 0419 | 23.8593 | . 0594 | 16.8319 |  |
|  | . 0072 | 137.507 | . 0247 | $40.43{ }^{\text {2 }} 8$ | . 0422 | 23.6945 | .0597 | 16.7496 |  |
|  | . 0075 | 132.219 | .0250 | 39.96 | .04250 | 23.5321 | . 05999 | 16.6681 |  |
|  | . 00785 | 127.321 | . 0253 | 39.5059 | .04279 | 23.3718 | . 0602 | 16.5874 |  |
|  | . 00315 | 122.764 | . 0256 | 39.0568 | . 0430 | 23.2137 | . 060 | 16.5075 |  |
|  | . 0084 | 118.540 | . 02589 | 38.6177 | . 0433 | 23.05 | G0 |  |  |
| 50 | . 00 | 11 | . 02619 | 38.1885 | . 04366 | 22. | . 06116 | 9 |  |
| 31 | . 00 | 110 | . 026 |  | . 04 | 22.7519 | 06 | 16 |  |
|  | . 00931 | 107.426 | .0267 | 37.3 | . 0442 | 22.6020 | . 061 | 16.1952 |  |
| 33 | . 00960 | 104.171 | .02706 | 36.9560 | . 0445 | 22.4541 | . 0620 | 16.1190 |  |
|  | . 00989 | 101.107 | .0273 | 36.5627 | . 0448 | 22.3081 | .0623 | 16.0435 |  |
|  | . 01018 | 93.21\%9 | . 02764 | 36.1\% ${ }^{\text {c }} 6$ | . 04512 | 22.1610 | . 0626 | 15.9687 |  |
|  | . 01047 | 93.4895 | . 0279 | 35.8006 | . 04541 | 22.0217 | .0699 | 15.8945 |  |
|  | . 01016 | 92.9085 | .0282 | 35.4313 | .045 | 21.8813 | . 063321 | 15.8211 |  |
|  | . 01105 | 90.4633 | . 0285 | 35.0395 | . 0459 | 21.7426 | . 06350 |  |  |
|  | . 01135 | 88.1436 | . 02881 | 34.7151 | - | 21.6056 | . 06379 |  |  |
| 40 | . 011 | 85 | . 0291 | 34.3678 |  | 21. |  | 15.6048 |  |
| 41 | . 0110 | 83.8 | . 0293 | 34.0 | . 046 | 21. | . 064 |  |  |
|  | .01222 | $81.84{ }^{\text {r }} 0$ | . 0296 | 33.6935 | .047 | 21.2049 | . | . |  |
|  | . 01251 | 79.9434 | .0299 | 33.3662 | . 0474 | 21.0747 | . 064 | 15.3943 | 7 |
|  | . 01280 | ${ }^{7} 8.1263$ | . 03026 | 33.0452 | . 04774 | 20.9460 | . 065 | 15.32 |  |
|  | . 01309 | \%6.3900 | . 03055 | 32.7303 | . 04803 | 20.8188 | . 065 | 15.2571 | 5 |
|  | .c1338 | 74.7292 | . 03084 | 32.4213 | . 0483 | 20.6032 | . 0658 | 15.1893 |  |
|  | . 01367 | 73.1390 | . 03114 | 32.1181 | . 0486 | 20.5691 | . 0661 | 15.12 | 13 |
|  | . 01396 | 71.6151 | . 0314 | 31.8205 | . 0489 | 20.4465 | . 0664 | 15.05 | 12 |
|  | . 01425 | 70.1533 | . 31 | 31.5284 | , | 20.3253 | . 060 | 14.9898 |  |
| 50 | . 0145 | 68.7501 | . 0320 | 31.2416 | 049 | 20.2056 | . 066 |  |  |
|  | . 0148 | 67.401 | .0323 | 30.9 | . 049 | 20.08 | . 067 | 14.8 |  |
|  | . 01513 | 66.1055 | .03259 | 30.6833 | .05007 | 19.9702 | . 0675 | 14.79 |  |
|  | . 01542 | 64.8580 | . 03288 | 30.4116 | . 0503 | 19.8546 | . 0668 | 14.7317 |  |
|  | . 01571 | 63.6567 | . 03317 | 30.1446 | . 050 | 19.7403 | . 0681 | 14.668 |  |
|  | . 01600 | 62.4992 | . 03346 | 29.8823 | . 0509 | 19.6273 | . 06847 | 14.605 |  |
|  | . 01629 | ${ }^{61.3829}$ | . 03376 | 29.6275 | . 0512 | 19.5156 | . 06876 | 14.543 |  |
|  | . 01658 | ${ }^{60.3058}$ | . 03405 | 29.3711 | . 05153 | 19.4051 | . 06995 | 14.4823 |  |
|  | . 01688 | 59.2659 | . 03434 | 29.1220 | . 05182 | 19.2959 | . 06934 | 14.4212 |  |
|  | . 01716 | 58.2612 | . 03463 | 28.8771 | 55212 | 19.1879 | . 06963 | 14.3607 |  |
| 60 | . 01746 | 57.2900 | . 03492 | 28.6363 | 524 | 19.0811 | . 06993 | 14.3007 | 0 |
|  |  |  |  | Tang | Cotang | Tang | Cotan | Tan |  |
|  |  |  |  |  |  |  |  |  |  |


|  | $4^{\circ}$ |  | $5{ }^{\circ}$ |  | $6^{\circ}$ |  | $7{ }^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| 0 | . 06993 | 14.3007 | . 08749 | 11.4301 | . 10510 | 9.51436 | . 12278 | 8.14435 | $\overline{60}$ |
| 1 | . 07022 | 14.2411 | . 08778 | 11.3919 | . 10540 | 9.48781 | . 12308 | 8.12481 | 59 |
| 2 | . 07051 | 14.1821 | . 08807 | 11.3540 | . 10569 | 9.46141 | . 12338 | 8.10536 | 58 |
| 3 | . 07080 | 14.1235 | . 08837 | 11.3163 | . 10599 | 9.43515 | .12367 | 8.08600 | 57 |
| 4 | . 07110 | 14.0655 | . 08866 | 11.2789 | . 10629 | 9.40904 | . 12397 | 8.06674 | 56 |
| 5 | . 07189 | 14.0079 | . 08895 | 11.2417 | . 10657 | 9.38307 | . 12426 | $8.04 \% 56$ | 55 |
| 6 | . 07168 | 13.9507 | . 08925 | 11.2048 | . 10687 | 9.35724 | . 12456 | 8.02848 | 54 |
| 7 | . 07197 | 13.8940 | . 08954 | 11.1681 | . 10716 | 9.33155 | . 12485 | 8.00948 | 53 |
| 8 | .07227 | 13.8378 | . 08983 | 11.1316 | . 10746 | 9.30599 | . 12515 | 7.99058 | 52 |
| 9 | . 07256 | 13.7821 | . 09013 | 11.0954 | 10775 | 9.28058 | .12544 | 7.97176 | 51 |
| 10 | .0\%285 | 13.7267 | . 09042 | 11.0594 | . 10805 | 9.25530 | 125\%4 | 7.95302 | 50 |
| 11 | . 07314 | 13.6719 | . 09071 | 11.0237 | . 10834 | 9.23016 | . 12603 | 7.93438 | 49 |
| 12 | . 07344 | 13.6174 | . 09101 | 10.9882 | . 10863 | 9.20516 | .12C33 | 7.91582 | 48 |
| 13 | . 07373 | 13.5634 | . 09130 | 10.9529 | . 10893 | 9.18028 | .12662 | 7.89734 | 47 |
| 14 | . 07402 | 13.5098 | . 09159 | $10.91 \% 8$ | . 10022 | 9.15554 | .12692 | 7.87895 | c |
| 15 | . 07431 | 13.4566 | . 09189 | 10.8829 | . 10952 | 9.13093 | .12\%22 | 7.86064 | 45 |
| 16 | . 07461 | 13.4039 | . 09218 | 10.8183 | . 10981 | 9.106 6 | .12\%51 | 7.84242 | 44 |
| 17 | . 07490 | 13.3515 | . $092 \times 47$ | 10.8139 | . 11011 | 9.08211 | . 12781 | 7.82428 | 43 |
| 18 | . 07519 | 13.2996 | . 09277 | 10.7797 | . $110<0$ | 9.05\% 9 | . 12810 | 7.80622 | 42 |
| 19 | . 07548 | 13.2480 | . 09306 | 10.7457 | .11c\% 0 | $9.033 \% 9$ | . 12840 | 7.78825 | 41 |
| 20 | . 07578 | 13.1969 | . 09335 | 10.7119 | . 11099 | 9.60983 | . 12869 | 7.7.035 | 40 |
| 21 | . 07607 | 13.1461 | . 09365 | 10.6783 | . 11128 | 8.98598 | . 12899 | 7.75254 | 39 |
| 22 | . 07636 | 13.0958 | . 09304 | 10.6450 | . 11158 | 8.96227 | .12929 | 7.73480 | , |
| 23 | . 07665 | 13.0458 | . 09423 | 10.6118 | . 11187 | 8.93867 | . 12958 | 7.71715 | 37 |
| 24 | . 07695 | 12.9962 | . 00453 | 10.5789 | . 11217 | 8.91520 | . 12988 | 7.69957 | 36 |
| 2 | . 07724 | -12.9469 | . 09482 | 10.5462 | . 11246 | 8.89185 | . 13017 | 7.68208 | 35 |
| 26 | . $07 \% 5$ | 12.8981 | . 09511 | 10.5136 | .112\%6 | 8.86862 | . 13047 | 7.66466 | 34 |
| 27 | . 07782 | 12.8196 | . 09541 | 10.4813 | . 11305 | 8.81551 | . 13076 | 7.64732 | 33 |
| 28 | . 07812 | 12.8014 | .09570 | 10.4491 | . 11335 | 8.82252 | . 13106 | 7.63005 | 32 |
| 29 | . 07841 | 12.7536 | . 08600 | 10.41\%2 | . 11364 | 8.79964 | . 13136 | 7.61287 | 31 |
| 30 | . 07870 | 12.7062 | . 09629 | 10.3854 | . 11394 | 8.77689 | . 13165 | $7.595 \%$ | 30 |
| 31 | . 07899 | 12.6591 | . 09658 | 10.3538 | . 11423 | 8.75425 | . 13195 | $7.5 \% 8 \% 2$ | 29 |
| 32 | . 07029 | 12.6124 | . 09638 | 10.3224 | .11452 | 8.73172 | . 13221 | 7.561\% 6 | 28 |
| 33 | . 07058 | 12.5660 | . 09717 | 10.2913 | . 11482 | 8.70931 | . 13254 | 7.54487 | 27 |
| 34 | . 07987 | 12.5199 | . 09746 | 10.2602 | . 11511 | 8.68\%01 | . 13281 | 7.52806 | 26 |
| 35 | . 08017 | 12.4742 | . 0976 | 10.2294 | . 11541 | 8.66482 | . 13313 | 7.51132 | 25 |
| 36 | . 08046 | 12.4288 | . 09805 | 10.1988 | . 11570 | 8.642\% 5 | . 13343 | 7.49465 | 24 |
| 37 | . 08075 | 12.3838 | . 09834 | 10.1683 | . 11600 | 8.620\%8 | . 13372 | 7.47806 | 3 |
| 38 | . 08104 | 12.3390 | . 09864 | 10.1381 | . 11629 | 8.59893 | . 13402 | 7.46154 | 2 |
| 39 | . 08134 | 12.2946 | . 09893 | 10.1080 | . 11659 | $8.5 \% 18$ | . 13432 | '7.44509 | 2 |
| 40 | . 08163 | 12.2505 | . 09923 | 10.0780 | . 11688 | 8.55555 | . 13461 | 7.42871 | 20 |
| 41 | . 08192 | 12.2067 | . 09952 | 10.0483 | . 11718 | 8.53402 | . 13491 | 7.41240 | 19 |
| 42 | .08221 | 12.1632 | . 09981 | 10.0187 | . 11747 | 8.51259 | . 13521 | 7.39616 | 18 |
| 43 | .08251 | 12.1201 | . 10011 | 9.98931 | . 11777 | 8.49128 | . 13550 | 7.37999 | 7 |
| 44 | . 08280 | 12.0\%\%2 | . 10040 | 9.96007 | . 11806 | 8.47007 | . 13550 | 7.36289 | 16 |
| 45 | . 08309 | 12.0346 | . 10069 | 9.93101 | . 11836 | 8.44896 | . 13609 | 7.34786 | 15 |
| 46 | . 08339 | 11.9923 | . 10099 | 9.90211 | . 11865 | 8.42\% 95 | .13639 | 7.33190 | 1 |
| 47 | . 08368 | 11.9504 | . 10128 | 9.87338 | . 11895 | 8.40\%05 | . 13669 | \%. 31600 | 13 |
| 48 | . 08397 | 11.9087 | . 10158 | 9.84482 | . 11924 | 8.38625 | . 13698 | 7.30018 | 12 |
| 49 | . 08427 | $11.86 \% 3$ | . 10187 | 9.81641 | . 11954 | 8.36555 | .13728 | 7.28442 | 11 |
| 50 | . 08456 | 11.8262 | . 10216 | 9.78817 | . 11983 | 8.34496 | . 13758 | 7.26873 | 10 |
| 51 | . 08485 | 11.7853 | . 10246 | 8.76009 | .12018 | 8.32446 | . 13787 | 7.25310 | 9 |
| 52 | . 08514 | 11.7448 | . $102 \%$ | 9.73217 | . 12042 | 8.30406 | . 13817 | 7.23754 |  |
| 5 | . 08544 | 11.6045 | . 10305 | 9.70441 | . $120 \%$ | 8.28376 | . 13846 | 7.22204 |  |
| 54 | . $085 \% 3$ | 11.6645 | . 10334 | 9.67680 | .12i01 | 8.26355 | . 13876 | 7.20661 |  |
| 55 | . 08602 | 11.6248 | . 10363 | 9.64935 | . 12131 | 8.24345 | . 13906 | 7.19125 | 5 |
| 56 | . 08632 | 11.5853 | . 10393 | 9.62205 | . 12160 | 8.22344 | . 13935 | 7.17594 |  |
| 57 | . 08661 | 11.5461 | . 10422 | 9.59490 | . 12190 | 8.20352 | . 13965 | 7.16071 | 3 |
| 58 | . 08690 | 11.5072 | . 10452 | $9.56 \% 91$ | .12219 | 8.183\%0 | . 13995 | 7.14553 | 2 |
| 59 | . 08720 | 11.4685 | . 10481 | 9.54106 | . 12249 | 8.16398 | . 14024 | 7.13042 | , |
| 60 | . 08749 | 11.4301 | . 10510 | 9.51436 | 12278 | 8.14435 | 14054 | 7.11537 | 0 |
|  | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang |  |
|  | $85^{\circ}$ |  | 84 |  | $83^{\circ}$ |  | 82 ${ }^{\circ}$ |  |  |


|  | $8^{\circ}$ |  | O |  | $10^{\circ}$ |  | $11^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cota |  | Co |  | g | Tang | Cotang |  |
|  | 0 . 140 | 7.11537 | . 15 | 6.3 | . 17 | 5.67128 | . 1 | 5.14455 | 60 |
|  | 1 . 14 | 7.1003 |  | 6.3 | . 17 | 5.6616 |  | 5.13658 | 59 |
|  | 2.14113 | 7.08546 | . 158 | 6. 29007 | . 17693 |  | 194 | 5.12862 |  |
|  | . 14143 | 7.07059 | . 1592 | 6.27829 | . 17723 | 5.64248 | . 19359 | 5.12069 |  |
|  | . 14173 | 7.05579 | . 15958 | 6.26655 | . 17753 | 5.63295 | . 19559 | 5.11279 | 56 |
|  | .14203 | 7.04105 | . 1598 | 6.25486 | . 1778 | 5.62344 | . 19589 | 5.10490 |  |
|  | . 14232 | 7.02637 | . 16017 | 6.24331 | . 17813 | 5.61397 | . 19619 | 5.09704 |  |
|  | . 14262 | 6.91174 | . 16047 | 6.23160 | . 17843 | 5.60452 | . 19649 | 5.08921 |  |
|  | . 14201 | 6.99718 | . 16077 | 6.22003 | . 17873 | 5.59511 | . 19680 | 5.08139 |  |
|  | . 14321 | 6.98268 | . 1610 | 6.20851 | . 17903 | 5.585 | . 19710 | 5.07360 |  |
| 0 | . 1435 | 6.96823 | . 1 | 6.19703 | . 17933 | 5.576 | . 197 |  |  |
| 1 | . 14381 | 6. | . 11 | 6.18 | . 17963 | 5.56\%06 | . 19770 | 5.05809 |  |
| 12 | . 14410 | 6.93952 | . 16100 | 6.17419 | . 17993 | 5.55777 | . 19801 | 5.05037 | 48 |
|  | . 14440 | 6.22525 | 1622 | 6.10283 | . 18023 | 5.54851 | . 19831 | 5.04267 | 47 |
| 14 | . 14470 | 6.91104 | .1625 | 6.15151 | . 18053 | 5.53927 | . 19861 | 5.03499 | 46 |
|  | . 14499 | 6.89688 | . 1628 | 6.14023 | . 18053 | 5.53007 | . 19891 | 5.02734 |  |
| 0 | . 14529 | 6.832\%8 | 1631 | 6.12899 | . 18113 | 5.52090 | . 1992 | 5.01971 |  |
| 17 | . 14559 | 6.85874 | . 1634 | 6.11779 | . 18143 | 5.51176 | . 199 | 5.01210 |  |
|  | . 1458 | 6.854 | $163 \% 6$ | 6.106 | . 18173 | 5.50264 | . 199 | 5.00451 | 41 |
| 19 | . 14618 | 6.84082 | . 1640 | 6.09552 | . 18203 | 5.43356 | . 20012 | 4.99695 | 41 |
| 20 |  | 6.82694 | . 16 | 6.08444 | . 1823 | 5.484 | . 2004 |  | 40 |
| 1 | . 1467 | 6.81312 | . 16 | 6.073 | 82 | 5.4 | 200 | 4.98188 |  |
|  | . 14707 | 6.79336 | . 16495 | 6.06240 | . 18293 | 5.46648 | . 20103 | 4.97438 |  |
|  | . 14737 | 6.73564 | . 1652 | 6.05143 | .18323 | 5.45751 | . 2013 | 4.96690 |  |
|  | . 14767 | 6. 77199 | . 1655 | 6.04051 | . 18353 | 5.44857 | . 20164 | 4.95945 | 36 |
|  | .14\%9 | 6.75838 | . 1658 | 6.02962 | . 18384 | 5.43960 | . 20194 | 4.95201 |  |
|  | . 14826 | 6.74483 | 16615 | 6.0187 | . 18414 | 5.430 | . 2022 | 4.94460 |  |
|  | . 14356 | 6.73133 | . 16645 | 6.00797 | . 18444 | 5.42192 | . 20254 | 4.93721 |  |
|  | . 14381 | 6.7178 | . 16674 | 5.99720 | . 18474 | 5.413 | . 202 | 4.92984 |  |
|  | . 1415 | 6.70450 | 1670 | 5.98646 | . 18504 | 5.40429 | . 203 | 4.92249 |  |
| 20 | . 14945 | 6.69 | 1673 | 5.97576 | . 1853 | 5.395 | . 203 | 4.91516 |  |
|  |  |  |  |  | . 1856 | 5.3 | 203 |  | 29 |
|  | . 15005 | 6.66463 | . 16794 | 5.95448 | . 18594 | 5.378 | . 204 | 4.90056 |  |
|  | . 15034 | 6.651 | .1682 | 5.94390 | . 18624 | 5.369 | . 204 | 4.89330 |  |
|  | . 1506 | 6.6383 | . 1685 | 5.9330 | . 1865 | 5.36070 | . 2040 | 4.88605 | 26 |
|  | . 1500 | 6.62523 | . 1688 | 5.922 | . 1868 | 5.352 | . 204 | 4.87882 |  |
|  | . 15124 | 6.61219 | 16914 | 5.91236 | . 1871 | 5.34345 | . 20527 | 4.87162 |  |
|  | . 15153 | 6.53321 | . 1694 | 5.90191 | . 18745 | 5.33487 | . 205 | 4.86444 |  |
|  | . 15183 | 6.536 | .169\% | 5.89151 | . 187 | 5.326 | . 205 | 4.85727 |  |
| O | .15213 | 6.57339 | .17094 | 5.88114 | . 1880 | 5.317 | . 20618 | 4.85013 |  |
| 40 | . 15 |  |  | 5.87080 | . 18835 | 5. | . 206 | 4.8 |  |
|  |  |  |  |  |  | 5.30 | . 20679 |  |  |
|  | . 15302 | 6.5350 | 17033 | 5.85024 | . 1889 | 5.29235 | . 20709 | 4.82882 |  |
|  | . 15332 | 6.52234 | . 17123 | 5.84001 | . 1892 | 5.28393 | . 2073 | 4.82175 | 17 |
|  | . 15362 | 6.50970 | . 1715 | 5.8298 | . 1895 | 5.27553 | . 207 | 4.81471 | 16 |
|  | . 15391 | 6.49710 | . 17183 | 5.8106 | . 18986 | E.20715 | . 2080 | 4.80769 |  |
|  | . 15421 | 6.48456 | . 17213 | 5.80953 | . 19016 | 5.25850 | . 2083 | 4.80068 |  |
| 47 | . 15451 | 6.47206 | . 17243 | 5.7994 | . 19046 | 5.250 | . 2086 | 4.79370 |  |
|  | . 1548 | 6.45961 | -1723 | 5.78933 | . 190 Í | 5.24218 | 20891 | 4.786 |  |
|  | . 15511 | 6.44720 | . 17303 | 5. 77936 | . 19106 | 5.23391 | . 20921 | 4.77978 | 11 |
| 50 | . 15540 | 484 | 173 | 5 | . 1913 | 5.225 | . 2095 | 4.772 | 10 |
|  |  | 6.42 |  |  |  | 5.21 | 20s |  |  |
|  | . 560 | 6.41020 | . 17393 | 5.74949 | . 1919 | 5.20925 | . 21013 | 4.75906 |  |
|  | . 1563 | 6.3980 | . 17423 | 5.73960 | . 19227 | 5.20107 | . 21043 | 4.75219 |  |
|  | -1506 | 6.38587 | . 1745 | 5.7290" | . 1925 | 5.19293 | . 21073 | 4.74 |  |
|  | . 15689 | 6.37374 | . 17483 | 5.71992 | . 19287 | 5.18480 | . 21104 | 4.73851 |  |
|  | . 15719 | 6.36165 | . 17513 | 5.71013 | . 19317 | 5.1767 | . 21134 | 4.73170 |  |
|  | . 15749 | 6.31961 | . 17543 | 5. 7003 | . 19347 | 5.168 | . 21164 | 4.77490 |  |
|  | . 15789 | 6.33761 | . 1757 | 5.6906 | . 1937 | 5.160 | . 2119 | 1.71813 | 1 |
|  | . 15809 | 6.32566 | . 17603 | 5.680 | . 19408 | 5.152 | . 2122 | 4.71137 | 1 |
| 6 | . 15838 | 6.31375 | 17633 | 5.67128 | . 19438 | 5.1445 | . 21256 | 4.70463 | 0 |
|  | Cotang | Tang | Cotang | Tang | ng | Tang | an | Tan |  |
|  |  |  |  |  |  |  |  |  |  |


| , | $12^{\circ}$ |  | $13^{\circ}$ |  | $14^{\circ}$ |  | $15^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| 0 | . 21256 | 4.70463 | . 23087 | 4.33148 | . 24933 | 4.01078 | . 26795 | 3.73205 | 60 |
|  | . 21286 | 4.69791 | . 23117 | 4.32573 | . 24964 | 4.00582 | . 26826 | 3.72771 | 59 |
| 2 | . 21316 | 4.69121 | . 23148 | 4.32001 | . 24995 | 4.00086 | . 26857 | 3. ${ }^{2} 23338$ | 58 |
| 3 | . 21347 | 4.68452 | . 23179 | 4.31430 | . 25026 | 3.99592 | . 26888 | 3.7190' | 57 |
| 4 | . 21377 | 4.67786 | . 23209 | 4.30860 | . 25056 | 3.99099 | . 26920 | 3.71476 | 56 |
| 5 | . 21408 | 4.67121 | . 23240 | 4.30291 | . 25087 | 3.98607 | . 26351 | 3.71046 | 55 |
| , | . 21438 | 4.66458 | . 23271 | 4.29724 | . 25118 | 3.98117 | . 26082 | 3.70616 | 51 |
| 7 | . 21469 | 4.65797 | . 23301 | 4.29159 | . 25149 | 3.97637 | . 27013 | 3.70188 | 53 |
| 8 | . 21499 | 4.65138 | . 23332 | 4.28595 | . 25180 | 3.97139 | . 27044 | 3.69761 | 53 |
| 9 | . $215 \% 9$ | 4.64480 | . 23363 | 4.28032 | . 25211 | 3.96651 | .2\%0\%6 | 3.69335 | 51 |
| 10 | . 21560 | 4.63825 | . 23393 | 4.27471 | . 25242 | 3.96165 | . 27107 | 3.68909 | 50 |
| 11 | . 21590 | 4.63171 | . 23424 | 4.26911 | . 25273 | 3.95680 | . 27138 | 3.68485 | 49 |
| 12 | . 21621 | 4.62518 | . 23455 | 4.26352 | . 25304 | 3.95196 | . 27169 | 3.68061 | 48 |
| 13 | . 21651 | 4.61868 | . 23485 | 4.25795 | . 25335 | $3.94{ }^{\prime} 13$ | . 27201 | 3.6\%638 | 47 |
| 14 | . 21682 | 4.61219 | . 23516 | 4.25239 | . 25366 | 3.94232 | . 27232 | 3.67217 | 46 |
| 15 | . 21712 | 4.60572 | . 23547 | 4.24685 | . 25397 | 3.93751 | . 27263 | $3.66 \% 96$ | 45 |
| 16 | . 21743 | 4.59927 | . 23578 | 4.24132 | . 25488 | 3.93271 | . 27294 | $3.663 \% 6$ | 44 |
| 17 | . 21773 | 4.59283 | . 23608 | 4.23580 | . 25459 | 3.92793 | . 27326 | 3.65957 | 43 |
| 18 | . 21804 | 4.58041 | . 23639 | 4.23030 | . 25490 | 3.92316 | . 27307 | 3.65538 | 42 |
| 19 | . 21834 | 4.58001 | . 23670 | 4.22481 | . 25521 | 3.91839 | . 27358 | 3.6512i | 41 |
| 20 | . 21864 | 4.57363 | . 23700 | 4.21933 | . 25552 | 3.91364 | . 27419 | 3.64705 | 40 |
| 21 | . 21895 | 4.56726 | . 23731 | 4.21387 | . 25583 | 3.90890 | . 27451 | 3.64289 | 39 |
| 22 | . 21225 | 4.56091 | . 23762 | 4.20842 | . 25614 | 3.90417 | . 27482 | 3.63874 | 38 |
| 23 | . 21056 | 4.55158 | . 23793 | 4.20238 | . 25645 | 3.89945 | . 27513 | 3.63461 | 37 |
| 24 | . 21986 | 4.54826 | . 23823 | 4.19756 | . 25676 | 3.89474 | . 27545 | 3.63048 | 36 |
| 25 | . 22017 | 4.54196 | . 23854 | 4.19215 | . 25707 | 3.89004 | . 27576 | 3.62636 | 35 |
| 20 | . 22047 | 4.53568 | . 23885 | 4.18675 | . 25738 | $3.885 \pm 6$ | . 27607 | 3.62224 | 34 |
| 27 | . 22078 | 4.53941 | . 23916 | 4.18137 | . 25769 | 3.88068 | . 27638 | 3.61814 | 33 |
| 23 | . 22108 | 4.53316 | . 23946 | 4.17600 | . 25800 | 3.87601 | . 2660 | 3.61405 | 32 |
| 23 | . 22139 | 4.51693 | . 23977 | 4.17064 | . 25831 | 3.87136 | . 27701 | 3.60996 | 31 |
| 30 | . 22169 | 4.51071 | . 24008 | 4.16530 | . 25862 | 3.86671 | . 27732 | 360588 | 30 |
| 31 | .22200 | 4.50451 | . 24039 | 4.15997 | . 25893 | 3.86208 | . 27764 | 3.60181 | 29 |
| ¢2 | . 22331 | 4.49832 | . 21069 | 4.15465 | . 25924 | 3.85745 | . 27495 | 3.59775 | 28 |
| 33 | . 22261 | 4.49215 | . 21100 | 4.14934 | . 25955 | 3.85284 | . 27826 | 3.59370 | 7 |
| 34 | .23292 | 4.48600 | . 24131 | 4.14405 | . 25986 | 3.84824 | . 27858 | 3.58966 | 26 |
| 35 | .22322 | 4.47986 | . 21162 | 4.13877 | . 26017 | 3.84364 | . 27889 | 3.58562 | 25 |
| 36 | . 22353 | 4.47374 | . 24193 | 4.13350 | . 26048 | 3.83906 | . 27921 | 3.58160 | 24 |
| 37 | . 23383 | 4.46764 | . 24223 | 4.12825 | . 26079 | 3.83412 | . 27952 | 3.57758 | 23 |
| 38 | . 22114 | 4.46155 | . 24254 | 4.12301 | . 26110 | 3.82992 | . 27983 | 3.57357 | 22 |
| 39 | . 22144 | 4.45548 | . 24285 | 4.11778 | . 26141 | 3.82537 | . 28015 | 3.56957 | 21 |
| 40 | . 22475 | 4.44942 | . 21316 | 4.11256 | . $261 \%$ | 3.82083 | . 28046 | 3.56557 | 20 |
| 41 | . 22505 | 4.44338 | . 24347 | 4.10736 | . 26203 | 3.81630 | . 28077 | 3.56159 | 19 |
| 42 | . 22536 | 4.43735 | . 24377 | 4.10216 | . 26235 | 3.81177 | . 28109 | 3.55761 | 18 |
| 43 | . 22567 | 4.43134 | . 24408 | 4.09699 | . 26266 | 3.80726 | . 28140 | 3.55364 | 17 |
| 44 | . 22597 | 4.42534 | . 24439 | 4.09182 | . 26297 | 3.80276 | . $281 \%$ | -3.54968 | 16 |
| 45 | . 22628 | 4.41936 | . 24470 | 4.08666 | . 26328 | 3.79827 | -28203 | $3.545 \% 3$ | 15 |
| 46 | . 22058 | 4.41340 | . 24501 | 4.08152 | . 26359 | 3.79378 | . 28234 | 3.54179 | 14 |
| 47 | . 22689 | $4.40 \% 45$ | . 24532 | 4.07639 | . 26390 | 3.78931 | . 28266 | 3.53785 | 13 |
| 48 | . 22719 | 4.40152 | . 24562 | 4.07127 | . 26421 | 3.78485 | . 28297 | 3.53393 | 12 |
| 49 | .22\%50 | 4.33560 | . 21593 | 4.06616 | . 26459 | 3.78040 | . 28329 | 3.53001 | 11 |
| 50 | . 22781 | 4.38969 | . 21624 | 4.06107 | . 26483 | 3.77595 | . 28360 | 3.52609 | 10 |
| 5 | . 22811 | 4.38381 | . 24655 | 4.05599 | . 26515 | 3.77152 | . 28391 | 3.52219 | 9 |
|  | . 22342 | 4.37793 | . 24686 | 4.05092 | . 26546 | 3.76709 | . 28423 | 3.51829 |  |
| 53 | . 228872 | 4.37207 | . 24717 | 4.04586 | . 26577 | 3.76268 | . 28454 | 3.51441 |  |
| 54 | . 22903 | 4.36633 | . 21747 | 4.04081 | . 26608 | 3.75828 | . 28486 | 3.51053 |  |
| 2 | . 22934 | 4.36040 | . 21778 | 4.03578 | . 26639 | 3.75388 | . 28517 | 3.50666 | 5 |
| E | . 22964 | 4.35459 | . 24809 | 4.03076 | . 26670 | 3.74950 | . 28549 | 3.50279 | 4 |
| 57 | . 22995 | 4.34879 | . 24840 | 4.02574 | . 26701 | 3.74512 | . 28580 | 3.49894 | 3 |
| 58 | . 23026 | 4.34300 | . 21871 | $4.020 \% 4$ | . 26733 | 3.74075 | . 28612 | 3.49509 | 2 |
| 59 | . 23056 | 4.33723 | . 21902 | $4.015{ }^{4} 6$ | . 26764 | 3.73640 | . 28643 | 3.49125 | 1 |
| 60 | . 23087 | 4.33148 | 4933 | $4.010 \% 8$ | 2f\%95 | 3.73205 | 2866 | 48،41 | 0 |
|  | Cotang | Tang | Cotang | Tang | Cotan | Tang | otan | Tang |  |
|  | $77^{\circ}$ |  | $76^{\circ}$ |  | $75^{\circ}$ |  | $74^{\circ}$ |  |  |


|  | $16^{\circ}$ |  | $17{ }^{\circ}$ |  | 18 |  | 19 |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| 0 | . 28 | 3.48741 | . 30 | 3.27085 | . 32492 | 3.07768 | . 34433 | 2.90421 | 60 |
|  | . 28706 | 3.48359 | 30 | 3.26745 | . 3252 | 3.07464 | . 34465 | 2.90147 | 59 |
|  | . 28738 | 3.47977 | . 30637 | 3.26406 | . 32556 | 3.07160 | . 3449 | 2.89873 |  |
| 3 | . 28769 | 3.47596 | . 30669 | 3.26067 | . 32588 | 3.06857 | . 3453 | 2.89600 |  |
|  | . 28800 | 3.47216 | . 30700 | 3.2572 | . 32621 | 3.06554 | . 345 | 2.89327 |  |
| 5 | . 28832 | 3.46837 | . 30732 | 3.25392 | . 32653 | 3.06252 | . 345 | 2.89055 |  |
| 6 | . 28864 | 3.4645 | . 3076 | 3.2505 | . 3268 | 3.05950 | . 3462 | 2.88783 |  |
| 7 | . 288895 | 3.46080 | .3079 | 3.24719 | . 32717 | 3.05649 | . 34661 | 2.88511 |  |
| 8 | .28997 | 34503 |  | 3.24383 | . 32749 | 3.05349 | . 3469 | 2.88240 |  |
| 10 | . 289958 | 3.153 | . 308080 | 3.2404 | . 327882 | 3.05049 |  | 2.87970 |  |
| 10 | . 28990 | 3.4 | . 3089 | $3.23$ | . 32814 | $3.04 \% 49$ | . 34 | 7700 |  |
| 11 | . 2 |  | 0023 | 3.2338 | . 32846 |  | . 34791 | 2.87430 |  |
| 12 | . 29053 | 3.44202 | . 0995 | 3.2 | . 32 | 3.041 | . 348 |  |  |
| 13 | . 29084 | 3.43829 | . 3098 | 3.22715 | . 32911 | 3.03854 | . 348 | 2.86892 |  |
| 14 | . 29116 | 3.434 | 3101 | 3.22384 | . 32943 | 3.03 3゙5 | . 348 | 2.86 |  |
| 15 | . 29147 | 3.43084 | . 31051 | 3.22053 | . 3297 | 3.03260 | . 3492 | 2.86356 |  |
| 16 | . 29179 | 3.42713 | . 3108 | 3.21722 | . 3300 | 3.02963 | . 3495 | 2.86089 |  |
| 17 | . 29210 | 3.42343 | . 31115 | 3.21392 | . 33040 | 3.02667 | . 3 | 2.85822 |  |
| 18 | . 29242 | 3.41973 | . 3114 | 3.2106 | . 33072 | 3.023\%2 | . 250 | 2.85555 |  |
| 19 | . 29274 | 3.4160 | 1178 | 3.20 | . 33104 | 3.02077 | 350 | 2.85289 |  |
| 20 | . 29305 | 3.412 | 1210 | 3.20 | . 33136 | 3.01 | 350 | . 8023 |  |
| 21 | . 2 |  |  |  |  |  |  |  |  |
|  | . 2936 | 3.405 | . 31 | 3.19 | . 3 | 3.01 |  |  |  |
| 23 | . 29400 | 3.40136 | . 3120 | 3.19426 | . 33233 | 3.00503 | . 351 | 2.84229 |  |
|  | . 29432 | 3.39771 | . 31338 | 3.10100 | . 3322 | 3.00611 | . 352 | 2.83965 |  |
| 25 | . 29463 | 3.39406 | . 31370 | 3.18775 | . 3323 | 3.00319 | . 35 | $2.83 \% 02$ |  |
| 26 | . 29495 | 3.39042 | . 31 | 3.1845 | . 33330 | 3.0008 | . $35 \% 3$ | 2.83439 |  |
| 27 | . 29526 | 3.38679 | . 314 | 3.1812 | . 33363 | 2.99738 | . 2531 | 2.83176 |  |
| 2 | . 29558 | 3.383 | . 314 | 3.17804, | . 33395 | 2.99 | . 353 | 2.82914 |  |
| 29 | . 29590 | 3.3795 | . 31408 | $\stackrel{0}{ }$ | . 334 | 2.991 | . 3537 |  |  |
| 30 | . 2962 | 3. | 31530 | 3 | . 3 | 2.98868 | . 35 |  |  |
| 31 | . 29 | 3. | . 315 | 3. | . 33492 | 2.98580 | . 3 |  |  |
|  | . 2968 | 3.3637 | . 315 | 3.16517 | . 3352 | 2.98292 | . 35 | 2.81870 |  |
| 3 | . 29716 | 3.30516 | . 3162 | 3.16197 | . 335 | 2.98004 | . 355 | 2.81610 |  |
|  | . 29748 |  | . 316 | 3.1587 | 335 | 2.97717 | 355 | 2.81350 |  |
| 3 | .29780 | 3.35800 | . 31690 | 3.155 | . 3362 | 2.97430 | . 355 | 2.81091 |  |
| 36 | . 2981 | 3.354 | . 3172 | 3.1524 | . 3363 | 2.97144 | . 3560 |  |  |
| 28 | . 2984 | 3.3508 | . 317 | 3.1492 | . 3368 | 2.96858 | . 356 | 2.80574 |  |
| 38 | . 29875 |  | . 317 | 3.146 | . 33718 | 2.00 | . 356 | 2.80316 |  |
| 39 | . 2990 | 3.3 |  | 3.1428 | 337 | 2.96288 | . 357 | 2.80059 |  |
| 40 | . 2993 | 3. | . 318 | 3.1397 | 378 | 2.06004 | . 357 | 2.79802 |  |
| 41 | . 29970 |  | . 318 | 3.13 | . 338 | 2.95 | . 357 |  |  |
|  | . 30001 | 3.333 | . 3191 | 3.1331 | . 338 | 2.95437 | . 358 | 2.79289 |  |
| 43 | . 3003 | 3.3296 | . 3194 | 3.13027 | . 33881 | 2.95155 | . 358 | 2.79033 |  |
| 45 | . 3006 | 3.32614 | .31001 | 3.12713 | . 33913 | 2.948\% | . 358 | 2. |  |
| 45 | . 30097 | 3.32264 | . 32010 | 3.12400 | . 33945 | 2.94591 | . 35904 | 2.78523 |  |
| , | . 30128 | 3. | . 20 | 3.12087 | . 33978 | 2.94309 | . 3593 | \%. 782 |  |
| 4. | . 30160 | 3.31565 | . 2207 | 3.117 | . 34010 | 2.94028 | . 3596 | 2.78014 |  |
| 48 | . 3019 | 3.31216 | . 3210 | 3.1146 | . 340 | 2.93748 | . 360 | 2.77 |  |
| 49 | . 30224 | 3.30368 | . 32139 | 3.1115 | . 340 | 2.93468 | . 360 | 2.7 |  |
| 50 | . 30255 | 3.30521 | . 3217 | 3.1084 | . 341 | 2.93189 | 360 | 2.75 | 10 |
| 51 |  |  |  |  |  |  | , |  |  |
| 52 | . 3031 | 3.22829 | . | 3.10223 | . 341 | 2.92632 | . 36134 | 2.6 |  |
|  | . 30351 | 3.29483 | . 3226 | 3.09914 | . 342 | 2.92354 | . 36167 | 2.76498 |  |
|  | . 30382 | 3.29139 | . 32299 | 3.096 | . 3423 | 2.920;6 | . 3619 | 2.76247 |  |
|  | . 30414 | 3.28795 | . 3233 | 3.09298 | . 34270 | 2.91799 | . 3623 | 2.75996 |  |
| 56 |  | 3.28452 | . 323 | 3.08991 | . 343 | 2.91523 | . 3 | 2.75 |  |
|  |  | 3.28109 | . 323 | 3.08 | . 343 | 2.91246 |  | 2.75 |  |
|  |  |  |  |  |  |  |  |  |  |
| 60 | . 30573 | 3.27085 | . 3249 | ${ }_{3}^{3.07768}$ |  | 2.90 | . 363 |  |  |
|  | Cotang | Tang |  | Tan | g | Tang | ang | Tan |  |
|  |  |  |  |  |  |  |  |  |  |


|  | $20^{\circ}$ |  | 21* |  | 220 |  | $23^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang ${ }^{\circ}$ | Cotang |  |
| 0 | . 36397 | 2.74748 | . 38386 | 2.60509 | . 40403 | 2.47509 | . 42447 | 2.35585 | 60 |
| 1 | . 36430 | 2.74499 | . 38420 | 2.60283 | . 40436 | 2.47302 | . 42482 | 2.35395 | 59 |
| 2 | . 36463 | 2.74251 | .38453 | 2.60057 | . 40470 | 2.47095 | . 42516 | 2.35205 | 58 |
| 3 | . 36496 | 2.74004 | . 38487 | 2.59831 | . 40504 | 2.46888 | . 42551 | 2.35015 | 57 |
| 4 | . 36529 | 2.73756 | . 38520 | 2.59608 | . 40538 | 2.46682 | . 42585 | 2.34825 | 56 |
| 5 | . 36562 | 2.73509 | . 38553 | 2.59381 | . 40572 | 2.46476 | . 42619 | 2.34636 | 55 |
| 6 | . 36595 | 2.73863 | . 38587 | 2.59156 | . 40606 | 2.46270 | . 42654 | 2.34447 | 54 |
| 7 | . 36628 | 2.73017 | . 38620 | 2.58932 | . 40643 | 2.46065 | . 42688 | 2.34258 | 53 |
| 8 | . 36661 | 2.72771 | . 38654 | 2.58708 | . 40674 | 2.45860 | .42\% 2 | 2.34069 | 52 |
| - | . 36694 | 2.72526 | . 38687 | 2.58484 | . 40707 | 2.45655 | .42\% ${ }^{\text {\% }} 7$ | 233881 | 51 |
| 10 | . 36727 | 2.72281 | . 38721 | 2.58261 | . 40741 | 2.45451 | . 42791 | 2.33693 | 50 |
| 11 | . 36760 | 2.72036 | . 38754 | 2.58038 | . 40775 | 2.45246 | . 42826 | 2.33505 | 49 |
| 12 | . 36793 | 2.71792 | . 38787 | 2.57815 | . 40809 | 2.45043 | . 42860 | 2.33317 | 48 |
| 13 | . 36826 | 2.71548 | . 38881 | 2.57593 | . 40843 | 2.44839 | . 42894 | 2.33130 | 47 |
| 14 | . 36859 | 2.71305 | . 38854 | 2.57371 | . 40877 | 2.44636 | . 42929 | 2.32943 | 46 |
| 15 | . 36892 | 2.71062 | . 38888 | 2.57150 | . 40911 | 2.44433 | . 42963 | 2.32756 | 45 |
| 16 | . 33925 | 2.70819 | . 38921 | 2.56928 | . 40945 | 2.44230 | . 42998 | 2.325\%0 | 44 |
| 17 | . 36958 | 2.70577 | . 38955 | 2.56707 | . 40979 | 2.44027 | . 43032 | 2.32383 | 43 |
| 18 | . 36991 | 2.70335 | . 38988 | 2.56487 | . 41013 | 2.43825 | . 43067 | 2.32197 | 42 |
| 19 | . 37024 | 2.70094 | . 39022 | 2.56256 | . 41047 | 2.43623 | . 43101 | 2.32012 | 41 |
| 20 | . 37057 | 2.69853 | . 89055 | 2.56046 | . 41081 | 2.43422 | . 43136 | 2.31826 | 40 |
| 21 | . 37090 | 2.69612 | . 39089 | 2.55827 | . 41115 | 2.43220 | . 43170 | 2.31641 | 39 |
| 22 | . 37123 | 2.63371 | . 39122 | 2.55608 | . 41149 | 2.43019 | . 43205 | 2.31456 | 38 |
| 23 | . 37157 | 2.63131 | . 39156 | 2.55389 | . 41183 | 2.42819 | . 43239 | 2.31271 | 37 |
| 24 | . 37190 | 2.63392 | . 39190 | 2.55170 | . 41217 | 2.42018 | . 43274 | 2.31086 | 36 |
| 25 | . 37223 | 2.63653 | . 39223 | 2.54952 | . 41251 | 2.42418 | . 43308 | 2.30902 | 35 |
| 26 | . 37256 | 2.63414 | . 39257 | 2.54734 | . 4125 | 2.42218 | . 43343 | 2.30718 | 34 |
| 27 | . 3 \%289 | 2.63175 | . 39290 | 2.54516 | . 41319 | 2.42019 | . 43378 | 2.30534 | 33 |
| 23 | . 37322 | $2.6703{ }^{\text {\% }}$ | . 39324 | 2.54299 | . 41353 | 2.41819 | . 43412 | 2.30351 | 1 |
| 23 | . 37355 | 2.6700 | . 33357 | 2.54082 | . 41337 | 2.41620 | . 43447 | 2.30167 | 31 |
| 30 | . 37388 | 2.67462 | . 39891 | 2.53865 | . 41421 | 2.41421 | . 43481 | 2.29984 | 30 |
| 31 | . 37422 | $2.67 \% 25$ | . 39425 | 2.53648 | . 41455 | 2.41223 | . 43516 | 2.29801 | 29 |
| 32 | . 37455 | 2.63939 | . 33458 | 2.53432 | . 41400 | 2.41025 | . 43550 | 2.29619 | 28 |
| 33 | . 37488 | 2.60752 | . 33492 | 2.53217 | . 41524 | 2.40827 | . 43585 | 2.29437 | 27 |
| 34 | . 37521 | 2.63516 | . 39526 | 2.53001 | . 4155 | 2.40629 | . 43620 | 2.29254 | 26 |
| 35 | . 3754 | 2.65281 | . 39559 | 2.527\%6 | . 41522 | 2.40432 | . 43654 | 2.29073 | 25 |
| 36 | . 37588 | 2.66046 | . 39593 | 2.525\%1 | .416~6 | 2.40235 | . 43689 | 2.28891 | 24 |
| 37 | . 3761 | 2.65811 | . 39626 | 2.52357 | . 41660 | 2.40038 | . 43724 | 2.28710 | 23 |
| 33 | . 37654 | 2.65576 | . 39660 | 2.52142 | . 41694 | 2.39841 | . 43758 | 2.28528 | 22 |
| 39 | . 37687 | 2.65342 | . 30094 | 2.519 9 | . 41778 | 2.39645 | .43793 | 2.28348 | 21 |
| 40 | . 37720 | 2.65109 | . 39787 | 2.51715 | . 41763 | 2.39449 | . 43828 | 2.28167 | 20 |
| 41 | . 3777 | 2.64873 | . 39761 | 2.51502 | . 41797 | 2.39253 | . 43862 | 2.27987 | 19 |
| 42 | . 3748 | 2.61042 | . 39795 | 2.51289 | . 41831 | 2.39058 | . 43897 | 227806 | 18 |
| 43 | . 37820 | 2.61410 | . 39829 | $2.510{ }^{6} 6$ | . 41865 | 2.38863 | . 43932 | 2.27626 | 17 |
| 44 | . 37853 | 2.64177 | . 39862 | 2.50864 | 41899 | 2.38668 | . 43966 | 2.27447 | 16 |
| 45 | . 37887 | 2.63045 | . 39896 | 2.50652 | . 41933 | 2.38473 | . 44001 | 2.27267 | 15 |
| 46 | . 37920 | 2.63714 | . 39950 | 2.50440 | . 41968 | 2.38279 | . 44036 | 2.27088 | 14 |
| 47 | . 37953 | 2.63483 | . 33963 | 2.50229 | . 42002 | 2.38084 | . 44071 | 2.26909 | 13 |
| 48 | . 37986 | 2.63252 | . 39997 | 2.50018 | . 42036 | 2.37891 | . 44105 | 2.26730 | 12 |
| 49 | . 33020 | 2.63021 | . 40031 | 2.49807 | . 42070 | 2.37697 | . 41140 | 2.26552 | 11 |
| 50 | . 38053 | 2.62791 | . 40065 | 2.49597 | . 42105 | 2.37504 | . 44175 | 2.26374 | 10 |
| 51 | . 38086 | 2.62561 | . 40098 | 2.49386 | . 42139 | 2.37311 | . 44210 | 2.26196 | 9 |
| 52 | . 38120 | 2.62332 | . 40132 | 2.49177 | . 42173 | 2.37118 | . 44214 | 2.26018 | 7 |
| 53 | . 38153 | 2.62103 | . 40166 | 2.48967 | . 42207 | 2.36925 | . 44279 | 2.25840 | , |
| 54 | . 33186 | 2.61874 | . 40200 | 2.48758 | . 42242 | 2.86733 | . 44314 | 2.25663 | 6 |
| 55 | . 38220 | 2.61646 | . 40234 | 2.48549 | . 42276 | 2.36541 | . 44349 | 2.25486 | 5 |
| 56 | . 38253 | 2.61418 | . 40267 | 248340 | . 42310 | 2.36349 | . 44384 | 2.25309 | 4 |
| 57 | . 38286 | 2.61190 | . 40301 | 2.48132 | . 42345 | 2.36158 | . 44418 | 2.25132 | 3 |
| 58 | . 38320 | 2.60963 | . 40335 | 2.47924 | . 42379 | 2.35967 | . 44453 | 2.24956 | 2 |
| 59 | . 38353 | 2.60736 | . 40369 | 2.47716 | . 42413 | 2.35776 | . 44488 | 2.24780 2.24604 | 1 |
| 60 | . 38386 | 2.60509 | 403 | 2.47509 | 42447 | 2.35585 | . 44523 | 2 | 0 |
| \% | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang |  |
|  |  |  |  |  |  |  |  |  |  |


|  | $24^{\circ}$ |  | $25^{\circ}$ |  | $26^{\circ}$ |  | $27^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tan | Co |  |  | Tang | C | Tang | g |  |
|  | . 445 | 2.24604 | . 4663 | 2.14 | . 48773 | 2.05030 | 00953 | 61 | 60 |
|  | . 4455 | 2.24428 | . 46666 | 2.1 | . 48809 | 2.04879 | 989 | 1.96120 | 5 |
|  | . 4459 | 2.24252 | .46\%02 | 2.1412 | . 48845 | $2.04 \% 28$ | . 51026 | 1.95979 |  |
|  | . 44627 | 2.24077 | . 46737 | 2.13963 | . 48881 | 2.04577 | . 51063 | 1.95838 |  |
|  | . 44662 | 2.23902 | .46772 | 2.13801 | . 48917 | 2.04426 | . 51099 | 1.95698 | 56 |
|  | . 44697 | 2.23727 | . 46808 | 2.13639 | . 48953 | 2,04226 | . 51136 | 1.95557 |  |
|  | . 44732 | 2.23553 | . 46843 | 2.13477 | . 4898 | 2.04125 | . 51173 | 1.95417 | 54 |
|  | . 4476 | 2.23378 | . 46879 | 2.13316 | . 4902 | 2.03975 | . 51209 | 1.95277 |  |
|  | . 448 | 2.232 | . 469 | 2.13154 | . 490 | 2.03 | . 51246 | 1.95 |  |
| 10 | . 448 | 2.228 | . 469985 | 2.12 | . 499134 | 2.0 | . 51319 |  | 51 |
| 11 | . 449 | 2.226 | 4,0 | 2.12 |  | 2.03 | . 513 |  |  |
|  | . 4494 | 2.22510 | . 40 | 2.125 | 49 | 2.03 | . 513 |  |  |
|  | . 4497 | 2.22337 | . 47092 | 2.12350 | . 4924 | $2.030 \% 8$ | . 514 | 1.94 | 4 |
|  | . 45012 | 2.22164 | . 47128 | 2.12190 | .492\% | 2.02929 | . 5146 | 1.943 | 46 |
|  | . 45047 | 2.21992 | . 47163 | 2.12030 | . 49315 | 2.02\%80 | . 51503 | 1.9416 |  |
|  | . 4508 | 2.21819 | . 4719 | 2.11871 | . 49351 | 2.02631 | . 51540 | 1.940 | 4 |
|  | . 45117 | 2.21647 | . 47234 | 2.11711 | . 49387 | 2.02483 | . 51577 | 1.93 |  |
|  | . 45152 | 2.21475 | .47270 | 2.11552 | . 49423 | 2.0233 | . 51614 | 1.93 | 42 |
|  | . 4518 | 2.21304 | .4705 | 2.11392 | . 49459 | 2.02187 | . 51651 | 1.93608 |  |
| 20 | . 4522 | 21132 | . 4734 | 2.11233 | - | 2.02039 | . 51688 | 1.93470 |  |
|  | . 4 | 2.20 | . 4 | 2.11 | 495 | 2.01 | . 517 |  |  |
|  | . 452 | 2.20790 | . 47412 | 2.10916 | . 495 C | 2.01 | . 51761 | 1.8 |  |
|  | . 453 | 2.20619 | . 47448 | 2.10758 | . 4960 | 2.01596 | . 51798 | 1.930 |  |
|  | . 4536 | -2.20449 | . 47483 | 2.10600 | . 49640 | 2.01449 | . 51835 | 1.92920 |  |
|  | . 45397 | 2.20278 | . 47519 | 2.10442 | . 49677 | 2.01302 | . $518 \% 2$ | 1.92782 |  |
|  | . 45432 | 2.20108 | . 4755 | 2.10284 | . 49713 | 2.01155 | . 51909 | 1.92645 |  |
|  | . 45467 | 2.19938 | . 4759 | 2.10126 | . 497 | 2.01008 | . 51946 | 1.92508 |  |
|  | . 45502 | 2.19769 | . 4762 | 2.09969 | . 497 | 2.00862 | . 51983 | 1.92371 |  |
|  | . 4553 | 2.19599 | . 4766 | 2.09811 | . 498 | 2.00715 | 2020 | 1.92 |  |
| 30 | . 45 | 2.19430 | . 4769 | 2.09654 | 49 | 2.00569 | . 52057 |  |  |
| 31 | . 4 | 2.19 | . 4773 | 2.09 | . 49 | 2.00423 | . 220 |  |  |
|  | . 456 | 2.1903 | . 4776 | 2.093 | . 49931 | 2.00277 | . 521 | 1.91 |  |
|  | . 45678 | 2.18923 | . 47805 | 2.09184 | . 49967 | 2.00131 | . 52168 | 1.91690 |  |
|  | . 45713 | 2.18755 | . 47840 | 2.09028 | . 5000 | 1.99986 | . 5220 | 1.91554 |  |
|  | .45\%48 | 2.18587 | . 41878 | 2.08872 | . 50040 | 1.99811 | . 52242 | 1.91418 |  |
| 30 | . 45784 | 2.18419 | . 47912 | 2.08716 | . 50076 | 1.09695 | . 52279 | 1.9128 | 2 |
| 37 | . 45819 | 2.18251 | . 47948 | 2.08560 | . 50113 | 1.05550 | . 52316 | 1.91147 |  |
|  | . 458 | 2.18084 | . 47984 | 2.08405 | . 50149 | 1.03406 | . 52353 | 1.91012 |  |
|  |  | 2.17916 | . 43019 | 2.03250 |  | 1. 59261 | 5239 |  |  |
| 40 | . 4592 | 2.17749 | . 43055 | 2.08094 | . 5022 | 1.93116 | . 5242 | 1.90741 |  |
|  | . 45 | 2.17 | . 4800 | 2.079 | . 502 | 1.98972 | . 224 |  | 19 |
|  | . | 2.17416 | . 43127 | 2.07785 | . 5029 | 1.98828 | . 25 | 1.90472 |  |
|  | . 46030 | 2.17249 | . 48163 | 2.07630 | . 50331 | 1.98684 | . 52538 | 1.90337 |  |
| 44 | . 46065 | 2.17083 | . 4819 | 2.07476 | . 5036 | 1.98540 | 52575 | 1.90203 |  |
|  | . 46101 | 2.10917 | . 48234 | 2.07321 | . 50404 | 1.98396 | . 52613 | 1.90069 |  |
|  | . 46136 | .2.16751 | . 48270 | 2.07167 | . 5044 | 1.98253 | . 52650 | 1.89935 |  |
|  | . 43171 | 2.16585 | . 48306 | 2.0\%014 | . 5047 | 1.98110 | . 52687 | 1.85801 |  |
|  | . 40206 | 2.16420 | . 48342 | 2.06860 | . 5051 | 1.97966 | . 5272 | 1.89667 |  |
| 5 | . 46242 | 2.16255 | . 48378 | 2.067 | 50 | 1.97823 | 5276 | 1.89533 |  |
| 50 | .46277 | 2.16090 | . 4841 | 2.0 |  | 1.97681 | . 5279 | 1.89400 |  |
|  |  | 2. |  | 2.06 | . 5062 | 1.975 | 528 |  |  |
|  | . 40348 | 2.15760 | . 43486 | 2.06247 | . 50660 | 1.97395 | . 5287 | 1.89133 |  |
|  | . 4638 | 2.15596 | . 43521 | 2.06094 | . 50696 | $1.9 \% 253$ | . 52910 | 1.89000 |  |
|  | . 46418 | 2.15432 | . 43357 | 2.05942 | .50733 | 1.97111 | . 52947 | 1.88867 |  |
|  | . 46454 | 2.15268 | . 48593 | 2.05790 | .50769 | 1.96969 | . 52985 | 1.88734 |  |
| 56 | . 46489 | 2.15104 | . 48629 | 2.05637 | . 50806 | 1.96827 | . 53022 | 1.88602 |  |
|  | . 4655 | 2.14940 | . 48665 | 2.05485 | . 50843 | 1.96685 | . 53059 | 1.88469 |  |
|  | . 46560 | 2.14777 | . 48701 | 2.05333 | . 50879 | 1.96544 | . 53096 | 1.88333 |  |
|  | . 46595 | 2.14614 | . 48737 | 2.05182 | . 50916 | 1.96402 | . 53134 | 1.88205 |  |
| 60 | 46631 | 2.14451 | 487\%3 | 2.05030 | . 50953 | 1.9626 | 53171 | 1.88073 | 0 |
|  | Cotang | Tang | Cotang | Tang | tang | Ta | Cotan | Tan |  |
|  |  |  |  |  |  |  |  |  |  |


|  | $28^{\circ}$ |  | $29^{\circ}$ |  | $30^{\circ}$ |  | $31^{\circ}$ |  | , |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| 0 | . 53171 | 1.88073 | . 55431 | 1.80405 | . 57735 | 1.73205 | . 60086 | 1.66428 | 60 |
|  | . 53208 | 1.87941 | . 55469 | 1.80281 | . 57774 | 1. 23089 | . 60126 | 1.66318 | 59 |
| 2 | . 53246 | 1.87809 | . 55507 | 1.80158 | . 57813 | 1.72973 | . 60165 | 1.66209 | 58 |
| 3 | . 53283 | $1.8{ }^{\text {r }} 677$ | . 55545 | 1.80034 | . 57851 | 1.72857 | . 60205 | 1.66099 | 57 |
| 4 | . 53320 | $1.8 \% 546$ | . 55583 | 1. 7.7911 | . 57890 | 1.72741 | . 60245 | 1.65990 | 56 |
| 5 | . 53358 | 1.87415 | . 55621 | 1.79788 | . 57929 | 1.72625 | . 60284 | 1.65881 | 55 |
| 6 | . 53395 | $1.8 \% 283$ | . 55659 | 1.79665 | . 57968 | 1. 72509 | . 60324 | $1.657 \% 2$ | 54 |
|  | . 53432 | 1.87152 | . 55697 | 1.79542 | . 58007 | 1.72393 | . 60364 | 1.65663 | 53 |
|  | . 53470 | 1.87021 | . 55736 | 1.79419 | . 58046 | 1.72278 | . 60403 | 1.65554 | 52 |
|  | . 53507 | 1.86891 | . 55774 | 1.79296 | . 58085 | 1.72163 | . 60443 | 1.65445 | 51 |
| 10 | . 53545 | 1.86760 | . 55812 | 1.79174 | . 58124 | 1.72047 | . 60483 | 1.65337 | 50 |
| 11 | . 53582 | 1.86630 | . 55850 | 1.79051 | . 58162 | 1.71932 | .60522 | 1.65228 | 49 |
| 12 | . 53620 | 1.86499 | . 55888 | 1.78929 | . 58201 | 1.71817 | . 60562 | 1.65120 | 48 |
| 13 | . 53657 | 1.86369 | . 55926 | 1.78807 | . 58240 | 1.71702 | . 60602 | 1.65011 | 47 |
| 14 | . 53694 | 1.86239 | . 55964 | 1.78685 | . 58279 | 1.71588 | . 60642 | 1.64903 | 46 |
| 15 | . 53732 | 1.86109 | . 56003 | 1.78563 | . 58318 | 1.71473 | . 60681 | 1.64795 | 45 |
| 16 | . 53769 | 1.85979 | . 56041 | 1.78441 | . 58357 | 1.71358 | . 60721 | 1.64687 | 44 |
| 17 | . 53807 | 1.85850 | . 56079 | 1.78319 | . 58396 | 1.71244 | . 60761 | 1.64579 | 43 |
| 18 | . 53844 | 1.85720 | .56117 | 1.78198 | . 58435 | 1.71129 | . 60801 | 1.64471 | 42 |
| 19 | . 53882 | 1.85591 | . 56156 | 1.78077 | . 58474 | 1.71015 | . 60841 | 1.64363 | 41 |
| 20 | . 53920 | 1.85462 | . 56194 | 1.77955 | . 58513 | 1.70901 | . 60881 | 1.64256 | 40 |
| 21 | . 53957 | 1.85333 | . 56232 | 1.77834 | . 58552 | 1.70787 | . 60921 | 1.64148 | 39 |
| 22 | . 53995 | 1.85204 | . $562 \% 0$ | 1.77713 | . 58591 | 1.70673 | . 60960 | 1.64041 | 38 |
| 23 | . 54032 | 1.85075 | . 56309 | 1.77592 | . 58631 | 1.70560 | . 61000 | 1.63934 | 37 |
| 24 | . 54070 | 1.81946 | . 56347 | 1.77471 | . 58670 | 1.70446 | . 61040 | 1.63826 | 36 |
| 25 | . 54107 | 1.84818 | . 56385 | 1.77351 | . 58709 | 1.70332 | . 61080 | 1.63719 | 35 |
| 26 | . 54145 | 1.84689 | . 56424 | 1.77230 | . 58748 | 1.70219 | . 61120 | 1.63612 | 34 |
| 27 | . 54183 | 1.84561 | . 56462 | 1.77110 | . 58787 | 1.70106 | . 61160 | 1.63505 | 33 |
| 28 | . 54220 | 1.81433 | . 56501 | 1.76990 | . 58826 | 1.69992 | . 61200 | 1.63398 | 32 |
| 29 | . 54258 | 1.84305 | . 56539 | 1.76869 | . 58865 | 1.69879 | . 61240 | 1.63292 | 31 |
| 30 | . 54296 | 1.81177 | . 56577 | 1.76\%49 | . 58905 | 1.69766 | . 61280 | 1.63185 | 30 |
| 31 | . 54333 | 1.84049 | . 56616 | 1.76629 | . 58944 | 1.69653 | . 61320 | 1.63079 | 29 |
| 32 | . 54371 | 1.83922 | . 56654 | 1.76510 | . 58983 | 1.69541 | . 61360 | 1.62972 | 28 |
| 33 | . 54409 | 1.83794 | . 56693 | 1.76390 | . 59022 | 1.69428 | . 61400 | 1.62866 | 27 |
| 34 | . 54446 | 1.83667 | . 56731 | 1.76271 | . 59061 | 1.69316 | . 61440 | 1.62760 | 26 |
| 35 | . 54484 | 1.83540 | . 56769 | 1.76151 | . 59101 | 1.69203 | . 61480 | 1.62654 | 25 |
| 36 | .54522 | 1.83413 | . 56808 | 1.76032 | . 59140 | 1.69091 | . 61520 | 1.62548 | 24 |
| 37 | . 54560 | 1.83286 | . 56846 | 1.75913 | . 59179 | 1.68979 | . 61561 | 1.62442 | 23 |
| 38 | . 54597 | 1.83159 | . 56885 | 1.75794 | . 59218 | 1.68866 | . 61601 | 1.62336 | 22 |
| 39 | . 54635 | 1.83033 | . 56923 | 1.756\% | . 59258 | 1.68754 | . 61641 | 1.62230 | 21 |
| 40 | . 54673 | 1.82906 | 56962 | 1.75556 | . 59297 | 1.68643 | . 61681 | 1.62125 | 20 |
| 41 | . 54711 | 1.82780 | . 57000 | 1.75437 | . 59336 | 1.68531 | . 61721 | 1.62019 | 19 |
| 42 | . 54748 | 1.82654 | . 57039 | 1.75319 | . 59376 | 1.68419 | . 61761 | 1.61914 | 18 |
| 43 | . 54786 | 1.82528 | . 57078 | 1.75200 | . 59415 | 1.68308 | . 61801 | 1.61808 | 17 |
| 44 | . 54924 | 1.82402 | . 57116 | 1.75082 | . 59454 | 1.68196 | . 61842 | 1.61703 | 16 |
| 45 | . 54862 | 1.82276 | . 57155 | 1.74964 | . 59494 | 1.68085 | . 61882 | 1.61598 | 15 |
| 46 | . 54900 | 1.82150 | . 57193 | 1.74846 | . 59533 | 1.67974 | . 61922 | 1.61493 | 14 |
| 47 | . 54938 | 1.82025 | . 57232 | 1.74728 | . $595 \% 3$ | 1.67863 | . 61962 | 1.61388 | 13 |
| 48 | . 54975 | 1.81899 | . 57271 | 1.74610 | . 59612 | 1.6\%'752 | . 62003 | 1.61283 | 12 |
| 49 | . 55013 | 1.81774 | . 57309 | 1.74492 | . 59651 | $1.6 \% 641$ | . 62043 | 1.61179 | 11 |
| 50 | . 55051 | 1.81649 | . 57348 | 1.74375 | . 59691 | 1.67530 | . 62083 | 1.61074 | 10 |
| 51 | . 55089 | 1.81524 | . 57386 | 1.74257 | . 59730 | 1.67419 | . 62124 | 1.60970 | 9 |
| 52 | . 55127 | 1.81399 | . $5 \sim 425$ | 1.74140 | . 59770 | 1.67309 | . 62164 | 1.60865 | 8 |
| 53 | . 55165 | 1.81274 | . 5 \% 464 | 1.74022 | . 59809 | 1.67198 | . 62204 | $1.60 \% 61$ | 7 |
| 54 | . 55203 | 1.81150 | . 57503 | 1.73905 | . 59849 | 1.67088 | . 62245 | 1.60657 | 6 |
| 55 | . 55241 | 1.81025 | . 5 \% 541 | 1.73788 | . 59888 | 1.66978 | . 62285 | 1.60553 | 5 |
| 56 | .552~9 | 1.80901 | . 57580 | 1.73671 | . 59928 | 1.66867 | . 62325 | 1.60449 | 4 |
| 57 | . 55317 | 1.80777 | . 57619 | 1.73555 | . 59967 | 1.66757 | . 62366 | 1.60345 | 3 |
| 58 | . 55355 | 1.80653 | . 57657 | 1.73438 | . 60007 | 1.66647 | . 62406 | 1.60241 | 2 |
| 59 | 55393 | 1.80529 | . 57696 | 1.73321 | . 60046 | 1.66538 | . 62446 | 1.60137 | 1 |
| 60 | 55431 | 1.80405 | . 57735 | 1.73205 | 60086 | 1.66428 | 62487 | 1.60033 | 0 |
|  | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang |  |
|  | $61^{\circ}$ |  | $60^{\circ}$ |  | 59* |  | $58^{\circ}$ |  |  |


|  | $32^{\circ}$ |  | 33 ${ }^{\circ}$ |  | $34^{\circ}$ |  | $35^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| $\overline{0}$ | . 62487 | 1.60033 | . 64941 | 1.53986 | . 67451 | 1.48256 | . 70021 | 1.42815 | $\overline{60}$ |
| 1 | . 62527 | 1.59930 | . 64982 | 1.53888 | . 67493 | 1.48163 | . 70064 | 1.42726 | 59 |
| 2 | . 62568 | 1.59826 | .65024 | 1.53791 | . 67536 | $1.480{ }^{\circ} 0$ | . 70107 | 1.42638 | 58 |
| 3 | . 62608 | 1.59723 | . 65065 | 1.53693 | . 67558 | 1.47977 | . 70151 | 1.42550 | 57 |
| 4 | . 62649 | 1.59620 | . 65106 | 1.53595 | . 67620 | 1.47885 | . 70194 | 1.42462 | 56 |
| 5 | . 62689 | 1.59517 | . 65148 | 1.53497 | . 67663 | 1.47792 | . 70238 | 1.42374 | 55 |
| ${ }_{6}^{6}$ | .62\%30 | 1.59414 | . 65189 | 1.53400 | . .67705 | 1.47699 | . 70281 | 1.42886 | 54 |
| 7 | .62770 | 1.59311 | . 65231 | 1.53302 | . 67748 | 1.47607 | . 70325 | 1.42198 | 53 |
| 8 | . 62811 | 1.59208 | .658in | 1.53305 | . 67790 | 1.47514 | . 70368 | 1.42110 | 52 |
| 9 | . 62852 | 1.59105 | . 65314 | 1.53107 | . 67832 | 1.47422 | . 70412 | 1.42022 | 51 |
| 10 | . 62892 | 1.59002 | . 65355 | 1.53010 | . 67885 | 1.47330 | . 70455 | 1.41934 | 50 |
| 11 | . 62933 | 1.58900 | . 65397 | 1.52913 | . 67917 | 1.47238 | . 70499 | 1.41847 | 49 |
| 12 | .62973 | 1.58797 | . 65438 | 1.52816 | . 6796 | 1.47146 | . 70542 | 1.41759 | 48 |
| 13 | . 63014 | 1.58695 | . 65480 | 1.52719 | . 68002 | 1.47053 | . 70586 | $1.4167 \%$ | 47 |
| 14 | . 63055 | 1.58593 | 65521 | 1.52622 | . 68045 | 1.46962 | . 70629 | 1.41584 | 46 |
| 15 | . 63095 | 1.58490 | . 65563 | 1.52525 | . 68088 | 1.46870 | . 70673 | 1.41497 | 45 |
| 16 | . 63136 | 1.58338 | . 65604 | 1.52429 | . 68130 | $1.467 \% 8$ | . 70717 | 1.41409 | 44 |
| 17 | . 63177 | 1.58286 | . 65646 | 1.52332 | . 68173 | 1.46686 | . 70760 | 1.41322 | 43 |
| 18 | . 63217 | 1.58184 | . 65688 | 1.52235 | . 68215 | 1.46595 | . 70804 | 1.41235 | 42 |
| 19 | .63258 | 1.58083 | . 65729 | 1.22139 | .682\%8 | 1.46503 | . 70948 | 1.41148 | 41 |
| 20 | .63299 | 1.05081 | .65761 | 1.52043 | . 68301 | 1.46411 | . 70801 | 1.41061 | 40 |
| 21 | . 63340 | 1.57879 | . 65813 | 1.51046 | . 68343 | 1.46320 | . 70935 | 1.40974 |  |
| 22 | . 63330 | 1.57743 | . 65854 | 1.51850 | . 68386 | 1.46229 | . 70979 | 1.40837 | 38 |
| 23 | . 63421 | 1.57676 | . 65896 | 1.51754 | . 68429 | 1.46137 | . 71023 | 1.40300 | 37 |
| 24 | . 63462 | 1.7575 | . 65938 | 1.51c\%8 | . 68471 | 1.46046 | . 71066 | 1.40714 | 36 |
|  | . 63503 | 1.57474 | . 65980 | 1.51562 | . 68514 | 1.45955 | . 71110 | 1.40627 | 35 |
| 26 | . 63544 | 1.57372 | . 66021 | 1.51466 | . 68557 | 1.45864 | . 71154 | 1.40540 | - |
| 27 | . 635884 | 1.57871 | . 66006 | 1.51370 | . 68600 | 1.45773 | . 71198 | 1.40454 | 33 |
|  | . 63625 | 1.57170 | . 66105 | 1.51275 | . 688642 | 1.45682 | . 71242 | 1.40367 | 32 |
| 29 | . 63666 | 1.57069 | . 66147 | 1.51179 | . 68685 | 1.45592 | . 71285 | 1.40281 | 31 |
| 30 | .63707 | 1.56969 | . 66189 | 1.51084 | . 68728 | 1.45501 | . 71329 | 1.40195 | 30 |
| 31 | . 63748 | 1.56868 | . 66230 | 1.50988 | . 68771 | 1.45410 | . 71373 | 1.40109 | 29 |
|  | . 63789 | 1.56767 | . 66272 | 1.50893 | . 68314 | 1.45320 | . 71417 | 1.400\% | 28 |
| ${ }_{3}^{33}$ | . 638380 | 1.56667 | . 66314 | 1.50797 | . 68857 | 1.45329 | . 71451 | 1.30936 | ${ }^{27}$ |
| 34 | . 63878 | 1.56566 | . 66356 | 1.50702 | . 68900 | 1.45139 | . 71505 | 1.33950 | 20 |
|  | . 63912 | 1.56466 | . 66398 | 1.50607 | . 68942 | 1.45049 | . 71549 | 1.39764 | 25 |
| 36 | . 63953 | 1.56366 | . 66440 | 1.50512 | . 68985 | 1.44958 | . 71593 | 1.39679 | 24 |
|  | . 63994 | 1.56265 | . 66482 | 1.50417 | . 69028 | 1.44868 | . 71637 | 1.39593 | 3 |
| 38 | . 64035 | 1.56165 | . 66554 | 1.50322 | . 69071 | $1.447{ }^{\text {P }}$ | . 71681 | 1.39507 | 22 |
| 39 | . 64076 | 1.56065 | . 66566 | 1.50228 | . 69114 | 1.44688 | . 71725 | 1.39421 | 21 |
| 40 | . 64117 | 1.55966 | . 66608 | 1.50133 | . 69157 | 1.44598 | . 71769 | 1.39336 | 20 |
| 41 | . 64158 | 1.55866 | . 66650 | 1.50038 | . 69200 | 1.44508 | . 71813 | 1.39250 | 19 |
| 42 | . 64199 | 1.55766 | . 66692 | 1.49944 | .69243 | 1.44418 | . 71857 | 1.30105 | 18 |
| 43 | . 64240 | 1.55666 | . 66734 | 1.49849 | . 69.988 | 1.44329 | . 71901 | 1.33059 | 17 |
| 44 | . 642882 | 1.55567 | . 6667818 | 1.49755 1.49661 | . 693329 | 1.44239 | . 71946 | 1.38994 | 16 |
| 45 | .$_{64363}$ | 1.55467 | . 668888 | 1.499566 | . 6939416 | 1.14149 | . 71290 | 1.38909 | 14 |
|  | . 644104 | 1.55263 | . 666890 | 1.49472 | . 6941 | 1.44060 | . 72034 | 1.38824 | 14 |
| 48 | . 64446 | $1.551 \% 0$ | . 66944 | 1.49378 | . 69502 | 1.43881 | . 72122 | $1.386{ }^{5} 3$ |  |
| 49 | . 64487 | 1.55071 | . $66 \sim 6$ | 1.49284 | . 69545 | $1.43 \% 02$ | . 22167 | 1.38508 | 11 |
| 50 | . 64528 | 1.54972 | . 67023 | 1.49190 | . 69588 | 1.43703 | . 72211 | 1.38484 | 10 |
| 51 | . 64569 | 1.54873 | .670\%1 | 1.49097 | . 69631 | 1.43614 | . 722 | 1.38399 | 9 |
|  | . 64610 | 1.54774 | . 67113 | 1.49003 | . 69675 | 1.43525 | . 72299 | 1.38314 | 8 |
|  | .64652 | 1.54675 | . 671 J | 1.48909 | . 69718 | 1.43436 | . 72344 | 1.38229 | 7 |
|  | . 64693 | $1.545 \% 6$ | .67197 | 1.48816 | . 69761 | 1.43347 | . 72388 | 1.38145 | 6 |
| ${ }_{58}^{55}$ | . 64734 | 1.54478 | . 67239 | 1.48722 | . 69804 | 1.43328 | . 72432 | 1.38060 | 5 |
|  | . 647 | 1.54379 | . 67283 | 1.48629 | . 69847 | 1.43169 | .7247 | 1.37976 | 4 |
|  | . 64858 | 1.54183 | . 67206 | 1.48442 | . 699931 | 1.438090 | -72565 |  | ${ }^{3}$ |
|  | . 64899 | 1.54085 | . 67409 | 1.48349 | . 69977 | 1.42003 | . 72610 | 1.37722 | 1 |
| 60 | . 64941 | 1.53986 | 7451 | 1.48256 | . 70021 | 1.42315 |  | 1.37638 | 0 |
|  | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang |  |
|  | $57^{\circ}$ |  | $56^{\circ}$ |  | $55^{\circ}$ |  | $54^{\circ}$ |  |  |


|  | $36^{\circ}$ |  | $37^{\circ}$ |  | 38 ${ }^{\circ}$ |  | $39^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| 0 | . 72654 | 1.37638 | . 75355 | 1.32704 | . 78129 | 1.27994 | . 80978 | 1.23490 | $\overline{00}$ |
| 1 | . 72699 | 1.37554 | . 75401 | 1.32624 | . 78175 | 1.27917 | . 81027 | 1.23416 | 59 |
|  | . 72743 | 1.37470 | . 75447 | 1.32544 | . 78222 | 1.27841 | . 81075 | 1.23343 | 58 |
| 3 | . 72788 | 1.37386 | .75492 | 1.32464 | . 78269 | 1.27764 | . 81123 | 1.23270 | 57 |
|  | . 72832 | 1.37302 | . 75538 | 1.32384 | . 78316 | 1.27688 | . 81171 | 1.23196 | 56 |
|  | . 72877 | 1.37218 | . 75584 | 1.32304 | . 78363 | 1.27611 | . 81220 | 1.23123 | 55 |
| 6 | . 72921 | 1.37134 | . 75629 | 1.32224 | . 78410 | 1.27535 | . 81268 | 1.23050 | 54 |
|  | . 72966 | 1.37050 | . 75675 | 1.32144 | . 78457 | 1.27458 | . 81316 | 1.22977 | 53 |
| 8 | . 73010 | 1.36967 | . 75721 | 1.32064 | . 78504 | 1.27382 | . 81364 | 1.22904 | 52 |
| 9 | . 73055 | 1.36883 | . 75767 | 1.31984 | . 78551 | 1.27306 | . 81413 | 1.22831 | 51 |
| 10 | . 73100 | 1.36800 | . 75812 | 1.31904 | . 78598 | 1.27230 | . 81461 | 1.22758 | 50 |
| 11 | . 73144 | 1.36716 | . 75858 | 1.31825 | . 78645 | 1.27153 | . 81510 | 1.22685 | 49 |
| 12 | . 73189 | 1.36633 | . 75904 | 1.31745 | . 78692 | 1.2i07 | . 81558 | 1.22612 | 48 |
| 13 | . 73234 | 1.36549 | . 75950 | 1.31666 | . 78739 | 1.27001 | . 81606 | 1.22539 | 47 |
| 14 | .73278 | 1.36466 | . 75996 | 1.31586 | . 78786 | 1.26925 | . 81655 | 1.22467 | 46 |
| 15 | . 73323 | 1.36383 | . 76042 | 1.31507 | . 78834 | 1.26849 | . 81703 | 1.22394 | 45 |
| 16 | . 73368 | 1.36300 | . 76088 | 1.31427 | . 78881 | 1.26774 | . 81752 | 1.22321 | 44 |
| 17 | . 73413 | 1.36217 | . 76134 | 1.31348 | . 78928 | 1.26698 | . 81800 | 1.22249 | 43 |
| 18 | . 73457 | 1.36134 | \%6180 | 1.31269 | . 78975 | 1.26622 | . 81849 | 1.22176 | 42 |
| 19 | . 73502 | 1.36051 | . 76226 | 1.31190 | . 79022 | 1.26546 | . 81898 | 1.22104 | 41 |
| 20 | . 73547 | 1.35968 | . 76272 | 1.31110 | . 79070 | $1.264 \% 1$ | . 81946 | 1.22031 | 40 |
| 21 | . 73592 | 1.35885 | . 69318 | 1.31031 | . 79117 | 1.26395 | . 81995 | 1.21959 | 39 |
| 22 | . 73637 | 1.35802 | . 6364 | 1.30952 | . 79164 | 1.26319 | . 82044 | 1.21886 | 38 |
| 23 | . 73681 | 1.35719 | . 76410 | 1.30873 | . 79212 | 1.26244 | . 82092 | 1.21814 | 3 T |
| 24 | . 73726 | 1.35637 | . 76456 | 1.30795 | . 79259 | 1.26169 | . 82141 | 1.21742 | 36 |
| 25 | . 73771 | 1.35554 | . 76502 | 1.30716 | . 79308 | 1.26093 | . 82190 | 1.216\%0 | 35 |
| 26 | .73816 | 1.35472 | . 76548 | 1.30637 | . 79354 | 1.26018 | . 82238 | 1.21598 | 34 |
| 27 | . 73861 | 1.35389 | . 76594 | 1.30558 | . 79401 | 1.25943 | . 82287 | 1.21526 | 33 |
| 8 | . 73906 | 1.35307 | .76640 | 1.30480 | . 79449 | 1.25867 | . 82336 | 1.21454 | 32 |
| 29 | . 73951 | 1.35224 | . 76686 | 1.30401 | . 79496 | 1.25792 | . 82385 | 1.21382 | 31 |
| 30 | . 73996 | 1.35142 | . 76733 | 1.30323 | . 79544 | 1.25717 | . 82434 | 1.21310 | 30 |
| 31 | . 74041 | 1.35060 | . 76779 | 1.30244 | . 79591 | 1.25642 | . 82483 | 1.21238 | 22 |
| 32 | . 74086 | 1.34978 | . 76825 | 1.30166 | . 79639 | 1.25567 | . 82531 | 1.21166 | 28 |
| 33 | . 74131 | 1.34896 | . 76871 | 1.30037 | . 79680 | 1.25492 | . 82380 | 1.21094 | 27 |
| 34 | . 74176 | 1.34814 | . 78918 | 1.30009 | . 79734 | 1.25417 | . 822629 | 1.21023 | 26 |
| 35 | . 74221 | 1.34732 | . 76964 | 1.29931 | . 79781 | 1.25343 | . 82678 | 1.20951 | 5 |
|  | 74267 | 1.34650 | . 77010 | 1.29853 | . 79829 | 1.25268 | . $82 \% 27$ | 1.20879 | 4 |
| 37 | . 74312 | 1.34568 | . 77057 | 1.29775 | . 79877 | 1.25193 | . 82776 | 1.20808 | 3 |
|  | . $\% 4357$ | 1.34487 | . 77103 | 1.29696 | . 79924 | 1.25118 | . 82825 | 1.20736 | 22 |
| 39 | . 74402 | 1.34405 | . 77149 | 1.29618 | . 79972 | 1.25044 | . 82874 | 1.20665 | 21 |
| 40 | . 744 | 1.34323 | . 7 | 1.295 | . 800 | 1.24 | . 829 | 1.20593 | 20 |
| 41 | . 74492 | 1.34242 | . 77242 | 1.29463 | . 80067 | 1.24895 | . 82972 | 1.20522 | 19 |
| 42 | . 74538 | 1.34160 | \%7289 | 1.29385 | . 80115 | 1.24820 | . 83022 | 1.20451 | 18 |
| 43 | . 74583 | 1.34079 | . 77335 | 1.29307 | . 80103 | 1.24 \% 46 | . 83071 | 1.20379 | 17 |
|  | 4 .74628 | 1.33998 | . 77382 | 1.29229 | . 80211 | 1.24672 | . 83120 | 1.20308 | 15 |
| 45 | 5.74674 | 1.33916 | . 77428 | 1.29152 | . 80258 | 1.24597 | . 83169 | 1.20237 | 5 |
|  | 6 . 74719 | 1.33835 | . 77475 | 1.29074 | . 80306 | 1.24523 | . 83218 | 1.20166 | 14 |
| 47 | 7.74764 | 1.33754 | . 77521 | 1.28997 | . 80354 | 1.24449 | . 83268 | 1.20095 | 13 |
| 48 | 8 .74810 | 1.33673 | . 77568 | 1.28919 | . 80402 | 1.24375 | . 83317 | 1.20024 | 12 |
| 40 | 9.74855 | 1.33592 | . 77615 | 1.28842 | . 80450 | 1.24301 | . 83366 | 1.19953 | 11 |
| 50 | 0 . 74900 | 1.33511 | . 77661 | 1.28764 | . 80498 | 1.24827 | . 83415 | 1.19882 | 10 |
|  | . 74946 | 1.33430 | . 77708 | 1.28687 | . 80546 | 1.24153 | . 83465 | 1.19811 | 9 |
|  | . 74991 | 1.33349 | . 77754 | 1.28610 | . 80594 | 1.24079 | . 83514 | 1.19740 | 8 |
|  | . 75037 | 1.33268 | . 77801 | 1.28533 | . 80642 | 1.24005 | . 82064 | 1.19669 | 7 |
|  | . 75082 | 1.33187 | . 77848 | 1.28456 | . 80690 | 1.23931 | . 83613 | 1.19599 | ${ }^{6}$ |
|  | . 75128 | 1.33107 | . 78895 | 1.28379 | . 80778 | 1.23858 | . 833662 | 1.19528 | 5 |
|  | . 75173 | 1.33026 | . 77994 | 1.28302 | . 80788 | 1.23784 | . 83712 | 1.19457 | 4 |
|  | . 775219 | 1.32946 1.32865 | .77988 .78035 | 1.28225 1.28148 | . 80883 | 1.23710 1.23637 | . 833811 | 1.19387 1.19316 | 3 |
|  | . 75310 | 1.32785 | . 78082 | 1.28071 | .80930 | 1.23563 | . 838860 | 1.19246 | 1 |
| 60 | . 0.75355 | $1.32 \% 04$ | . 78129 | 1.27994 | . 80978 | 1.23490 | . 83910 | 1.19175 | 0 |
|  | Cotang | Tang | $\overline{\text { Cotang }}$ | Tang | Cotang | Tang | $\overline{\text { Cotang }}$ | Tang |  |
|  |  | ${ }^{\circ}$ |  | $2 \cdot$ |  | $1{ }^{\circ}$ |  | $0{ }^{\circ}$ |  |


|  | $40^{\circ}$ |  | $41^{\circ}$ |  | 420 |  | $43^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang |  |
| 0 | . 83910 | 1.19175 | . 86929 | 1.15037 | . 90040 | 1.11061 | . 93252 | 1.07237 | 60 |
| 1 | . 83960 | 1.19105 | . 86980 | 1.14969 | . 90093 | 110996 | . 93306 | 1.07174 | 59 |
| a | . 84009 | 1.19035 | . 87031 | 1.14902 | . 90146 | 1.10931 | . 93360 | 1.07112 | 58 |
| 3 | . 84059 | 1.18964 | . 87082 | 1.14804 | . 90199 | 1.10867 | . 93415 | 1.07049 | 57 |
| 4 | . 84108 | 1.18894 | . 87133 | 1.14767 | . 90251 | 1.10802 | . 93469 | 1.06987 | 56 |
| 5 | . 81158 | 1.18824 | . 87184 | 1.14699 | . 90304 | 1.10737 | . 93524 | 1.06925 | 55 |
| 6 | . 84208 | 1.18754 | . 87236 | 1.14632 | . 90357 | 1.10672 | . 93578 | 1.06862 | 54 |
| 7 | . 84258 | 1.18684 | . 87887 | 1.14565 | . 90410 | 1.10607 | . 93633 | 1.06800 | 53 |
| 8 | . 81307 | 1.18614 | . 87338 | 1.14498 | . 90463 | 1.10543 | . 93688 | 1.06738 | 52 |
| 9 | . 84357 | 1.18544 | . 87389 | 1.14430 | . 90516 | 1.10478 | . 93742 | 1.06676 | 51 |
| 10 | . 84407 | 1.18474 | . 87441 | 1.14363 | . 90569 | 1.10414 | . 98797 | 1.06613 | 50 |
| 11 | . 84457 | 1.18404 | . 87492 | 1.14896 | . 00621 | 1.10349 | . 93852 | 1.06551 | 49 |
| 12 | . 84507 | 1.18334 | . 87543 | 1.14229 | . 90674 | 1.10285 | . 93906 | 1.06489 | 48 |
| 13 | . 84556 | 1.18264 | . 87595 | 1.14162 | . 90727 | 1.10220 | . 93961 | 1.06427 | 47 |
| 14 | . 84606 | 1.18194 | . 87646 | 1.14095 | . 90781 | 1.10156 | . 94016 | 1.06365 | 46 |
| 15 | . 84656 | 1.18125 | . 87698 | 1.14028 | . 90834 | 1.10091 | . 94071 | 1.06303 | 45 |
| 16 | . 84706 | 1.18055 | . 87749 | 1.13961 | . 90887 | $1.100 \% 7$ | . 94125 | 1.06241 | 44 |
| 17 | . 84756 | 1.17986 | . 87801 | 1.13894 | . 90940 | 1.09963 | . 94180 | 1.06179 | 43 |
| 18 | . 84806 | 1.17916 | . 87852 | 1.13828 | . 90993 | 1.09899 | . 94235 | 1.06117 | 42 |
| 19 | . 84856 | 1.17846 | . 87904 | 1.13761 | . 91046 | 1.09834 | . 94290 | 1.06056 | 41 |
| 20 | . 84906 | $1.17 \% 7$ | . 87955 | 1.13694 | . 91099 | $1.097 \% 0$ | . 94345 | 1.05994 | 40 |
| 21 | . 84956 | 1.17\%08 | . 88007 | 1.13627 | . 91153 | 1.09706 | . 94400 | 1.05932 | 39 |
| 22 | . 85006 | 1.17638 | . 88059 | 1.13561 | . 91206 | 1.09642 | . 94455 | 1.05870 | 38 |
| 23 | . 85057 | 1.17569 | . 88110 | 1.13494 | . 91259 | 1.09578 | . 94510 | 1.05809 | 37 |
| 24 | . 85107 | 1.17500 | . 88162 | 1.13428 | . 91313 | 1.09514 | . 94565 | 1.05747 | 36 |
| 25 | . 85157 | 1.17430 | . 88214 | 1.13361 | . 91366 | 1.09450 | . 94620 | 1.05685 | 35 |
| 26 | . 85207 | 1.17361 | . 88265 | 1.13295 | . 91419 | 1.09386 | . 94676 | 1.05624 | 34 |
| 27 | . 85257 | 1.17292 | . 88317 | 1.13228 | . 91473 | 1.09322 | . 94731 | 1.05562 | 33 |
| 28 | . 85308 | 1.17283 | . 88369 | 1.13162 | . 91526 | 1.09258 | . 94786 | 1.05501 | 32 |
| 29 | . 85358 | 1.17154 | . 88421 | 1.13096 | . 91580 | 1.09195 | . 94841 | 1.05439 | 31 |
| 30 | . 85408 | 1.17085 | . 88473 | 1.13029 | . 91633 | 1.09131 | . 94896 | 1.05378 | 30 |
| 31 | . 85458 | 1.17016 | . 88524 | 1.12963 | . 91687 | 1.09067 | . 94952 | 1.65317 | 29 |
| 32 | . 85509 | 1.16947 | . 88576 | 1.12897 | . 91740 | 1.09003 | . 95007 | 1.05255 | 28 |
| 33 | . 85559 | 1.16878 | . 88628 | 1.12831 | . 91794 | 1.08940 | . 95062 | 1.05194 | 27 |
| 34 | . 85609 | 1.16809 | . 88680 | 1.12765 | . 91847 | $1.088 \% 6$ | . 95118 | 1.05133 | 26 |
| 35 | . 85660 | $1.16 \% 41$ | . 88732 | 1.12699 | . 91901 | 1.03813 | . 95173 | 1.05072 | 25 |
| 36 | . 85710 | 1.16672 | . 88784 | 1.12633 | . 91955 | 1.c8849 | . 95229 | 1.05010 | 24 |
| 37 | . 85761 | 1.16603 | . 88836 | 1.12567 | . 92008 | 1.08686 | . 95284 | 1.04949 | 23 |
| 38 | . 85811 | 1.16535 | . 88888 | 1.12501 | . 92002 | 1.08622 | . 95340 | 1.04888 | 22 |
| 39 | . 85862 | 1.16466 | . 88940 | 1.12435 | . 02116 | 1.03559 | . 95395 | 1.04827 | 21 |
| 40 | . 85912 | 1.16398 | . 88992 | 1.12369 | . 92170 | 1.08496 | . 95451 | 1.04766 | 20 |
| 41 | . 85963 | 1.16329 | . 89045 | 1.12303 | . 92224 | 1.08432 | . 95506 | 1.04705 | 19 |
| 42 | . 86014 | 1.16261 | . 89097 | 1.12238 | . 92237 | 1.08369 | . 95562 | 1.04644 | 18 |
| 43 | . 86064 | 1.16192 | . 89149 | 1.12172 | . 92331 | 1.08306 | . 95618 | 1.04583 | 17 |
| 44 | . 86115 | 1.16124 | . 89201 | 1.12106 | . 92385 | 1.08243 | . 95673 | 1.04522 | 16 |
| 45 | . 86166 | 1.16056 | . 89253 | 1.12041 | . 92439 | 1.08179 | . $95 \% 29$ | 1.04461 | 15 |
| 46 | . 86216 | 1.15987 | . 89306 | 1.11975 | . 92493 | 1.08116 | . 95785 | 1.04401 | 14 |
| 47 | . 86267 | 1.15919 | . 89358 | 1.11909 | . 92547 | 1.08053 | . 95841 | 1.04340 | 13 |
| 48 | . 86318 | 1.15851 | . 89410 | 1.11844 | . 92601 | 1.07990 | . 95897 | 1.04279 | 12 |
| 49 | . 86368 | 1.15783 | . 89463 | 111778 | . 92655 | 1.07927 | . 95952 | 1.04218 | 11 |
| 50 | . 86419 | 1.15715 | . 89515 | 1.11713 | . 92709 | 1.07864 | . 96008 | 1.04158 | 10 |
| 51 | . 86470 | 1.15647 | . 89567 | 1.11648 | . 92763 | 1.07801 | . 96064 | 1.04097 | 9 |
| 52 | . 86521 | 1.15579 | . 89620 | 1.11582 | . 92817 | 1.07738 | . 96120 | 1.04036 | 8 |
| 53 | . 86572 | 1.15511 | . 89672 | 1.11517 | . 99872 | 1.07676 | . 96176 | 1.03976 | 7 |
| 54 | . 86623 | 1.15443 | . 89725 | 1.11452 | . 92926 | 1.07613 | . 96232 | 1.03915 | 6 |
| 55 | . 86674 | 1.15375 | . 89777 | 1.11387 | . 92980 | 1.07550 | . 96288 | 1.03855 | 5 |
| 56 | . 86725 | 1.15308 | . 89830 | 1.11321 | . 93034 | 1.07487 | . 96344 | 1.03794 | 4 |
| 57 | . 86776 | 1.15240 | . 89883 | 1.11256 | . 93088 | 1.07425 | . 96400 | 1.03734 | 3 |
| 58 | . 88887 | 1.15172 | . 89935 | 1.11191 | . 93143 | 1.07362 | . 96457 | 1.03674 | 2 |
| 59 | . 86878 | 1.15104 | . 89998 | 1.11126 | . 93197 | 1.07299 | . 96513 | 1.03613 | 1 |
| 60 | . 86929 | 1.15037 | 90040 | 1.11061 | 9325 | 1.07237 | 96569 | 1.03553 | 0 |
|  | Cotang | Tang | Cotang | Tang | Cotang | Tang | Cotang | Tang |  |
|  |  |  |  |  |  |  |  |  |  |


| , | $44^{\circ}$ |  |  | , | $44^{\circ}$ |  |  | , | $44^{\circ}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Tang | Cotang |  |  | Tang | Cotang |  |  | Tang | Cotang |  |
| 0 | . 96569 | 1.03553 | 60 | 20 | . 97700 | 1.02355 | 40 | 40 | . 98843 | 1.01170 | 20 |
| 1 | . 96625 | 1.03493 | 59 | 21 | . 97756 | 1.02295 | 39 | 41 | . 98901 | 1.01112 | 19 |
| 2 | . 96681 | 1.03433 | 58 | 22 | . 97813 | 1.02236 | 38 | 42 | . 98958 | 1.01053 | 18 |
| 3 | . 96738 | $1.0337 \%$ | 57 | 23 | . $978 \% 0$ | $1.021 \% 6$ | 37 | 43 | . 99016 | 1.00994 | 17 |
| 4 | . 96791 | 1.03312 | 56 | 24 | . 97927 | 1.02117 | 36 | 44 | . 99073 | 1.00935 | 16 |
| 5 | . 96850 | 1.03252 | 55 | 25 | . 97984 | 1.02057 | 35 | 45 | . 99131 | 1.00876 | 15 |
| 6 | .96907 | 1.03192 | 54 | 26 | . 98041 | 1.01998 | 34 | 46 | . 99189 | 1.00818 | 14 |
| 7 | . 96963 | $1.0315 \%$ | 53 | 27 | . 98098 | 1.01939 | 33 | 47 | . 99247 | 1.00759 | 13 |
| 8 | . 97020 | 1.03072 | 52 | 28 | . 98155 | 1.01879 | 32 | 48 | . 99304 | 1.00701 | 12 |
| 9 | . 97066 | 1.03012 | 51 | 29 | . 98213 | 1.01820 | 31 | 49 | . 99362 | 1.00642 | 11 |
| 10 | . 97133 | 1.02952 | 50 | 30 | . $982 \% 0$ | 1.01761 | 30 | 50 | . 99420 | 1.00583 | 10 |
| 11 | . 97189 | 1.02892 | 49 | 31 | . 98327 | 1.01702 | 29 | 51 | . 99478 | 1.00525 | 9 |
| 12 | . 97246 | 1.02832 | 48 | 3. | . 98384 | 1.01642 | 28 | 52 | . 99536 | 1.00467 | 8 |
| 13 | . 97302 | 1.02\% | 47 | 33 | . 98441 | 1.01583 | 27 | 53 | . 99594 | 1.00408 | 7 |
| 14 | . 97359 | 1.02713 | 46 | 34 | . 98499 | 1.01524 | 26 | 54 | . 99652 | 1.00350 | 6 |
| 15 | . 97416 | 1.02653 | 45 | 35 | . 98556 | 1.01465 | 25 | 55 | . 99710 | 1.00291 | 5 |
| 16 | . 97472 | 1.02593 | 44 | 36 | . 98613 | 1.01406 | 24 | 56 | . 99768 | 1.00233 | 4 |
| 17 | . 97529 | 1.02533 | 43 | 37 | . 98671 | 1.01347 | 23 | 57 | . 99896 | $1.001 \%$ | 3 |
| 18 | . 97586 | 1.02474 | 42 | 38 | . 98728 | 1.01288 | 22 | 58 | . 99884 | 1.00116 | 2 |
| 19 | . 97643 | 1.02414 | 41 | 39 | . 98786 | 1.01229 | 21 | 59 | . 99942 | 1.00058 | 1 |
| 20 | . 97700 | 1.02355 | 40 | 40 | . 98813 | 1.01170 | 20 | 60 | 1.00000 | 1.00000 | 0 |
|  | Cotang Tang |  |  | 1 | Cotang Tang |  |  | 1 | Cotang $\mid$ Tang |  |  |
|  | $45^{\circ}$ |  |  |  |  |  |  |  | 45 |  |  |

THIS BOOK IS DUE ON THE LAST DATE STAMPED BELOW

##  <br> APR 161945

YB 11067


## $2$


[^0]:    * See "Theorie des Erddruckes auf Grund der neueren Anschauungen," by Prof. Weyrauch, 1881.

[^1]:    * In all the demonstrations which follow, the dimension perpendicular to the page will be considered as unity.

[^2]:    * For comparison, see the "Technic," 1888; a construction by Prof. Greene.

    The construction follows (see Fig. 4, above) directly from Raukine's Ellipse of Stress.

[^3]:    * See Rankine's Applied Mechanics; Alexander's Applied Mechanics; Theories of Winkler and Mohr.

[^4]:    * See Van Nostrand's Magazine, xvir, 1877, p. 5. "New Constructions in Graphical Statics," by H. T. Eddy, C.E., Ph.D.

[^5]:    * Zeitschrift für Baukunde, Band I. Heft 2, 1878.

[^6]:    * Annales des Ponts et Chaussées. $\dagger$ Van Nostrand's Magazine.

