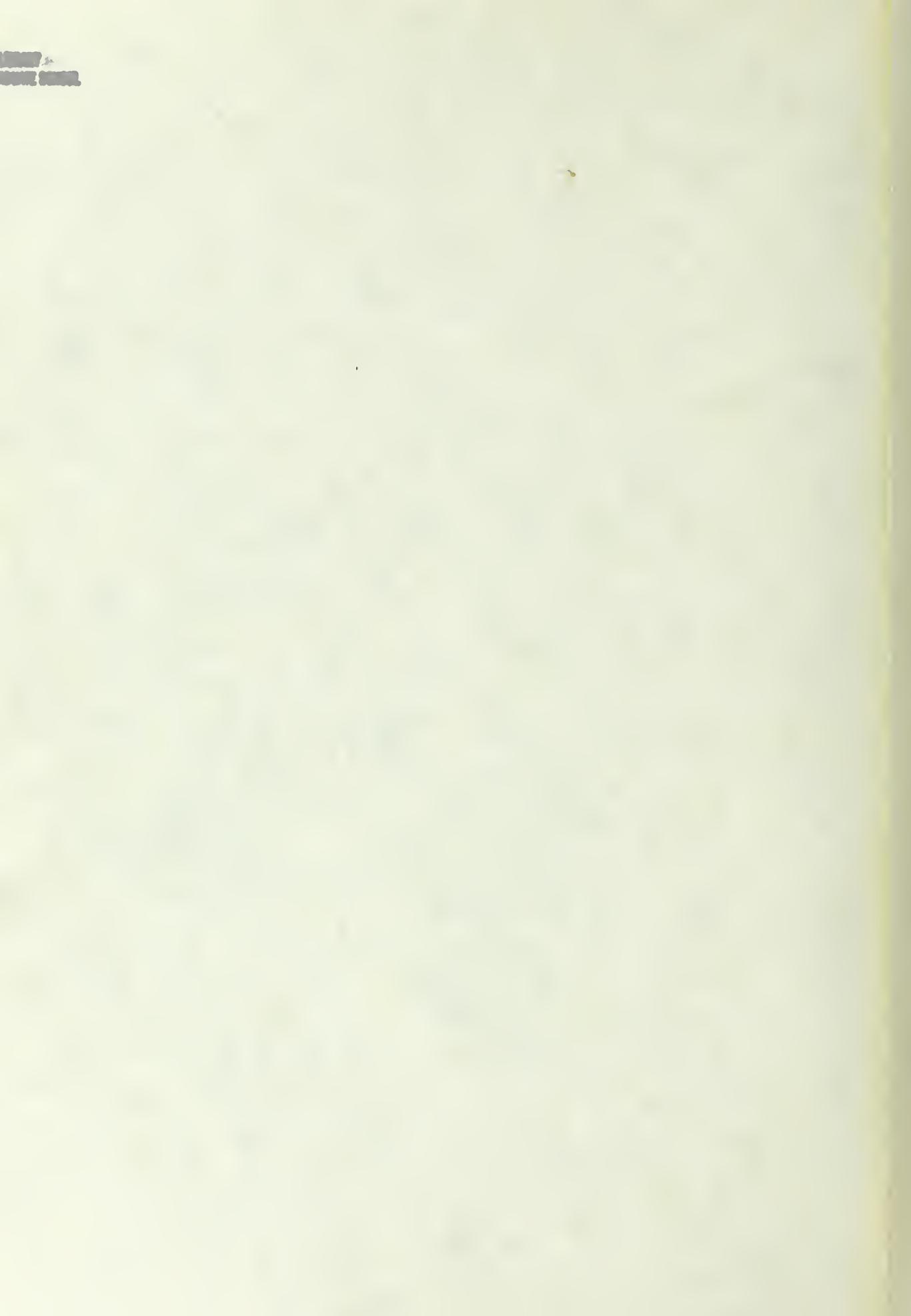


EA-6B MISSION PLANNING PROGRAM

Carl Alan Beaudet



# NAVAL POSTGRADUATE SCHOOL

## Monterey, California



# THESIS

EA-6B MISSION PLANNING PROGRAM

by

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June 1977

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T178628



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REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle)  EA-6B Mission Planning Program		5. TYPE OF REPORT & PERIOD COVERED  Master's Thesis; June 1977
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s)  Carl Alan Beaudet		8. CONTRACT OR GRANT NUMBER(s)
9. PERFORMING ORGANIZATION NAME AND ADDRESS  Naval Postgraduate School Monterey, California 93940		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
11. CONTROLLING OFFICE NAME AND ADDRESS  Naval Postgraduate School Monterey, California 93940		12. REPORT DATE  June 1977
		13. NUMBER OF PAGES 50
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report)  Unclassified
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report)  Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number)  EA-6B Mission Planning Electronic Warfare		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number)  The EA-6B Mission Planning Program is designed for use by aircrewmen deployed on board aircraft carriers. It is an interactive computer program for automated sorting,		



## (20. ABSTRACT Continued)

retrieval, display, and plotting of information. All decision making is done by the aircrewman. The desired goals of this program are increased mission planning efficiency and effectiveness through automation of the clerical tasks of the planning process.



EA-6B Mission Planning Program

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Submitted in partial fulfillment of the  
requirements for the degree of

MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL

June 1977



### ABSTRACT

The EA-6B Mission Planning Program is designed for use by aircrewmen deployed on board aircraft carriers. It is an interactive computer program for automated sorting, retrieval, deisplay, and plotting of information. All decision making is done by the aircrewman. The desired goals of this program are increased mission planning efficiency and effectiveness through automation of the clerical tasks of the planning process.



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## I. INTRODUCTION

The process of planning the electronic warfare support for an air strike is complex and time consuming. There are sophisticated methods for data collection, analysis, and storage, and the effective employment of today's weapon systems depends on such data. However, no automated interface has been developed to assist the Electronic Warfare Officer in his efforts to correlate the two. It's all done by hand.

The mission planning process reveals a pattern common to many technical problem solving efforts. The majority of one's time is spent gathering information and setting up the problem to be solved. Very little time is devoted directly to solving it. The data and information needed for EA-6B mission planning must be retrieved, by hand, from numerous source documents such as Kilting lists, TACMANUALS, computer printouts of the Electronic Order of Battle (EOB), radar handbooks, etc. Much of the information contained in these documents is non-essential to the immediate problem of mission planning, and only adds to the time required for sorting and retrieval. Data lists must be made for later reference in flight, and charts must be marked showing the EOB, route of flight, and emitter detection envelopes. Only when all this preliminary work is complete can the scenario



be subjectively analyzed by the planner, and jammer positioning optimized. Some calculations of jamming effectiveness may be performed, but these are so cumbersome and time consuming that an operator can seldom afford to make more than one or two such calculations, just to obtain a "feel" for the situation. The inefficiency is apparent, and under some circumstances, unaffordable.



## II. DESIGN CONSIDERATIONS

The primary objective was to automate as much of the planning process as possible, and allow the operator to devote more of his time to solving the problem of asset optimization. Speed and simplicity were the foundation for design considerations. The following is a description of the rationale used, and the decisions made which led to the final system design.

Sorting, plotting, and calculating can be most effectively accomplished by a computer. Making decisions, or choosing between options is a relatively simple task for a human. Therefore, the software system is not a completely hands off, optimization routine. Rather, it is a series of steps requiring decisions from the planner at various points, indicating in which direction to proceed. The decision points arise whenever it is more efficient for the planner to perform some portion of the planning process rather than generating a computer code to do the same thing. All probable situations have been included in the system design.

To minimize cost, the system is intended to utilize equipment and facilities currently on board aircraft carriers. Access to the ship's main computer at arbitrarily random times is not considered practical. Therefore, the system is designed for a small, peripheral type, general purpose computer. All Pacific Fleet carriers have been funded for



the WANG 2200 system, which is such a computer. The WANG system has:

- 1) Central processor of moderate capacity - 32K
- 2) Auxiliary storage of three 250K floppy disks
- 3) Video display
- 4) Typewriter for hard copy output
- 5) X-Y plotter capable of accommodating aeronautical charts.

Naval Ocean Systems Command has developed an interface between the WANG computer and the ship's main computer for fast transfer of information, with no apparent interruption of the main computer's functions.

The ship's computer contains the EOB for areas of the world in which the Navy has responsibility. The information contained in other reference documents is not presently in computer files, and the EA-6B TACMANUAL lists 13 separate publications for reference during mission planning. Therefore, an Emitter Parameter Library file was developed containing information retrieved from these various sources and stored in the floppy disk for reference. The library is an array containing EA-6B pertinent information on each type radar listed in the ship's EOB. When site locations are retrieved from the EOB in the main computer, the site type is retrieved also. The listing routine matches site types retrieved with the corresponding type in the Emitter Library and builds a "working" EOB. The parameters listed when the EOB is printed out are:



- 1) Site number
- 2) Latitude and longitude
- 3) Threat type, e.g. Low Blow or Tall King
- 4) Emitter function, e.g. Fire Control or Early Warning
- 5) Frequency band and frequency range of emitter
- 6) PRF range
- 7) Automatic and manual jamming codes against the emitter
- 8) Percent of frequency band of the emitter
- 9) Pertinent remarks, e.g. "HOJ against noise", or "SA-2, DLJ beacon at    MHZ".

The listing would be used by the operator to determine computer lists to generate for the mission, site locations to be programmed, and any preemptive jamming assignments to be made. The information included in the Emitter Library is arbitrary, and items can be added or deleted to suit the preferences of the squadron or community. Computer space may be at a premium, so the intent here is to include information of primary importance for quick reaction planning. No in-depth information on how the various codes were generated is presented.

This first generation program produces a "Flat Earth" solution. This was necessary to keep solution time to a minimum. Geographical features are not presently stored in the ship's computer, and an algorithm will have to be



devised to accommodate this problem. One approach will be discussed in the proposals for system expansion. At present, it would require a great deal of time and inputs from the operator to include terrain features, which is at cross purpose with the guidelines of speed and simplicity. All conflicts concerning whether or not to automate a certain phase of the mission planning process were resolved within this framework.



### III. SYSTEM DESCRIPTION

A computer simulation was accomplished utilizing:

- 1) IBM 360/67 general purpose computer
- 2) Tektronix 4012 graphics terminal (30/12 system)
- 3) Tektronix 4610 hard copy unit.

FORTRAN language was used in the simulation and conversion to BASIC language used on the WANG 2200 is relatively straightforward due to the similarities of the two languages.

The system that has been designed assists the operator as follows: The operator initiates the planning process by choosing one of three basic mission profiles; escort, modified escort, or standoff. Next, either a route of flight for the strike group is entered, if it has already been chosen, or just the latitude and longitude of the target may be entered, allowing the operator to choose and enter a route later. The system produces a printout of the area's EOB. Next, the EOB, detection envelopes of the various emitters, and the route of flight are visually presented to the operator. This is a key point in the planning process, for once the visual presentation is available, the optimum route is often apparent at a glance. A large X-Y plotter appropriately marks the operator's chart with this presentation, if and when desired. If the operator wishes to consider a number of alternatives, each may be plotted on a transparent overlay, several of which may be presented to



the strike leader or staff for final decision. Then hard copies of the complete navigation solution and the Time Scenario of the route are printed. The Time Scenario is a minute-by-minute listing of the sites within detection range of the strike group and/or EA-6B. It contains all necessary information to react as quickly as possible to an onboard jamming system malfunction which would cause operation in a degraded mode.

The operator may consider as many combinations of routes and mission profiles as he desires. The point to remember is that this is not an optimization routine. The aircrewman must make all the decisions. The success of his planning efforts will depend on how he uses his training, experience, and imagination, which is no different from the way things have always been. Hopefully he will have the chance to be considerably more effective by utilizing a system which performs most of the clerical tasks of mission planning for him.



#### IV. DETAILED PLANNING SESSION

##### A. INTRODUCTION

The following is a description of all aspects of system assistance available in a complete planning session. The assumption made is that the strike aircraft route of flight to and from the target has already been designated. The operator's task is to pick the most appropriate EA-6B mission profile, and optimize his assets accordingly.

##### B. ESCORT MISSION

Upon initiating the program, the operator can select any of the mission profiles for consideration. This example assumes consideration of the Escort mission profile first. The system then asks the operator to enter the strike aircraft route of flight to and from the target, including turnpoints, speeds on each leg of the route, and the magnetic variation of the area. At this point the system takes the given L/L's of the route, determines the maximum and minimum of each, and adds and subtracts 120 NM to the maximum and minimum, respectively. This sets the geographical limits of search in the EOB to be obtained from the ship's main computer. The system then accesses the current listing of the EOB. It searches through the EOB and retrieves all sites that fall within the geographical limits of the maximum and minimum latitudes and longitudes previously calculated.



It retrieves the site L/L and the site type (Fansong, Barlock, etc.). All these sites are stored in an array (list) in the peripheral machine, and matched with the appropriate parameters in the Emitter Parameter Library, as described in Design Considerations.

The "working" EOB now contains all sites listed in the ship's EOB within the geographical limits set. The operator is asked if he wants to add any additional sites to the EOB that may be a result of recent intelligence (VQ, returning strike a/c, RA-5C missions, etc.). He also has the option to build his own complete EOB, ignoring the ship's listing entirely. This would be useful in exercises against friendly EOB's such as U.S. coastlines, EW ranges at Fallon and Pinecastle, etc.

When the "working" EOB is complete, the system asks if the operator wishes a printout of the EOB. If a listing is desired, the operator can be selective by choosing to list all emitters, just EW/Acq type emitters, or just Terminal Threat types. The information presented comes from the "working" EOB and the Emitter Parameter Library. The listing contains information pertinent to EA-6B operators (see Fig. 1).

The next step in the process is to display the EOB and route of flight to the operator. He can display the route of flight and radar detection envelopes of all emitters, EW/Acq types only, or terminal threat types only. Once the desired combinations are entered, the sites, route of flight,



and detection envelopes are displayed (see Figs. 2,3,4), and the margin is scaled with appropriate L/L. The L/L convention used is + for N and E, - for S and W. The operator may choose to have this presentation drawn on his aeronautical chart, or on a transparent overlay. The display and chart can be studied for terrain features, appropriateness of route, etc.

The next choice offered the operator is 'Do you wish a Navigation solution?' for the route currently being considered. It is necessary to have the NAV solution if a time scenario (discussed in System Description) is desired also. Several items calculated in the NAV solution, such as speeds, headings, times, etc., are used in determining present position of the EA-6B as it proceeds around the route. If the operator doesn't want the NAV solution, he is then asked if he wishes to consider a different route, or a different mission profile. If he does want the solution, it is displayed for him, and a hard copy is produced (see Fig. 8).

Next, the system asks "Do you want a listing of the Time Scenario?" If desired, the system proceeds as follows:

- 1) It asks operator to indicate emitters of interest by type (all, EW/Acq only, Terminal Threat only) and by frequency band.
- 2) It calculates present position (p/p) of the EA-6B beginning at the first point of the route.



- 3) It filters the "working" EOB for the proper type and frequency bands, calculates the range from p/p to each emitter having passed the filter, checks that distance against the "threat" range for that emitter listed in the Emitter Parameter Library. If the distance is within "threat" range, the important inflight parameters are displayed.
- 4) After all sites are checked, time is incremented by a minute, and a new p/p is calculated using headings and speeds obtained from the NAV solution.
- 5) There are detailed routines in this portion to check if the p/p increment goes around a turnpoint and if heading speed changes occur, or if the end of the route is reached.
- 6) The process iterates around the entire route, minute by minute.
- 7) The parameters printed out for the operator (see Fig. 9) in hard copy form are:
  - a) Time
  - b) Present Position
  - c) Emitter Type
  - d) "Working" EOB Number
  - e) Range and Bearing to the emitter
  - f) Automatic and Degraded jamming modes to use
  - g) Relative percent of frequency band of the emitter.



The entire purpose of producing this scenario is to provide the operator with all information necessary to devote a minimum of "inflight" time searching, analyzing, and reacting to known sites, allowing more time to concentrate on the unknown or unexpected emitters. The information will allow him to handle system degradations with as little confusion and consternation as possible. This program does not attempt to solve the problem of having fewer assets than number of sites in range. Here, again, the operator must decide relative priorities using information available such as ranges to the various sites (just entering or about to exit an envelope), type emitters (AAA vs SAM, or SA-2 vs SA-6), and choose his asset deployment scheme accordingly. If the route is too saturated, perhaps a case can be made for an alternative route.

After the Time Scenario is complete, the system offers the operator the options to consider a different strike route, or a different mission profile. If he wishes to consider a new strike route, he enters it and the system returns to the Display portion of the program. This example will retain the same strike route and next, consider the Standoff mission profile.

#### C. STANDOFF MISSION

A standoff mission is hereby defined as using an EA-6B to primarily jam EW/Acq type radars as the strike group proceeds to and from the target. The "raids" can usually be



divided into three phases; ingress, over the target, and egress. The standoff objective is to optimize jamming against EW/Acq emitters during ingress and egress. While the strike aircraft are over the target there is little in the way of effective jamming that can be accomplished (from a standoff orbit) against narrow beam fire control and missile control radars. Therefore, the operator usually attempts to optimize his track or orbit to cover the first and last phases of the strike.

The program offers the operator a chance to view the EOB and various standoff stations. Often a single orbit for each phase (ingress and egress) is the optimum jamming position. Standoff jamming positioning is dictated by strike aircraft location. Therefore the system takes as inputs for the standoff solution, the strike aircraft route (in this example it has already been entered and need not be done so again), and the latitude and longitude of a standoff jamming point. Standoff jamming orbits are usually short enough in length with respect to distances to target emitters that jamming effectiveness will not vary significantly from one point in the orbit to another. Therefore, the midpoint of an orbit will suffice for most cases.

Next the system displays the strike aircraft route, the EOB sites and their detection envelopes, and the standoff point for the EA-6B. The display may be filtered by the operator, as before, by type emitter groupings. Additionally,



detection envelopes depressed by jamming may be displayed. The operator can assess the effectiveness of this jamming orbit for various strike group locations around the route. Several different standoff points may be considered before the operator decides on the optimum position(s). As an example, perhaps a stationary orbit is optimum for the first 15 minutes of the strike. Then the EA-6B must transit to an orbit some distance away, say 100 NM, for optimum positioning to cover the egress. The operator can display the first orbit, the last orbit, and as many points between the two as desired. He would then have a visual indication of how his jamming effectiveness will be affected during the transit phase, and where the strike group is most vulnerable to EW/Acq emitters during the strike (see Figs. 5,6).

As with the Escort mission discussed earlier, the operator would receive a hardcopy printout of:

- 1) Strike group navigation solution
- 2) List of jamming parameters for use in flight  
(see Fig. 11)
- 3) Chart appropriately marked with the EOB, route of flight, and emitter detection envelopes.

Operator judgment will play a large part in determining the success and speed at which optimum positioning of the standoff EA-6B is accomplished. He must vary the parameters, consider the options, and then make the decision. The system will not do it for him. It does provide enough speed



and ease of computation to allow the operator the luxury of considering many alternatives before making his selection, something seldom affordable with current planning procedures.

#### D. MODIFIED ESCORT

The Modified Escort mission profile is one in which the jamming aircraft directly accompanies the strike group on its route until such time as the group must penetrate AAA or SAM weapon envelopes. At such time, the EA-6B parallels the strike group just outside the weapon envelopes. Timing and positioning of the EA-6B is critical if any measurable success is to be achieved against fire control and missile control radars.

If the operator wishes to plan for this profile, he proceeds as mentioned in the previous sections. He may enter the strike group route and the EA-6B route, or just the strike group route, or neither. In this example, the strike group route has been entered previously. The operator has the option of viewing the EOB and strike route before entering the modified escort route for the EA-6B. He also has available the NAV solution for the strike group, with the times at various turnpoints on the route. He may use this information to coordinate the timing and positioning of the EA-6B, consistent with the strike group route. Once this route has been entered, the Modified Escort route is added to the visual presentation (see Fig. 7). If this



route is acceptable to the operator, he may have his chart marked with that route. He receives a printout of the EA-6B navigation solution, and a Time Scenario for the mission, if desired (see Figs. 8,10). The Time Scenario considers threats to both the strike group and the EA-6B, and lists the emitters "in range" accordingly. At the end of this sequence, the operator can alter the strike route, change the EA-6B route, change mission profiles again, or terminate the planning process.



## V. PROPOSALS FOR PROGRAM EXPANSION

- 1) The data base (Emitter Parameter Library) could be expanded to include EA-6B information grouped by weapon platforms such as ships, aircraft, and missile threat (ASM, SSM, and AAM). This information is easily compiled and requires no additional computer calculations. Current auxiliary storage space is sufficient. A simple call for information on a particular ship or aircraft would produce a printout similar to the EOB listing available in the current program (see Fig. 12).
- 2) Terrain consideration is one of the most important aspects of EW mission planning. The U.S. Geological Survey has developed a procedure to store geographical features in computer format that may prove adaptable to this planning program. The approach would be to store the terrain features of various areas of the world on cassette tapes and load the particular area of interest into the computer when planning a mission. The amount of computer space and complexity of application may prove to be beyond the capabilities of a mini computer, but that should be investigated.
- 3) If a refresh graphics display is available (the Tektronix is a storage tube, i.e. once the picture is drawn, it cannot be altered without redrawing the entire presentation), it



may be possible to present a dynamic visual display of jamming effectiveness as a mission proceeds from beginning to end. The current program presents "snapshots" of the situation at various points selected by the planner.

- 4) A current proposal for the ICAP II version of the EA-6B is to load a complete mission plan into the aircraft with a cassette tape. The intent is to go through the complete mission planning procedure in the Ready Room, compile all necessary data, enter it on a cassette tape, then take it to the aircraft and load it. The entire known scenario would then be stored in the aircraft's computer, lists would be automatically activated and deactivated, pre-programmed jamming assignments made, etc. Practially all the information required for such an effort is available in its current status. There would be a necessity to develop a language interface between the WANG and the aircraft computer.



SITE	NAME	HIGH	LOW	RNGE	LST	FLO	FHI	PRF1	PRF1	PRF2	PRF2	AUTO	DEGR	PRCT	REMARKS	
1	FANGONG D-F	NO.	7	29.	11	2700	2750	1000	1050	2000	2100	T050	WSS	44	SA-2, CCI PCH AT 800HZ	
		26.05	-07.30													
2	TALL KING	E/W	EN	2	100.	8	100	200	100	200	200	240	FT7	WPT	40	NONE
3	FANGONG C-E	HC	EN	3	20.	15	4200	4250	300	650	1000	1700	T002	WSS	36	SA-2, CCI PCH AT 800HZ, E-O BACKUP
		27.20	-09.23													
4	SPARIST AG	EN	EN	2	150.	8	100	130	120	180			FT2	WPT	40	SA-2 ACQUISITION
		27.27	-09.28													
5	LOW BLOW	HC	EN	0	10.	17	8725	8775	1.20	++50			T900	WSS	83	SA-3, LOW ALT 150FT, E-O BACKUP
		27.36	-07.33													
6	SPRINT 8	HC	EN	1	120.	6	50	50	CO	30			FT4	WPT	28	NONE
		27.38	-07.37													
7	FIREWHEEL	FG	EN	7	5.	11	2600	2720	1800	1900			T483.SS1	18	AAA, CONSCAN	
		27.54	-07.21													
8	FANGONG C-E	HC	EN	3	20.	15	4200	4250	200	850	1600	1700	T902	WSS	36	SA-2, CCI PCH AT 800HZ, F-O BACKUP
		27.57	-07.12													
9	SPRINT FLUSH	HC	EN	3	10.	15	4210	4200	3080	4000			T385	WSS	12	SA-C, ACQ-TRKP, FAST FLYER, 3-6 SEC
		27.33	-07.12													
10	DRILL-PCB1P	HC	EN	7	20.	11	2050	2000	420	430	840	850	FT3	WPT	21	ERLN, G PEAMS, STAPLE PRF 300
		22.31	-09.00													
11	TALL KING	EN	EN	2	100.	3	100	230	100	200	200	240	FT7	WPT	40	NONE
		23.41	-09.48													
12	FIRECIV	FG	EN	7	5.	11	2710	2730	2500	2520			T377	SS3	32	AAA, CONSCAN
		22.44	-07.03													
13	SPRINT FLUSH	HC	EN	8	10.	15	4210	4200	3080	4000			T385	WSS	12	SA-C, ACQ-TRKP, FAST FLYER, 3-6 SEC
		29.16	-09.01													
14	LOW BLOW	HC	EN	0	10.	17	8725	8775	1.20	++50			T200	WSS	83	SA-3, LOW ALT 150FT, E-O BACKUP
		29.20	-08.12													
15	SPRINT	FG	EN	7	5.	11	2020	2040	2000	2050			FT3	SS2	27	AAA, CONSCAN
		22.23	-08.23													
16	LOW BLOW	HC	EN	0	10.	17	8725	8775	1.20	++50			T200	WSS	83	SA-3, LOW ALT 150FT, E-O BACKUP
		29.28	-08.32													
17	LOW BLOW	HC	EN	0	10.	17	8725	8775	1.20	++50			T200	WSS	83	SA-3, LOW ALT 150FT, E-O BACKUP
		29.19	-08.38													

FIGURE - 1 LISTING OF E.O.B. WITH EA-6B PERTINENT PARAMETERS.

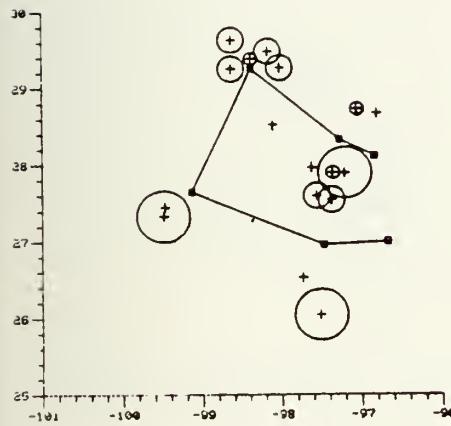


FIGURE - 2 VISUAL PRESENTATION FOR AN ESCORT MISSION PROFILE WITH ONLY TERMINAL THREAT (FIRE CONTROL, MISSILE CONTROL) Emitter DETECTION ENVELOPES DISPLAYED. SITE LOCATIONS +, AND ROUTE OF FLIGHT —■— ARE SHOWN. THE SCALE INDICATES LAT/LONG WITH THE CONVENTION N/S = +/-, AND E/W = +/-.

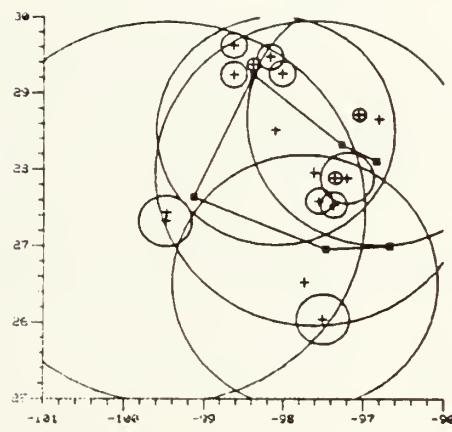


FIGURE - 3 VISUAL PRESENTATION FOR AN ESCORT MISSION PROFILE WITH ALL EMITTERS AND THEIR DETECTION ENVELOPES PRESENTED. SITE LOCATIONS +, AND ROUTE OF FLIGHT —■— ARE SHOWN, AND THE SCALE INDICATES LAT/LONG WITH THE CONVENTION N/S = +/-, AND E/W = +/-.

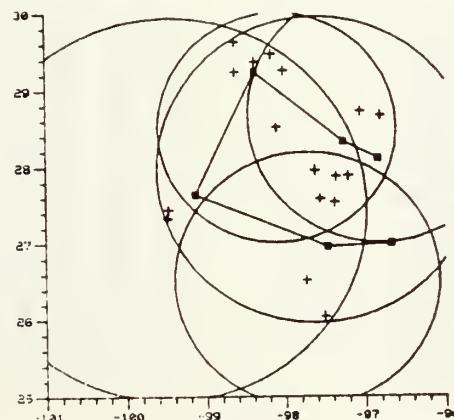


FIGURE - 4 VISUAL PRESENTATION FOR AN ESCORT MISSION PROFILE WITH ONLY EARLY WARNING/Acquisition TYPE RADAR DETECTION ENVELOPES PRESENTED. SITE LOCATIONS +, AND ROUTE OF FLIGHT —■— ARE SHOWN. SCALE INDICATES LAT/LONG WITH THE CONVENTION N/S = +/-, AND E/W = +/-.



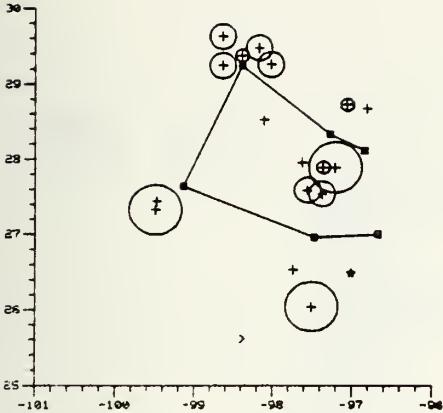


FIGURE - 5 SHOWS STRIKE GROUP ROUTE, STANDOFF JAMMER POSITION +, AND TERMINAL THREAT Emitter ENVELOPES (NOT DEPRESSED BY JAMMING).

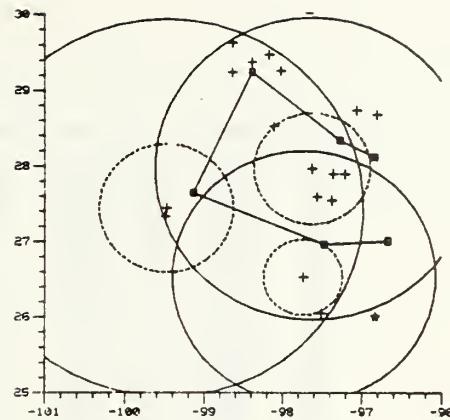


FIGURE - 6 SHOWS STRIKE GROUP ROUTE, STANDOFF JAMMER POSITION +, EV/AQ Emitter ENVELOPES UNJAMMED (SOLID LINES) AND JAMMED (DASHED LINES).

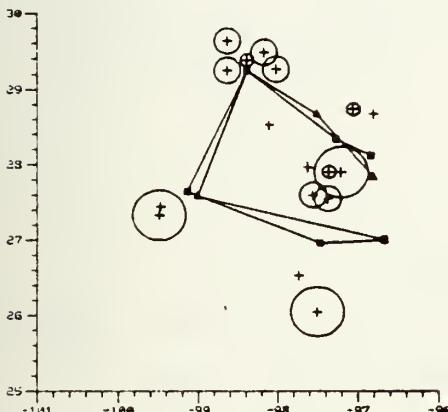


FIGURE - 7 SHOWS STRIKE GROUP ROUTE X, EA-6B ROUTE O, AND TERMINAL THREAT Emitter ENVELOPES (NOT JAMMED).

#### Strike Group Navigation Solution

Leg	Dist	Time	Th	Mn	TAS	TOPD	TOPD	To Turnpt L/L
1	3.0	7	67	350	310	7.1	43.0	26.65 -97.25
2	95.2	14	245	237	420	21.2	141.2	27.39 -99.03
3	10.1	13	22	14	180	34.2	249.1	29.15 -98.43
4	90.6	10	133	125	180	44.3	329.7	29.80 -97.16
5	26.4	3	116	111	480	47.6	392.1	23.07 -96.50

#### Ed Escort Navigation Solution

Leg	Dist	Time	Th	Mn	TAS	TOPD	TOPD	To Turnpt L/L
1	129.7	19	286	278	420	18.2	129.7	27.35 -99.00
2	105.2	13	18	10	480	31.7	234.9	29.15 -98.63
3	98.3	7	127	119	480	39.0	293.2	29.40 -97.30
4	62.3	3	143	135	480	46.3	367.5	27.0 -96.43

FIGURE - 8 THIS IS A LISTING OF THE STRIKE GROUP AND EA-6B IAVL SOLUTIONS FOR A MODIFIED ESCORT MISSION.

8 27 00 -0P								
TIME	PRES POS	TYPE	EOB	RNGE	BPG	AUTO	DEGR	PRCT
1	27 00 -96 40	TALL KING	2	63	244	FT7	IFT	40
		SPNRST 8	6	77	319	FT4	IFT	28
		TALL KING	2	58	242	FT7	IFT	40
2	26 59 -96 40	SPNRST AC	4	146	281	FT2	IFT	40
		SPNRST 8	6	74	328	FT4	IFT	28
		TALL KING	2	54	239	FT7	IFT	40
3	26 59 -96 50	SPNRST AC	4	141	281	FT2	IFT	40
		SPNRST 8	6	71	326	FT4	IFT	28
		TALL KING	2	49	236	FT7	IFT	40
4	26 59 -96 50	SPNRST AC	4	136	282	FT2	IFT	40
		SPNRST 8	6	68	320	FT4	IFT	28
		TALL KING	2	44	233	FT7	IFT	40
5	26 59 -97 04	SPNRST AC	4	131	281	FT2	IFT	40
		SPNRST 8	6	66	314	FT4	IFT	28
		TALL KING	2	40	229	FT7	IFT	40
6	26 58 -97 16	SPNRST AC	4	126	283	FT2	IFT	40
		SPNRST 8	6	64	313	FT4	IFT	28
		TALL KING	2	36	224	FT7	IFT	40
7	26 58 -97 22	SPNRST AC	4	121	284	FT2	IFT	40
		SPNRST 8	6	62	313	FT4	IFT	28
		TALL KING	2	32	217	FT7	IFT	40
8	26 58 -97 22	SPNRST AC	4	116	285	FT2	IFT	40
		SPNRST 8	6	60	314	FT4	IFT	28

FIGURE - 9 THIS IS A PORTION OF THE TIME SCENARIO FOR AN ESCORT MISSION SHOWING PARAMETERS NECESSARY TO ANTICIPATE ALL KNOWN SITES, AND TO REACT TO SYSTEM MALFUNCTIONS CAUSING DEGRADED MODE OPERATION WITH A MINIMUM OF CALCULATION. THE SITES ARE ONLY LISTED IF THE STRIKE GROUP IS WITHIN THE DESIGNATED DETECTION RANGE.

FIGURE - 10 THIS IS A PORTION OF THE TIME SCENARIO PRINTOUT FOR A MODIFIED ESCORT MISSION. ANGLES ARE FROM THE -00 TO VARIOUS EMITTERS. EMITTERS ARE LISTED IF EITHER THE STRIKE GROUP OR THE EA-6B ARE WITHIN THE DESIGNATED ENCOUNTER RANGE.



STANDOFF L/L 26.00 -96 59	TYPE	EOB	RNGE	BRQ	AUTO	DEGR	PRCT
	FANSONG B-F	1	36	275	T960	WSS	44
	TALL KING	2	58	304	FT7	WFT	40
	SPNRST AC	4	165	302	FT2	WFT	46
	LOW BLOW	5	103	338	T900	WSS	83
	SPNRST B	6	125	340	FT4	WFT	28
	FIREWHEEL	7	117	346	T483	SS1	18
	BRLK-BGBAR B	10	165	336	FT3	WFT	21
	TALL KING	11	161	1	FT7	WFT	40
	FIRECAN	12	164	356	T377	SS3	32
	LOW BLOW	14	221	341	T900	WSS	83
	WHIFF	15	219	338	FT3	SS2	27
	LOW BLOW	16	238	336	T900	WSS	83
	LOW BLOW	17	222	332	T900	WSS	83

FIGURE - 11 JAMMING PARAMETERS FOR A STANDOFF MISSION FROM ORBIT POINT INDICATED.

KASHIN DLG

ARMAMENT: SAM 20 x SA-N-15 (2 TWIN)

GUNS 4 x 50MM (TWIN MOUNT)

ASW 2 x RBU- 1000  
2 x RBU- 2000  
4 x 10 IN. TORPEDOES

A/C 1 x HORMONE

ELECTRONICS:

EMITTER	FUNC	BAND	LIST	RNGE	FLO	FHI	PRF1	PRF1	PRF2	PRF2	AUTO	DEGR	PRCT	REMARKS
BIG BOY	EW	1	4	100	25	50	100	110	----	----	S123	WSS	12	PRIMARY AIR SCH
BAD NEWS	EW	4	2	50	100	200	250	269	----	----	FT20	WFT	23	NONE
POPCORN	FC	7	15	22	2000	2100	1000	1010	----	----	T321	WSS	44	AAA, E-O ALSO
DON-2	NAV	9	12	8	4000	4400	8800	8900	9300	9400	FTC3	NFT2	67	NONE
FOOLYA	MC	8	30	45	6000	7000	1800	1850	----	----	T456	NSS	97	SA-N-15, DLJ

FIGURE - 12 TYPICAL PRINTOUT OF EA-6B PERTINENT INFORMATION BY WEAPON PLATFORM.



THIS IS THE MAIN PROGRAM FOR EA-6B MISSION PLANNING.  
FOR TEST PURPOSES, ENTER FILE DEF 4 DSK BEFORE EACH  
EXECUTION.

MA100010  
MA100020  
MA100030  
MA100040

## COMPUTER PROGRAM

```

DIMENSION X(10), Y(10), SX(20), SY(20), ST(20), CL(9), SPD(9), TL(9),
           MH(9), AS(9), SR(20), TR(9), NB(8),
           D5(30), D6(30), D7(30), D8(30), D9(30),
           D10(30), D11(30), D12(30), TN(30,3), TRMKS(30), MAIJDS83,
           MTAS(9), SPDM(9), PJ(30), PR(30), GJR(30), GAIN(MA100090),
           DLM(9), DTW(9), MTH(9), MA100100,
           MA100110, MA100120, MA100130, MA100140, MA100150, MA100160,
           MA100170, MA100180, MA100190, MA100200, MA100210, MA100220,
           MA100230, MA100240, MA100250, MA100260, MA100270, MA100280,
           MA100290, MA100310, MA100320, MA100330, MA100340, MA100350,
           MA100360, MA100370, MA100380, MA100390, MA100400, MA100410,
           MA100420, MA100430, MA100440, MA100450
1 2* HF, IZ, 3*B120, 2*D4*4, 5*7, 8, 8, 3*9, 6/
1 2* DATA TRG/3*120, 150, 2*100, 100, 3*5, 2*20, 30, 10, 5, MA100440
      
```



```

3 CCNTINUE
DC 313 I=1,3J
READ(4,311)(TRMKS(I,J),J=1,9)
311 FCRMAT(9A4)
311 CCNTINUE
DATA P J/10*20.,10*50.,10*10./
DATA B /30*20./

```



DATA GJR/30\*10./  
DATA PR/3J\*100./  
DATA GAIN/30\*36./

YMAX=0.  
PI=3.14159  
CS=9.  
CNFLG=1.  
XWIN=300.  
XNIN=300.  
INTEGER(6,1)  
FORMAT(6,1) FOR ALL QUESTIONS: ENTER 1 FOR YES, 0 FOR NO  
1 WRITE(6,299)  
WRITE(6,299)  
FORMAT(1X,1)  
1 IF(STIME? )  
READ(5,2) MSN  
IF(MSN.EQ.0)GO TO 99  
WRITE(6,300)  
FORMAT(1X,1)  
1 READ(5,2) MSN  
1 READ(5,2)MSN  
1 READ(5,2)MSN  
99 WRITE(6,101)  
FORMAT(1,1) DO YOU HAVE A STRIKE ROUTE TO ENTER AT THIS TIME?  
111 READ(5,2)LRTE  
FORMAT(5,1)LRTE  
2 IF(LRTE.EQ.1)GO TO 113  
N=2  
WRITE(6,103)  
FORMAT(5,4)(Y(I),X(I),I=1,N) START PT LAT/LONG, THEN TGT LAT/LONG  
103 READ(5,F7.2)  
4 FORMAT(4,2)  
CALL LL(X,Y,2)  
GC TO 42  
113 WRITE(6,115)  
FORMAT(4,6)N,K  
READ(4,6)N,K  
FORMAT(12)  
6 N=N-1  
WRITE(6,113)  
FORMAT(4,4)(Y(I),X(I),I=1,N) THEN LONG OF EACH TURNPT IN ORDER.  
13 READ(4,4)(Y(I),X(I),I=1,N)  
CALL LL(X,Y,N)  
113 WRITE(6,40)  
FORMAT(4,1)  
140 FORMAT(4,1)  
ENTER MAG VAR TO NEAREST DEGREE,+ FOR W,- E, EX:-12=12E+1 MA101380  
MA100940  
MA100980  
MA100990  
MA101000  
MA101020  
MA101030  
MA101040  
MA101050  
MA101060  
MA101070  
TMA101080  
MA101090  
MA101100  
MA101110  
MA101120  
MA101130  
MA101140  
MA101150  
MA101160  
MA101170  
MA101180  
MA101190  
MA101200  
MA101210  
MA101220  
MA101230  
MA101240  
MA101250  
MA101260  
MA101270  
MA101280  
MA101290  
MA101300  
MA101310  
MA101320  
MA101330  
MA101340  
MA101350  
MA101360  
MA101370  
MA101380



```

41 READ(4,41) VAR
42 FCRMAT(13)
43 WRITE(6,17)
44 FCRMAT(6,17)
45 READ(4,41)(ITAS(I),I=1,M)
46 SPD(I)=ITAS(I)/60.
47 CCNTINUE
48 CALL MAX(X,N,FR)
49 CALL MAX(Y,N,TOP)
50 CALL MIN(X,N,FL)
51 CALL MIN(Y,N,BOT)
52 TCF=TOP+2.
53 BCT=BOT-2.
54 FR=FR+2.
55 FL=FL-2.
56 READ(4,16)(SY(I),SX(I),ST(I),I=1,K)
57 FFORMAT(2F7.2,I3)
58 CALL LL(SX,SY,K)
59 IF(LRT-EQ.0) GO TO 44
60 IF(MSN-2)44 400,298
61 WRITE(6,401)
62
63 FORMAT(1X,'DO YOU HAVE A MOD ESCORT ROUTE TO ENTER NOW?')
64 READ(5,2)MERTF
65 IF(MERTF-EQ.0) GO TO 44
66 WRITE(6,402)
67 FFORMAT(1X,'ENTER NUMBER OF TURNPTS IN MOD ESCORT ROUTE')
68 READ(4,2)MEN
69 WRITE(6,403)
70 FORMAT(1X,'ENTER LAT, THEN ENTER LONG OF EACH POINT IN ORDER')
71 READ(4,4)(YME(I),XME(I),I=1,MEN)
72 CALL LL(XME,YME,MEN)
73 MEM=MEN-1
74 WRITE(6,404)
75 FFORMAT(1X,'ENTER TAS FOR EACH LEG, IN ORDER')
76 READ(4,41)(MTAS(I),I=1,MEM)
77 DC405 I=1,MEM
78 SPDM(I)=MTAS(I)/60.
79 GC TO 44
80 WRITE(6,301)
81
82 FORMAT(1X,'DO YOU HAVE A STANDOFF POINT TO CONSIDER YET?')
83 READ(5,2)LSD

```



```

      IF (LSO.EQ.0) GO TO 44
327  WRITE(6,302)
      FORMAT(1X,'ENTER LAT, THEN LONG OF STANOFF POINT')
332  READ(5,1)STOX
      CALL LL(STOX,STOY,1)
44   WRITE(6,56)

56  FORMAT(' DO YOU WISH TO USE SHIPS F.O.B. ?')
      READ(5,2)L
      IF(L.EQ.0) GO TO 55
55  WRITE(6,51)
51  FORMAT(' OPERATOR WISH TO ENTER SITES IN ADDITION TO OR INSTEAD OF
      FORMAT THE SHIPS E.C.B.?')
      READ(5,2)L
      IF(L.EQ.0) GO TO 57
57  WRITE(6,52)
52  FORMAT(' ENTER TOTAL NUMBER OF SITES OPERATOR IS ENTERING, USE 2 DIGITS,
      F6.34, OR 07, OR 13.')
      READ(5,6)J
      NL=K+J
      K1=K+1
      WRITE(6,53)
53  FORMAT(' ENTER LAT, THEN LONG OF EACH SITE IN ORDER')
      READ(5,4)(SY(I),SX(I),I=K1,NL)
      WRITE(6,54)
54  FORMAT(' ENTER SITE TYPE ACCORDING TO Emitter LIST NUMBER IN SAME
      ORDER AS YOU ENTERED THE SITES')
      READ(5,4)(ST(I),I=K1,NL)
      K=NL
      WRITE(6,58)
58  FORMAT(' DO YOU WISH A LISTING OF THE F.O.B.?')
      READ(5,2)L
      IF(L.EQ.0) GO TO 35
      CALL RLL(SX,SY,K)
      WRITE(6,70)
70  FORMAT(' CHOOSE TYPE EmitterS TO LIST: 1=ALL; 2=EW/ACQ ONLY; 3=TERM
      1 THREAT ONLY')
      READ(5,2)L
      WRITE(6,59)
59  FORMAT(' SITE',1X,' NAME',1X,' MSN',1X,' BAND',1X,' RANGE',1X,
      1,1X,' LIST',1X,' FHI',1X,' PRF1',1X,' PRF2',1X,' PRF3',1X,' PRF4',1X,
      2,1X,' AUTO',1X,' DEGR',1X,' REMARK$')
      IF(L-2)71,72,73
71  DO 74 I=1,K
      WRITE(6,75)I,(TN(ST(I),J),J=1,3),(TYP(ST(I),J),J=1,2),(ST(I),J1,J=1,9),
      1,(TRMKS(ST(I),JN),JN=5,14),(TRMKS(ST(I),J1),J1,SY(I),SX(I))

```



```

75 FORMAT(1X,I3,2X,3A4,1X,A4,3X,11,2X,F4.0,10(1X,A4),1X,9A4,/5X,F7.2,
    1 2X,F7.2)
74 CNTINUE
    CALL LL(SX,SY,K)
    GO TO 35
72 DO 76 I=1,K
    IF(ST(1).GT.1.0)GO TO 76
    WRITE(6,75)I,(TN(ST(I),J),J=1,3),(TRMK(S(ST(I),J),J=1,14),JN=5,14),
    1 (TYP(ST(I),JN),JN=5,14),(TRMK(S(ST(I),J),J=1,9),SY(I)),SX(I))
76 CNTINUE
    CALL LL(SX,SY,K)
    GO TO 35
73 DO 77 I=1,K
    IF(ST(1).LT.1.0)GO TO 77
    WRITE(6,75)I,(TN(ST(I),J),J=1,3),(TYP(ST(I),J),J=1,14),JN=5,14),
    1 (TYP(ST(I),JN),JN=5,14),(TRMK(S(ST(I),J),J=1,9),SY(I)),SX(I))
77 CNTINUE
    CALL LL(SX,SY,K)
    GO TO 35
35 WRITE(6,32)
32 FORMAT(1X,J,' DO YOU WISH TO DISPLAY S0B & RTE?')
    READ(5,2)J
    IF(J.EQ.0.)GO TO 34
    WRITE(6,31)
31 FORMAT(1AFTER DISPLAY PRESENTED, ENTER ANY SINGLE CHARACTER TO C
    1 NTINUE.)
    WRITE(6,63)
63 FORMAT(1DO YOU WISH TO DISPLAY FOR THREAT ENVELOPES ALSO?')
    READ(5,2)LDISP
    IF(LDISP.EQ.0.)GO TO 66
    WRITE(6,61)
61 FORMAT(1TO DISPLAY WEAPON/RADAR ENVELOPES, ENTER APPROPRIATE NUM
    1 BER: 1=ALL EMITTERS, 2=EW/ACQ ONLY, 3=TERMINAL THREAT ONLY.)
    READ(5,2)LEMIT
    IF(LEMIT.NE.3)GO TO 66
    WRITE(6,62)
600 FORMAT(1X,J,' DO YOU WANT TO SEE ENVELOPES DEPRESSED BY JAMMING?')
    READ(5,2)JAM
    66 CALL MAX(X,N,XMAX)
    CALL MAX(SX,K,XMAX)
    XMAX=XMAX+2
    CALL MAX(Y,N,YMAX)
    CALL MAX(SY,K,YMAX)
    YMAX=YMAX+2
    CALL MTN(IX,N,XMIN)

```



```

CALL MIN(XMIN-.2,K,XMIN)
CALL MIN(YMIN,YMIN)
CALL MIN(SY,K,YMIN)
YMIN=SCALE(XMAX,XMIN,YMAX,YMIN)
CALL INIT LINE(0)
CALL SLIMX(150,850)
CALL SLIMY(050,700)
CALL XFRM(2)
CALL YFRM(2)
CALL DLIMX(XMIN,XMAX)
CALL DLIMY(YMIN,YMAX)
CALL NPTS(K)
CALL SYMBL(8)
CALL SIZES(1.)
CALL PLDT(SX,SY)
CALL NPTS(N)
CALL SYMBL(4)
IF(LRTE.EQ.0) GO TO 45
CALL LINE(1)
GO TO 46
45 CALL LINE(0)
CALL CPLOT(X,Y)
46 IF(MSN-2)47,420,380
420 CALL NPTS(MEN)
CALL SYMBL(3)
CALL CPLOT(XME,YME)
GO TO 47
380 CALL NPTS(1)
CALL LINE(0)
CALL SYMBL(5)
CALL CPLOT(STOX,STOY)
CALL LINF(1)
47 IF((LDISP-EQ.0) GO TO 36
81 DO 80 I=1,K
YC=SY(I)
RAD=TRG(ST(1))/60
IF(TRG(ST(1));GE.100; ) GO TO 90
IF(TRG(ST(1));GE.50.; ) GO TO 100
J=31
GC TO 110
J=51
GO TO 110
100 J=41

```



```

110 CALL CIRCLE(X0,Y0,RAD,J)
IF(MSN.NE.3)GO TO 80
IF(JAM.EQ.0)GO TO 80
CALL DIST(PJ(ST(I)),B(ST(i)),STOY,RJ)
RAD=BTHRU(PJ(ST(I)),B(ST(i)),STOY,RJ),GJR(ST(II)),PR(ST(II)),CS,RJ,GAIN(ST(I))
1,CMFLG)
J=41 LINE(121)
CALL CIRCLE(X0,Y0,RAD,J)
CALL LINE(1)
CALL LINE(1)
CONTINUE
80 GO TO 36
82 DO 84 I=1,K
IF(ST(I).GT.10)GO TO 84
X0=SY(I)
Y0=SY(I)
RAD=TRG(ST(I))/60
IF(TRG(ST(I)).GE.100.)GO TO 85
J=41 GO TO 86
85 J=51 CALL CIRCLE(X0,Y0,RAD,J)
IF(MSN.NE.3)GO TO 84
IF(JAM.EQ.0)GO TO 84
CALL DIST(PJ(Y0-ST(I)),B(ST(i)),STOY,RJ)
RAD=BTHRU(PJ(ST(I)),B(ST(i)),STOY,RJ),GJR(ST(II)),PR(ST(II)),CS,RJ,GAIN(ST(I))
1,CMFLG)
J=41 LINE(121)
CALL CIRCLE(X0,Y0,RAD,J)
CALL LINE(1)
CALL LINE(1)
CONTINUE
84 GO TO 36
85 DO 88 I=1,K
IF((ST(I).LE.10).OR.(ST(I).GT.20))GO TO 88
X0=SY(I)
Y0=SY(I)
RAD=TRG(ST(I))/60.
J=31
CALL CIRCLE(X0,Y0,RAD,J)
IF(MSN.NE.3)GO TO 88
IF(JAM.EQ.0)GO TO 88
CALL DIST(PJ(Y0-ST(I)),B(ST(i)),STOY,RJ)
RAD=BTHRUPJ(ST(I)),B(ST(i)),STOY,RJ),GJR(ST(II)),PR(ST(II)),CS,RJ,GAIN(ST(I))
1,CMFLG)
J=41 LINE(121)
CALL CIRCLE(X0,Y0,RAD,J)

```



CALL LINE(1)  
CONTINUE  
CALL PAUSE  
CALL FIN  
MAI03670  
MAI03680  
MAI03690  
MAI03700

```

34 IF(LRTE•EQ•0)GO TO 331
IF(MSN•EQ•0)GO TO 322
CALL TD(X,Y,M,SPD,TLM,DL,DT)
CALL HDG(X,Y,M,VAR,NTH,MH,PI)
WRITE(6,21)
FORMAT(6,21)
1 WANT NAV SOLN? ITS NECESS. FOR TIME SCENARIO')
2 READ(5,2)L
1 IF(L•EQ•0) GO TO 331
1 WRITE(6,500)
1 FORMAT(1X,'/ STRIKE GROUP NAVIGATION SOLUTION')
1 WRITE(6,23)
1 FORMAT(1X,'LEG',3X,'DIST',2X,'TIME',3X,'TH',3X,'MH',2X,'TAS',3X)
1 FORMAT(1X,'TOT',3X,'TOTD',6X,'TC TURNPT L/L')
1 CALL RLL(X,Y,N)
1 DO 25 I=1,M
1 WRITE(6,24)I,DL(I),TL(I),NTH(I),MH(I),ITAS(I),TT(I),DT(I),Y(I+1),
1 X(I+1)
1 FORMAT(12,4X,F5.1,3X,F3.0,2X,I3,2X,I3,2X,F4.1,3X,F5.1,2X,F7.
1 25 CONTINUE
1 CALL RLL(X,Y,N)
1 IF(MSN-2)93,450,330
1 CALL TD(XME,YME,MEM,SPDM,TLM,TTM,DL,DTM)
1 CALL HDG(XME,YME,MEM,VAR,NTHM,MHM,PI)
1 CALL RLL(XME,YME,MEN)
1 WRITE(6,451)
1 FORMAT(1X,'/ MOD ESCORT NAVIGATION SOLUTION')
1 WRTIE(6,23)
1 DO 452 I=1,MEM
1 WRITE(6,24)I,DLM(I),TLM(I),NTHM(I),NMH(I),MTAS(I),TTM(I),DTM(I),
1 E(I+1),XME(I+1)
1 CONTINUE
452 CONTINUE

```

```

93 WRITE(6,92)          ! DO YOU WISH A TIME SCENARIO?')
92 FORMAT(1X,/,1)
READ(5,2)L
IF(L.EQ.0)GO TO 331
WRITE(6,121)
121 FORMAT(1X,ENTER NUMBER OF DIFFERENT BANDS INTERESTED IN')

```



```

READ(5,2)NUMBND
WRITE(6,122)
FORMAT(6,122)ENTER THE BAND NO. S ONE AT A TIME, NO COMBOS LIKE 5/6')
122 READ(5,2)(NB(I),I=1,NUMBND)
WRITE(6,123)
FORMAT(6,123)INDICATE TYPE EMITTERS INTERESTED IN FOR THE BANDS YOU HAMA
123 1VE SELECTED: 1=ALL, 2=EW/ACQ ONLY, 3=TERM TREAT ONLY.
READ(5,2)NTYP
PRINT OUT HEADINGS FOR TIME SCENARIO
124 WRITE(6,124)
FORMAT(1X,'TIME',5X,'PRESS POS',8X,'TYPE',6X,'EOB',2X,'RNGE',2XMA
1, 'BRG',2X,'AUTO',2X,'DÉGR',2X,'PRCT')
SET INITIAL CONDITIONS FOR TIME SOLUTION
IF(MSN.EQ.2)GO TO 700
TIME=0.
PPX=X(1)
PPY=Y(1)
CALL RLL(PPX,PPY,1)
WRITE(6,125)TIME,PPY,PPX
125 FORMAT(1X,F4.0,2X,F7.2,1X,F7.2)
CALL LL(PPX,PPY,1)
CALL PPS(PPX,PPY,SX,SY,ST,K,NB,NUMBND,NTYP,TYP,IZ,TN,TRG,MSN,XSTK,
1YSTK)
C NOW PROCEEDED AROUND THE ROUTE 1 MIN AT A STEP
129 I=1
TIME=TIME+1/60.
DINC=SPD(I)/60.
IF(TT(I)*GT(TIME))GO TO 126
C NOT TRUE MEANS TURNPT WAS LE TO ONE MINUTE FROM LAST COMPUTED POSIT.
C FIGURE SPEED TO TURNPT, SUBTRACT FROM 1 MIN, INCREMENT I TO GET NEW PP.
C DING, SPEED=USE TIME REMAINING IN MINUTE TO GET NEW PP.
PARTM=TIME-TT(I)
I=I+1
C IS THIS LAST TURNPOINT?
IF(I.EQ.N)GO TO 127
PPX=X(I)
PPY=Y(I)
CALL RADN(LINTH,H)
PPX=PPX+PARTM*SPD(I)/60.*COS(H)
PPY=PPY+PARTM*SPD(I)/60.*SIN(H)
GO TO 128
C IT WAS THE LAST TURNPOINT
127 PPX=X(N)
PPY=Y(N)
TIME=TT(M)
GO TO 128
C NEXT SECTION MEANS HAVE NOT GONE PAST A TP, JUST ADD INCREMENT FOR THI
C LEG AND CONTINUE.

```



```

126 CALL RADN(I,NTH,H)
PPX=PPX+DINC*COS(H)
PPY=PPY+DINC*SIN(H)
128 CALL RLL(PPX,PPY,1)
WRITE(6,125)TIME,PPY,PPX
CALL LL(PPX,PPY,SX,SY,ST,K,NB,NUMBND,NTYP,TYP,IZ,TN,TRG,MSN,XSTK,MAI04620
1YSTK)
1IF(TIME.GE.TT(M))GO TO 331
GO TO 129

PAGE
SKIPO03
TIME=0.
PPX=XME(1)
PPY=YME(1)
XSTK=X(1)
YSTK=Y(1)
CALL RLL(PPX,PPY,1)
WRITE(6,125)TIME,PPY,PPX
CALL LL(PPX,PPY,1)
CALL PPS(PPX,PPY,SX,SY,ST,K,NB,NUMBND,NTYP,TYP,IZ,TN,TRG,MSN,XSTK,MAI04640
1YSTK)
1IF(TIME.GE.TT(M))GO TO 331
GO TO 129

725 TIME=TIME+1./60.
DINC=SPD(I)/60.
DINC=M/SPD(I)/60.
IF(TT(I)*GT*TIME)GO TO 726
PARTIN=TIME-TT(I)
I=I+1
IF(I.EQ.N)GO TO 727
XSTK=X(I)
YSTK=Y(I)
CALL RADN(I,NTH,H)
XSTK=XSTK+PARTIM*SPD(I)/60.*COS(H)
YSTK=YSTK+PARTIM*SPD(I)/60.*SIN(H)
GO TO 728
727 XSTK=X(N)
YSTK=Y(N)
GO TO 728
726 CALL RADN(I,NTH,H)
XSTK=XSTK+DINC*COS(H)
YSTK=YSTK+DINC*SIN(H)
IF(TTM(J)*GT*TIME)GO TO 736
PARTIN=TIME-TTM(J)
J=J+1
IF(J.EQ.MEN)GO TO 737
PPX=XME(J)

```



```

PY=YME(J)
CALL RADN(J,NTHM,H)
PX=PPX+PARTIM*SPDM(J)/60.*COS(H)
PY=PPY+PARTIM*SPDM(J)/60.*SIN(H)
GO TO 738
737 PPX=XMF(MEN)
PY=YME(MEN)
GO TO 738
736 CALL RADN(J,NTHM,H)
PX=PPX+DINC*CO$H)
PY=PPY+DINC*SIN(H)
CALL RLL(PPX,PPY,1)
WRITE(6,125)TIME,PPY,PPX
CALL PPS(PPX,PPY,SX,SY,ST,K,NB,NUMBND,NTYP,TYP,IZ,TN,TRG,MSN,XSTK,
1YSTK)
IF(TIME>TTM(MEM)) GO TO 331
GO TO 729

```

```

330 WRITE(6,315)
315 FORMAT(1X,'! DO YOU WANT A PRINTOUT OF JAMMING PARAMETERS FOR THIS MAI05200
1STANDOFF POINT?')
1 READ(5,2)L
1 IF(L.EQ.0) GO TO 331
1 WRITE(6,121)
1 READ(5,2)NUMBND
1 WRITE(6,122)
1 READ(5,2)(NB(I),I=1,NUMBND)
1 WRITE(6,123)
1 READ(5,2)NTYP
1 WRITE(6,316)
1 FORMAT(1X,' STANDOFF L/L',12X,'TYPE',6X,'E0B ',2X,'RNGE',2X,'BRGM
1 AUTOT/2X,DEGR',2X,'PRCT')
1 CALL RLL(STO$X$TOY$T)
1 WRITE(6,317)STOY,STOX
1 FORMAT(1XF7.2$1X,F7.2)
1 CALL LL(STO$X$TOY$T)
1 CALL PPS(STO$X$TOY$T)
1 K,YSTK)

```

```

331 WRITE(6,329)
329 FORMAT(1X,' ARE YOU THROUGH PLANNING?')
1 IF(L.EQ.1) GC TO 331
MAI05400
MAI05410
MAI05420
MAI05430

```



```

324 WRITE(6,324) WANT TO CONSIDER A (DIFFERENT) MISSION PROFILE?"'
      FORMAT(1X,/)
      READ(5,2)L
      IF(L.EQ.0) GO TO 91
      WRITE(6,300)
      READ(5,2)MSN

```

```

91  WRITE(6,26) // DO YOU WISH TO CONSIDER (ANOTHER) STRIKE ROUTE?"'
      FORMAT(5,2)LAGAIN
      IF(LAGAIN.EQ.0)GO TO 322
      WRITE(6,15)
      READ(4,2)N
      N=N-1
      WRITE(6,13)
      RFAD(4,4)(Y(I),X(I),I=1,N)
      CALL LL(X,Y,N)
      WRITE(6,17)
      READ(4,4)(ITAS(I),I=1,M)
      DO 28 I=1,M
      SPD(I)=ITAS(I)/6.
28   LRT(E=1
      322 IF(MSN-2)35,400,318

```

```

318 WRITE(6,319) WANT TO CONSIDER ANOTHER STANDOFF POINT?"'
319 FORMAT(1X,/)
      READ(5,2)LSC
      IF(LSD.EQ.0)GO TO 331
      WRITE(6,302)
      READ(5,4)STOX,STOX
      CALL LL(STOX,STOX,1)
      GO TO 35

```

```

33 STOP
END

```



THIS IS THE SUBROUTINE TO CALCULATE DISTANCE FROM A/C PRESENT POSITION TO EACH OF THE SITES IN THE EOB. IT PRINTS OUT INFO NECESSARY TO KEEP A RUNNING ACCOUNT OF UPCOMING ACTION REQUIRED BY THE OPERATOR DURING THE FLIGHT.

PARAMETERSS PASSED:  
 PPY, PPX; A/C PRESENT POSIT' LAT AND LONG  
 SX, SY; ARRAY CONTAINING LAT/LONG OF EACH SITE IN THE EOB  
 ST; TYPE OF EACH SITE IN THE EOB

```

K; NUMBER OF SITES IN THE EOB
NB; ARRAY CONTAINING THE SPECIFIC BANDS INTERESTED IN
NUMBND; HOW MANY DIFFERENT BANDS INTERESTED IN
NTYP; INDICATES TYPE OF EMITTER INTERESTED IN, 1 ALL, 2 EW, 3 TTP
TYP; ARRAY CONTAINING ENTIRE LIBRARY
SUBROUTINE PPS(PPX,PPY,SX,SY,ST,K,NB,NUMBND,NTYP,TYP,IZ,TN,TRG,MSN)
1 DIMENSION SX(1),SY(1),NB(1),ST(1),TYP(30,15),IZ(1),TN(30,3),TRG(1)
      INTEGER ST
      PI=3.14159
      DC=180./PI
      DC=100. I=1,K
      IC=ST(I)
      IF(INTYP-2)10,20,30
      10 DO11 J=1,NUMBND
      11 IF(IZ(IC).EQ.NB(J))GO TO 110
      11 CONTINUE
      GO TO 100
      C CHECK ONLY EW/AQO TYPE EMITTERS
      20 IF((IC.GT.10))GO TO 100
      DO21 J=1,NUMBND
      21 IF((IC.LT.11).OR.(IC.GT.20))GO TO 100
      DO31 J=1,NUMBND
      31 IF((IZ(IC).EQ.NB(J))GO TO 110
      31 CONTINUE
      C Emitter IS RIGHT TYPE AND BAND NOW
      110 SDF=1.J24-.0J23*ABS((SY(I+1)+PPY)/2.)-.0001*((SY(I+1)+PPY)/2.*)**2
      DX=(SX(I)-PPY)
      DY=(SY(I)-PPY)
      DS=(SQRT(DX**2+DY**2))**60.
      IF(MSN.EQ.3)GO TO 111
      IF(DS.GT.TRG(IC))GO TO 111
      GC TO 111
      IF(MSN.NE.2)GO TO 100
    
```



```

PPS00490
PPS00500
PPS00510
PPS00520
PPS00530
PPS00540
PPS00550
PPS00560
PPS00570
PPS00580
PPS00590
PPS00600
PPS00610
PPS00620
PPS00630
PPS00640
PPS00650
PPS00660
PPS00670
PPS00680
PPS00690
PPS00700
PPS00710

DXX=(SX(1)-XSTK)*SDF
DY=SY(1)-YSTK
DSS=(SQRT(DXX**2+DYY**2))*60.
IF(DSS.GT.100)GO TO 100
WENT BACK TO 100 ANOTHER SITE BECAUSE ITS NOT IN RANGE.
BUT IF IT IS IN RANGE, CONTINUE BELOW. CALC BEARING, PRINT OUT EVRY T
111 BRG=ATAN(DY/DX)
IF(DX.GE.0)GO TO 60
BRG=(1.5*pi-BRG)*DG
GO TO 80
60 BRG=(PI/2.-BRG)*DG
8C IF(NBRG.GE.0)GO TO 40
NBRG=NBRG+360
GC TO 89
40 IF(NBRG.LT.360)GO TO 89
NBRG=NBRG-360
89 WRITE(6,90)(TN(IC,J),J=1,3),I,DS,NBRG,TYP(IC,12),TYP(IC,13),TYP(ICPPS00660
114)
90 FORMAT(21X,3A4,3X,I2,3X,F4.0,3X,I3,2X,A4,2X,A4)
100 CONTINUE
RETURN
END

```



```

SCA00010
SCA00020
SCA00030
SCA00040
SCA00050
SCA00060
SCA00070
SCA00080
SCA00090
SCA00100
SCA00110
SCA00120
SCA00130
SCA00140
SCA00150
SCA00160
SCA00170

THIS SUBROUTINE TAKES THE MAX AND MIN OF THE X AND Y COORDINATES,
C CHOOSES THE LARGER OF THE TWO SPREADS AND ADJUSTS THE OTHER
C ACCORDINGLY SO THAT THE PLOT ON THE GRAPH IS ALWAYS SQUARE AND
C THE DISTANCE SCALE IN THE ORZ. IS THE SAME AS IN THE VERT.
SUBROUTINE SCALE(XMAX,XMIN,YMAX,YMIN)
DX=ABS(XMAX-XMIN)
CY=ABS(YMAX-YMIN)
IF(CX.GE.DY)GO TO 10
XMP=XMIN+DX/2.
XNAX=XMP+DY/2.
GC TO 20
10 YNP=YMIN+DY/2.
YNIN=YMP-DX/2.
YMAX=YMP+DX/2.
20 RETURN
END

```

```

TD 00010
TD 00020
TD 00030
TD 00040
TD 00050
TD 00060
TD 00070
TD 00080
TD 00090
TD 00100
TD 00110
TD 00120
TD 00130
TD 00140
TD 00150
TD 00160
TD 00170
TD 00180
TD 00190
TD 00200
TD 00210
TD 00220
TD 00230
TD 00240

THIS SUBROUTINE COMPUTES THE TIME AND DISTANCE AROUND A PARTICULAR
ROUTE. COMPUTES BOTH TOTAL AND INDIVIDUAL LEG VALUES.
PARAMETERS PASSED ARE: LONG AND LAT OF TRNPTS, NUMBER OF PTS
SPEED (NAUT MILES/MIN) FOR EACH LEG, TIME/LEG, TOTAL TIME, DIST/LEG
AND TOTAL DISTANCE.
SUBROUTINE TD(X,Y,M,SPD,TL,TT,DL)
DIMENSION X(1),Y(1),SPD(1),TL(1),DT(1),TT(1)
T=0.
D=0.
DO 10 I=1,M
C MUST CALCULATE A SCALE DOWN FACTOR SDF, LAT*LONG IN DIST
C SDF=1.0294-.0023*ABS((Y(I+1)+Y(I))/2.)*SDF
C DX=(X(I+1)-X(I))*SDF
C DY=Y(I+1)-Y(I)
C DL(I)=SQR(T(DX**2+DY**2))
C DL(I)=DL(I)**60.
C DT(I)=D/60.
C TL(I)=DL(I)/SPD(I)
C TT(I)=T+TL(I)
10 CCNTINUE
END

```



```

C THIS SUBRTN CONVERTS TRUE HEADINGS FROM DEGREES TO RADIANS FOR USE IN RAD00010
C COMPUTING THE DISTANCE INCREMENT OF LAT/LONG IN THE TIME SCENARIO.
C SPECIFICALLY: SIN AND COS FUNCTIONS REQUIRE RAD00020
C SUBROUTINE RADN(I,NTH,H)
C DIMENSION NTH(1)
C X=2.*3.14159/360.
C IF((NTH(I)*GE.0).AND.(NTH(I)*LE.90))GO TO 10
C H=(450-NTH(I))*X
C GO TO 20
C H=(90-NTH(I))*X
C 10 RETURN
C END

```

```

C THIS SUBRTN COMPUTES TRUE HDG AND MAG HDG FOR A ROUTE OF FLIGHT.
C PARAMETERS PASSED ARE: X AND Y COORDS OF TURNPTS,
C NUMBER OF TRNPTS, MAGNETIC VARIATION, DUMMY LISTS FOR TH AND MH,
C AND VALUE OF PI.
C SUBROUTINE HDG(X(I),Y(I),MVAR,NTH,MH,PI)
C DIMENSION X(1),Y(1),NTH(1),MH(1)
C INTEGER VAR
C DG=180./PI
C DO 10 I=1,M
C     MUST CALC A SCALE DOWN FACTOR SDF, SINCE LAT.*NE LONG IN DIST
C     SDF=1.0294-.0023*ABS((Y(I+1)+Y(I))/2.)-.0001*((Y(I+1)+Y(I))/2.)***2
C     CX=(X(I+1)-X(I))*SDF
C     DY=Y(I+1)-Y(I)
C     TH=ATAN(DY/DX)
C     IF(CX*GE.0)GO TO 60
C     TH=(1.*5.*PI-TH)*DG
C     GO TO 80
C 10 TH=(PI/2.-TH)*DG
C 80 MT(I)=NTH(I)+VAR
C     MC=MH(I)
C     IF(MD*GE.0)GO TO 40
C     MT(I)=MH(I)+360
C     GO TO 10
C 40 IF(MD.LT.-360)GO TO 10
C     MH(I)=MH(I)-360
C     CONTINUE
C END

```



```

C THIS ROUTINE CONVERTS DEGREES AND MINUTES TO DEGREES AND TENTHS OF
C DEGREES.
SUBROUTINE LL(X,Y,N)
DIMENSION X(1),Y(1)
DO 10 I=1,N
  NX=X(I)
  RX=X(I)-NX
  RX=RX/.6
  X(I)=NX+RX
  NY=Y(I)
  RY=RY-.NY
  RY=RY/.6
  Y(I)=NY+RY
  10 CCNTINUE
      RETURN
END

```

```

C THIS ROUTINE RE-CODENTS LATITUDES AND LONGITUDES FROM DEGREES AND
C TENTHS OF DEG. TO DEGREES AND MINUTES FOR PRINTOUT.
SUBROUTINE RL(X,Y,N)
DIMENSION X(1),Y(1)
DO 10 I=1,N
  NX=X(I)
  RX=X(I)-NX
  RX=RX*.6
  X(I)=NX+RX
  NY=Y(I)
  RY=Y(I)-NY
  RY=RY*.6
  Y(I)=NY+RY
  10 CCNTINUE
      RETURN
END

```



```

      THIS ROUTINE SEARCHES FOR THE MAX VALUE OF THE LIST PASSED TO,
      AND RETURNS THIS VALUE TO THE MAIN PROGRAM. IT IS USED TO
      OBTAIN THE MAX LAT AND LONG FOR USE IN ADJUSTING THE SCALE
      DURING THE PLOTTING SEQUENCE.
      PARAMETERS PASSED ARE: ARRAY OF LATITUDES OR LONGITUDES, NUMBER
      OF ITEMS IN THE ARRAY, DUMMY VALUE OF MAX WITH TRUE MAXIMUM TO BE
      RETURNED.
      SUBROUTINE MAX(X,N,XMAX)
      DIMENSION X(1)
      XMAX=X(1)
      DO 10 I=1,N
      XMAX=AMAX1(XMAX,X(I))
      10 RETURN
      END

```

```

      THIS SUBROUTINE SEARCHES LIST PASSED FOR THE MIN VALUE AND
      RETURNS IT TO THE MAIN PROGRAM. USED TO SET LIMITS IN SCALING
      DURING THE PLOTTING ROUTINE.
      PARAMETERS PASSED ARE: ARRAY OF LATITUDES OR LONGITUDES,
      NUMBER OF ITEMS IN THE ARRAY, DUMMY VALUE OF MIN WITH THE REAL
      VALUE OF MIN TO BE RETURNED TO THE MAIN PROGRAM.
      SUBROUTINE MIN(X,N,XMIN)
      DIMENSION X(1)
      XMIN=X(1)
      DO 10 I=1,N
      XMIN=AMINI(XMIN,X(I))
      10 RETURN
      END

```



```

C THIS ROUTINE CALCULATES THE DISTANCE FROM A STANOFF POINT TO EACH
C Emitter of interest, when printing parameters for a standoff msn.
C SUBROUTINE DIST(X0,Y0,STOX,STOY,RJ)
C SF=1.0294-.0023*ABS((Y0+STOY)/2.)-.001*((YC+STOY)/2.)**2
C DX=(X0-STOX)*SDF
C DY=Y0-STOY
C RJ=SQRT(DX**2+DY**2)
C RETURN
C END

```

```

C THIS FUNCTION CALCULATES THE BURNTHRU RANGE FOR JAMMING FROM A STANOFF
C PCINT TO ANY Emitter of interest. USED IN PLOTTING DEPRESSED JAMMING
C ENVELOPES IN THE DISPLAY PORTION OF THE PROGRAM.
C FUNCTION BTHRU(PJ,B,GJR,PR,CS,RJ,GAIN,CMFLG)
C RJ=RJ*1852.
C PR=PR*1000.
C BTHR=(((PR*GAIN)**2*CMFLG*CS*RJ**2)/(12.56637062*pj
C 1*B*((10.*((GJR/10.))**.25))/1852.
C RETURN
C END

```



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